

## PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

### § 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

**2019-07-03 Zodiac Seats France (formerly SICMA Aero Seat):** Amendment 39-19614; Docket No. FAA-2017-0839; Product Identifier 2017-NE-31-AD.

#### (a) Effective Date

This AD is effective May 16, 2019.

#### (b) Affected ADs

None.

#### (c) Applicability

This AD applies to all Zodiac Seats France, 536-Series Cabin Attendant Seats, part number (P/N) 53600, all dash numbers, all serial numbers. These appliances are installed on, but not limited to: Avions de transport regional (ATR) 42 and ATR 72 model airplanes of U.S. registry.

#### (d) Subject

Joint Aircraft System Component (JASC) Code 2500, Cabin Equipment/Furnishings.

#### (e) Unsafe Condition

This AD was prompted by corrosion found on the seat structure or on clamps of the Zodiac Seats France 536-Series Cabin Attendant Seats. We are issuing this AD to prevent failure of these seats. The unsafe condition, if not addressed, could result in failure of the seat occupied by the cabin attendant, and possible injury to the cabin attendant.

#### (f) Compliance

Comply with this AD within the compliance times specified, unless already done.

#### (g) Required Actions

(1) Within 14 months after the first installation of the seat on an aircraft, or within 3 months after the effective date of this AD, whichever occurs later, remove the seat from the aircraft and perform a detailed visual inspection in accordance with the Accomplishment Instructions, Paragraph 2.B., of Zodiac Seats France Service Bulletin (SB) No. 536-25-002, Revision 3, dated November 2, 2016. If the date of the first installation of a seat on an airplane is unknown, use the date of manufacture of the seat (which can be found on the ID placard of the seat) to determine when the inspection must be accomplished.

(2) Within 3 months after the inspection required by paragraph (g)(1) of this AD, and, thereafter, at intervals not to exceed 3 months, perform a detailed visual inspection in accordance with the Accomplishment Instructions, Paragraphs 2.A. and 2.B., of Zodiac Seats France SB No. 536-25-002, Revision 3, dated November 2, 2016.

(3) If corrosion or other damage is found, before next flight or before (re)installation of the seat on an aircraft, as applicable, repair the seat in accordance with the Accomplishment Instructions, Paragraphs 2.B. and 2.C., of Zodiac Seats France SB No. 536-25-002, Revision 3, dated November 2, 2016.

(4) Temporarily stowing and securing a damaged attendant seat in a retracted position to prevent occupancy, in accordance with the provisions and limitations applicable Master Minimum Equipment List item, is an acceptable alternative method to defer compliance with the requirements of paragraph (g)(3) of this AD.

#### (h) Installation Prohibition

After the effective date of this AD, do not install on any aircraft an affected Zodiac Seats France 536-Series Cabin Attendant Seat that has accumulated more than 14 months since first installation on any aircraft, unless it has passed an inspection in accordance with the Accomplishment Instructions, Paragraph 2.B., of Zodiac Seats France SB No. 536-25-002, Revision 3, dated November 2, 2016.

#### (i) Optional Terminating Action

Modification and re-identification (P/N change) of a seat in accordance with the Accomplishment Instructions, paragraph 2.A., of Zodiac Seats France SB No. 536-25-004, Rev. 0, dated October 19, 2017, constitutes a terminating action for the repetitive inspections as required by this AD. Operators are not required to perform the steps in Sections A6 and A9 in paragraph 2.A. of the SB to complete this terminating action.

#### (j) Credit for Previous Actions

You may take credit for actions required by paragraph (g) of this AD if you performed these actions before the effective date of this AD using Zodiac Seats France SB No. 536-25-002, Revision 2, dated August 29, 2016.

#### (k) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Boston ACO Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the ACO Branch, send it to the attention of the person identified in paragraph (l)(1) of this AD.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

#### (l) Related Information

(1) For more information about this AD, contact Dorie Resnik, Aerospace Engineer, Boston ACO Branch, FAA, 1200 District Avenue, Burlington, MA 01803; phone: 781-238-7693; fax: 781-238-7199; email: [dorie.resnik@faa.gov](mailto:dorie.resnik@faa.gov).

(2) Refer to European Union Aviation Safety Agency AD 2016-0167R1, dated

February 2, 2018 (corrected March 1, 2018) for more information. You may examine the EASA AD in the AD docket on the internet at <http://www.regulations.gov> by searching for and locating it in Docket No. FAA-2017-0839.

#### (m) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Zodiac Seats France Service Bulletin (SB) No. 536-25-002, Revision 3, dated November 2, 2016.

(ii) Zodiac Seats France SB No. 536-25-004, Rev. 0, dated October 19, 2017.

(3) For Zodiac Seats France service information identified in this AD, contact Safran Seats France, 61, Rue Pierre Curie, CS20001, Plaisir Cedex, France phone: + 33 977 428 378; email: [AOG.3S@safrangroup.com](mailto:AOG.3S@safrangroup.com); website: <https://www.safran-group.com>.

(4) You may view this service information at FAA, Engine & Propeller Standards Branch, 1200 District Avenue, Burlington, MA 01803. For information on the availability of this material at the FAA, call 781-238-7759.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Burlington, Massachusetts, on April 2, 2018.

**Karen M. Grant,**

*Acting Manager, Engine and Propeller Standards Branch, Aircraft Certification Service.*

[FR Doc. 2019-07164 Filed 4-10-19; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2019-0191; Product Identifier 2018-NM-161-AD; Amendment 39-19610; AD 2019-06-12]

RIN 2120-AA64

#### Airworthiness Directives; Airbus SAS Airplanes

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule; request for comments.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for all

Airbus SAS Model A330–201, –202, and –203; and Model A330–301, –302, and –303 airplanes. This AD was prompted by reports of disbonds on the engine air inlet cowl inner barrel lower panel between the back skins and the honeycomb core of airplanes equipped with certain engines. This AD requires repetitive detailed inspections of the engine air inlet cowls, and corrective actions if necessary, as specified in an European Aviation Safety Agency (EASA) AD, which is incorporated by reference. We are issuing this AD to address the unsafe condition on these products.

**DATES:** This AD becomes effective April 26, 2019.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of April 26, 2019.

We must receive comments on this AD by May 28, 2019.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202–493–2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For the incorporation by reference (IBR) material described in the “Related IBR Material Under 1 CFR part 51” section in **SUPPLEMENTARY INFORMATION**, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 89990 1000; email: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu); internet: [www.easa.europa.eu](http://www.easa.europa.eu).

You may find this IBR material on the EASA website at <https://ad.easa.europa.eu>. You may view this IBR material at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195. It is also available in the AD docket on the internet at <http://www.regulations.gov>.

### Examining the AD Docket

You may examine the AD docket on the internet at <http://www.regulations.gov> by searching for

and locating Docket No. FAA–2019–0191; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The street address for Docket Operations (phone: 800–647–5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

### FOR FURTHER INFORMATION CONTACT:

Vladimir Ulyanov, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; phone and fax: 206–231–3229.

### SUPPLEMENTARY INFORMATION:

#### Discussion

The EASA, which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2018–0228, dated October 22, 2018 (“EASA AD 2018–0228”) (also referred to as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for all Airbus SAS Model A330–201, –202, and –203; and Model A330–301, –302, and –303 airplanes. The MCAI states:

Occurrences were reported on A330 aeroplanes fitted with General Electric CF6–80E1 engines, where the air inlet cowl inner barrel lower panel, was found disbonded between the back skins and the honeycomb core. The technical investigation results revealed that this occurrence may have been caused by freezing of water, accumulated in the non-drained honeycomb cells, damaging the adhesive bond between the panel core and the back skin.

This condition, if not corrected, in combination with an engine surge condition, could lead to in-flight detachment of an air inlet cowl inner barrel, possibly resulting in damage to the aeroplane, and/or injury to persons on the ground.

To address this potential unsafe condition, Airbus issued the inspection SB [Airbus Service Bulletin A330–71–3035] to provide instructions for [repetitive] detailed inspections (DET, tap test inspection method) of the air inlet cowl inner barrel lower panels, based on the Goodrich SB [Goodrich Aerospace Service Bulletin CF6–80E1–NAC–71–048]. Airbus also developed mod [modification] 203989, installation of improved inner barrel lower panels, based on Goodrich Aerospace SB CF6–80E1–NAC–71–052, and published the modification SB [Airbus SB A330–71–3036] as terminating action.

For the reasons described above, this EASA AD requires repetitive DET (tap tests) of the inner barrel lower panels of each affected part on both engines and, depending on findings, replacement with an improved inner barrel lower panel [which terminates the repetitive inspections]. This [EASA] AD

also allows, in case of no findings, the modification SB [Airbus SB A330–71–3036 (replacement with an improved inner barrel lower panel) to be accomplished] as optional terminating action for the repetitive inspections.

### Related IBR Material Under 1 CFR Part 51

EASA AD 2018–0228 describes procedures for repetitive detailed inspections of the engine air inlet cowls, and corrective actions. Corrective actions include an inspection of the inlet anti-ice seal installation, modification of affected parts, and replacement of damaged inlet anti-ice seals. This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section and it is publicly available through the EASA website.

### FAA’s Determination

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI referenced above. We are issuing this AD because we evaluated all pertinent information and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

### Requirements of This AD

This AD requires accomplishing the actions specified in EASA AD 2018–0228 described previously, as incorporated by reference, except for any differences identified as exceptions in the regulatory text of this AD.

### Explanation of Required Compliance Information

In the FAA’s ongoing efforts to improve the efficiency of the AD process, the FAA worked with Airbus and EASA to develop a process to use certain EASA ADs as the primary source of information for compliance with requirements for corresponding FAA ADs. As a result, EASA AD 2018–0228 is incorporated by reference in the FAA final rule. This AD, therefore, requires compliance with the provisions specified in EASA AD 2018–0228, except for any differences identified as exceptions in the regulatory text of this AD. Service information specified in EASA AD 2018–0228 that is required for compliance with EASA AD 2018–0228 is available on the internet at <http://www.regulations.gov> by searching for

and locating Docket No. FAA-2019-0191.

**FAA’s Justification and Determination of the Effective Date**

Since there are currently no domestic operators of this product, notice and opportunity for public comment before issuing this AD are unnecessary. In addition, for the reasons stated above, we find that good cause exists for making this amendment effective in less than 30 days.

**Comments Invited**

This AD is a final rule that involves requirements affecting flight safety, and

we did not precede it by notice and opportunity for public comment. We invite you to send any written relevant data, views, or arguments about this AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA-2019-0191; Product Identifier 2018-NM-161-AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this AD. We will consider all comments received by the closing date and may amend this AD based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this AD.

**Costs of Compliance**

Currently, there are no affected U.S.-registered airplanes. If an affected airplane is imported and placed on the U.S. Register in the future, we provide the following cost estimates to comply with this AD:

**ESTIMATED COSTS FOR REQUIRED ACTIONS**

Labor cost	Parts cost	Cost per product
10 work-hours × \$85 per hour = \$850 per inspection cycle ...	\$2,814 per inspection cycle .....	\$3,664 per inspection cycle.

**ESTIMATED COSTS FOR OPTIONAL ACTIONS**

Labor cost	Parts cost	Cost per product
Up to 172 work-hours × \$85 per hour = \$14,620 per engine	Up to \$184,124 per engine .....	Up to \$198,744 per engine.

We estimate the following costs to do any necessary on-condition actions that would be required based on the results

of any required actions. We have no way of determining the number of aircraft

that might need these on-condition actions:

**ESTIMATED COSTS OF ON-CONDITION ACTIONS**

Labor cost	Parts cost	Cost per product
Up to 172 work-hours × \$85 per hour = \$14,620 per engine	Up to \$184,124 per engine .....	Up to \$198,744 per engine.

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: “General requirements.” Under that

section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

This AD is issued in accordance with authority delegated by the Executive Director, Aircraft Certification Service,

as authorized by FAA Order 8000.51C. In accordance with that order, issuance of ADs is normally a function of the Compliance and Airworthiness Division, but during this transition period, the Executive Director has delegated the authority to issue ADs applicable to transport category airplanes and associated appliances to the Director of the System Oversight Division.

**Regulatory Findings**

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

1. Is not a “significant regulatory action” under Executive Order 12866;

2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);

3. Will not affect intrastate aviation in Alaska; and

4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**Adoption of the Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

### § 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

**2019-06-12 Airbus SAS:** Amendment 39-19610; Docket No. FAA-2019-0191; Product Identifier 2018-NM-161-AD.

#### (a) Effective Date

This AD becomes effective April 26, 2019.

#### (b) Affected ADs

None.

#### (c) Applicability

This AD applies to the airplanes identified in paragraphs (c)(1) and (c)(2) of this AD, certificated in any category, all manufacturer serial numbers.

(1) Airbus SAS Model A330-201, -202, and -203 airplanes.

(2) Airbus SAS Model A330-301, -302, and -303 airplanes.

#### (d) Subject

Air Transport Association (ATA) of America Code 71, Powerplant.

#### (e) Reason

This AD was prompted by reports of disbands on the engine air inlet cowl inner barrel lower panel between the back skins and the honeycomb core of airplanes equipped with certain engines. We are issuing this AD to address such disbands, which, in combination with an engine surge, could lead to in-flight detachment of an air inlet cowl inner barrel, possibly resulting in damage to the airplane.

#### (f) Compliance

Comply with this AD within the compliance times specified, unless already done.

#### (g) Requirements

Except as specified in paragraphs (h) and (i) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, European Aviation Safety Agency (EASA) AD 2018-0228, dated October 22, 2018 ("EASA AD 2018-0228").

#### (h) Exceptions to EASA AD 2018-0228

(1) For purposes of determining compliance with the requirements of this AD: Where EASA AD 2018-0228 refers to its effective date, this AD requires using the effective date of this AD.

(2) The "Remarks" section of EASA AD 2018-0228 does not apply to this AD.

#### (i) No Reporting Requirement

Although the service information referenced in EASA AD 2018-0228 specifies to submit certain information to the manufacturer, this AD does not include that requirement.

#### (j) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International

Section, Transport Standards Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Section, send it to the attention of the person identified in paragraph (k) of this AD. Information may be emailed to: [9-ANM-116-AMOC-REQUESTS@faa.gov](mailto:9-ANM-116-AMOC-REQUESTS@faa.gov). Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) *Contacting the Manufacturer:* For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, International Section, Transport Standards Branch, FAA; or EASA; or Airbus SAS's EASA DOA. If approved by the DOA, the approval must include the DOA-authorized signature.

(3) *Required for Compliance (RC):* For any service information referenced in EASA AD 2018-0228 that contains RC procedures and tests: Except as required by paragraph (j)(2) of this AD, RC procedures and tests must be done to comply with this AD; any procedures or tests that are not identified as RC are recommended. Those procedures and tests that are not identified as RC may be deviated from using accepted methods in accordance with the operator's maintenance or inspection program without obtaining approval of an AMOC, provided the procedures and tests identified as RC can be done and the airplane can be put back in an airworthy condition. Any substitutions or changes to procedures or tests identified as RC require approval of an AMOC.

#### (k) Related Information

For more information about this AD, contact Vladimir Ulyanov, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; phone and fax: 206-231-3229

#### (l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) European Aviation Safety Agency (EASA) AD 2018-0228, dated October 22, 2018.

(ii) [Reserved]

(3) For EASA AD 2018-0228, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 89990 6017; email: [ADS@easa.europa.eu](mailto:ADS@easa.europa.eu); Internet: [www.easa.europa.eu](http://www.easa.europa.eu). You may find this EASA AD on the EASA website at <https://ad.easa.europa.eu>.

(4) You may view this EASA AD at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this

material at the FAA, call 206-231-3195. EASA AD 2018-0228 may be found in the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2019-0191.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Des Moines, Washington, on March 26, 2019.

**Michael Kaszycki,**

*Acting Director, System Oversight Division, Aircraft Certification Service.*

[FR Doc. 2019-07187 Filed 4-10-19; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2019-0190; Product Identifier 2018-NM-177-AD; Amendment 39-19607; AD 2019-06-09]

RIN 2120-AA64

#### Airworthiness Directives; Airbus SAS Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule; request for comments.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for certain Airbus SAS Model A350-941 airplanes. This AD was prompted by reports that baby bassinet inserts installed on airplane stowages and partitions were found loose because a self-securing fixation device (Loctite) had not been applied. This AD requires repetitive tightness checks of the baby bassinet inserts installed on stowages and partitions and, depending on findings, accomplishment of applicable corrective actions, as specified in an European Aviation Safety Agency (EASA) AD, which is incorporated by reference. We are issuing this AD to address the unsafe condition on these products.

**DATES:** This AD becomes effective April 26, 2019.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of April 26, 2019.

We must receive comments on this AD by May 28, 2019.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods: