

document by which a Federal agency makes know its intentions to award discretionary grants or cooperative agreements, usually as a result of competition for funds. FOA announcements may be known as program announcements, notices of funding availability, solicitations, or other names depending on the agency and type of program. FOA announcements can be found at *www.Grants.gov* in the Search Grants tab and on the funding agency's or program's website.

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■ 14. Revise § 1776.6 to read as follows:

§ 1776.6 Funding availability.

A FOA will be posted to *www.Grants.gov* in fiscal years that funds are available for this program. The FOA will establish the period during which applications for such funds may be submitted for consideration.

■ 15. Amend § 1776.8 by revising paragraph (d) to read as follows:

§ 1776.8 Methods for submitting applications.

* * * * *

(d) The methods of submitting applications may be changed from time to time to reflect changes in addresses and electronic submission procedures. The applicant should refer to the most recent FOA for notice of any such changes. In the event of any discrepancy, the most recent FOA must be followed.

PART 1783—REVOLVING FUNDS FOR FINANCING WATER AND WASTEWATER PROJECTS (REVOLVING FUND PROGRAM)

■ 16. The authority citation for part 1783 continues to read as follows:

Authority: 7 U.S.C. 1926 (a)(2)(B).

Subpart A—General

■ 17. Amend § 1783.3 by adding a definition for “Funding opportunity announcement” in alphabetical order to read as follows:

§ 1783.3 What definitions are used in this regulation?

* * * * *

Funding opportunity announcement (FOA) means a publicly available document by which a Federal agency makes know its intentions to award discretionary grants or cooperative agreements, usually as a result of competition for funds. FOA announcements may be known as program announcements, notices of funding availability, solicitations, or other names depending on the agency

and type of program. FOA announcements can be found at *Grants.gov* in the Search Grants tab and on the funding agency's or program's website.

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Subpart B—Revolving Loan Program Grants

■ 18. Revise § 1783.6 to read as follows:

§ 1783.6 When will applications for grants be accepted?

A FOA will be posted to *www.Grants.gov* in fiscal years that funds are available for this program. The FOA will establish the period during which applications for such funds may be submitted for consideration.

■ 19. Amend § 1783.8 by revising the second sentence of paragraph (c) to read as follows:

§ 1783.8 What are the acceptable methods for submitting applications?

* * * * *

(c) * * * Applicants should refer to the most recent FOA for notice of any such changes. In the event of any discrepancy, the information contained in the FOA must be followed. * * *

Dated: August 27, 2018.

Christopher A. McLean,
Acting Administrator, Rural Utilities Service.
[FR Doc. 2018–19199 Filed 9–4–18; 8:45 am]

BILLING CODE 3410–15–P

**DEPARTMENT OF TRANSPORTATION
Federal Aviation Administration**

14 CFR Part 25

[Docket No. FAA–2015–0309; Special Conditions No. 25–594A–SC]

Special Conditions: The Boeing Company (Boeing) Model 747–8 Airplane, Dynamic Test Requirements for Single-Occupant, Oblique (Side-Facing) Seats, With or Without Airbag Devices or 3-Point Restraints

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Amended final special conditions; request for comments.

SUMMARY: These amended special conditions are issued for the Boeing Model 747–8 airplane. This amendment states that the Boeing Model 747–8 airplane oblique (side-facing) seats may be installed at an angle of 18 to 45 degrees to the airplane centerline and may include a 3-point or airbag restraint system, or both, for occupant restraint and injury protection. Additionally, this

amendment changes paragraphs 4 through 8 of the special conditions section. This airplane will have novel or unusual design features when compared to the state of technology envisioned in the airworthiness standards for transport category airplanes. These design features are oblique (side-facing) single-occupant seats equipped with or without airbag devices or 3-point restraints.

The applicable airworthiness regulations do not contain adequate or appropriate safety standards for these design features. These special conditions contain the additional safety standards the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

DATES: This action is effective on Boeing on September 5, 2018. Send comments on or before October 22, 2018.

ADDRESSES: Send comments identified by Docket No. FAA–2015–0309 using any of the following methods:

- *Federal eRegulations Portal:* Go to <http://www.regulations.gov/> and follow the online instructions for sending your comments electronically.

- *Mail:* Send comments to Docket Operations, M–30, U.S. Department of Transportation (DOT), 1200 New Jersey Avenue SE, Room W12–140, West Building Ground Floor, Washington, DC, 20590–0001.

- *Hand Delivery or Courier:* Take comments to Docket Operations in Room W12–140 of the West Building Ground Floor at 1200 New Jersey Avenue SE, Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

- *Fax:* Fax comments to Docket Operations at 202–493–2251.

Privacy: The FAA will post all comments it receives, without change, to <http://www.regulations.gov/>, including any personal information the commenter provides. Using the search function of the docket website, anyone can find and read the electronic form of all comments received into any FAA docket, including the name of the individual sending the comment (or signing the comment for an association, business, labor union, etc.). DOT's complete Privacy Act Statement can be found in the **Federal Register** published on April 11, 2000 (65 FR 19477–19478).

Docket: Background documents or comments received may be read at <http://www.regulations.gov/> at any time. Follow the online instructions for accessing the docket or go to Docket Operations in Room W12–140 of the West Building Ground Floor at 1200 New Jersey Avenue SE, Washington,

DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

FOR FURTHER INFORMATION CONTACT: John Shelden, Airframe and Cabin Safety Section, AIR-675, Transport Standards Branch, Policy and Innovation Division, Aircraft Certification Service, Federal Aviation Administration, 2200 South 216th Street, Des Moines, Washington 98198; telephone and fax 206-231-3214; email John.Shelden@faa.gov.

SUPPLEMENTARY INFORMATION: The substance of these special conditions has been published in the **Federal Register** for public comment in several prior instances with no substantive comments received. The FAA therefore finds it unnecessary to delay the effective date and finds that good cause exists for making these special conditions effective upon publication in the **Federal Register**.

Comments Invited

We invite interested people to take part in this rulemaking by sending written comments, data, or views. The most helpful comments reference a specific portion of the special conditions, explain the reason for any recommended change, and include supporting data.

We will consider all comments we receive by the closing date for comments. We may change these special conditions based on the comments we receive.

Background

On February 3, 2014, Boeing applied for an amendment to Type Certificate no. A20WE to allow installation of single-occupant, oblique (side-facing) seats with or without airbag devices or 3-point restraints in the Boeing Model 747-8 airplanes.

Boeing requested special conditions to allow installation of oblique business-class passenger seats in the Boeing Model 747-8 airplane. The seating configuration Boeing proposes in Certification Plan no. 15090, "Installation of Business Class Zodiac Seats and Furniture for 747-8 TRX RC076," consists of Zodiac Cirrus III model oblique (side-facing), pod-style, business-class seats (with surrounding shells and front-row furniture) installed at an angle of up to 30 degrees to the airplane longitudinal centerline. These seats will include inflatable restraint (airbag) systems for occupant restraint and injury protection.

On November 22, 2017, Boeing applied for a change to Type Certificate No. A20WE for the installation of oblique (side-facing) passenger seats and surrounding furniture in the Boeing

Model 747-8 airplane. These oblique (side-facing) seats may be installed at an angle of 18 to 45 degrees to the airplane centerline and may include a 3-point or airbag restraint system, or both, for occupant restraint and injury protection.

The Boeing Model 747-8 airplane is a four-engine, transport category airplane with a maximum certified passenger capacity of 605, and a maximum takeoff weight of 987,000 lbs.

Type Certification Basis

Under the provisions of title 14, Code of Federal Regulations (14 CFR) 21.101, Boeing must show that the Model 747-8 airplane meets the applicable provisions of the regulations listed in Type Certificate no. A20WE, or the applicable regulations in effect on the date of application for the change, except for earlier amendments as agreed upon by the FAA. The regulations listed in the type certificate are commonly referred to as the "original type certification basis." The regulations listed in Type Certificate no. A20WE are as follows:

14 CFR part 25, Amendments 25-1 through 25-120, with exceptions permitted by § 21.101. In addition, the certification basis includes certain special conditions, exemptions, or later amended sections of the applicable part that are not relevant to these special conditions.

If the Administrator finds that the applicable airworthiness regulations (*i.e.*, 14 CFR part 25) do not contain adequate or appropriate safety standards for Boeing Model 747-8 airplanes because of a novel or unusual design feature, special conditions are prescribed under § 21.16.

Special conditions are initially applicable to the model for which they are issued. Should the type certificate for that model be amended later to include any other model that incorporates the same novel or unusual design feature, or should any other model already included on the same type certificate be modified to incorporate the same novel or unusual design feature, these special conditions would also apply to the other model under § 21.101. In addition to the applicable airworthiness regulations and special conditions, the Boeing Model 747-8 airplane must comply with the fuel-vent and exhaust-emission requirements of 14 CFR part 34, and the noise-certification requirements of 14 CFR part 36.

The FAA issues special conditions as defined in 14 CFR 11.19, in accordance with § 11.38, and they become part of the type certification basis under § 21.101.

Novel or Unusual Design Features

The business-class seating configuration Boeing proposes is novel or unusual due to the seat installation at 30 degrees to the airplane centerline, the airbag-system installation, and the seat/occupant interface with the surrounding furniture that introduces occupant alignment and loading concerns. The proposed business-class seating configuration also is beyond the limits of current acceptable equivalent-level-of-safety findings. These oblique (side-facing) seats may be installed at an angle of 18 to 45 degrees to the airplane centerline and may include a 3-point or airbag restraint system, or both, for occupant restraint and injury protection.

The existing regulations do not provide adequate or appropriate safety standards for occupants of oblique-angled seats with airbag systems. To provide a level of safety that is equivalent to that afforded occupants of forward- and aft-facing seats, additional airworthiness standards, in the form of special conditions, are necessary. These special conditions supplement part 25 and, more specifically, supplement §§ 25.562 and 25.785. The requirements contained in these special conditions consist of both test conditions and injury pass/fail criteria.

Discussion

The FAA has been conducting and sponsoring research on appropriate injury criteria for oblique (side-facing) seat installations. However, the FAA research program is not complete and we may update these criteria as we obtain further research results. To reflect current research findings, the FAA issued policy statement PS-ANM-25-03-R1 to update injury criteria for fully side facing seats, and the policy statement PS-AIR-25-27, to define injury criteria for oblique (side-facing) seats.

The proposed Boeing Model 747-8 airplanes business-class seat installation is novel such that the current Boeing Model 747-8 airplane certification basis does not adequately address protection of the occupant's neck and spine for seat configurations that are positioned at an angle greater than 18 degrees from the airplane centerline. These special conditions for oblique (side-facing) seat installations do not adequately address oblique seats, reflecting the current research results, with or without 3-point or airbag restraint systems. Therefore, Boeing's proposed configuration will require amended special conditions.

The installation of passenger seats at angles of 18 to 45 degrees to the airplane centerline are unique due to the seat/

occupant interface with the surrounding furniture that introduces occupant alignment/loading concerns with or without the installation of a 3-point or airbag restraint system, or both. Ongoing research has invalidated previously released special conditions for oblique seat installations. These updated special conditions further address potential injuries to the occupant's neck and spine. As a result, this special condition replaces special condition 25-594-SC.

FAA-sponsored research has found that an un-restrained flailing of the upper torso, even when the pelvis and torso are nearly aligned, can produce serious spinal and torso injuries. At lower impact severities, even with significant misalignment between the torso and pelvis, these injuries did not occur. Tests with an FAA H-III anthropomorphic test device (ATD) have identified a level of lumbar spinal tension corresponding to the no-injury impact severity. This level of tension is included as a limit in the special conditions. The spine tension limit selected is conservative with respect to other aviation injury criteria since it corresponds to a no-injury loading condition.

As noted in the special conditions for each airbag restraint system, because an airbag restraint system is essentially a single use device, there is the potential that it could deploy under crash conditions that are not sufficiently severe as to require head injury protection from the airbag restraint system. Since an actual crash is frequently composed of a series of impacts before the airplane comes to rest, this could render the airbag restraint system useless if a larger impact follows the initial impact. This situation does not exist with energy absorbing pads or upper torso restraints, which tend to provide protection according to the severity of the impact. Therefore, the installation of the airbag restraint system should be such that the airbag restraint system will provide protection when it is required, and will not expend its protection when it is not needed.

Because these airbag restraint systems may or may not activate during various crash conditions, the injury criteria listed in these special conditions and in § 25.562 must be met in an event that is slightly below the activation level of the airbag restraint system. If an airbag restraint system is included with the oblique seats, the system must meet the requirements in one of the airbag (inflatable restraint) special conditions applicable to the Boeing Model 747-8 airplane.

These amended special conditions will provide head injury criteria, neck injury criteria, spine injury criteria, and body-to-wall contact criteria. They contain the additional safety standards that the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

Applicability

As discussed above, these special conditions are applicable to the Boeing Model 747-8 airplanes. Should Boeing apply at a later date for a change to the type certificates to include another model incorporating the same novel or unusual design feature, or should any other model already included on the same type certificate be modified to incorporate the same novel or unusual design feature, these special conditions would apply to the other model as well.

Conclusion

This action affects only certain novel or unusual design features on one model airplane. It is not a rule of general applicability.

The substance of these special conditions has been subjected to the notice and comment period in several prior instances and has been derived without substantive change from those previously issued. It is unlikely that prior public comment would result in a significant change from the substance contained herein. Therefore, because a delay would significantly affect the certification of the airplane, which is imminent, the FAA has determined that prior public notice and comment are unnecessary and impracticable, and good cause exists for adopting these special conditions upon publication in the **Federal Register**. The FAA is requesting comments to allow interested persons to submit views that may not have been submitted in response to the prior opportunities for comment described above.

List of Subjects in 14 CFR Part 25

Aircraft, Aviation safety, Reporting and recordkeeping requirements.

Authority Citation

The authority citation for these special conditions is as follows:

Authority: 49 U.S.C. 106(f), 106(g), 40113, 44701, 44702, 44704.

The Special Conditions

Accordingly, pursuant to the authority delegated to me by the Administrator, the following special conditions are issued as part of the type certification basis for the Boeing Model 747-8 airplane.

Side-Facing Seats Special Conditions

In addition to the requirements of § 25.562:

1. Head Injury Criteria:

Compliance with § 25.562(c)(5) is required, except that, if the ATD has no apparent contact with the seat/structure but has contact with an airbag, a head-injury criterion (HIC) unlimited score in excess of 1000 is acceptable, provided the HIC15 score (calculated in accordance with 49 CFR 571.208) for that contact is less than 700.

2. Body-to-Wall/Furnishing Contact:

If a seat is installed aft of structure (e.g., an interior wall or furnishing) that does not provide a homogenous contact surface for the expected range of occupants and yaw angles, then additional analysis and/or test(s) may be required to demonstrate that the injury criteria are met for the area that an occupant could contact. For example, if different yaw angles could result in different airbag performance, then additional analysis or separate test(s) may be necessary to evaluate performance.

3. Neck Injury Criteria:

The seating system must protect the occupant from experiencing serious neck injury. The assessment of neck injury must be conducted with the airbag device activated, unless there is reason to also consider that the neck-injury potential would be higher for impacts below the airbag-device deployment threshold.

a. The N_{ij} (calculated in accordance with 49 CFR 571.208) must be below 1.0, where $N_{ij} = F_z/F_{zc} + M_y/M_{yc}$, and N_{ij} critical values are:

- i. $F_{zc} = 1,530$ lb for tension
- ii. $F_{zc} = 1,385$ lb for compression
- iii. $M_{yc} = 229$ lb-ft in flexion
- iv. $M_{yc} = 100$ lb-ft in extension

b. In addition, peak F_z must be below 937 lb in tension and 899 lb in compression.

c. Rotation of the head about its vertical axis, relative to the torso, is limited to 105 degrees in either direction from forward-facing.

d. The neck must not impact any surface that would produce concentrated loading on the neck.

4. Spine and Torso Injury Criteria:

a. The lumbar spine tension (F_z) cannot exceed 1200 lb.

b. Significant concentrated loading on the occupant's spine, in the area between the pelvis and shoulders during impact, including rebound, is not acceptable. During this type of contact, the interval for any rearward (X direction) acceleration exceeding 20g must be less than 3 milliseconds as measured by the thoracic

instrumentation specified in 49 CFR part 572, subpart E filtered in accordance with SAE International (SAE) recommended practice J211/1, "Instrumentation for Impact Test—Part 1—Electronic Instrumentation."

c. The occupant must not interact with the armrest or other seat components in any manner significantly different than would be expected for a forward-facing seat installation.

5. *Pelvis Criteria:*

Any part of the load-bearing portion of the bottom of the ATD pelvis must not translate beyond the edges of the seat bottom seat-cushion supporting structure.

6. *Femur Criteria:*

Axial rotation of the upper leg (about the z-axis of the femur per SAE Recommended Practice J211/1) must be limited to 35 degrees from the nominal seated position. Evaluation during rebound does not need to be considered.

7. *ATD and Test Conditions:*

Longitudinal tests conducted to measure the injury criteria above must be performed with the FAA Hybrid III ATD, as described in SAE 1999-01-1609, "A Lumbar Spine Modification to the Hybrid III ATD for Aircraft Seat Tests." The tests must be conducted with an undeformed floor, at the most-critical yaw cases for injury and with all lateral structural supports (e.g. armrests or walls) installed.

Note: Boeing must demonstrate that the installation of seats via plinths or pallets meets all applicable requirements. Compliance with the guidance contained in policy memorandum PS-ANM-100-2000-00123, "Guidance for Demonstrating Compliance with Seat Dynamic Testing for Plinths and Pallets," dated February 2, 2000, is acceptable to the FAA.

8. *Inflatable Airbag Restraint Systems Special Conditions:*

If inflatable airbag restraint systems are installed, the airbag systems must meet the requirements in one of the airbag (inflatable restraint) special conditions applicable to the Boeing Model 747-8 airplane.

Issued in Des Moines, Washington, on August 22, 2018.

Victor Wicklund,

Manager, Transport Standards Branch, Policy and Innovation Division, Aircraft Certification Service.

[FR Doc. 2018-19216 Filed 9-4-18; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 25

[Docket No. FAA-2018-0335; Special Conditions No. 25-725-SC]

Special Conditions: Bombardier Inc., Model BD-700-2A12 and BD-700-2A13 Series Airplanes; Flight Envelope Protection: High Incidence Protection System

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final special conditions; correction.

SUMMARY: The FAA is correcting an error that appeared in the **Federal Register** on May 1, 2018, for special conditions No. 25-725-SC, Docket No. FAA-2018-0335. As published, there was an error in the citation and the correct citation has been added.

DATES: Effective on Bombardier on September 5, 2018.

FOR FURTHER INFORMATION CONTACT: Joe Jacobsen, Airplane and Flight Crew Interface, AIR-671, Transport Standards Branch, Policy and Innovation Division, Aircraft Certification Service, Federal Aviation Administration, 2200 South 216th Street, Des Moines, Washington 98198; telephone and fax 206-231-3158; email Joe.Jacobsen@faa.gov.

SUPPLEMENTARY INFORMATION:

Background

On April 25, 2018, the FAA issued Special Conditions No. 25-725-SC, Docket No. FAA-2018-0335, which was published in the **Federal Register** on May 1, 2018 (83 FR 18934). Those special conditions pertain to the high incidence protection system that replaces the stall warning system during normal operating conditions, prohibits the airplane from stalling, limits the angle of attack at which the airplane can be flown during normal low speed operation, and cannot be overridden by the flight crew for Bombardier Model BD-700-2A12 and BD-700-2A13 series airplanes. As published, part II, paragraph 7 of the final special conditions cited § 25.143(j)(2)(i) instead of § 25.143(j)(1). There are no substantive changes to the document and it was apparent that § 25.143(j)(1) should have been referenced from the beginning.

Correction

In the final special conditions document FR Doc. 2018-09126 (Filed 4-30-2018; 8:45 a.m.), published on

May 1, 2018 (83 FR 18934), make the following correction:

On page 18938, column 2, under part II, paragraph 7, correct "§ 25.143(j)(2)(i)" to read "§ 25.143(j)(1)".

Issued in Des Moines, Washington, on August 27, 2018.

Victor Wicklund,

Manager, Transport Standards Branch, Policy and Innovation Division, Aircraft Certification Service.

[FR Doc. 2018-19215 Filed 9-4-18; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2018-0163; Product Identifier 2017-NM-168-AD; Amendment 39-19386; AD 2018-18-07]

RIN 2120-AA64

Airworthiness Directives; The Boeing Company Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for certain The Boeing Company Model 757 airplanes. This AD was prompted by an evaluation by the design approval holder (DAH) indicating that the longitudinal lap splices of the fuselage skin are subject to widespread fatigue damage (WFD). This AD requires repetitive inspections of the longitudinal lap splices of the fuselage skin for cracking and protruding fasteners, and applicable corrective actions. We are issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective October 10, 2018.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of October 10, 2018.

ADDRESSES: For service information identified in this final rule, contact Boeing Commercial Airplanes, Attention: Contractual & Data Services (C&DS), 2600 Westminister Blvd., MC 110-SK57, Seal Beach, CA 90740-5600; telephone 562-797-1717; internet <https://www.myboeingfleet.com>. You may view this service information at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195.