■ 8. Revise § 621.9 to read as follows:

### §621.9 Reinstatement to accrual status.

(a) Before being reinstated to accrual status, a loan must be current on contractual payments and the borrower offered servicing in accordance with the institution's policies maintained under either § 614.4170 or part 617 of this chapter, whichever is applicable. Additional reinstatement eligibility requirements are dependent upon certain characteristics of the loan under review.

(1) A loan that was current when placed in nonaccrual status pursuant to  $\S$  621.6(a)(1) may be reinstated to accrual status if the known risks to the continued collection of principal or interest have been mitigated. If the loan was past due when placed in nonaccrual status, it may only be reinstated under either paragraph (a)(2) or (a)(3) of this section, as applicable.

(2) A loan placed in nonaccrual status when past due and not adequately secured must remain current on contractual payments for a period of sustained performance before it may be reinstated.

(3) A loan placed in nonaccrual status when past due and adequately secured must have a recent repayment pattern demonstrating future repayment capacity to make on-time payments before it may be reinstated. The repayment pattern is established in one of two ways:

(i) Sustained performance in making on-time contractual payments, or

(ii) A recent history of making on-time partial payments in amounts the same or greater than newly restructured payment amounts.

(b) Nothing in this section prevents a current loan from being reinstated to accrual status in response to a Credit Review Committee decision issued under section 4.14D(d) of the Farm Credit Act of 1971, as amended, when that decision was made in compliance with applicable laws, regulations, and in accordance with generally accepted accounting principles.

Dated: July 21, 2020.

#### Dale Aultman,

Secretary, Farm Credit Administration Board. [FR Doc. 2020–16135 Filed 8–24–20; 8:45 am] BILLING CODE 6705–01–P

# DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2020-0777; Project Identifier MCAI-2020-01071-T; Amendment 39-21217; AD 2020-17-12]

## RIN 2120-AA64

# Airworthiness Directives; Dassault Aviation Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT. **ACTION:** Final rule; request for comments.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain Dassault Aviation Model MYSTERE-FALCON 900, FALCON 900EX, FALCON 2000, and FALCON 2000EX airplanes. This AD was prompted by reports of loose or missing nuts on the pilot and co-pilot ventral seat belt attachment points. This AD requires a detailed inspection of certain seat belt attaching point nuts for any loose or missing nuts and replacement, as specified in a European Union Aviation Safety Agency (EASA) AD, which is incorporated by reference. The FAA is issuing this AD to address the unsafe condition on these products.

**DATES:** This AD becomes effective September 9, 2020.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of September 9, 2020.

The FAA must receive comments on this AD by October 9, 2020.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

• *Federal eRulemaking Portal:* Go to *https://www.regulations.gov.* Follow the instructions for submitting comments.

• *Fax:* 202–493–2251.

• *Mail:* U.S. Department of Transportation, Docket Operations, M– 30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

• *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M– 30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For material incorporated by reference (IBR) in this AD, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email *ADs@easa.europa.eu;*  internet *www.easa.europa.eu*. You may find this IBR material on the EASA website at *https://ad.easa.europa.eu*. You may view this IBR material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195. It is also available in the AD docket on the internet at *https:// www.regulations.gov* by searching for and locating Docket No. FAA–2020– 0777.

## **Examining the AD Docket**

You may examine the AD docket on the internet at *https:// www.regulations.gov* by searching for and locating Docket No. FAA–2020– 0777; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, any comments received, and other information. The street address for Docket Operations is listed above. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Tom Rodriguez, Aerospace Engineer, Large Aircraft Section, International Validation Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone and fax 206–231–3226; email *Tom.Rodriguez@faa.gov.* 

## SUPPLEMENTARY INFORMATION:

#### Discussion

The EASA, which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2020–0168R1, dated July 29, 2020 ("EASA AD 2020–0168R1") (also referred to as the Mandatory Continuing Airworthiness Information, or "the MCAI"), to correct an unsafe condition for certain Dassault Aviation Model MYSTERE–FALCON 900, FALCON 900EX, FALCON 2000, and FALCON 2000EX airplanes.

This AD was prompted by reports of loose or missing nuts on the pilot and co-pilot ventral seat belt attachment points. The FAA is issuing this AD to address this condition, which could lead to detachment of the seat belt at a critical phase of flight, such as landing or, in the case of turbulence or emergency landing, resulting in the flight crew becoming unrestrained from their seat, causing injury to the flight crew and/or subsequent loss of control of the airplane. This condition could impede the continued safety of flight. See the MCAI for additional background information.

# Related IBR Material Under 1 CFR Part 51

EASA AD 2020–0168R1 describes procedures for a detailed inspection of certain seat belt attaching point nuts for any loose or missing nuts and corrective action, which is replacement. This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

## **FAA's Determination**

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to the FAA's bilateral agreement with the State of Design Authority, the FAA has been notified of the unsafe condition described in the MCAI referenced above. The FAA is issuing this AD because the FAA evaluated all pertinent information and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

### **Requirements of This AD**

This AD requires accomplishing the actions specified in EASA AD 2020–0168R1, described previously, as incorporated by reference, except for any differences identified as exceptions in the regulatory text of this AD.

## Explanation of Required Compliance Information

In the FAA's ongoing efforts to improve the efficiency of the AD process, the FAA initially worked with Airbus and EASA to develop a process to use certain EASA ADs as the primary source of information for compliance with requirements for corresponding FAA ADs. The FAA has since coordinated with other manufacturers and civil aviation authorities (CAAs) to use this process. As a result, EASA AD 2020-0168R1 is incorporated by reference in this final rule. This AD, therefore, requires compliance with EASA AD 2020-0168R1 in its entirety, through that incorporation, except for any differences identified as exceptions in the regulatory text of this AD. Using common terms that are the same as the heading of a particular section in the EASA AD does not mean that operators

need comply only with that section. For example, where the AD requirement refers to "all required actions and compliance times," compliance with this AD requirement is not limited to the section titled "Required Action(s) and Compliance Time(s)" in the EASA AD. Service information specified in EASA AD 2020–0168R1 that is required for compliance with EASA AD 2020– 0168R1 is available on the internet at *https://www.regulations.gov* by searching for and locating Docket No. FAA–2020–0777.

# FAA's Justification and Determination of the Effective Date

An unsafe condition exists that requires the immediate adoption of this AD without providing an opportunity for public comments prior to adoption. The FAA has found that the risk to the flying public justifies foregoing notice and comment prior to adoption of this rule because the detachment of the seat belt at a critical phase of flight such as landing or, in the case of turbulence or emergency landing, could result in the flight crew becoming unrestrained from their seat, causing injury to the flight crew and/or subsequent loss of control of the airplane, thereby impeding continued safety of flight. In addition, the compliance time for the required action is shorter than the time necessary for the public to comment and for publication of the final rule. Therefore, the FAA finds good cause that notice and opportunity for prior public comment are impracticable. In addition, for the reasons stated above, the FAA finds that good cause exists for making this amendment effective in less than 30 days.

## **Comments Invited**

This AD is a final rule that involves requirements affecting flight safety, and the FAA did not precede it by notice and opportunity for public comment. The FAA invites you to send any written relevant data, views, or arguments about this AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA–2020–0777; Project Identifier MCAI–2020–01071–T" at the beginning of your comments. The most helpful comments reference a specific portion of the proposal, explain the reason for any recommended change, and include supporting data. To ensure the docket does not contain duplicate comments, commenters should submit only one copy of the comments.

Except for Confidential Business Information (CBI) as described in the following paragraph, and other information as described in 14 CFR 11.35, the FAA will post all comments the FAA receives, without change, to *https://www.regulations.gov*, including any personal information you provide. The FAA will also post a report summarizing each substantive verbal contact the FAA receives about this AD.

## **Confidential Business Information**

CBI is commercial or financial information that is both customarily and actually treated as private by its owner. Under the Freedom of Information Act (FOIA) (5 U.S.C. 552), CBI is exempt from public disclosure. If your comments responsive to this AD contain commercial or financial information that is customarily treated as private, that you actually treat as private, and that is relevant or responsive to this AD, it is important that you clearly designate the submitted documents as CBI. Please mark each page of your submission containing CBI as "PROPIN." The FAA will treat such marked submissions as confidential under the FOIA, and they will not be placed in the public docket of this AD. Submissions containing CBI should be sent to the person identified in the FOR FURTHER INFORMATION **CONTACT** section. Any commentary that the FAA receives that is not specifically designated as CBI will be placed in the public docket for this rulemaking.

## **Regulatory Flexibility Act (RFA)**

The requirements of the RFA do not apply when an agency finds good cause pursuant to 5 U.S.C. 553 to adopt a rule without prior notice and comment. Because the FAA has determined that it has good cause to adopt this rule without notice and comment, RFA analysis is not required.

### **Costs of Compliance**

The FAA estimates that this AD affects 793 airplanes of U.S. registry. The FAA estimates the following costs to comply with this AD:

## ESTIMATED COSTS FOR REQUIRED ACTIONS

Labor cost	Parts cost	Cost per product	Cost on U.S. operators
4 work-hours $\times$ \$85 per hour = \$340	\$4	\$344	\$272,792

52256 Federal Register/Vol. 85, No. 165/Tuesday, August 25, 2020/Rules and Regulations

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

## **Regulatory Findings**

The FAA determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

(1) Is not a "significant regulatory action" under Executive Order 12866, and

(2) Will not affect intrastate aviation in Alaska.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

## Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

#### §39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

2020–17–12 Dassault Aviation: Amendment 39–21217; Docket No. FAA–2020–0777; Project Identifier MCAI–2020–01071–T.

#### (a) Effective Date

This AD becomes effective September 9, 2020.

## (b) Affected ADs

None.

#### (c) Applicability

This AD applies to Dassault Aviation Model MYSTERE–FALCON 900, FALCON 900EX, FALCON 2000, and FALCON 2000EX airplanes, certificated in any category, as identified in European Union Aviation Safety Agency (EASA) AD 2020–0168R1, dated July 29, 2020 ("EASA AD 2020–0168R1").

## (d) Subject

Air Transport Association (ATA) of America Code 25, Equipment/furnishings.

#### (e) Reason

This AD was prompted by reports of loose or missing nuts on the pilot and co-pilot ventral seat belt attachment points. The FAA is issuing this AD to address this condition, which could lead to detachment of the seat belt at a critical phase of flight such as landing or, in the case of turbulence or emergency landing, resulting in the flight crew becoming unrestrained from their seat, causing injury to the flight crew and/or subsequent loss of control of the airplane.

#### (f) Compliance

Comply with this AD within the compliance times specified, unless already done.

#### (g) Requirements

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, EASA AD 2020–0168R1.

# (h) Exceptions to EASA AD EASA AD 2020–0168R1

(1) Where EASA AD 2020–0168R1 refers to its effective date, this AD requires using the effective date of this AD.

(2) Where paragraph (2) of EASA AD 2020– 0168R1 specifies actions if, "deficiencies (as defined in the applicable inspection SB) are found on," for this AD deficiencies are defined as any missing nuts or any axial end play on any bolt assemblies (loose nuts).

(3) Paragraph (4) of EASA AD 2020– 0168R1 provides credit for "accomplishment of a 'C' check (as per Dassault definition)." For this AD, that credit is allowed provided it can be conclusively determined that each affected part was inspected as specified in that 'C' check.

(4) The "Remarks" section of EASA AD 2020–0168R1 does not apply to this AD.

#### (i) No Reporting Requirement

Although the service information referenced in EASA AD 2020–0168R1 specifies to submit certain information to the manufacturer, this AD does not include that requirement.

#### (j) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, Large Aircraft

Section, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the Large Aircraft Section, International Validation Branch, send it to the attention of the person identified in paragraph (k) of this AD. Information may be emailed to: 9-AVS-AIR-730-AMOC@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) Contacting the Manufacturer: For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, Large Aircraft Section, International Validation Branch, FAA; or EASA; or Dassault Aviation's EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

#### (k) Related Information

For more information about this AD, contact Tom Rodriguez, Aerospace Engineer, Large Aircraft Section, International Validation Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone and fax 206–231–3226; email *Tom.Rodriguez@ faa.gov.* 

#### (l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) European Union Aviation Safety Agency (EASA) AD 2020–0168R1, dated July 29, 2020.

(ii) [Reserved]

(3) For information about EASA AD 2020– 0168R1, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email *ADs*@ *easa.europa.eu;* internet *www.easa.europa.eu.* You may find this EASA AD on the EASA website at *https:// ad.easa.europa.eu.* 

(4) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195. This material may be found in the AD docket on the internet at *https:// www.regulations.gov* by searching for and locating Docket No. FAA–2020–0777.

(5) You may view this material that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email *fedreg.legal@ nara.gov*, or go to: *https://www.archives.gov/ federal-register/cfr/ibr-locations.html.*  Issued on August 13, 2020. Lance T. Gant, Director, Compliance & Airworthiness Division, Aircraft Certification Service. [FR Doc. 2020–18488 Filed 8–24–20; 8:45 am] BILLING CODE 4910–13–P

# DEPARTMENT OF TRANSPORTATION

## Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2020-0690; Project Identifier AD-2020-00860-T; Amendment 39-21207; AD 2020-17-02]

#### RIN 2120-AA64

## Airworthiness Directives; The Boeing Company Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT. **ACTION:** Final rule; request for comments.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain The Boeing Company Model 747–8 and -8F series airplanes, and Model 787-8, –9, and –10 airplanes. This AD requires removing Kathon FP 1.5 biocide from the fuel tanks and engines, installing a fuel limitation placard, and revising the existing airplane flight manual (AFM) to prohibit operation of the airplane with Kathon FP 1.5 biocide in a fuel tank or engine. This AD was prompted by a report indicating that Kathon FP 1.5 biocide added to fuel and running through the engines can lead to significant engine anomalies. The FAA is issuing this AD to address the unsafe condition on these products.

**DATES:** This AD is effective August 25, 2020.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of August 25, 2020.

The FĂA must receive comments on this AD by October 9, 2020.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

 Federal eRulemaking Portal: Go to https://www.regulations.gov. Follow the instructions for submitting comments.
Fax: 202-493-2251.

Mail: U.S. Department of

Transportation, Docket Operations, M– 30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

• *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this final rule, contact Boeing Commercial Airplanes, Attention: Contractual & Data Services (C&DS), 2600 Westminster Blvd., MC 110-SK57, Seal Beach, CA 90740–5600; telephone 562-797-1717; internet https:// www.myboeingfleet.com. You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195. It is also available on the internet at https://www.regulations.gov by searching for and locating Docket No. FAA-2020-0690.

## **Examining the AD Docket**

You may examine the AD docket on the internet at *https:// www.regulations.gov* by searching for and locating Docket No. FAA–2020– 0690; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, any comments received, and other information. The street address for Docket Operations is listed above. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Tak Kobayashi, Aerospace Engineer, Propulsion Section, FAA, Seattle ACO Branch, 2200 South 216th St., Des Moines, WA 98198; phone and fax: 206– 231–3553; email: *takahisa.kobayashi@ faa.gov.* 

#### SUPPLEMENTARY INFORMATION:

#### Discussion

The FAA has received a report indicating that a The Boeing Company Model 787 airplane equipped with General Electric Company (GE) GEnx– 1B model turbofan engines experienced temporary thrust anomalies on both engines during descent into Kansai International Airport in Japan, on March 29, 2019. Specifically, both engines briefly fell below idle thrust, and the flightcrew received failure messages for both engines.

The FAA's review of the data from this incident indicated the thrust anomalies resulted from fuel control instability. The fuel tanks of the event airplane had recently been treated with Kathon FP 1.5 biocide for suspected microbial growth contamination. Salt crystals can form in the fuel under certain conditions after Kathon FP 1.5 biocide is applied. These salt crystals have the potential to cause slow response of engine hydromechanical control features, resulting in compressor stalls or flameouts, potentially on both engines.

This condition, if not addressed, could result in malfunction of the engine's control system hydromechanical unit due to undispersed Kathon FP 1.5 biocide contaminating and restricting the movement of internal parts. Because the fuel systems for both engines on an affected airplane are likely to be similarly affected, there is the potential for loss of thrust control on both engines. Loss of thrust control on both engines could result in failure to climb on takeoff, a forced off-airport landing, or an unacceptably high flightcrew workload.

However, after this biocide is added to the fuel tanks, adding fuel without biocide diminishes the hazard. Eventually, after the tanks have been refilled a sufficient number of times with untreated fuel, enough of the treated fuel would be dissipated, and the unsafe condition would be removed. Specifically, Boeing determined that operating the airplane, or any individual engine, for at least 30 flight cycles, while adding only fuel that has not been treated with this biocide, would flush the biocide from the fuel tank system and the engines. The FAA finds this number of flight cycles to be sufficiently conservative, and therefore has incorporated it the requirements of this AD.

The FAA's analysis of the risks posed by this issue has been ongoing, as has the information available to the agency. On March 10, 2020, the manufacturer of Kathon FP 1.5 issued a letter recommending an immediate halt of using Kathon FP 1.5 biocide for aviation fuel applications. A copy of that letter is in the docket for this rulemaking. On March 25, 2020, the FAA issued a Special Airworthiness Information Bulletin (SAIB), which is in the docket for this rulemaking, regarding the use of Kathon FP 1.5 and another biocide. Most recently, on June 25, 2020, the Japan Transport Safety Board issued an "Aircraft Serious Incident Investigation Report" regarding the March 29, 2019 incident. That report is in the docket for this rulemaking.

The engine and aircraft manufacturers also evaluated the potential of Kathon FP 1.5 biocide application resulting in adverse effects on the engines besides GEnx–1B model engines installed on Model 787 airplanes. Based on this evaluation, the FAA has determined that the unsafe condition also exists on The Boeing Company Model 747–8 and –8F series airplanes powered by GEnx– 2B model engines.