

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by:
 ■ a. Removing Airworthiness Directive 2019–16–11, Amendment 39–19714 (84 FR 45061, August 28, 2019); and
 ■ b. Adding the following new Airworthiness Directive:

2024–01–08 Airbus SAS: Amendment 39–22659; Docket No. FAA–2023–2136; Project Identifier MCAI–2023–00759–T.

(a) Effective Date

This airworthiness directive (AD) is effective March 21, 2024.

(b) Affected ADs

This AD replaces AD 2019–16–11, Amendment 39–19714 (84 FR 45061, August 28, 2019) (AD 2019–16–11).

(c) Applicability

This AD applies to Airbus SAS Model A300 F4–605R and F4–622R airplanes, certificated in any category, as identified in European Union Aviation Safety Agency (EASA) AD 2023–0117, dated June 13, 2023 (EASA AD 2023–0117).

(d) Subject

Air Transport Association (ATA) of America Code: 52, Doors.

(e) Unsafe Condition

This AD was prompted by a report of two adjacent frame forks that were found cracked on the aft lower deck cargo door (LDCD) of two airplanes during scheduled maintenance, and a determination that certain compliance times need to be revised. The FAA is also issuing this AD to address the susceptibility of the frame forks of affected LDCDs to develop cracks, which could lead to additional rupture of one or more LDCD frame forks, compromising the structural integrity of the LDCD and therefore of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Requirements

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, EASA AD 2023–0117.

(h) Exceptions to EASA AD 2023–0117

(1) Where EASA AD 2023–0117 refers to its effective date, this AD requires using the effective date of this AD.

(2) Where Table 2 of EASA AD 2023–0117 refers to the effective date of EASA AD 2015–0152R1, dated May 23, 2017, this AD requires using November 5, 2018 (the effective date of AD 2018–20–06, Amendment 39–19440 (83 FR 49265, October 1, 2018)).

(3) Where Table 2 of EASA AD 2023–0117 refers to the effective date of EASA AD 2015–0152, dated July 24, 2015, this AD requires using January 26, 2017 (the effective date of AD 2016–25–03, Amendment 39–18729 (81 FR 93801, December 22, 2016)).

(4) Where paragraph (6) of EASA AD 2023–0117 uses the phrase “before next flight, contact Airbus for approved corrective action instructions, and within the compliance time specified therein, accomplish those instructions accordingly,” this AD requires replacing those words with “repair cracking before further flight using a method approved by the Manager, International Validation Branch, FAA; or EASA; or Airbus SAS’s EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.”
 (5) This AD does not adopt the “Remarks” section of EASA AD 2023–0117.

(i) No Reporting Requirement

Although the service information referenced in EASA AD 2023–0117 specifies to submit certain information to the manufacturer, this AD does not include that requirement.

(j) Additional AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the International Validation Branch, send it to the attention of the person identified in paragraph (k) of this AD. Information may be emailed to: 9-AVS-AIR-730-AMOC@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(2) *Contacting the Manufacturer:* For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, International Validation Branch, FAA; or EASA; or Airbus SAS’s EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(3) *Required for Compliance (RC):* Except as required by paragraphs (h)(4) and (j)(2) of this AD, if any service information contains procedures or tests that are identified as RC, those procedures and tests must be done to comply with this AD; any procedures or tests that are not identified as RC are

recommended. Those procedures and tests that are not identified as RC may be deviated from using accepted methods in accordance with the operator’s maintenance or inspection program without obtaining approval of an AMOC, provided the procedures and tests identified as RC can be done and the airplane can be put back in an airworthy condition. Any substitutions or changes to procedures or tests identified as RC require approval of an AMOC.

(k) Additional Information

For more information about this AD, contact Dan Rodina, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 206–231–3225; email dan.rodina@faa.gov.

(l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) European Union Aviation Safety Agency (EASA) AD 2023–0117, dated June 13, 2023.

(ii) [Reserved]

(3) For EASA AD 2023–0117, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu; website easa.europa.eu. You may find this EASA AD on the EASA website ad.easa.europa.eu.

(4) You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th Street, Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit www.archives.gov/federal-register/cfr/ibr-locations, or email fr.inspection@nara.gov.

Issued on January 9, 2024.

Caitlin Locke,

Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2024–03080 Filed 2–14–24; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA–2023–2145; Project Identifier MCAI–2023–00358–T; Amendment 39–22660; AD 2024–01–09]

RIN 2120–AA64

Airworthiness Directives; Bombardier, Inc., Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for all Bombardier, Inc., Model BD-100-1A10 airplanes. This AD was prompted by a report of a steering control unit (SCU) filter plate connector that does not meet the certification requirements for exposure of electronic components to high intensity radiated field environments, which could result in malfunction of the nose wheel steering (NWS) system. This AD requires determining if the SCU is an affected SCU, replacing all affected SCUs, and rigging and testing the NWS control. This AD also prohibits installing an affected SCU on any airplane. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective March 21, 2024.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of March 21, 2024.

ADDRESSES:

AD Docket: You may examine the AD docket at *regulations.gov* under Docket No. FAA-2023-2145; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the mandatory continuing airworthiness information (MCAI), any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

Material Incorporated by Reference:

- For service information identified in this final rule, contact Bombardier Business Aircraft Customer Response Center, 400 Côte-Vertu Road West, Dorval, Québec H4S 1Y9, Canada; telephone 514-855-2999; email *ac.yul@aero.bombardier.com*; website *bombardier.com*.
- You may view this service information at the FAA, Airworthiness

Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. It is also available at *regulations.gov* under Docket No. FAA-2023-2145.

FOR FURTHER INFORMATION CONTACT:

William Reisenauer, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516-228-7301; email *9-avs-nyaco-cos@faa.gov*.

SUPPLEMENTARY INFORMATION:

Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to all Bombardier, Inc., Model BD-100-1A10 airplanes. The NPRM published in the **Federal Register** on November 13, 2023 (88 FR 77538). The NPRM was prompted by AD CF-2023-13, dated February 24, 2023, issued by Transport Canada, which is the aviation authority for Canada (referred to after this as the MCAI). The MCAI states that the manufacturer of SCU part number (P/N) 46000-01 introduced a new filter plate connector that does not meet the certification requirements related to the susceptibility of electronic components to high intensity radiated field. According to the MCAI, this non-compliant filter plate connector, if not replaced, could result in a malfunction of the NWS system and cause un-commanded steering or lateral excursion from the runway.

In the NPRM, the FAA proposed to require determining if the SCU is an affected SCU, replacing all affected SCUs, and rigging and testing the NWS control. In the NPRM, the FAA also proposed to prohibit installing an affected SCU on any airplane. The FAA is issuing this AD to address the unsafe condition on these products.

You may examine the MCAI in the AD docket at *regulations.gov* under Docket No. FAA-2023-2145.

Discussion of Final Airworthiness Directive

Comments

The FAA received no comments on the NPRM or on the determination of the cost to the public.

Conclusion

This product has been approved by the aviation authority of another country and is approved for operation in the United States. Pursuant to the FAA's bilateral agreement with this State of Design Authority, it has notified the FAA of the unsafe condition described in the MCAI referenced above. The FAA reviewed the relevant data and determined that air safety requires adopting this AD as proposed. Accordingly, the FAA is issuing this AD to address the unsafe condition on this product. Except for minor editorial changes, this AD is adopted as proposed in the NPRM. None of the changes will increase the economic burden on any operator.

Related Service Information Under 1 CFR Part 51

The FAA reviewed Bombardier Service Bulletin 100-32-34, dated October 18, 2021, and Bombardier Service Bulletin 350-32-010, dated October 18, 2021. This service information specifies procedures for determining the serial number of SCU P/N 46000-01, replacing any affected SCU, and rigging and testing the NWS control. These documents are distinct since they apply to different airplane serial numbers. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

Costs of Compliance

The FAA estimates that this AD affects 725 airplanes of U.S. registry. The FAA estimates the following costs to comply with this AD:

ESTIMATED COSTS FOR REQUIRED ACTIONS

Labor cost	Parts cost	Cost per product	Cost on U.S. operators
9 work-hours × \$85 per hour = \$765	\$44,950	\$45,715	\$33,143,375

The FAA has included all known costs in its cost estimate. According to the manufacturer, however, some of the costs of this AD may be covered under warranty, thereby reducing the cost impact on affected operators.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII:

Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section

44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a “significant regulatory action” under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

2024-01-09 Bombardier, Inc.: Amendment 39-22660; Docket No. FAA-2023-2145; Project Identifier MCAI-2023-00358-T.

(a) Effective Date

This airworthiness directive (AD) is effective March 21, 2024.

(b) Affected ADs

None.

(c) Applicability

This AD applies to all Bombardier, Inc., Model BD-100-1A10 airplanes, certificated in any category.

(d) Subject

Air Transport Association (ATA) of America Code 32, Landing gear.

(e) Unsafe Condition

This AD was prompted by a report of a steering control unit (SCU) filter plate connector that does not meet the certification requirements for exposure of electronic components to high intensity radiated field environments. The FAA is issuing this AD to prevent malfunction of the nose wheel steering (NWS) system. The unsafe condition, if not addressed, could result in uncommanded steering or lateral runway excursion.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Inspection or Records Review

Within 24 months after the effective date of this AD: Inspect the SCU to determine if SCU part number (P/N) 46000-01 with a serial number listed in Figure 1 to paragraph (g) of this AD is installed on the airplane. A review of the airplane maintenance records is acceptable in lieu of the inspection if the SCU P/N and serial number can be conclusively determined from that review. If an SCU P/N 46000-01 with a serial number listed in Figure 1 to paragraph (g) of this AD is not installed on the airplane, or if the SCU identification plate is marked with SB100-32-030, then no further action is required by this AD; however, the installation prohibition in paragraph (i) of this AD still applies.

Figure 1 to Paragraph (g)—Affected SCU Serial Numbers

06-188, 16042190, 16047519, 16047521, 16047522, 17053050, 17053052, 17054845, 17054846, 17054847, 17055273, 17055274, 18057933, 18057934, 18058184, 18058185, 18058599, 18058600, 18058601, 18058602, 18058884, 18058885, 18058886, 18059357, 18059358, 18059359, 18059360, 18059814, 18059815, 18059816, 18059817, 18060201, 18060202, 18060582, 18060789, 18060982, 18060983, 18061292, 18061572, 18061573, 18061770, 18061771, 18061880, 18061881, 18061995, 18061996, 18062731, 18062732, 18062733, 18062734, 18063183, 18063184, 18063520, 18063521, 18064776, 18064777, 18064778, 18064779, 18065323, 18065324, 18065325, 18065326, 18065327, 18065331, 18065332, 19068045, 19068046, 19068047, 19068048, 19068049, 19068050, 19068051, 19068052, 19068053, 19068054, 19068055, 19068056, 19068057, 19068058, 19068059, 19068060, 19068061, 19068062, 19068063, 19068064, 19068065, 19068066, 19068067, 19068068, 19068069, 19068070, 19068071, 19068072, 19068073, 19068074, 19072062, 19072063, 19072067, 19072068, 19072069, 19072070, 19072071, 19072072, 19072073, 19072074, 19072075, 19072076, 19072077, 19072078, 19075712, 19075713, 19075714, 19075715, 19075716, 19075718, 19075719, 20077461, 20077462, 20077463, 20077464, 20077465, 20077466, 20077467, 20077470, 20077472, 20077473, 20077474, 20077476, 20077477, 20077478, 20077480.

(h) Replacement

For airplanes with SCU P/N 46000-01 with a serial number listed in Figure 1 to paragraph (g) of this AD installed and not marked on the SCU identification plate with SB100-32-030: Within 24 months after the effective date of this AD, replace the SCU and rig and test the NWS control, in accordance with the instructions in paragraph (h)(1), (2), or (3) of this AD, as applicable.

(1) For airplane serial numbers 20001 through 20500 inclusive: Steps 2.C.(1) and 2.C.(3) of Section 2.C., "Part B—Modification," and Section 2.D., "Testing,"

of the Accomplishment Instructions in Bombardier Service Bulletin 100-32-34, dated October 18, 2021.

(2) For airplane serial numbers 20501 through 20893 inclusive: Steps 2.C.(1) and 2.C.(3) of Section 2.C., "Part B—Modification," and Section 2.D., "Testing," of the Accomplishment Instructions in Bombardier Service Bulletin 350-32-010, dated October 18, 2021.

(3) For airplane serial numbers 20894 and larger: A method approved by the Manager, International Validation Branch, FAA; or Transport Canada; or Bombardier, Inc.'s Transport Canada Design Approval

Organization (DAO). If approved by the DAO, the approval must include the DAO-authorized signature.

(i) Parts Installation Prohibition

As of the effective date of this AD, do not install SCU P/N 46000-01 on any airplane if the serial number of the SCU is listed in figure 1 to paragraph (g) of this AD, unless the SCU identification plate has been marked with SB100-32-030.

(j) Additional AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the manager of the International Validation Branch, mail it to the address identified in paragraph (k)(2) of this AD. Information may be emailed to: 9-avs-nyaco-cos@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(2) *Contacting the Manufacturer*: For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, International Validation Branch, FAA; or Transport Canada or Bombardier, Inc.'s Transport Canada Design Approval Organization (DAO). If approved by the DAO, the approval must include the DAO-authorized signature.

(k) Additional Information

(1) Refer to Transport Canada AD CF-2023-13, dated February 24, 2023, for related information. This Transport Canada AD may be found in the AD docket at [regulations.gov](https://www.regulations.gov) under Docket No. FAA-2023-2145.

(2) For more information about this AD, contact William Reisenauer, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516-228-7301; email 9-avs-nyaco-cos@faa.gov.

(l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) Bombardier Service Bulletin 100-32-34, dated October 18, 2021.

(ii) Bombardier Service Bulletin 350-32-010, dated October 18, 2021.

(3) For service information identified in this AD, contact Bombardier Business Aircraft Customer Response Center, 400 Côte-Vertu Road West, Dorval, Québec H4S 1Y9, Canada; telephone 514-855-2999; email ac.yul@aero.bombardier.com; website [bombardier.com](https://www.bombardier.com).

(4) You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit www.archives.gov/federal-register/cfr/ibr-locations or email fr.inspection@nara.gov.

Issued on January 9, 2024.

Caitlin Locke,

Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2024-03047 Filed 2-14-24; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 97

[Docket No. 31530; Amdt. No. 4099]

Standard Instrument Approach Procedures, and Takeoff Minimums and Obstacle Departure Procedures; Miscellaneous Amendments

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: This rule establishes, amends, suspends, or removes Standard Instrument Approach Procedures (SIAPs) and associated Takeoff Minimums and Obstacle Departure Procedures (ODPs) for operations at certain airports. These regulatory actions are needed because of the adoption of new or revised criteria, or because of changes occurring in the National Airspace System, such as the commissioning of new navigational facilities, adding new obstacles, or changing air traffic requirements. These changes are designed to provide safe and efficient use of the navigable airspace and to promote safe flight operations under instrument flight rules at the affected airports.

DATES: This rule is effective February 15, 2024. The compliance date for each SIAP, associated Takeoff Minimums, and ODP is specified in the amendatory provisions.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of February 15, 2024.

ADDRESSES: Availability of matters incorporated by reference in the amendment is as follows:

For Examination

1. U.S. Department of Transportation, Docket Ops-M30, 1200 New Jersey Avenue SE, West Bldg., Ground Floor, Washington, DC 20590-0001.

2. The FAA Air Traffic Organization Service Area in which the affected airport is located;

3. The office of Aeronautical Information Services, 6500 South MacArthur Blvd., Oklahoma City, OK 73169 or,

4. The National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit www.archives.gov/federal-register/cfr/ibr-locations or email fr.inspection@nara.gov.

Availability

All SIAPs and Takeoff Minimums and ODPs are available online free of charge. Visit the National Flight Data Center at nfdc.faa.gov to register. Additionally, individual SIAP and Takeoff Minimums and ODP copies may be obtained from the FAA Air Traffic Organization Service Area in which the affected airport is located.

FOR FURTHER INFORMATION CONTACT:

Thomas J. Nichols, Flight Procedures and Airspace Group, Flight Technologies and Procedures Division, Flight Standards Service, Federal Aviation Administration, Mailing Address: FAA Mike Monroney Aeronautical Center, Flight Procedures and Airspace Group, 6500 South MacArthur Blvd., STB Annex, Bldg. 26, Room 217, Oklahoma City, OK 73099. Telephone (405) 954-1139.

SUPPLEMENTARY INFORMATION: This rule amends 14 CFR part 97 by establishing, amending, suspending, or removes SIAPs, Takeoff Minimums and/or ODPS. The complete regulatory description of each SIAP and its associated Takeoff Minimums or ODP for an identified airport is listed on FAA form documents which are incorporated by reference in this amendment under 5 U.S.C. 552(a), 1 CFR part 51, and 14 CFR 97.20. The applicable FAA Forms 8260-3, 8260-4, 8260-5, 8260-15A, 8260-15B, when required by an entry on 8260-15A, and 8260-15C.

The large number of SIAPs, Takeoff Minimums and ODPs, their complex nature, and the need for a special format make publication in the **Federal Register** expensive and impractical. Further, pilots do not use the regulatory text of the SIAPs, Takeoff Minimums or ODPs, but instead refer to their graphic depiction on charts printed by publishers or aeronautical materials. Thus, the advantages of incorporation by reference are realized and publication of the complete description of each SIAP, Takeoff Minimums and ODP listed on FAA form documents is unnecessary. This amendment provides the affected CFR sections and specifies the types of SIAPs, Takeoff Minimums and ODPs with their applicable effective dates. This amendment also identifies the airport and its location, the procedure, and the amendment number.