

The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

2024–08–03 Britten-Norman Aircraft, Ltd.:
Amendment 39–22736; Docket No. FAA–2024–0044; Project Identifier MCAI–2023–00629–A.

(a) Effective Date

This airworthiness directive (AD) is effective June 3, 2024.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Britten-Norman Aircraft Ltd airplanes, all serial numbers, certificated in any category, identified in paragraphs (c)(1) and (2) of this AD.

(1) Model BN–2, BN–2A, BN–2A–2, BN–2A–3, BN–2A–6, BN–2A–8, BN–2A–9, BN–2A–20, BN–2A–21, BN–2A–26, BN–2A–27, BN–2B–20, BN–2B–21, BN–2B–26, BN–2B–27, BN–2T, BN2T–4R, and BN2T–4S airplanes.

(2) Model BN2A MK. III, BN2A MK. III–2, and BN2A MK. III–3 airplanes.

(d) Subject

Joint Aircraft System Component (JASC) Code 2797, Flight Control System Wiring.

(e) Unsafe Condition

This AD was prompted by reports of electrical cable (Koiled Kord) and flight control cables interference with the control column. The FAA is issuing this AD to address interference between the Koiled Kord, flight control cables, and the control column, which could restrict the full and free movement of the flight controls. This unsafe condition, if not addressed, could result in loss of control of the airplane during flight.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Definition

For the purposes of this AD, a Koiled Kord is the coiled electrical cable that carries the wires from switches on the control yoke, through the control column tube, to the rear of the instrument panel. It exits the control column tube behind the instrument panel and continues to a terminal block.

(h) Required Actions

(1) Within 100 hours time-in-service (TIS) after the effective date of this AD, inspect for interference between the control column, rudder pedal adjuster cable, and any other wiring, including the Koiled Kord, in accordance with Sections 6 and 7(1) of Britten-Norman Service Bulletin SB 398, Issue 2, dated May 30, 2022 (Britten-Norman SB 398, Issue 2), while concurrently performing a control column full and free movement inspection, in accordance with Section 8 of Britten-Norman SB 398, Issue 2, to inspect for free play, friction, binding, non-linear forces, and any remaining interference.

(2) If interference between the control column, the rudder pedal adjuster cable, and any other wiring, including the Koiled Kord, or any free play, friction, binding, non-linear forces, or any remaining interference was found during the inspections required by paragraph (h)(1) of this AD, before further flight, securely tie any interfering electrical cables clear of the control column for its full range of motion and perform a final full and free movement inspection in accordance with Section 8 of Britten-Norman SB 398, Issue 2, to inspect for free play, friction, binding, non-linear forces, and any remaining interference. If there is any free play, friction, binding, non-linear forces, or any remaining interference, before further flight resolve these issues in accordance with a method approved by the Manager, International Validation Branch, FAA; or the Civil Aviation Authority United Kingdom (CAA UK); or Britten-Norman Aircraft Ltd.'s CAA UK Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(i) Alternative Methods of Compliance (AMOCs)

The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Validation Branch, mail it to the address identified in paragraph (j)(2) of this AD or email to: *9-AVS-AIR-730-AMOC@faa.gov*. If mailing information, also submit information by email. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local Flight Standards District Office/certificate holding district office.

(j) Additional Information

(1) Refer to CAA UK AD G–2022–0017, dated September 20, 2022, for related information. This CAA UK AD may be found in the AD docket at [regulations.gov](https://www.regulations.gov) under Docket No. FAA–2024–0044.

(2) For more information about this AD, contact Penelope Trease, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: (303) 342–1094; email: *penelope.trease@faa.gov*.

(k) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Britten-Norman Service Bulletin SB 398, Issue 2, dated May 30, 2022.

(ii) [Reserved]

(3) For service information, contact Britten-Norman Aircraft Ltd., Bembridge Airport, Bembridge, Isle of Wight, PO35 5PR United Kingdom; phone: +44 20 3371 4000; email: *customer.support@britten-norman.com*; website: *britten-norman.com/approvals-technical-publications*.

(4) You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 901 Locust, Kansas City, MO 64106. For information on the availability of this material at the FAA, call (817) 222–5110.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit www.archives.gov/federal-register/cfr/ibr-locations or email *fr.inspection@nara.gov*.

Issued on April 15, 2024.

Victor Wicklund,

Deputy Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2024–09083 Filed 4–26–24; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2024–0035; Project Identifier MCAI–2023–00986–A; Amendment 39–22728; AD 2024–07–07]

RIN 2120–AA64

Airworthiness Directives; GA 8 Airvan (Pty) Ltd Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is superseding Airworthiness Directive (AD) 2010–18–06, which applied to all GA8 Airvan (Pty) Ltd Model GA8 and GA8–TC320 airplanes. AD 2010–18–06 required inspections and a minor design change to the forward slide of the cargo door with corrective action as necessary. Since the FAA issued AD 2010–18–06, the Civil Aviation Safety Authority (CASA), which is the aviation authority for Australia, superseded the previous CASA Australia AD to incorporate more detailed inspections and additional modifications as specified in updated

service information published by the manufacturer. This AD was prompted by reports of in-flight cargo door separation. This AD requires inspections and rework (modifications) of the cargo door with corrective action as necessary. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective June 3, 2024.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of June 3, 2024.

ADDRESSES:

AD Docket: You may examine the AD docket at [regulations.gov](https://www.regulations.gov) under Docket No. FAA-2024-0035; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the mandatory continuing airworthiness information (MCAI), any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

Material Incorporated by Reference:

- For service information contact GA8 Airvan (Pty) Ltd, PO Box 881, Morwell, Victoria 3840, Australia; phone: +61 03 5172 1200; website: [gippsaero.com.au](https://www.gippsaero.com.au); email: TECHPUBS@gippsaero.com.au.

- You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 901 Locust, Kansas City, MO 64106. For information on the availability of this material at the FAA call (817) 222-5110. It is also available at [regulations.gov](https://www.regulations.gov) under Docket No. FAA-2024-0035.

FOR FURTHER INFORMATION CONTACT:

Doug Rudolph, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: (816) 329-4059; email: doug.rudolph@faa.gov.

SUPPLEMENTARY INFORMATION:

Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to supersede AD 2010-18-06, Amendment 39-16419 (75 FR 52253, August 25, 2010) (AD 2010-18-06). AD 2010-18-06 applied to all GA8 Airvan (Pty) Ltd Model GA8 and GA8-TC320 airplanes. AD 2010-18-06 was prompted by MCAI originated by CASA, which is the aviation authority for Australia. CASA Australia issued CASA Australia AD AD/GA8/3 Amdt 2, dated August 11, 2010 (CASA Australia AD/

GA8/3 Amdt 2) to correct an unsafe condition identified as excessive wear in the forward cargo door slide, which could result in an in-flight separation of the cargo door, with possible loss of control of the airplane. CASA Australia AD AD/GA8/3 Amdt 2 was issued to require the actions in service information updated by the manufacturer to remove any ambiguities in the previous revision and provide an improved inspection method and a minor design change to the forward slide of the cargo door (inclusion of a slide backing plate, castellated nut, and split pin).

AD 2010-18-06 required doing all of Action 1 (measuring the groove width of the forward cargo door slide and if it exceeds 0.145 inch at any point along the slide, or is cracked, installing a new slider assembly) and Action 2 (inspecting wear of the forward slide of the cargo door and doing applicable corrective action steps specified in Action 1) of GippsAero Pty. Ltd. Mandatory Service Bulletin SB-GA8-2005-23, Issue 3, dated August 5, 2010. The FAA issued AD 2010-18-06 to address excessive wear in the forward cargo door slide.

The NPRM published in the **Federal Register** on January 23, 2024 (89 FR 4211). The NPRM was prompted by CASA Australia AD AD/GA8/3 amdt 3, dated August 18, 2023 (also referred to as the MCAI). The MCAI states that inspections revealed cases of excessive wear in the forward slide of the cargo door. Excessive wear in the forward slide of the cargo door may result in the cargo door separating from the airplane in flight with potentially catastrophic results. The MCAI requires accomplishing the actions specified in GippsAero Service Bulletin SB-GA8-2005-23, Issue 7, dated May 30, 2023 (GippsAero Service Bulletin SB-GA8-2005-23, Issue 7). This service bulletin includes procedures for revised inspections of the door mechanism, installing a stop on the forward slide of the cargo door and reworking the door slide to suit (accommodate) the track stop installation. Depending on the findings of the inspections, additional actions might be necessary including reworking the door mechanism pivot, upgrading the door operating rod, or fitting a door handle with an integral stop.

The FAA is issuing this AD to address excessive wear in the forward slide of the cargo door. The unsafe condition, if not addressed, could result in the cargo door separating from the airplane during flight, with potential loss of control of the airplane.

You may examine the MCAI in the AD docket at [regulations.gov](https://www.regulations.gov) under Docket No. FAA-2024-0035.

Discussion of Final Airworthiness Directive

Comments

The FAA received no comments on the NPRM or on the determination of the costs.

Conclusion

These products have been approved by the aviation authority of another country and are approved for operation in the United States. Pursuant to the FAA's bilateral agreement with this State of Design Authority, it has notified the FAA of the unsafe condition described in the MCAI referenced above. The FAA reviewed the relevant data and determined that air safety requires adopting this AD as proposed. Accordingly, the FAA is issuing this AD to address the unsafe condition on these products. This AD is adopted as proposed in the NPRM.

Related Service Information Under 1 CFR Part 51

The FAA reviewed GippsAero Service Bulletin SB-GA8-2005-23, Issue 8, dated October 11, 2023 (GippsAero SB-GA8-2005-23, Issue 8). This service information specifies procedures for installing a backing plate on the forward slide of the cargo door; inspecting the forward slide of the cargo door for excessive wear; inspecting the cargo door latching mechanism for contact between the operating rod and door handle pivot post, inspecting the threaded studs and rod ends at both ends of the operating rod for bending, and checking the cargo door handle engagement with the catch; reworking the cargo door handle pivot post; reworking the door operating rod; inspecting the door handle to determine if an integrated stop is installed and checking for excessive play; and inspecting the center rail of the cargo door to determine if an aft stop is installed, installing an aft stop, and reworking the center rail of the cargo door to accommodate the track stop.

This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in **ADDRESSES**.

Differences Between This AD and the MCAI

The MCAI applicability is Gippsland Aeronautics Model GA8 Series airplanes, all serial numbers. The applicability in this AD is GA8 Airvan (Pty) Ltd Model GA8 and GA8-TC320

airplanes because the FAA type certificate specifies GA8 Airvan (Pty) Ltd instead of Gippsland Aeronautics and specifies Model GA8 and GA8-TC320 airplanes instead of Model GA8 Series airplanes.

The MCAI requires doing the actions in Gippsland Aeronautics mandatory service bulletin SB-GA8-2005-23 Issue 7, dated May 30, 2023. This AD requires

doing the actions in GippsAero SB-GA8-2005-23, Issue 8. After the MCAI was published, the manufacturer issued GippsAero SB-GA8-2005-23, Issue 8, which was revised to provide clarification regarding the actions and compliance schedule. The title page of GippsAero SB-GA8-2005-23, Issue 8, specifies GippsAero instead of Gippsland Aeronautics.

Costs of Compliance

The FAA estimates that this AD affects 61 airplanes of U.S. registry.

The FAA estimates the following costs to comply with this AD. The corresponding letter and number in parenthesis refer to the specific paragraph in GippsAero SB-GA8-2005-23, Issue 8.

ESTIMATED COSTS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Installing forward cargo door slide backing plate (A1).	0.50 work-hour × \$85 per hour = \$42.50.	\$175	\$217.50	\$13,267.50.
Inspecting forward cargo door slide wear (A2).	0.25 work-hour × \$85 per hour = \$21.25 per inspection cycle.	0	\$21.25 per inspection cycle ...	\$1,296.25 per inspection cycle.
Inspecting cargo door latching mechanism (B1).	1 work-hour × \$85 per hour = \$85 per inspection cycle.	0	\$85 per inspection cycle	\$5,185 per inspection cycle.
Inspecting cargo door handle and inspecting for excessive play (C).	0.75 work-hour × \$85 per hour = \$63.75.	0	\$63.75	\$3,88.75.
Inspecting cargo door center rail (D1).	1 work-hour × \$85 per hour = \$85.	0	\$85	\$5,185.

The FAA estimates the following costs to do any necessary actions that would be required based on the results of the inspections. The agency has no

way of determining the number of airplanes that might need these actions. The corresponding letter and number in parenthesis refer to the specific

paragraph in GippsAero SB-GA8-2005-23, Issue 8.

ON-CONDITION COSTS

Action	Labor cost	Parts cost	Cost per product
Inspecting/replacing forward cargo door slide (A1, Steps 2 through 4), corrective action for (A2).	0.50 work-hour × \$85 per hour = \$42.50 ...	\$175	\$217.50.
Reworking cargo door pivot (B2) and reworking/replacing door operating rod assembly (B3).	2 work-hours × \$85 per hour = \$170	630	800.
Replacing door handle/handle bush (C)	1 work-hour × \$85 per hour = \$85	267	352.
Replacing cargo door center rail/slide-center and backing plate (D1) and reworking cargo door center rail and backing plate (D2).	2 work-hours × \$85 per hour = \$170	152	322.

The FAA has included all known costs in its cost estimate. According to the manufacturer, however, some of the costs of this AD may be covered under warranty, thereby reducing the cost impact on affected operators.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency’s authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA

with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

The FAA has determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a “significant regulatory action” under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Amendment

Accordingly, under the authority delegated to me by the Administrator,

the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by:

■ a. Removing Airworthiness Directive 2010–18–06, Amendment 39–16419 (75 FR 52253, August 25, 2010); and

■ b. Adding the following new airworthiness directive:

2024–07–07 GA 8 Airvan (Pty) Ltd:
Amendment 39–22728; Docket No.

FAA–2024–0035; Project Identifier MCAI–2023–00986–A.

(a) Effective Date

This airworthiness directive (AD) is effective June 3, 2024.

(b) Affected ADs

This AD replaces AD 2010–18–06, Amendment 39–16419 (75 FR 52253, August 25, 2010).

(c) Applicability

This AD applies to GA 8 Airvan (Pty) Ltd Model GA8 and GA8–TC320 airplanes, all serial numbers, certificated in any category.

(d) Subject

Joint Aircraft System Component (JASC) Code 5230, Cargo/Baggage Doors

(e) Unsafe Condition

This AD was prompted by reports of in-flight cargo door separation. The FAA is issuing this AD to detect and correct excessive wear in the forward cargo door slide, which could result in an in-flight separation of the cargo door, with possible loss of control of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Required Actions

Do the applicable actions specified in Table 1 to paragraph (g) of this AD at the times in Table 1 to paragraph (g) of this AD, in accordance with the Accomplishment Instructions of GippsAero Service Bulletin SB–GA8–2005–23, Issue 8, dated October 11, 2023 (GippsAero SB–GA8–2005–23, Issue 8).

TABLE 1 TO PARAGRAPH (g)

Paragraphs in Accomplishment Instructions of GippsAero SB–GA8–2005–23, Issue 8	Action	Compliance time
12.1, A1, steps 1 and 2, for backing plate inspection, except where Figure 1 in step 1 specifies to remove and discard the vertical bolt, remove the vertical bolt from service. Steps 3 through 7, if a backing plate is not installed.	Inspect for the existence of a backing plate on the forward slide of the cargo door. If a backing plate is not installed, install a backing plate on the forward slide of the cargo door, measure the groove width of the forward slide, and replace the slide if it exceeds 0.145 inch at any point or is cracked or worn beyond limits.	Inspect within 50 hours time-in-service (TIS) or 2 months after the effective date of this AD, whichever occurs first. Install, measure, and replace before further flight after the inspection.
12.2, A2, steps 1 and 2 for the inspection 12.2, A2, step 3 or 4, and 12.1, A1, steps 2 through 4, for the follow-on inspection and replacement.	Inspect for wear of the forward slide of the cargo door by inserting a slide gauge or feeler gauge to measure the clearance between the forward slide and the cargo door track. If a gap is found, measure the groove width of the forward slide and replace the slide if the groove width exceeds 0.145 inch at any point or is cracked or worn beyond limits.	Inspect for wear within 100 hours TIS or 2 months after the effective date of this AD, whichever occurs first, and thereafter at intervals not to exceed 100 hours TIS or 12 months, whichever occurs first after the most recent inspection. Measure the groove width and replace the slide before further flight after each inspection as necessary.
12.3, B1, steps 1 through 6 for the inspections 12.3, B1, steps 2, 3i, and 3ii; 12.4, B2, steps 1 through 5; and 12.5, B3, steps 1 through 12 for the corrective actions.	Inspect the cargo door mechanism for contact between the operating rod and cargo door handle pivot post, inspect the threaded studs and rod ends at both ends of the operating rod for bending, and inspect the cargo door handle engagement with the catch. Perform all applicable corrective actions.	Inspect within 50 hours TIS or 2 months after the effective date of this AD, whichever occurs first and thereafter at intervals not to exceed 100 hours TIS or 12 months, whichever occurs first after the most recent inspection. Perform all applicable corrective actions before further flight.
12.6, C, steps 1 through 6	Inspect the cargo door handle to determine if an integrated stop is installed and if an integrated stop is not installed, install a cargo door handle with an integrated stop. Inspect the cargo door handle for beyond normal play and replace the handle bush if the door handle has beyond normal play.	Within 150 hours TIS or 4 months after the effective date of this AD, whichever occurs first. Perform the installation and replacement, as necessary, before further flight after the inspection.
12.7, D1, steps 1 through 10 for the center rail cargo door inspection and installation. 12.8, D2, steps 1 through 2, for any necessary follow-on rework.	Inspect the center rail of the cargo door to determine if a center rail aft stop is installed and if a center rail aft stop is not installed, install an aft stop before further flight.	Within 50 hours TIS or 2 months after the effective date of this AD, whichever occurs first.

(h) Alternative Methods of Compliance (AMOCs)

The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as

appropriate. If sending information directly to the manager of the International Validation Branch, mail it to the address identified in paragraph (i)(2) of this AD or email to: 9-AVS-AIR-730-AMOC@faa.gov. If mailing information, also submit information by email. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager

of the local Flight Standards District Office/certificate holding district office.

(i) Additional Information

(1) Refer to Civil Aviation Safety Authority (CASA) Australia AD AD/GA8/3 amdt 3, dated August 18, 2023, for related information. This CASA Australia AD may be

found in the AD docket at regulations.gov under Docket No. FAA-2024-0035.

(2) For more information about this AD, contact Doug Rudolph, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: (816) 329-4059; email: doug.rudolph@faa.gov.

(j) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) GippsAero Service Bulletin SB-GA8-2005-23, Issue 8, dated October 11, 2023.

(ii) [Reserved]

(3) For service information contact GA8 Airvan (Pty) Ltd, PO Box 881, Morwell, Victoria 3840, Australia; phone: +61 03 5172 1200; website: gippsaero.com.au; email: TECHPUBS@gippsaero.com.au.

(4) You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 901 Locust, Kansas City, MO 64106. For information on the availability of this material at the FAA, call (817) 222-5110.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit www.archives.gov/federal-register/cfr/ibr-locationsoremailfr.inspection@nara.gov.

Issued on March 29, 2024.

Victor Wicklund,

Deputy Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2024-09087 Filed 4-26-24; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 91

[Docket No. FAA-2023-1836; Amdt. No. 91-371B]

RIN 2120-AL70

Inclusion of Additional Automatic Dependent Surveillance-Broadcast (ADS-B) Out Technical Standard Orders; Incorporation by Reference; Correction

AGENCY: Federal Aviation Administration (FAA), U.S. Department of Transportation (DOT).

ACTION: Direct final rule; correction.

SUMMARY: On October 17, 2023, the Federal Aviation Administration (FAA) published the subject direct final rule in the Federal Register, confirmed on November 29, 2023. In that direct final rule, the FAA redesignated two paragraphs in one section of the Code of

Federal Regulations but failed to amend cross-references to those paragraphs. This document corrects those errors.

DATES: This correction is effective April 29, 2024.

FOR FURTHER INFORMATION CONTACT: Juan Sebastian Yanguas, Airspace Rules & Regulations, AJV-P21, Federal Aviation Administration, 800 Independence Avenue SW, Washington, DC 20591; telephone (202) 267-8783; email Juan.S.Yanguas@faa.gov.

SUPPLEMENTARY INFORMATION: On October 17, 2023, the Federal Aviation Administration (FAA) published the direct final rule, Inclusion of Additional Automatic Dependent Surveillance-Broadcast (ADS-B) Out Technical Standard Orders; Incorporation by Reference, in the Federal Register.¹ In that direct final rule the FAA incorporated by reference two new Technical Standard Orders (TSOs), and three RTCA documents—TSO-C166c, TSO-C154d, section 2 of RTCA DO-260C, RTCA DO-260C Change 1, and section 2 of RTCA DO-282C into 14 CFR 91.225 and 91.227. The FAA responded to comments and confirmed the direct final rule on November 28, 2023.²

In the direct final rule, the FAA amended § 91.225 by redesignating paragraphs (h) and (i) as set out in the following redesignation table:

Table with 2 columns: Old paragraph, New paragraph. Row 1: paragraph (h) redesignated to paragraph (i). Row 2: paragraph (i) redesignated to paragraph (h).

While the FAA discussed the redesignation of paragraphs (h) and (i), it failed to revise cross-references to those paragraphs appearing elsewhere in § 91.225. This document corrects those errors.

List of Subjects in 14 CFR Part 91

Air traffic control, Aircraft, Airports, Aviation safety, Incorporation by reference, Transportation.

Accordingly, 14 CFR part 91 is corrected by making the following correcting amendments:

PART 91—GENERAL OPERATING AND FLIGHT RULES

■ 1. The authority citation for part 91 continues to read as follows:

Authority: 49 U.S.C. 106(f), 106(g), 40101, 40103, 40105, 40113, 40120, 44101, 44111, 44701, 44704, 44709, 44711, 44712, 44715, 44716, 44717, 44722, 46306, 46315, 46316, 46504, 46506-46507, 47122, 47508, 47528-

¹ 88 FR 71468; FR Doc. 2023-22710.

² 88 FR 83022.

47531, 47534, Pub. L. 114-190, 130 Stat. 615 (49 U.S.C. 44703 note); articles 12 and 29 of the Convention on International Civil Aviation (61 Stat. 1180), (126 Stat. 11).

■ 2. Amend § 91.225 by revising the introductory text of paragraphs (d) and (f) to read as follows:

§ 91.225 Automatic Dependent Surveillance-Broadcast (ADS-B) Out equipment and use.

* * * * *

(d) After January 1, 2020, except as prohibited in paragraph (h)(2) of this section or unless otherwise authorized by ATC, no person may operate an aircraft in the following airspace unless the aircraft has equipment installed that meets the requirements in paragraph (b) of this section:

* * * * *

(f) Except as prohibited in paragraph (h)(2) of this section, each person operating an aircraft equipped with ADS-B Out must operate this equipment in the transmit mode at all times unless—

* * * * *

Issued under the authority of 49 U.S.C. 106(f) in Washington, DC.

Brandon Roberts,

Executive Director, Office of Rulemaking.

[FR Doc. 2024-08885 Filed 4-26-24; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF COMMERCE

Bureau of Industry and Security

15 CFR Parts 740, 744, 746, and 762

[Docket No. 240423-0115]

RIN 0694-AJ59

Amendment to Existing Controls on Russia and Belarus Under the Export Administration Regulations (EAR) Adding New License Exception Medical Devices (MED); Corrections

AGENCY: Bureau of Industry and Security, Department of Commerce.

ACTION: Final rule.

SUMMARY: In this final rule, the Bureau of Industry and Security (BIS) makes changes to the Russia and Belarus sanctions under the Export Administration Regulations (EAR) to add a new license exception for EAR99 medical devices and related parts, components, accessories, and attachments for use in or with medical devices that are destined for both countries and the temporarily occupied Crimea region of Ukraine, or the covered regions of Ukraine. The purpose of this final rule is to authorize under a license