

(202) 741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Fort Worth, Texas, on April 18, 2017.

Scott A. Horn,

Acting Manager, Rotorcraft Directorate, Aircraft Certification Service.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2017-0019; Directorate Identifier 2016-CE-038-AD; Amendment 39-18861; AD 2017-08-12]

RIN 2120-AA64

Airworthiness Directives; GROB Aircraft AG Gliders

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for GROB Aircraft AG Models GROB G 109 and GROB G 109B gliders. This AD results from mandatory continuing airworthiness information (MCAI) issued by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as broken pivots of the tail wheel mounting bracket resulting from corrosion and damage due to wear. We are issuing this AD to require actions to address the unsafe condition on these products.

DATES: This AD is effective May 31, 2017.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of May 31, 2017.

ADDRESSES: You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2017-0019; or in person at Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

For service information identified in this AD, contact GROB Aircraft AG, Product Support, Lettenbachstrasse 9, D-86874 Tussenhausen-Mattsies, Germany, telephone: + 49 (0) 8268-998-105; fax: + 49 (0) 8268-998-200; email:

productsupport@grob-aircraft.com; Internet: grob-aircraft.com. You may view this referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148. It is also available on the Internet at <http://www.regulations.gov> by searching for Docket No. FAA-2017-0019.

FOR FURTHER INFORMATION CONTACT: Jim Rutherford, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4165; fax: (816) 329-4090; email: jim.rutherford@faa.gov.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to GROB Aircraft AG Models GROB G 109 and GROB G 109B gliders. The NPRM was published in the **Federal Register** on January 18, 2017 (82 FR 5456). The NPRM proposed to correct an unsafe condition for the specified products and was based on mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country. The MCAI states:

Occurrences were reported of broken pivots of the tail wheel mounting bracket. Subsequent investigation attributed these events to corrosion and damage due to wear.

This condition, if not detected and corrected, could lead to loss of rudder control, resulting in reduced control of the powered sailplane.

To address this potentially unsafe condition, Grob Aircraft AG issued Mandatory Service Bulletin (MSB) 817-70 (hereafter referred to as ‘the MSB’ in this [EASA] AD) to provide inspection and repair instructions.

For the reasons described above, this [EASA] AD requires repetitive inspections of the tail wheel mounting bracket and, depending on findings, accomplishment of applicable corrective action(s).

The MCAI can be found in the AD docket on the Internet at: <https://www.regulations.gov/document?D=FAA-2017-0019-0002>.

Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM or on the determination of the cost to the public.

Conclusion

We reviewed the relevant data and determined that air safety and the

public interest require adopting the AD as proposed except for minor editorial changes. We have determined that these minor changes:

- Are consistent with the intent that was proposed in the NPRM for correcting the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM.

Related Service Information Under 1 CFR Part 51

We reviewed GROB Aircraft AG Service Bulletin No. MSB817-70, dated September 28, 2016, which describes procedures for inspection of the tail mounting bracket; and GROB Aircraft AG Repair Instruction RI 817-015, dated September 16, 2016, which provides instructions for any necessary repair. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section of this final rule.

Costs of Compliance

We estimate that this AD will affect 57 products of U.S. registry. We also estimate that it would take about 3 work-hours per product to comply with the basic requirements of this AD. The average labor rate is \$85 per work-hour. Required parts would cost about \$50 per product.

Based on these figures, we estimate the cost of this AD on U.S. operators to be \$17,385, or \$305 per product.

In addition, we estimate that any necessary follow-on actions would take about 5 work-hours and require parts costing \$100, for a cost of \$525 per product. We have no way of determining the number of products that may need these actions.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition

that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

(1) Is not a “significant regulatory action” under Executive Order 12866,

(2) Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),

(3) Will not affect intrastate aviation in Alaska, and

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2017-0019; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

2017-08-12 GROB Aircraft AG:

Amendment 39-18861; Docket No.

FAA-2017-0019; Directorate Identifier 2016-CE-038-AD.

(a) Effective Date

This airworthiness directive (AD) becomes effective May 31, 2017.

(b) Affected ADs

None.

(c) Applicability

This AD applies to GROB Aircraft AG Models GROB G 109 and GROB G 109B gliders, all serial numbers, certificated in any category.

(d) Subject

Air Transport Association of America (ATA) Code 32: Landing Gear.

(e) Reason

This AD was prompted by mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as broken pivots of the tail wheel mounting bracket resulting from corrosion and damage due to wear. We are issuing this AD to detect and correct if necessary any corrosion or damage to the tail wheel mounting bracket, which could cause loss of rudder control and result in reduced control.

(f) Actions and Compliance

Unless already done, do the following actions:

(1) Within the next 3 months after May 31, 2017 (the effective date of this AD) or 100 hours time-in-service (TIS) after May 31, 2017 (the effective date of this AD), whichever occurs first, and repetitively thereafter at intervals not to exceed every 100 hours TIS or 12 months, whichever occurs first, inspect the tail wheel mounting bracket following the Accomplishment Instructions in section 1.8 of GROB Aircraft AG Service Bulletin (SB) No. MSB817-70, dated September 28, 2016.

(2) If any damage is found during any inspection required in paragraph (f)(1) of this AD, before further flight, repair following GROB Aircraft AG Repair Instruction RI 817-015, dated September 16, 2016.

Note 1 to paragraph (f)(2) of this AD: The bolt in Figure 1, Pos. 10 of GROB Aircraft AG Repair Instruction RI 817-015, dated September 16, 2016, is welded into place onto the steel base plate. Therefore, in order to facilitate the removal of the bolt, the welding seams may be carefully ground off using caution to not damage the steel base plate, instead of completely cutting off the bolt head.

(3) Repairs made as required by paragraph (f)(2) of this AD do not qualify as terminating action for the repetitive inspections required in paragraph (f)(1) of this AD.

(g) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures

found in 14 CFR 39.19. Send information to ATTN: Jim Rutherford, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4165; fax: (816) 329-4090; email: jim.rutherford@faa.gov. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(h) Related Information

Refer to MCAI European Aviation Safety Agency (EASA) AD No.: 2016-0228, dated November 14, 2016, for related information. The MCAI can be found in the AD docket on the Internet at: <https://www.regulations.gov/document?D=FAA-2017-0019-0002>.

(i) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) GROB Aircraft AG Service Bulletin (SB) No. MSB817-70, dated September 28, 2016.

(ii) GROB Aircraft AG Repair Instruction RI 817-015, dated September 16, 2016.

(3) For GROB Aircraft AG service information identified in this AD, contact GROB Aircraft AG, Product Support, Lettenbachstrasse 9, D-86874 Tussenhausen-Mattsies, Germany, telephone: + 49 (0) 8268-998-105; fax: + 49 (0) 8268-998-200; email: productsupport@grob-aircraft.com; Internet: grob-aircraft.com.

(4) You may review this referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148. In addition, you can access this service information on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2017-0019.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Kansas City, Missouri, on April 13, 2017.

Brian Yanez,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

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