

the material listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this material as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) European Union Aviation Safety Agency (EASA) AD 2023-0218, dated December 19, 2023.

(ii) [Reserved]

(3) For EASA material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 8999 000; email: ADs@easa.europa.eu; website: ad.easa.europa.eu. You may find this EASA AD on the EASA website at ad.easa.europa.eu.

(4) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 1200 District Avenue, Burlington, MA 01803. For information on the availability of this material at the FAA, call (817) 222-5110.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit www.archives.gov/federal-register/cfr/ibr-locations or email fr.inspection@nara.gov.

Issued on September 13, 2024.

Steven W. Thompson,

Acting Deputy Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2024-24365 Filed 10-21-24; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2024-1685; Project Identifier MCAI-2024-00076-T; Amendment 39-22855; AD 2024-19-13]

RIN 2120-AA64

Airworthiness Directives; Airbus SAS Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is superseding Airworthiness Directive (AD) 2023-04-06 and AD 2024-04-07, which applied to certain Airbus SAS Model A318, A319, A320 and A321 series airplanes. AD 2023-04-06 and AD 2024-04-07 required revising the existing maintenance or inspection program, as applicable, to incorporate new or more restrictive airworthiness limitations. This AD was prompted by a determination that new or more restrictive airworthiness limitations are necessary. This AD continues to require certain actions in AD 2023-04-06 and AD 2024-04-07 and requires revising the existing maintenance or inspection

program, as applicable, to incorporate new or more restrictive airworthiness limitations; as specified in the European Union Aviation Safety Agency (EASA) AD, which is incorporated by reference. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective November 26, 2024.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of November 26, 2024.

The Director of the Federal Register approved the incorporation by reference of a certain other publication listed in this AD as of April 22, 2024 (89 FR 19234, March 18, 2024).

The Director of the Federal Register approved the incorporation by reference of a certain other publication listed in this AD as of April 10, 2023 (88 FR 13665, March 6, 2023).

ADDRESSES:

AD Docket: You may examine the AD docket at regulations.gov under Docket No. FAA-2024-1685; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the mandatory continuing airworthiness information (MCAI), any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

Material Incorporated by Reference:

- For EASA material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu; website easa.europa.eu. You may find this material on the EASA website at ad.easa.europa.eu.

- You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. It is also available at regulations.gov under Docket No. FAA-2024-1685.

FOR FURTHER INFORMATION CONTACT: Timothy Dowling, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 206-231-3367; email timothy.p.dowling@faa.gov.

SUPPLEMENTARY INFORMATION:

Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to supersede AD 2023-04-06,

Amendment 39-22353 (88 FR 13665, March 6, 2023) (AD 2023-04-06), and AD 2024-04-07, Amendment 39-22686 (89 FR 19234, March 18, 2024). AD 2023-04-06 and AD 2024-04-07 applied to Airbus SAS Model A318-111, -112, -121, and -122 airplanes; Model A319-111, -112, -113, -114, -115, -131, -132, -133, -151N, -153N, and -171N airplanes; Model A320-211, -212, -214, -216, -231, -232, -233, -251N, -252N, -253N, -271N, -272N, and -273N airplanes; and Model A321-111, -112, -131, -211, -212, -213, -231, -232, -251N, -252N, -253N, -271N, -272N, -271NX, -272NX, -252NX, -253NX, -271NX, and -272NX airplanes. AD 2023-04-06 and AD 2024-04-07 required revising the existing maintenance or inspection program, as applicable, to incorporate additional new or more restrictive airworthiness limitations. The FAA issued AD 2023-04-06 and AD 2024-04-07 to address a safety-significant latent failure.

The NPRM published in the **Federal Register** on June 18, 2024 (89 FR 51471). The NPRM was prompted by AD 2024-0030, dated January 31, 2024, issued by EASA, which is the Technical Agent for the Member States of the European Union (EASA AD 2024-0030) (also referred to as the MCAI). The MCAI states that new or more restrictive airworthiness limitations have been developed.

In the NPRM, the FAA proposed to require revising the existing maintenance or inspection program, as applicable, to incorporate additional new or more restrictive airworthiness limitations, as specified in EASA AD 2024-0030. The FAA is issuing this AD to address a safety significant latent failure (that is not annunciated), which, in combination with one or more other specific failures or events, could result in a hazardous or catastrophic failure condition.

You may examine the MCAI in the AD docket at regulations.gov under Docket No. FAA-2024-1685.

Discussion of Final Airworthiness Directive

Comments

The FAA received comments from the Air Line Pilots Association, International (ALPA) who supported the NPRM without change.

Additional Changes Made To This AD

The proposed AD did not retain the terminating action specified in paragraph (j) of AD 2024-04-07. The FAA has added that terminating action to paragraph (j) of this AD and redesignated subsequent paragraphs of this AD accordingly.

Conclusion

This product has been approved by the aviation authority of another country and is approved for operation in the United States. Pursuant to the FAA's bilateral agreement with this State of Design Authority, it has notified the FAA of the unsafe condition described in the MCAI referenced above. The FAA reviewed the relevant data, considered the comment received, and determined that air safety requires adopting this AD as proposed. Accordingly, the FAA is issuing this AD to address the unsafe condition on this product. Except for minor editorial changes, and any other changes described previously, this AD is adopted as proposed in the NPRM. None of the changes will increase the economic burden on any operator.

Material Incorporated by Reference Under 1 CFR Part 51

EASA AD 2024-0030 specifies procedures for new or more restrictive airworthiness limitations for airplane structures and safe life limits.

This AD also requires EASA AD 2023-0138, dated July 13, 2023, which the Director of the Federal Register approved for incorporation by reference as of April 22, 2024 (89 FR 19234, March 18, 2024).

This AD also requires EASA AD 2022-0091, dated May 20, 2022, which the Director of the Federal Register approved for incorporation by reference as of April 10, 2023 (88 FR 13665, March 6, 2023).

This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section.

Costs of Compliance

The FAA estimates that this AD affects 1,898 airplanes of U.S. registry. The FAA estimates the following costs to comply with this AD:

The FAA estimates the total cost per operator for the retained actions from AD 2023-04-06 and AD 2024-04-07 to be \$7,650 (90 work hours × \$85 per work hour) per AD.

The FAA has determined that revising the existing maintenance or inspection program takes an average of 90 work hours per operator, although the agency recognizes that this number may vary from operator to operator. Since operators incorporate maintenance or inspection program changes for their affected fleet(s), the FAA has determined that a per-operator estimate is more accurate than a per-airplane estimate.

The FAA estimates the total cost per operator for the new proposed actions to

be \$7,650 (90 work hours × \$85 per work hour).

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by:

- a. Removing Airworthiness Directive (AD) 2023-04-06, Amendment 39-22353 (88 FR 13665, March 6, 2023); and AD 2024-04-07, Amendment 39-22686 (89 FR 19234, March 18, 2024); and

- b. Adding the following new AD:

2024-19-13 Airbus SAS: Amendment 39-22855; Docket No. FAA-2024-1685; Project Identifier MCAI-2024-00076-T.

(a) Effective Date

This airworthiness directive (AD) is effective November 26, 2024.

(b) Affected ADs

This AD replaces AD 2023-04-06, Amendment 39-22353 (88 FR 13665, March 6, 2023) (AD 2023-04-06); and AD 2024-04-07, Amendment 39-22686 (89 FR 19234, March 18, 2024) (AD 2024-04-07).

(c) Applicability

This AD applies to the Airbus SAS airplanes specified in paragraphs (c)(1) through (4) of this AD, certificated in any category, with an original airworthiness certificate or original export certificate of airworthiness issued on or before December 15, 2023.

(1) Model A318-111, -112, -121, and -122 airplanes.

(2) Model A319-111, -112, -113, -114, -115, -131, -132, -133, -151N, -153N, and -171N airplanes.

(3) Model A320-211, -212, -214, -216, -231, -232, -233, -251N, -252N, -253N, -271N, -272N, and -273N airplanes.

(4) Model A321-111, -112, -131, -211, -212, -213, -231, -232, -251N, -251NX, -252N, -252NX, -253N, -253NX, -271N, -271NX, -272N, and -272NX airplanes.

(d) Subject

Air Transport Association (ATA) of America Code 05, Time Limits/Maintenance Checks.

(e) Unsafe Condition

This AD was prompted by a determination that new or more restrictive airworthiness limitations are necessary. The FAA is issuing this AD to address a safety significant latent failure (that is not unannounced), which, in combination with one or more other specific failures or events, could result in a hazardous or catastrophic failure condition.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Retained Revision of the Existing Maintenance or Inspection Program, With AD 2024-04-07, With No Changes

This paragraph restates the requirements of paragraph (g) of AD 2024-04-07, with no changes. For airplanes with an original airworthiness certificate or original export certificate of airworthiness issued on or before May 12, 2023: Except as specified in paragraph (h) of this AD, comply with all required actions and compliance times specified in, and in accordance with EASA AD 2023-0138, dated July 13, 2023 (EASA

AD 2023–0138). Accomplishing the revision of the existing maintenance or inspection program required by paragraph (n) of this AD terminates the requirements of this paragraph.

(h) Retained Exceptions to AD 2024–04–07, With No Changes

This paragraph restates the exceptions specified in paragraph (h) of AD 2024–04–07, with no changes.

(1) This AD does not adopt the requirements specified in paragraphs (1) and (2) of EASA AD 2023–0138.

(2) Paragraph (3) of EASA AD 2023–0138 specifies revising “the approved AMP” within 12 months after its effective date, but this AD requires revising the existing maintenance or inspection program, as applicable, within 90 days after April 22, 2024 (the effective date of AD 2024–04–07).

(3) The initial compliance time for doing the tasks specified in paragraph (3) of EASA AD 2023–0138 is at the applicable “associated thresholds” as incorporated by the requirements of paragraph (3) of EASA AD 2023–0138, or within 90 days after April 22, 2024 (the effective date of AD 2024–04–07), whichever occurs later.

(4) This AD does not adopt the provisions specified in paragraphs (4) of EASA AD 2023–0138.

(5) This AD does not adopt the “Remarks” section of EASA AD 2023–0138.

(i) Retained Restrictions on Alternative Actions and Intervals, With a New Exception

This paragraph restates the requirements of paragraph (i) of AD 2024–04–07, with a new exception. Except as required by paragraph (n) of this AD, after the existing maintenance or inspection program has been revised as required by paragraph (g) of this AD, no alternative actions (e.g., inspections) and intervals are allowed unless they are approved as specified in the provisions of the “Ref. Publications” section of EASA AD 2023–0138.

(j) Retained Terminating Action for Certain Tasks Required by AD 2023–04–06

Accomplishing the actions required by paragraph (g) of this AD terminates the corresponding requirements of paragraph (k) of this AD for the tasks identified in the material referenced in EASA AD 2023–0138 only.

(k) Retained Revision of the Existing Maintenance or Inspection Program, With AD 2023–04–06, With No Changes

This paragraph restates the requirements of paragraph (o) of AD 2023–04–06, with no changes. For airplanes with an original airworthiness certificate or original export certificate of airworthiness issued on or before February 18, 2022: Except as specified in paragraph (l) of this AD, comply with all required actions and compliance times specified in, and in accordance with EASA AD 2022–0091, dated May 20, 2022 (EASA AD 2022–0091). Accomplishing the revision of the existing maintenance or inspection program required by paragraph (n) of this AD terminates the requirements of this paragraph.

(l) Retained Exceptions to AD 2023–04–06, With No Changes

This paragraph restates the exceptions specified in paragraph (p) of AD 2023–04–06, with no changes.

(1) The requirements specified in paragraphs (1) and (2) of EASA AD 2022–0091 do not apply to this AD.

(2) Paragraph (3) of EASA AD 2022–0091 specifies revising “the approved AMP” within 12 months after its effective date, but this AD requires revising the existing maintenance or inspection program, as applicable, within 90 days after April 10, 2023 (the effective date of AD 2023–04–06).

(3) The initial compliance time for doing the tasks specified in paragraph (3) of EASA AD 2022–0091 is at the applicable “associated thresholds” as incorporated by the requirements of paragraph (3) of EASA AD 2022–0091, or within 90 days after April 10, 2023 (the effective date of AD 2023–04–06), whichever occurs later.

(4) The provisions specified in paragraphs (4) and (5) of EASA AD 2022–0091 do not apply to this AD.

(5) This AD does not adopt the “Remarks” section of EASA AD 2022–0091.

(m) Retained Restrictions on Alternative Actions and Intervals, With a New Exception

This paragraph restates the requirements of paragraph (q) of AD 2023–04–06, with a new exception. Except as required by paragraphs (j) and (n) of this AD, after the existing maintenance or inspection program has been revised as required by paragraph (k) of this AD, no alternative actions (e.g., inspections) and intervals are allowed unless they are approved as specified in the provisions of the “Ref. Publications” section of EASA AD 2022–0091.

(n) New Revision of the Existing Maintenance or Inspection Program

Except as specified in paragraph (o) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, EASA AD 2024–0030, dated January 31, 2024 (EASA AD 2024–0030). Accomplishing the revision of the existing maintenance or inspection program required by this paragraph terminates the requirements of paragraphs (g) and (k) of this AD.

(o) Exceptions to EASA AD 2024–0030

(1) This AD does not adopt the requirements specified in paragraphs (1) and (2) of EASA AD 2024–0030.

(2) Paragraph (3) of EASA AD 2024–0030 specifies revising “the approved AMP,” within 12 months after its effective date, but this AD requires revising the existing maintenance or inspection program, as applicable, within 90 days after the effective date of this AD.

(3) The initial compliance time for doing the tasks specified in paragraph (3) of EASA AD 2024–0030 is at the applicable “associated thresholds” as incorporated by the requirements of paragraph (3) of EASA AD 2024–0030, or within 90 days after the effective date of this AD, whichever occurs later.

(4) This AD does not adopt the provisions specified in paragraphs (4) and (5) of EASA AD 2024–0030.

(5) This AD does not adopt the “Remarks” section of EASA AD 2024–0030.

(p) New Provisions for Alternative Actions and Intervals

After the existing maintenance or inspection program has been revised as required by paragraph (n) of this AD, no alternative actions (e.g., inspections) and intervals are allowed unless they are approved as specified in the provisions of the “Ref. Publications” section of EASA AD 2024–0030.

(q) Additional AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the manager of the International Validation Branch, mail it to the address identified in paragraph (r) of this AD. Information may be emailed to: 9-AVS-AIR-730-AMOC@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(2) *Contacting the Manufacturer*: For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, International Validation Branch, FAA; or EASA; or Airbus SAS’s EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(r) Additional Information

For more information about this AD, contact Timothy Dowling, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 206–231–3367; email timothy.p.dowling@faa.gov.

(s) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the material listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this material as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(3) The following material was approved for IBR on November 26, 2024.

(i) European Union Aviation Safety Agency (EASA) AD 2024–0030, dated January 31, 2024.

(ii) [Reserved]

(4) The following material was approved for IBR on April 22, 2024 (89 FR 19234, March 18, 2024).

(i) EASA AD 2023–0138, dated July 13, 2023.

(ii) [Reserved]

(5) The following material was approved for IBR on April 10, 2023 (88 FR 13665, March 6, 2023).

(i) EASA AD 2022–0091, dated May 20, 2022.

(ii) [Reserved]

(6) For EASA material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu; website easa.europa.eu. You may find these EASA ADs on the EASA website ad.easa.europa.eu.

(7) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th Street, Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

(8) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit www.archives.gov/federal-register/cfr/ibr-locations or email fr.inspection@nara.gov.

Issued on September 20, 2024.

Peter A. White,

Deputy Director, Integrated Certificate Management Division, Aircraft Certification Service.

[FR Doc. 2024–24385 Filed 10–21–24; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2024–1480; Project Identifier MCAI–2023–00930–A; Amendment 39–22851; AD 2024–19–09]

RIN 2120–AA64

Airworthiness Directives; DAHER (Type Certificate Previously Held by SOCATA) Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is superseding Airworthiness Directive (AD) 2016–08–08 for all SOCATA (type certificate now held by DAHER) Model MS 880B, MS 885, MS 892A–150, MS 892E–150, MS 893A, MS 893E, MS 894A, MS 894E, Rallye 100S, Rallye 150ST, Rallye 150T, Rallye 235C, and Rallye 235E airplanes. AD 2016–08–08 required doing repetitive detailed visual inspections of the intersection between the axle radius and the nose landing gear (NLG) fork area for chafing; doing repetitive dye penetrant inspections on the NLG wheel axle for cracks, distortion, and nicks or wear; doing corrective actions if necessary; and replacing the NLG wheel axle attachment screws with new screws. This AD is prompted by the introduction of a new inspection method and a new design for the

affected NLG wheel axle that provides terminating action for the repetitive inspections, as specified in updated service information published by the manufacturer. This AD requires preparing, inspecting, replacing, reconditioning the NLG wheel axle, and prohibiting affected parts. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective November 26, 2024.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of November 26, 2024.

ADDRESSES:

AD Docket: You may examine the AD docket at regulations.gov under Docket No. FAA–2024–1480; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the mandatory continuing airworthiness information (MCAI), any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

Material Incorporated by Reference:

- For Daher Aerospace material identified in this AD, contact Daher Aerospace, Customer Support, Airplane Division, 65921 Tarbes Cedex 9, France; phone: 33 (0)5 62.41.73.00; email: tbmcare@daher.com; website: daher.com.

- You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 901 Locust, Kansas City, MO 64106. For information on the availability of this material at the FAA, call (817) 222–5110. It is also available at regulations.gov under Docket No. FAA–2024–1480.

FOR FURTHER INFORMATION CONTACT: Fred Guerin, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: (206) 231–2346; email: fred.guerin@faa.gov.

SUPPLEMENTARY INFORMATION:

Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to supersede AD 2016–08–08, Amendment 39–18484 (81 FR 21711, April 13, 2016) (AD 2016–08–08). AD 2016–08–08 applied to SOCATA (type certificate now held by DAHER) Model MS 880B, MS 885, MS 892A–150, MS 892E–150, MS 893A, MS 893E, MS 894A, MS 894E, Rallye 100S, Rallye 150ST, Rallye 150T, Rallye 235C, and Rallye 235E airplanes.

AD 2016–08–08 was prompted by MCAI originated by European Union Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union. EASA issued AD 2015–0203, dated October 7, 2015 (EASA AD 2015–0203) to correct an unsafe condition identified as fatigue failure of the NLG wheel axle. EASA AD 2015–0203 was prompted by an NLG wheel axle failure that occurred in service. The investigation revealed that this failure was due to fatigue caused by premature wear. EASA AD 2015–0203 required retaining the requirements of Direction Générale de l’Aviation Civile (DGAC) France AD 91–163(A)R2, dated March 30, 1994, which was superseded, but required those actions to be accomplished within reduced intervals.

AD 2016–08–08 required doing repetitive detailed visual inspections of the intersection between the axle radius and the NLG fork area for chafing; doing repetitive dye penetrant inspections on the NLG wheel axle for cracks, distortion, and nicks or wear; corrective actions if necessary; and replacing the NLG wheel axle attachment screws with new screws. The FAA issued AD 2016–08–08 to address chafing and cracking of the NLG wheel axle, which could lead to failure of the NLG with consequent damage to the airplane and/or injury to occupants.

Since the FAA issued AD 2016–08–08, EASA superseded EASA AD 2015–0203 and issued EASA AD 2023–0080, dated April 17, 2023 (EASA AD 2023–0080) (also referred to as the MCAI) to correct an unsafe condition for all DAHER AEROSPACE Model MS 880 B, MS 880 B–D, MS 881, MS 883, MS 884, MS 885, MS 886, MS 887, MS 890 A, MS 890 B, MS 892 A.150, MS 892 B.150, MS 892 E.150, MS 892 E–D.150, MS 893 A, MS 893 B, MS 893 E, MS 893 E–D, MS 894 A, MS 894 E, RALLYE 100 S, RALLYE 100 S–D, RALLYE 100 ST, RALLYE 100 ST–D, RALLYE 110 ST, RALLYE 150 ST, RALLYE 150 ST–D, RALLYE 150 T, RALLYE 150 T–D, RALLYE 150 SV, RALLYE 150 SVS, RALLYE 180 T, RALLYE 180 TS, RALLYE 180 T–D, RALLYE 235 A, RALLYE 235 F, RALLYE 235 E, and RALLYE 235 E–D airplanes.

The MCAI states that it retains the requirements of EASA AD 2015–0203 but introduces a new inspection method and a new design of the NLG wheel axle as a terminating action for the repetitive detailed inspections. You may examine the MCAI in the AD docket at regulations.gov under Docket No. FAA–2024–1480.

The NPRM published in the **Federal Register** on June 4, 2024 (89 FR 47879). In the NPRM, the FAA proposed to