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This section of the FEDERAL REGISTER contains regulatory documents having general applicability and legal effect, most of which are keyed to and codified in the Code of Federal Regulations, which is published under 50 titles pursuant to 44 U.S.C. 1510.

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## NUCLEAR REGULATORY COMMISSION

### 10 CFR Part 72

[NRC-2016-0200]

RIN 3150-AJ86

#### List of Approved Spent Fuel Storage Casks: AREVA Inc., Standardized NUHOMS® Cask System, Certificate of Compliance No. 1004, Amendment No. 14, and Revision 1 of the Initial Certificate, Amendment Nos. 1 Through 11, and Amendment No. 13

**AGENCY:** Nuclear Regulatory Commission.

**ACTION:** Direct final rule; confirmation of effective date.

**SUMMARY:** The U.S. Nuclear Regulatory Commission (NRC) is confirming the effective date of April 25, 2017, for the direct final rule that was published in the **Federal Register** on January 25, 2017. The direct final rule amended the NRC's spent fuel storage regulations by revising the "List of approved spent fuel storage casks" to include Amendment No. 14 and Revision 1 to the Initial Certificate, Amendment Nos. 1 through 11, and Amendment No. 13 to Certificate of Compliance (CoC) No. 1004 for the AREVA Inc., Standardized NUHOMS® Cask System.

**DATES:** *Effective Date:* The effective date of April 25, 2017, for the direct final rule published January 25, 2017 (82 FR 8353), is confirmed.

**ADDRESSES:** Please refer to Docket ID NRC-2016-0200 when contacting the NRC about the availability of information for this action. You may obtain publicly-available information related to this action by any of the following methods:

- *Federal Rulemaking Web site:* Go to <http://www.regulations.gov> and search for Docket ID NRC-2016-0200. Address questions about NRC dockets to Carol Gallagher; telephone: 301-415-3463; email: [Carol.Gallagher@nrc.gov](mailto:Carol.Gallagher@nrc.gov). For

technical questions, contact the individual listed in the **FOR FURTHER INFORMATION CONTACT** section of this document.

- *NRC's Agencywide Documents Access and Management System (ADAMS):* You may obtain publicly-available documents online in the ADAMS Public Documents collection at <http://www.nrc.gov/reading-rm/adams.html>. To begin the search, select "ADAMS Public Documents" and then select "Begin Web-based ADAMS Search." For problems with ADAMS, please contact the NRC's Public Document Room (PDR) reference staff at 1-800-397-4209, 301-415-4737, or by email to [pdr.resource@nrc.gov](mailto:pdr.resource@nrc.gov). The ADAMS accession number for each document referenced (if it is available in ADAMS) is provided the first time that it is mentioned in the **SUPPLEMENTARY INFORMATION** section.

- *NRC's PDR:* You may examine and purchase copies of public documents at the NRC's PDR, Room O1-F21, One White Flint North, 11555 Rockville Pike, Rockville, Maryland 20852.

**FOR FURTHER INFORMATION CONTACT:** Edward Lohr, Office of Nuclear Material Safety and Safeguards, U.S. Nuclear Regulatory Commission, Washington, DC 20555-0001; telephone: 301-415-0253; email: [Edward.Lohr@nrc.gov](mailto:Edward.Lohr@nrc.gov).

**SUPPLEMENTARY INFORMATION:** On January 25, 2017 (82 FR 8353), the NRC published a direct final rule amending § 72.214 of title 10 of the *Code of Federal Regulations* by revising the "List of approved spent fuel storage casks" to include Amendment No. 14 and Revision 1 to the Initial Certificate, Amendment Nos. 1 through 11, and Amendment No. 13 to CoC No. 1004 for the AREVA Inc., Standardized NUHOMS® Cask System. Amendment No. 14 revises multiple items in the technical specifications (TSs) for dry shielded canister (DSC) models listed under CoC No. 1004; most of these revisions involve changes to the authorized contents. The revisions to the Initial Certificate, Amendment Nos. 1 through 11, and Amendment No. 13 remove language in the TSs that requires a transfer cask containing a DSC to be returned to the spent fuel pool following a drop of over 15 inches.

In the direct final rule, the NRC stated that if no significant adverse comments were received, the direct final rule would become effective on April 25,

2017. As described more fully in the direct final rule, a significant adverse comment is a comment where the commenter explains why the rule would be inappropriate, including challenges to the rule's underlying premise or approach, or would be ineffective or unacceptable without a change. Because no significant adverse comments were received, the direct final rule will become effective as scheduled.

During the comment period for the direct final rule, the NRC identified misspellings and page numbering errors in the draft CoCs and TSs associated with Revision 1 of the Initial Certificate, Amendment Nos. 1 through 11, and Amendment No. 13. The final CoCs and TSs have been corrected for these administrative errors. Also, the draft CoCs were changed to update the signature block for issuing authority. The final CoCs, TSs, and the final Safety Evaluation Report for Revision 1 of the Initial Certificate, Amendment Nos. 1 through 11, and Amendment No. 13 can be viewed in ADAMS under Accession No. ML17067A412.

The final CoC, TSs, and the final Safety Evaluation Report for Amendment No. 14 to CoC No. 1004 can be viewed in ADAMS under Accession No. ML17093A261.

Dated at Rockville, Maryland, this 7th day of April, 2017.

For the Nuclear Regulatory Commission.

**Cindy Bladey,**

*Chief, Rules, Announcements, and Directives Branch, Division of Administrative Services, Office of Administration.*

[FR Doc. 2017-07422 Filed 4-12-17; 8:45 am]

**BILLING CODE 7590-01-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2017-0245; Directorate Identifier 2017-NM-023-AD; Amendment 39-18841; AD 2017-07-03]

RIN 2120-AA64

#### Airworthiness Directives; Airbus Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule; correction.

**SUMMARY:** The FAA is correcting an airworthiness directive (AD) that published in the **Federal Register**. That AD applies to all Airbus Model A330–243, –243F, –341, –342, and –343 airplanes. As published, the AD number specified in the preamble and regulatory text is incorrect. This document corrects that error. In all other respects, the original document remains the same.

**DATES:** This correction is effective April 17, 2017.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of April 17, 2017 (82 FR 15985, March 31, 2017).

**ADDRESSES:** For service information identified in this final rule, contact Airbus SAS, Airworthiness Office—EAL, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 45 80; email [airworthiness.A330-A340@airbus.com](mailto:airworthiness.A330-A340@airbus.com); Internet <http://www.airbus.com>. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221. It is also available on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2017–0245.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (phone: 800–647–5527) is Docket Management Facility, U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590.

**FOR FURTHER INFORMATION CONTACT:** Vladimir Ulyanov, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057–3356; telephone 425–227–1138; fax 425–227–1149.

**SUPPLEMENTARY INFORMATION:** As published, Airworthiness Directive 2017–07–05, Amendment 39–18841 (82 FR 15985, March 31, 2017), requires an inspection to determine if affected hydraulic pressure tube assemblies are installed, and replacement with serviceable hydraulic pressure tube

assemblies if necessary, for all Airbus Model A330–243, –243F, –341, –342, and –343 airplanes. That AD also requires repetitive replacements of serviceable hydraulic pressure tube assemblies.

#### Need for the Correction

As published, the AD number specified in the preamble and regulatory text is incorrect. The incorrectly specified number was AD 2017–07–05, which is assigned to another AD; the correct number is AD 2017–07–03.

#### Related Service Information Under 14 CFR Part 51

Airbus has issued Alert Operators Transmission (AOT) A71L012–16, Revision 01, dated February 24, 2017. The service information describes procedures for replacing hydraulic pressure tube assembly, part number (P/N) AE711121–18, and hydraulic pressure tube assembly, P/N AE711121–18 Rev A. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

#### Correction of Publication

This document corrects an error and correctly adds the AD as an amendment to 14 CFR 39.13. Although no other part of the preamble or regulatory information has been corrected, we are publishing the entire rule in the **Federal Register**.

The effective date of this AD remains April 17, 2017.

Since this action only corrects an AD number, it has no adverse economic impact and imposes no additional burden on any person. Therefore, we have determined that notice and public procedures are unnecessary.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Correction

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Corrected]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

**2017–07–03 Airbus:** Amendment 39–18841; Docket No. FAA–2017–0245; Directorate Identifier 2017–NM–023–AD.

#### (a) Effective Date

This AD is effective April 17, 2017.

#### (b) Affected ADs

None.

#### (c) Applicability

This AD applies to Airbus Model A330–243, –243F, –341, –342, and –343 airplanes, certificated in any category, all manufacturer serial numbers.

#### (d) Subject

Air Transport Association (ATA) of America Code 71, Powerplant.

#### (e) Reason

This AD was prompted by a determination that cracks can develop on the ripple damper of the hydraulic pressure tube assembly, which could lead to hydraulic leakage and consequent loss of the green hydraulic system. This AD was also prompted by reports of failure of the ripple damper of the hydraulic pressure tube assembly. We are issuing this AD to prevent cracking and failure of the ripple damper of the hydraulic pressure tube assembly, which could, in combination with other system failures, result in reduced control of the airplane.

#### (f) Compliance

Comply with this AD within the compliance times specified, unless already done.

#### (g) Definition of Affected Part

For the purpose of this AD, a hydraulic pressure tube assembly, part number (P/N) AE711121–18, as introduced by Airbus mod 205242, is hereafter referred to as an “affected part” in this AD.

#### (h) Definition of Serviceable Part

For the purpose of this AD, a “serviceable part” is a hydraulic pressure tube assembly (which has a double-welded ripple damper installed), P/N AE711121–18 Rev A, that has accumulated fewer than 800 total flight cycles since first installation on an airplane. The hydraulic pressure tube assembly, P/N AE711121–18 Rev A, is introduced by Airbus mod 206979 on the production line.

#### (i) Identification of Affected Parts

Within 15 days after April 17, 2017 (the effective date of this AD), inspect to determine the part number of the hydraulic pressure tube assembly that is installed on each engine. A review of airplane maintenance records is acceptable in lieu of this inspection if the part number of the hydraulic pressure tube assembly can be conclusively determined from that review.

#### (j) Replacement of Affected Parts

Within the compliance time specified in table 1 to paragraph (j) of this AD, as

applicable, or within 4 months after April 17, 2017 (the effective date of this AD), whichever occurs first, replace each affected

part (see paragraph (g) of this AD) with a serviceable part (see paragraph (h) of this AD), in accordance with the instructions of

Airbus Alert Operators Transmission (AOT) A71L012-16, Revision 01, dated February 24, 2017.

TABLE 1 TO PARAGRAPH (j) OF THIS AD—REPLACEMENT COMPLIANCE TIMES

Flight cycles accumulated *	Compliance time
Fewer than 775 total flight cycles .....	Before exceeding 800 total flight cycles on the affected hydraulic pressure tube assembly since first installation on an airplane.
775 total flight cycles or more .....	Within 25 flight cycles after April 17, 2017 (the effective date of this AD).
An unknown number of flight cycles accumulated .....	Within 25 flight cycles after April 17, 2017 (the effective date of this AD).

\* Unless specified otherwise, the flight cycles in the “flight cycles accumulated” column of table 1 to paragraph (j) of this AD are those accumulated by an affected hydraulic pressure tube assembly, on April 17, 2017 (the effective date of this AD), since first installation on an airplane.

**(k) Repetitive Replacement of Serviceable Parts—Life Limit**

Before a serviceable part (see paragraph (h) of this AD) exceeds 800 total flight cycles since first installation on an airplane, replace it with a serviceable part, in accordance with the instructions of Airbus AOT A71L012-16, Revision 01, dated February 24, 2017.

**(l) Engine Installation Limitation**

As of April 17, 2017 (the effective date of this AD), except as required by paragraph (m) of this AD, it is allowed to install on any airplane a replacement engine having an affected part (see paragraph (g) of this AD) installed, provided that, before that affected part exceeds 800 total flight cycles since first installation on an airplane, or within 4 months after April 17, 2017 (the effective date of this AD), whichever occurs first, the part is replaced with a serviceable part (see paragraph (h) of this AD), in accordance with the instructions of Airbus AOT A71L012-16, Revision 01, dated February 24, 2017.

**(m) Parts and Engine Installation Prohibition**

As of 4 months after April 17, 2017 (the effective date of this AD): Do not install on any airplane an affected part (see paragraph (g) of this AD), or an engine having an affected part installed.

**(n) Credit for Previous Actions**

This paragraph provides credit for actions required by paragraph (j) of this AD, if those actions were performed before April 17, 2017 (the effective date of this AD) using Airbus AOT A71L012-16, dated December 22, 2016.

**(o) Other FAA AD Provisions**

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Vladimir Ulyanov, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone 425-227-1138; fax 425-227-1149.

Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) *Contacting the Manufacturer*: For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA; or the European Aviation Safety Agency (EASA); or Airbus’s EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

**(p) Related Information**

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA Airworthiness Directive 2017-0041, dated February 24, 2017; corrected February 28, 2017, for related information. This MCAI may be found in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2017-0245.

(2) For more information about this AD, contact Vladimir Ulyanov, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone 425-227-1138; fax 425-227-1149. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov.

(3) Service information identified in this AD that is not incorporated by reference is available at the addresses specified in paragraphs (q)(4) and (q)(5) of this AD.

**(q) Material Incorporated by Reference**

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(3) The following service information was approved for IBR on April 17, 2017 (82 FR 15985, March 31, 2017).

(i) Airbus Alert Operators Transmission (AOT) A71L012-16, Revision 01, dated February 24, 2017.

(ii) Reserved.

(4) For service information identified in this AD, contact Airbus SAS, Airworthiness Office—EAL, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 45 80; email [airworthiness.A330-A340@airbus.com](mailto:airworthiness.A330-A340@airbus.com); Internet <http://www.airbus.com>.

(5) You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

(6) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Renton, Washington, on April 5, 2017.

**Michael Kaszycki,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2017-07442 Filed 4-12-17; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF HOMELAND SECURITY**

**Coast Guard**

**33 CFR Part 100**

[Docket No. USCG-2017-0096]

**RIN 1625-AA08**

**Special Local Regulation; Red Bull Air Race—San Diego 2017; San Diego Bay, CA**

**AGENCY:** Coast Guard, DHS.

**ACTION:** Temporary final rule.

**SUMMARY:** The Coast Guard is temporarily establishing special local regulations for the Red Bull Air Race—San Diego 2017 event held on the navigable waters of San Diego Bay, California. This action is necessary to provide for the safety of life on navigable waters during the event. This