

(a) Effective Date

This airworthiness directive (AD) is effective March 17, 2022.

(b) Affected ADs

None.

(c) Applicability

This AD applies to the following gliders, all serial numbers, certificated in any category:

(1) Fiberglas-Technik Rudolf Lindner GmbH & Co. KG (type certificate previously held by GROB Aircraft AG, Grob Aerospace GmbH i.l., Grob Aerospace GmbH, Burkhart Grob Luft—und Raumfahrt GmbH & Co. KG, GROB TFE, GROB—WERKE GMBH & CO KG (a division of Burkhart Grob Flugzeugbau)) Model G102 ASTIR CS.

(2) Fiberglas-Technik Rudolf Lindner GmbH & Co. KG (type certificate previously held by GROB Aircraft AG, Grob Aerospace GmbH i.l., Grob Aerospace GmbH, Burkhart Grob Luft—und Raumfahrt GmbH & Co. KG) Model G103 TWIN ASTIR, G103 TWIN II, G103A TWIN II ACRO, G103 C TWIN III ACRO, and G 103 C TWIN III SL.

(d) Subject

Joint Aircraft System Component (JASC) Code 2730, Elevator Control System.

(e) Unsafe Condition

This AD was prompted by mandatory continuing airworthiness information (MCAI) issued by the aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as corrosion on the elevator control pushrod. The unsafe condition, if not addressed, could result in failure of the elevator control pushrod and loss of control of the glider.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Required Actions

(1) Within 25 hours time in service (TIS) after the effective date of this AD, inspect the elevator control pushrod in the vertical fin for water and corrosion and replace the elevator control pushrod before further flight if there is any water or corrosion in accordance with the Actions and Instructions, paragraph 3, of Fiberglas-Technik Rudolf Lindner Anweisung/Instructions (A/I—G09), Revision 1, dated May 14, 2020.

(2) If no water and no corrosion is detected, before further flight, treat the inside of the elevator control pushrod with corrosion preventative LPS 3 or equivalent.

(3) If required by paragraph (g)(1) of this AD, you must replace the elevator control pushrod before further flight with an elevator control pushrod that has zero hours TIS or with an elevator control pushrod that has passed the inspection in accordance with paragraphs (g)(1) and (2) of this AD.

(h) Credit for Previous Actions

You may take credit for the actions required by paragraphs (g)(1) and (2) of this AD if you performed these actions before the

effective date of this AD using Fiberglas-Technik Rudolf Lindner Anweisung/Instructions (A/I—G09), dated April 8, 2020.

(i) Alternative Methods of Compliance (AMOCs)

(1) The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the certification office, send it to the attention of the person identified in paragraph (j)(1) of this AD and email to: 9-AVS-AIR-730-AMOC@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(j) Related Information

(1) For more information about this AD, contact Jim Rutherford, Aviation Safety Engineer, General Aviation & Rotorcraft Section, International Validation Branch, FAA, 901 Locust, Room 301, Kansas City, MO 64106; phone: (816) 329-4165; email: jim.rutherford@faa.gov.

(2) Refer to European Union Aviation Safety Agency (EASA) AD 2020-0138, dated June 19, 2020, for more information. You may examine the EASA AD in the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2021-0944.

(k) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Fiberglas-Technik Rudolf Lindner Anweisung/Instructions (A/I—G09), Revision 1, dated May 14, 2020.

Note 1 to paragraph (k)(2)(i): This service information contains German to English translation. EASA used the English translation in referencing the document from Fiberglas-Technik Rudolf Lindner. For enforceability purposes, the FAA will cite the service information in English as it appears on the document.

(ii) [Reserved]

(3) For service information identified in this AD, contact Fiberglas-Technik Rudolf Lindner GmbH & Co. KG, Steige 3, D-88487 Walpertschhofen, Germany; phone: +49 (0) 7353 22 43; email: info@LTB-Lindner.com; website: <https://www.ltb-lindner.com>.

(4) You may view this service information at FAA, Airworthiness Products Section, Operational Safety Branch, 901 Locust, Kansas City, MO 64106. For information on the availability of this material at the FAA, call (817) 222-5110.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on

the availability of this material at NARA, email: fr.inspection@nara.gov, or go to: <https://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued on January 20, 2022.

Lance T. Gant,

Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2022-02717 Filed 2-9-22; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2021-1012; Project Identifier MCAI-2021-00697-R; Amendment 39-21916; AD 2022-02-19]

RIN 2120-AA64

Airworthiness Directives; Airbus Helicopters Deutschland GmbH (AHD) Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Airbus Helicopters Deutschland GmbH (AHD) Model EC135P1, EC135P2, EC135P2+, EC135P3, EC135T1, EC135T2, EC135T2+, and EC135T3 helicopters. This AD was prompted by a report of restricted collective lever movement caused by entanglement of the emergency flashlight strap with the cargo hook emergency release lever, causing the emergency flashlight to leave its seat. This AD requires replacing each affected emergency flashlight with a serviceable part, as specified in a European Union Aviation Safety Agency (EASA) AD, which is incorporated by reference. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective March 17, 2022.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of March 17, 2022.

ADDRESSES: For EASA material incorporated by reference (IBR) in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu; internet www.easa.europa.eu. You may find the EASA material on the EASA website at <https://ad.easa.europa.eu>. For Airbus Helicopters Deutschland GmbH (AHD) service information identified in this final rule, contact Airbus Helicopters,

2701 North Forum Drive, Grand Prairie, TX 75052; telephone (972) 641-0000 or (800) 232-0323; fax (972) 641-3775; or at <https://www.airbus.com/helicopters/services/technical-support.html>. You may view this material at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N-321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222-5110. Service information that is IBRed is also available in the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2021-1012.

Examining the AD Docket

You may examine the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2021-1012; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the EASA AD, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Hal Jensen, Aerospace Engineer, Operational Safety Branch, FAA, 950 L'Enfant Plaza SW, Washington, DC 20024; telephone (202) 267-9167; email hal.jensen@faa.gov.

SUPPLEMENTARY INFORMATION:

Background

EASA, which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2021-0149, dated June 21, 2021 (EASA AD 2021-0149), to correct an unsafe condition for Airbus Helicopters Deutschland GmbH (AHD) Model EC135 P1, EC135 P2, EC135 P2+, EC135 P3, EC135 T1, EC135 T2, EC135 T2+, EC135 T3, EC635 P2+, EC635 P3, EC635 T1, EC635 T2+, and EC635 T3 helicopters, all variants, all serial numbers up to 820 inclusive. Model EC635 P2+, EC635 P3, EC635 T1, EC635 T2+, and EC635 T3 helicopters are not certificated by the FAA and are not included on the U.S. type certificate data sheet (TCDS), except where the TCDS explains that the Model EC635T2+ helicopter having serial number 0858 was converted from Model EC635T2+ to Model EC135T2+. This AD, therefore, does not include Model EC635 P2+, EC635 P3, EC635 T1, EC635 T2+, and EC635 T3 helicopters in the applicability.

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to Airbus Helicopters Deutschland GmbH Model EC135P1, EC135P2, EC135P2+, EC135P3, EC135T1, EC135T2, EC135T2+, and EC135T3 helicopters, certificated in any category, as identified in EASA 2021-0149. The NPRM published in the **Federal Register** on November 23, 2021 (86 FR 66474). The NPRM was prompted by a report of restricted collective lever movement caused by entanglement of the emergency flashlight strap with the cargo hook emergency release lever, causing the emergency flashlight to leave its seat. The NPRM proposed to require replacing each affected emergency flashlight with a serviceable part, as specified in EASA AD 2021-0149.

The FAA is issuing this AD to address entanglement of the emergency flashlight strap with the cargo hook emergency release lever. See EASA AD 2021-0149 for additional background information.

Discussion of Final Airworthiness Directive

Comments

The FAA received no comments on the NPRM or on the determination of the costs.

Conclusion

These helicopters have been approved by EASA and are approved for operation in the United States. Pursuant to the FAA's bilateral agreement with the European Union, EASA has notified the FAA about the unsafe condition described in its AD. The FAA reviewed the relevant data and determined that air safety requires adopting this AD as proposed. Accordingly, the FAA is issuing this AD to address the unsafe condition on these helicopters. Except for minor editorial changes, including updating the name "Airbus Helicopters Deutschland GmbH" to "Airbus Helicopters Deutschland GmbH (AHD)" to match its FAA TCDS type certificate holder name, adding the Other Related Service Information section to describe Airbus Helicopters service information, correcting the issuance date of EASA AD 2021-0149 in the Applicability paragraph, and reformatting and updating the Costs of Compliance section, this AD is adopted as proposed in the NPRM.

Related Service Information Under 1 CFR Part 51

EASA AD 2021-0149 requires replacing each affected emergency flashlight with a serviceable part. EASA

AD 2021-0149 also specifies that an affected part can be modified and re-identified into a serviceable part. EASA AD 2021-0149 also prohibits the installation of an affected part.

This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

Other Related Service Information

The FAA reviewed Airbus Helicopters Alert Service Bulletin ASB EC135-25A-032, Revision 0, dated May 20, 2021. This service information specifies procedures to remove the strap from the emergency flashlight and what part number to write on the flashlight.

Costs of Compliance

The FAA estimates that this AD affects 335 helicopters of U.S. Registry. Labor rates are estimated at \$85 per work-hour. Based on these numbers, the FAA estimates the following costs to comply with this AD.

Replacing an emergency flashlight takes about 1 work-hour and parts cost about \$219 for an estimated cost of \$304 per flashlight. Alternatively, modifying an emergency flashlight takes about 1 work-hour for an estimated cost of \$85 per flashlight.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and

responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a “significant regulatory action” under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

2022-02-19 Airbus Helicopters

Deutschland GmbH (AHD): Amendment 39-21916; Docket No. FAA-2021-1012; Project Identifier MCAI-2021-00697-R.

(a) Effective Date

This airworthiness directive (AD) is effective March 17, 2022.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Airbus Helicopters Deutschland GmbH (AHD) Model EC135P1, EC135P2, EC135P2+, EC135P3, EC135T1, EC135T2, EC135T2+, and EC135T3 helicopters, certificated in any category, as identified in European Union Aviation Safety Agency (EASA) AD 2021-0149, dated June 21, 2021 (EASA 2021-0149).

(d) Subject

Joint Aircraft Service Component (JASC) Code: 2510, Flight Compartment Equipment.

(e) Unsafe Condition

This AD was prompted by a report of restricted collective lever movement. Subsequent inspection determined that the emergency flashlight was stuck under that lever caused by entanglement of the emergency flashlight strap with the cargo hook emergency release lever, causing the emergency flashlight to leave its seat. The FAA is issuing this AD to address entanglement of the emergency flashlight

strap with the cargo hook emergency release lever. The unsafe condition, if not addressed, could result in reduced control of the helicopter, resulting in damage to the helicopter and injury to occupants.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Requirements

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, EASA AD 2021-0149.

(h) Exceptions to EASA AD 2021-0149

(1) Where EASA AD 2021-0149 refers to its effective date, this AD requires using the effective date of this AD.

(2) This AD does not mandate compliance with the “Remarks” section of EASA AD 2021-0149.

(i) Alternative Methods of Compliance (AMOCs)

(1) The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Validation Branch, send it to the attention of the person identified in paragraph (j) of this AD. Information may be emailed to: 9-AVS-AIR-730-AMOC@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(j) Related Information

For more information about this AD, contact Hal Jensen, Aerospace Engineer, Operational Safety Branch, FAA, 950 L'Enfant Plaza SW, Washington, DC 20024; telephone (202) 267-9167; email hal.jensen@faa.gov.

(k) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) European Union Aviation Safety Agency (EASA) AD 2021-0149, dated June 21, 2021.

(ii) [Reserved]

(3) For EASA AD 2021-0149, EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu; internet www.easa.europa.eu. You may find the EASA material on the EASA website at <https://ad.easa.europa.eu>.

(4) You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N-321, Fort Worth, TX 76177. For

information on the availability of this material at the FAA, call (817) 222-5110. This material may be found in the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2021-1012.

(5) You may view this material that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email fr.inspection@nara.gov, or go to: <https://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued on January 18, 2022.

Lance T. Gant,

Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2022-02752 Filed 2-9-22; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2021-0887; Project Identifier MCAI-2021-00045-R; Amendment 39-21910; AD 2022-02-13]

RIN 2120-AA64

Airworthiness Directives; Airbus Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for Airbus Helicopters Model EC120B helicopters. This AD was prompted by a report of corrosion found on the external tail boom skin, under the Very High Frequency (VHF) antenna. This AD requires inspecting the tail boom at the VHF antenna attachments and depending on the results, repairing or modifying the tail boom skin, as specified in a European Union Aviation Safety Agency (EASA) AD, which is incorporated by reference. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective March 17, 2022.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of March 17, 2022.

ADDRESSES: For EASA material incorporated by reference (IBR) in this final rule, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu; internet www.easa.europa.eu. You may find the EASA material on the EASA website at <https://ad.easa.europa.eu>. For Airbus