

Continuing Operational Safety, at the address identified in paragraph (j)(2) of this AD or email to: 9-avs-nyaco-cos@faa.gov. If mailing information, also submit information by email. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(2) *Contacting the Manufacturer:* For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, New York ACO Branch, FAA; or Transport Canada or Bombardier, Inc.'s Transport Canada Design Approval Organization (DAO). If approved by the DAO, the approval must include the DAO-authorized signature.

(j) Additional Information

(1) Refer to Transport Canada AD CF-2022-15, dated April 7, 2022, for related information. This Transport Canada AD may be found in the AD docket at [regulations.gov](https://www.regulations.gov) under Docket No. FAA-2023-0169.

(2) For more information about this AD, contact Gabriel Kim, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516-228-7300; email 9-avs-nyaco-cos@faa.gov.

(k) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) Task 32-34-00-101, "Functional Test of the Landing-Gear Emergency Extension," of Part 2, "Airworthiness Limitations," of the Bombardier Global Express Time Limit/Maintenance Check manual (TLMC), Publication No. BD-700 TLMC, Revision 34, dated March 1, 2022.

NOTE 1 TO PARAGRAPH (K)(2)(I): For obtaining the tasks specified in paragraphs (k)(2)(i) through (iii) of this AD for Bombardier Global Express TLMC, Publication No. BD-700 TLMC, Revision 34, dated March 1, 2022, use Document Identification No. GL 700 TLMC.

(ii) Task 32-43-25-101, "Operational Test of the Brake Shutoff Valve," of Part 2, "Airworthiness Limitations," of the Bombardier Global Express TLMC, Publication No. BD-700 TLMC, Revision 34, dated March 1, 2022.

(iii) Task 52-11-00-106, "Visual Check of the Passenger-Door Vent-Flap Mechanism," of Part 2, "Airworthiness Limitations," of the Bombardier Global Express TLMC, Publication No. BD-700 TLMC, Revision 34, dated March 1, 2022.

(iv) Task 32-34-00-101, "Functional Test of the Landing-Gear Emergency Extension," of Part 2, "Airworthiness Limitations," of the Bombardier Global Express XRS TLMC, Publication No. BD-700 XRS TLMC, Revision 21, dated March 1, 2022.

NOTE 2 TO PARAGRAPH (K)(2)(IV): For obtaining the tasks specified in paragraphs (k)(2)(iv) through (vi) of this AD for Bombardier Global Express XRS TLMC, Publication No. BD-700 XRS TLMC, use Document Identification No. GL XRS TLMC.

(v) Task 32-43-25-101, "Operational Test of the Brake Shutoff Valve," of Part 2, "Airworthiness Limitations," of the Bombardier Global Express XRS TLMC, Publication No. BD-700 XRS TLMC, Revision 21, dated March 1, 2022.

(vi) Task 52-11-00-106, "Visual Check of the Passenger-Door Vent-Flap Mechanism," of Part 2, "Airworthiness Limitations," of the Bombardier Global Express XRS TLMC, Publication No. BD-700 XRS TLMC, Revision 21, dated March 1, 2022.

(vii) Task 32-34-00-101, "Functional Test of the Landing-Gear Emergency Extension," of Part 2, "Airworthiness Limitations," of the Bombardier Global 5000 TLMC, Publication No. BD-700 TLMC, Revision 25, dated March 1, 2022.

NOTE 3 TO PARAGRAPH (K)(2)(VII): For obtaining the tasks specified in paragraphs (k)(2)(vii) through (ix) of this AD for Bombardier Global 5000 TLMC, Publication No. BD-700 TLMC, use Document Identification No. GL 5000 TLMC.

(viii) Task 32-43-25-101, "Operational Test of the Brake Shutoff Valve," of Part 2, "Airworthiness Limitations," of the Bombardier Global 5000 TLMC, Publication No. BD-700 TLMC, Revision 25, dated March 1, 2022.

(ix) Task 52-11-00-106, "Visual Check of the Passenger-Door Vent-Flap Mechanism," of Part 2, "Airworthiness Limitations," of the Bombardier Global 5000 TLMC, Publication No. BD-700 TLMC, Revision 25, dated March 1, 2022.

(x) Task 32-34-00-101, "Functional Test of the Landing-Gear Emergency Extension," of Part 2, "Airworthiness Limitations," of the Bombardier Global 5000 Featuring Global Vision Flight Deck (GVFD) TLMC, Publication No. GL 5000 GVFD TLMC, Revision 15, dated March 1, 2022.

NOTE 4 TO PARAGRAPH (K)(2)(X): For obtaining the tasks specified in paragraphs (k)(2)(x) through (xii) of this AD for Bombardier Global 5000 Featuring GVFD TLMC, Publication No. GL 5000 GVFD TLMC, use Document Identification No. GL 5000 GVFD TLMC.

(xi) Task 32-43-25-101, "Operational Test of the Brake Shutoff Valve," of Part 2, "Airworthiness Limitations," of the Bombardier Global 5000 Featuring GVFD TLMC, Publication No. GL 5000 GVFD TLMC, Revision 15, dated March 1, 2022.

(xii) Task 52-11-00-106, "Visual Check of the Passenger-Door Vent-Flap Mechanism," of Part 2, "Airworthiness Limitations," of the Bombardier Global 5000 Featuring GVFD, Publication No. GL 5000 GVFD TLMC, Revision 15, dated March 1, 2022.

(xiii) Task 32-34-00-101, "Functional Test of the Landing-Gear Emergency Extension," of Part 2, "Airworthiness Limitations," of the Bombardier Global 6000 TLMC, Publication No. GL 6000 TLMC, Revision 15, dated March 1, 2022.

NOTE 5 TO PARAGRAPH (K)(2)(XIII): For obtaining the tasks specified in paragraphs (xiii) through (xv) of this AD for Bombardier Global 6000 TLMC, use Document Identification No. GL 6000 TLMC.

(xiv) Task 32-43-25-101, "Operational Test of the Brake Shutoff Valve," of Part 2, "Airworthiness Limitations," of the Bombardier Global 6000 TLMC, Publication

No. GL 6000 TLMC, Revision 15, dated March 1, 2022.

(xv) Task 52-11-00-106, "Visual Check of the Passenger-Door Vent-Flap Mechanism," of Part 2, "Airworthiness Limitations," of the Bombardier Global 6000 TLMC, Publication No. GL 6000 TLMC, Revision 15, dated March 1, 2022.

(3) For service information identified in this AD, contact Bombardier Business Aircraft Customer Response Center, 400 Côte-Vertu Road West, Dorval, Québec H4S 1Y9, Canada; telephone 514-855-2999; email ac.yul@aero.bombardier.com; website bombardier.com.

(4) You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email fr.inspection@nara.gov, or go to: www.archives.gov/federal-register/cfr/ibr-locations.html.

Issued on June 12, 2023.

Ross Landes,

Deputy Director for Regulatory Operations, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2023-14001 Filed 6-30-23; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2023-0927; Project Identifier MCAI-2023-00013-T; Amendment 39-22461; AD 2023-12-03]

RIN 2120-AA64

Airworthiness Directives; Airbus SAS Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Airbus SAS Model A350-941 and -1041 airplanes. This AD was prompted by a determination that new or more restrictive airworthiness limitations are necessary. This AD requires revising the existing maintenance or inspection program, as applicable, to incorporate new or more restrictive airworthiness limitations, as specified in a European Union Aviation Safety Agency (EASA) AD, which is incorporated by reference. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective August 7, 2023.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of August 7, 2023.

ADDRESSES:

AD Docket: You may examine the AD docket at *regulations.gov* under Docket No. FAA-2023-0927; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the mandatory continuing airworthiness information (MCAI), any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

Material Incorporated by Reference:

- For material incorporated by reference in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email *ADs@easa.europa.eu*; website *easa.europa.eu*. You may find this material on the EASA website at *ad.easa.europa.eu*.

- You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. It is also available in the AD docket at *regulations.gov* under Docket No. FAA-2023-0927.

FOR FURTHER INFORMATION CONTACT: Dat Le, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516-228-7317; email *dat.v.le@faa.gov*.

SUPPLEMENTARY INFORMATION:

Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to certain Airbus SAS Model A350-941 and -1041 airplanes. The NPRM published in the **Federal Register** on April 14, 2023 (88 FR 22923). The NPRM was prompted by AD 2023-0004, dated January 6, 2023, issued by EASA, which is the Technical Agent for the Member States of the European Union (EASA AD 2023-0004) (also referred to as the MCAI). The MCAI states that new or more restrictive airworthiness limitations are necessary.

EASA AD 2023-0004 specifies that it requires tasks (limitations) already in Airbus A350 Airworthiness Limitations Section (ALS), Part 2, Revision 08, dated May 2, 2022, that is required by EASA

AD 2022-0125, dated June 28, 2022 (which corresponds to FAA AD 2023-04-05, Amendment 39-22352 (88 FR 13668, March 6, 2023) (AD 2023-04-05)), and that incorporation of EASA AD 2023-0004 invalidates (terminates) prior instructions for those tasks. This AD therefore terminates the limitations for the tasks identified in the service information referenced in EASA AD 2023-0004 only, as required by paragraph (j) of AD 2023-04-05.

In the NPRM, the FAA proposed to require revising the existing maintenance or inspection program, as applicable, to incorporate new or more restrictive airworthiness limitations, as specified in EASA AD 2023-0004. The FAA is issuing this AD to address reduced structural integrity of the airplane.

You may examine the MCAI in the AD docket at *regulations.gov* under Docket No. FAA-2023-0927.

Discussion of Final Airworthiness Directive

Comments

The FAA received comments from The Air Line Pilots Association, International (ALPA) who supported the NPRM without change.

Conclusion

This product has been approved by the aviation authority of another country and is approved for operation in the United States. Pursuant to the FAA's bilateral agreement with this State of Design Authority, it has notified the FAA of the unsafe condition described in the MCAI referenced above. The FAA reviewed the relevant data, considered the comment received, and determined that air safety requires adopting this AD as proposed. Accordingly, the FAA is issuing this AD to address the unsafe condition on this product. Except for minor editorial changes, this AD is adopted as proposed in the NPRM. None of the changes will increase the economic burden on any operator.

Related Service Information Under 14 CFR Part 51

EASA AD 2023-0004 describes new or more restrictive airworthiness limitations for airplane structures and safe life limits. This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

Costs of Compliance

The FAA estimates that this AD affects 31 airplanes of U.S. registry. The

FAA estimates the following costs to comply with this AD.

The FAA has determined that revising the existing maintenance or inspection program takes an average of 90 work-hours per operator, although the agency recognizes that this number may vary from operator to operator. Since operators incorporate maintenance or inspection program changes for their affected fleet(s), the FAA has determined that a per-operator estimate is more accurate than a per-airplane estimate. Therefore, the agency estimates the average total cost per operator to be \$7,650 (90 work-hours × \$85 per work-hour).

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

2023–12–03 Airbus SAS: Amendment 39–22461; Docket No. FAA–2023–0927; Project Identifier MCAI–2023–00013–T.

(a) Effective Date

This airworthiness directive (AD) is effective August 7, 2023.

(b) Affected ADs

This AD affects AD 2023–04–05, Amendment 39–22352 (88 FR 13668, March 6, 2023) (AD 2023–04–05).

(c) Applicability

This AD applies to Airbus SAS Model A350–941 and –1041 airplanes, certificated in any category, with an original airworthiness certificate or original export certificate of airworthiness issued on or before November 1, 2022.

(d) Subject

Air Transport Association (ATA) of America Code: 05, Time Limits/Maintenance Checks.

(e) Unsafe Condition

This AD was prompted by a determination that new or more restrictive airworthiness limitations are necessary. The FAA is issuing this AD to address reduced structural integrity of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Requirements

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, European Union Aviation Safety Agency (EASA) AD 2023–0004, dated January 6, 2023 (EASA AD 2023–0004).

(h) Exceptions to EASA AD 2023–0004

(1) This AD does not adopt the requirements specified in paragraphs (1) and (2) of EASA AD 2023–0004.

(2) Paragraph (3) of EASA AD 2023–0004 specifies revising “the approved AMP” within 12 months after its effective date, but this AD requires revising the existing maintenance or inspection program, as applicable, within 90 days after the effective date of this AD.

(3) The initial compliance time for doing the tasks specified in paragraph (3) of EASA 2023–0004 is on or before the applicable “associated thresholds” as incorporated by the requirements of paragraph (3) of EASA AD 2023–0004, or within 90 days after the effective date of this AD, whichever occurs later.

(4) This AD does not adopt the provisions specified in paragraphs (4) of EASA AD 2023–0004.

(5) This AD does not adopt the “Remarks” section of EASA AD 2023–0004.

(i) Provisions for Alternative Actions and Intervals

After the existing maintenance or inspection program has been revised as required by paragraph (g) of this AD, no alternative actions (e.g., inspections) and intervals are allowed unless they are approved as specified in the provisions of the “Ref. Publications” section of EASA AD 2023–0017.

(j) Terminating Action for AD 2023–04–05

Accomplishing the actions required by this AD terminates the corresponding requirements of AD 2023–04–05, for the tasks identified in the service information referenced in EASA AD 2023–0004 only.

(k) Additional AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the International Validation Branch, send it to the attention of the person identified in paragraph (l) of this AD. Information may be emailed to: 9-AVS-AIR-730-AMOC@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(2) *Contacting the Manufacturer:* For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, International Validation Branch, FAA; or EASA; or Airbus SAS’s EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(l) Additional Information

For more information about this AD, contact Dat Le, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516–228–7317; email dat.v.le@faa.gov.

(m) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) European Union Aviation Safety Agency (EASA) AD 2023–0004, dated January 6, 2023.

(ii) [Reserved]

(3) For EASA AD 2023–0004, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu; website easa.europa.eu. You may find this EASA AD on the EASA website at ad.easa.europa.eu.

(4) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

(5) You may view this material that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email fr.inspection@nara.gov, or go to: www.archives.gov/federal-register/cfr/ibr-locations.html.

Issued on June 12, 2023.

Ross Landes,

Deputy Director for Regulatory Operations, Compliance & Airworthiness Division, Aircraft Certification Service.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2023–0926; Project Identifier MCAI–2022–01583–A; Amendment 39–22462; AD 2023–12–04]

RIN 2120–AA64

Airworthiness Directives; Pilatus Aircraft Ltd. Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Pilatus Aircraft Ltd. (Pilatus) Model PC–24 airplanes. This AD was prompted by a report that an incorrect wiring arrangement was detected around the weather radar system. This AD requires modifying the weather radar redundant wiring, as specified in a European Union Aviation Safety Agency (EASA) AD, which is incorporated by reference (IBR). The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective August 7, 2023.

The Director of the Federal Register approved the incorporation by reference