

(since first installation on an airplane, or since last NBB replacement, whichever occurs later) is 16,000 total flight cycles.

(2) As of July 31, 2017: The THSA flight-cycle limit (since first installation on an airplane, or since last NBB replacement, whichever occurs later) is 14,000 total flight cycles.

(3) As of July 31, 2018: The THSA flight-cycle limit (since first installation on an airplane, or since last NBB replacement, whichever occurs later) is 12,000 total flight cycles.

(j) THSA Replacement for Airbus Model A340-500 and -600 Series Airplanes

For Airbus Model A340-500 and A340-600 series airplanes having a THSA with a part number specified in paragraph (g)(2) of this AD: Not later than the date specified in paragraphs (j)(1), (j)(2), (j)(3), and (j)(4) of this AD, as applicable: For any THSA having reached or exceeded on that date the corresponding number of total flight cycles as specified in paragraphs (j)(1), (j)(2), (j)(3), and (j)(4) of this AD, as applicable, replace each affected THSA with a serviceable THSA, in accordance with the Accomplishment Instructions of Airbus Service Bulletin A340-27-5062, dated July 15, 2014.

(1) As of the effective date of this AD: The THSA flight-cycle limit (since first installation on an airplane, or since last NBB replacement, whichever occurs later) is 6,000 total flight cycles.

(2) As of April 30, 2017: The THSA flight-cycle limit (since first installation on an airplane, or since last NBB replacement, whichever occurs later) is 5,200 total flight cycles.

(3) As of April 30, 2018: The THSA flight-cycle limit (since first installation on an airplane, or since last NBB replacement, whichever occurs later) is 4,400 total flight cycles.

(4) As of April 30, 2019: The THSA flight-cycle limit (since first installation on an airplane, or since last NBB replacement, whichever occurs later) is 3,500 total flight cycles.

(k) THSA Replacement Intervals for All Airbus Airplanes Identified in Paragraph (c) of This AD

For any part installed as required by this AD having a part number identified in paragraph (g)(1) or (g)(2) of this AD: From the dates specified in paragraphs (i) and (j) of this AD, as applicable, and prior to exceeding the accumulated number of total flight cycles corresponding to each time, replace each affected THSA with a serviceable part, in accordance with the Accomplishment Instructions of the applicable service information identified in paragraphs (k)(1), (k)(2), and (k)(3) of this AD.

(1) Airbus Service Bulletin A330-27-3199, dated July 15, 2014.

(2) Airbus Service Bulletin A340-27-4190, dated July 15, 2014.

(3) Airbus Service Bulletin A340-27-5062, dated July 15, 2014.

(l) Definition of Serviceable THSA

For the purposes of this AD a serviceable THSA is a THSA:

(1) Having a part number identified in paragraph (g)(1) or (g)(2) of this AD that has not exceeded any of the total accumulated flight cycles identified in paragraphs (i)(1) through (i)(3) of this AD, or paragraphs (j)(1) through (j)(4) of this AD, as applicable; or

(2) Having a part number that is not identified in paragraph (g)(1) or (g)(2) of this AD.

(m) Parts Installation Limitation

From each date specified in paragraphs (i)(1), (i)(2), and (i)(3) of this AD, and paragraphs (j)(1) through (j)(4) of this AD, as applicable, a THSA having a part number identified in paragraph (g)(1) or (g)(2) of this AD may be installed on any airplane, provided the THSA has not exceeded the corresponding number of accumulated total flight cycles.

(n) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Vladimir Ulyanov, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone 425-227-1138; fax 425-227-1149. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Contacting the Manufacturer*: For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA; or the European Aviation Safety Agency (EASA); or Airbus's EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(3) *Required for Compliance (RC)*: If any service information contains procedures or tests that are identified as RC, those procedures and tests must be done to comply with this AD; any procedures or tests that are not identified as RC are recommended. Those procedures and tests that are not identified as RC may be deviated from using accepted methods in accordance with the operator's maintenance or inspection program without obtaining approval of an AMOC, provided the procedures and tests identified as RC can be done and the airplane can be put back in an airworthy condition. Any substitutions or changes to procedures or tests identified as RC require approval of an AMOC.

(o) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA Airworthiness Directive 2014-0257R1, dated May 29, 2015, for related information. This MCAI may be found in the AD docket on the Internet at <http://www.regulations.gov/#!documentDetail;D=FAA-2014-0006-0002>.

(2) For service information identified in this AD, contact Airbus SAS, Airworthiness Office—EAL, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 45 80; email airworthiness.A330-A340@airbus.com; Internet <http://www.airbus.com>. You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Issued in Renton, Washington, on October 30, 2015.

Jeffrey E. Duven,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2015-30822 Filed 12-22-15; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2015-7526; Directorate Identifier 2014-NM-217-AD]

RIN 2120-AA64

Airworthiness Directives; Airbus Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for all Airbus Model A318, A319, A320, and A321 series airplanes. This proposed AD was prompted by the discovery of corroded circlips in fuel vent protectors (FVP) having a certain part number. This proposed AD would require an inspection to determine the part number and serial number of the FVP, and replacement if necessary. We are proposing this AD to detect and correct corroded circlips. Corroded circlips could lead to failure of the circlips and consequent movement of the FVP, resulting in a reduction of the flame protector capability of the FVP cartridge, which could result in damage to the airplane in case of lightning impact or fire on the ground.

DATES: We must receive comments on this proposed AD by February 8, 2016.

ADDRESSES: You may send comments, using the procedures found in 14 CFR

11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal*: Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax*: 202-493-2251.

- *Mail*: U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

- *Hand Delivery*: U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Airbus, Airworthiness Office—ELAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email account.airworth-eas@airbus.com; Internet <http://www.airbus.com>. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-7526; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:

Sanjay Ralhan, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone 425-227-1405; fax 425-227-1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA-2015-7526; Directorate Identifier 2014-NM-217-AD” at the beginning of your comments. We specifically invite

comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union, has issued EASA Airworthiness Directive 2014-0234R1, dated December 11, 2014 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for all Airbus Model A318, A319, A320, and A321 series airplanes. The MCAI states:

On each aeroplane wing, a NACA [National Advisory committee for Aeronautics] duct assembly is installed, including a Fuel Vent Protector (FVP) which is used as flame arrester. This FVP is maintained in its NACA duct assembly by a circlip (also known as C-clip). Following a wing water pressure test, the FVP is removed and dried with heat. During an inspection after this test, several circlips were reported to be discoloured. Investigation revealed that a batch of circlips fitted on some FVP Part Number (P/N) 786073-1-0 have an increased risk of corrosion due to a manufacturing quality issue.

This condition, if not detected and corrected, could lead to circlip failure and consequent FVP movement, reducing the flame protector capability of the FVP cartridge, possibly resulting in damage to the aeroplane in case of lightning impact or fire on ground.

For the reason described above, EASA issued AD 2014-0234 to require identification by serial number (s/n) of the affected FVP P/N 786073-1-0 and removal from service [and replacement with a serviceable part].

This [EASA] AD is revised to clarify that only maintenance records since August 2012 should be consulted to demonstrate, as required by paragraph (2), that not replacement FVP has been installed.

You may examine the MCAI in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-7526.

Related Service Information Under 1 CFR Part 51

Airbus has issued Service Bulletin A320-28-1221, dated July 21, 2014. The service information describes procedures for inspecting the FVP to determine the part number, and

replacing the FVP if necessary. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section of this NPRM.

FAA’s Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of these same type designs.

Costs of Compliance

We estimate that this proposed AD affects 7 airplanes of U.S. registry.

We also estimate that it would take about 5 work-hours per product to comply with the basic requirements of this proposed AD. The average labor rate is \$85 per work-hour. Parts would cost \$25,640. Based on these figures, we estimate the cost of this proposed AD on U.S. operators to be up to \$182,455, or \$26,065 per product.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on

the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

Airbus: Docket No. FAA–2015–7526; Directorate Identifier 2014–NM–217–AD.

(a) Comments Due Date

We must receive comments by February 8, 2016.

(b) Affected ADs

None.

(c) Applicability

This AD applies to the airplanes specified in paragraphs (c)(1), (c)(2), (c)(3), and (c)(4) of this AD, certificated in any category, all manufacturer serial numbers.

- (1) Airbus Model A318–111, –112, –121, and –122 airplanes.
- (2) Airbus Model A319–111, –112, –113, –114, –115, –131, –132, and –133 airplanes.
- (3) Airbus Model A320–211, –212, –214, –231, –232, and –233 airplanes.
- (4) Airbus Model A321–111, –112, –131, –211, –212, –213, –231, and –232 airplanes.

(d) Subject

Air Transport Association (ATA) of America Code 28, Fuel.

(e) Reason

This AD was prompted by the discovery of corroded circlips in fuel vent protectors (FVP) having a certain part number. We are

issuing this AD to detect and correct corroded circlips. Corroded circlips could lead to failure of the circlips and consequent movement of the FVP, resulting in a reduction of the flame protector capability of the FVP cartridge, which could result in damage to the airplane in case of lightning impact or fire on the ground.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Inspection of FVP and Corrective Action

For airplanes having a manufacturer serial number specified in figure 1 to paragraphs (g) and (i) of this AD: At the earliest of the times specified in paragraphs (h)(1), (h)(2), and (h)(3) of this AD, do an inspection to determine the part number and serial number of the FVP. If the FVP has part number (P/ N) 786073–1–0 with a serial number that is specified in figure 2 to paragraphs (g) and (i) of this AD, and the FVP is not marked “Amdt B,” replace the FVP with a serviceable part, at the earliest of the times specified in paragraphs (h)(1), (h)(2), and (h)(3) of this AD, in accordance with the Accomplishment Instructions of Airbus Service Bulletin A320–28–1221, dated July 21, 2014. A review of airplane maintenance records is acceptable in lieu of this inspection if the part number and serial number of the FVP can be conclusively determined from that review.

FIGURE 1 TO PARAGRAPHS (g) AND (i) OF THIS AD—AFFECTED AIRPLANE MANUFACTURER SERIAL NUMBERS

5438	5461	5485 through 5488 inclusive	5536
5441	5463	5490 through 5493 inclusive	5539
5444	5464	5495 through 5505 inclusive	5541
5445	5469	5507 through 5515 inclusive	5544
5447	5473 through 5478 inclusive	5517	5547
5457	5481	5518	5551
5459	5482	5520 through 5527 inclusive	5553
5460	5483	5530	5556

FIGURE 2 TO PARAGRAPHS (g) AND (i) OF THIS AD—AFFECTED SERIAL NUMBERS FOR PART NUMBER 786073–1–0
[Manufactured during August 2012]

Serial number 786073IN0xxxx (xxxx indicates the last four digits)					
3752	3821	3868	3911	3966	4010
3753	3826	3871	3914	3967	4011
3754	3827	3874	3922	3969	4013
3755	3829	3877	3925	3971	4017
3756	3830	3878	3927	3972	4019
3757	3833	3882	3930	3977	4023
3758	3834	3893	3937	3978	4024
3759	3836	3897	3938	3980	4025
3760	3839	3898	3940	3981	4026
3761	3840	3899	3945	3982	4039
3787	3848	3900	3946	3983	4048
3788	3849	3901	3947	3984	4065
3810	3850	3904	3948	3985	4066
3812	3851	3905	3951	3986	4068
3814	3853	3906	3961	3987	4070
3817	3859	3907	3962	3996	4184
3819	3860	3908	3964	3997	4187
3820	3867	3910	3965	4009	None

(h) Compliance Times for the Requirements of Paragraph (g) of This AD

Do the actions required by paragraph (g) of this AD at the earliest of the times specified in paragraphs (h)(1), (h)(2), and (h)(3) of this AD.

(1) Before the accumulation of 5,000 total flight cycles after the date of manufacture of the airplane.

(2) Before the accumulation of 7,500 total flight hours after the date of manufacture of the airplane.

(3) Within 30 months after the date of manufacture of the airplane.

(i) Exclusion From Actions Required by Paragraph (g) of This AD

An airplane that does not have a manufacturer serial number specified in figure 1 to paragraphs (g) and (i) of this AD is excluded from the requirements of paragraph (g) of this AD, provided that, a FVP having P/N 786073-1-0 with a serial number specified in figure 2 to paragraphs (g) and (i) of this AD has not been installed on that airplane after July 2012. If a FVP having P/N 786073-1-0 with a serial number specified in figure 2 to paragraphs (g) and (i) of this AD is installed, or the serial number cannot be identified: Within 12 months after the effective date of this AD, replace the FVP with a serviceable part, in accordance with the Accomplishment Instructions of Airbus Service Bulletin A320-28-1221, dated July 21, 2014. A review of airplane maintenance records is acceptable if it can be conclusively determined from that review that a FVP having a serial number specified in figure 2 to paragraphs (g) and (i) of this AD has not been installed on that airplane after July 2012.

(j) Parts Installation Limitation

As of the effective date of this AD, a FVP having P/N 786073-1-0 and a serial number listed in figure 2 to paragraphs (g) and (i) of this AD may be installed on any airplane, provided the FVP is marked with "Amdt B."

(k) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Sanjay Ralhan, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone 425-227-1405; fax 425-227-1149. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Contacting the Manufacturer*: For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA; or the European Aviation Safety Agency (EASA); or Airbus's EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(l) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA Airworthiness Directive 2014-0234R1, dated December 11, 2014, for related information. This MCAI may be found in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-7526.

(2) For service information identified in this AD, contact Airbus, Airworthiness Office—EIAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email account.airworth-eas@airbus.com; Internet <http://www.airbus.com>. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Issued in Renton, Washington, on December 11, 2015.

Michael Kaszycki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2015-32082 Filed 12-22-15; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2015-7528; Directorate Identifier 2015-NM-004-AD]

RIN 2120-AA64

Airworthiness Directives; Airbus Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for all Airbus Model A300 B4-600, B4-600R, and F4-600R series airplanes, and Model A300 C4-605R Variant F airplanes (collectively called Model A300-600 series airplanes); and Model A310 series airplanes. This proposed AD was prompted by reports of premature aging of certain passenger chemical oxygen generators that resulted in the generators failing to

activate. This proposed AD would require an inspection to determine if certain passenger chemical oxygen generators are installed and replacement of affected passenger chemical oxygen generators. We are proposing this AD to prevent failure of the passenger chemical oxygen generator to activate and consequently not deliver oxygen during an emergency, possibly resulting in injury to airplane occupants.

DATES: We must receive comments on this proposed AD by February 8, 2016.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal*: Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax*: 202-493-2251.

- *Mail*: U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

- *Hand Delivery*: U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For Airbus service information identified in this proposed AD, contact Airbus SAS, Airworthiness Office—EAW, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone: +33 5 61 93 36 96; fax: +33 5 61 93 44 51; email: account.airworth-eas@airbus.com; Internet <http://www.airbus.com>.

For B/E Aerospace service information identified in this proposed AD, contact B/E Aerospace Inc., 10800 Pflumm Road, Lenexa, KS 66215; telephone: 913-338-9800; fax: 913-469-8419; Internet <http://beaerospace.com/home/globalsupport>.

You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-7528; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The