

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

Airbus: Docket No. FAA–2015–3990; Directorate Identifier 2014–NM–255–AD.

(a) Comments Due Date

We must receive comments by December 3, 2015.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Airbus Model A320–214, –232, and –233 airplanes; and Airbus Model A321–211 and –231 airplanes, certificated in any category, having manufacturer serial numbers (MSNs) 5583, 5598, 5602, 5604, 5608, 5610, 5613 through 5622 inclusive, 5624 through 5627 inclusive, 5629 through 5632 inclusive, 5634 through 5636 inclusive, 5638, 5640 through 5644 inclusive, 5646 through 5649 inclusive, 5651 through 5653 inclusive, 5655, 5657 through 5661 inclusive, 5663, 5665, 5667, 5670, 5672, 5673, and 5675.

(d) Subject

Air Transport Association (ATA) of America Code 25, Equipment/Furnishings.

(e) Reason

This AD was prompted by reports of incorrect installation of jiffy joint connectors on cables connected to certain passenger service units (PSU), which could cause the passenger oxygen container to malfunction if the connector becomes disengaged during flight due to vibration. We are issuing this AD to prevent failure of the door of the passenger oxygen container to open in the event of airplane decompression, resulting in lack of oxygen supply and consequent injury to occupants.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Inspection and Related Investigative and Corrective Actions

Within 7,500 flight hours or 26 months after the effective date of this AD, whichever

occurs first, do an inspection to identify the part number and serial number of each PSU and if an affected part number or serial number is found, do all applicable related investigative and corrective actions, in accordance with the Accomplishment Instructions of Airbus Service Bulletin A320–25–1B20, dated October 9, 2014. Do all applicable related investigative and corrective actions within 7,500 flight hours or 26 months after the effective date of this AD, whichever occurs first. An affected PSU part number or serial number is one listed in Appendix 1 of Airbus Operations GmbH Vendor Service Bulletin Z315H–25–004, dated September 26, 2014. A review of airplane maintenance records is acceptable in lieu of this inspection if the part number and serial number of the PSU can be conclusively determined from that review.

(h) Clarification of Vendor Service Information

Appendix 1 of Airbus Operations GmbH Vendor Service Bulletin Z315H–25–004, dated September 26, 2014, identifies Attachment 1 as the list of affected PSU part numbers and serial numbers. Also, the “List of Attachments” in Appendix 1, specifies Attachment 1 as Table 4, however “Attachment 1” and “Table 4” do not appear on any of the pages of the list of affected PSU part numbers and serial numbers, nor does a date. Furthermore, the pagination of the list of affected PSU part numbers and serial numbers is independent of the pagination of Airbus Operations GmbH Vendor Service Bulletin Z315H–25–004, dated September 26, 2014.

(i) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Sanjay Ralhan, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057–3356; telephone 425–227–1405; fax 425–227–1149. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Contacting the Manufacturer:* For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA; or the European Aviation Safety Agency (EASA); or Airbus’s EASA Design Organization Approval (DOA). If approved by

the DOA, the approval must include the DOA-authorized signature.

(3) *Required for Compliance (RC):* If any service information contains procedures or tests that are identified as RC, those procedures and tests must be done to comply with this AD; any procedures or tests that are not identified as RC are recommended. Those procedures and tests that are not identified as RC may be deviated from using accepted methods in accordance with the operator’s maintenance or inspection program without obtaining approval of an AMOC, provided the procedures and tests identified as RC can be done and the airplane can be put back in a serviceable condition. Any substitutions or changes to procedures or tests identified as RC require approval of an AMOC.

(j) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA Airworthiness Directive 2014–0256, dated November 26, 2014, for related information. This MCAI may be found in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2015–3990.

(2) For Airbus service information identified in this AD, contact Airbus, Airworthiness Office—EIAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email account.airworth-eas@airbus.com; Internet <http://www.airbus.com>. For Airbus Operations GMBH service information identified in this AD, contact Airbus Operations GMBH, Cabin Electronics, Lueneburger Schanze 30, 21614 Buxtehude, Germany; telephone +49 40 7437 46 32; telefax +49 40 7437 16 80; email ruediger.jansen@airbus.com. You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221.

Issued in Renton, Washington, on October 6, 2015.

Jeffrey E. Duven,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2015–26223 Filed 10–16–15; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2015–3989; Directorate Identifier 2014–NM–250–AD]

RIN 2120–AA64

Airworthiness Directives; Airbus Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for all Airbus Model A318; A319; A320; and A321 series airplanes. This proposed AD was prompted by reports of premature aging of certain passenger chemical oxygen generators that resulted in the generators failing to activate. This proposed AD would require an inspection to determine if certain passenger chemical oxygen generators are installed and replacement of affected passenger chemical oxygen generators. We are proposing this AD to prevent failure of the passenger chemical oxygen generator to activate and consequently not deliver oxygen during an emergency, possibly resulting in injury to the airplane occupants.

DATES: We must receive comments on this proposed AD by December 3, 2015.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.
- *Fax:* 202-493-2251.
- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.
- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For Airbus service information identified in this proposed AD, contact Airbus, Airworthiness Office—EIAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone: +33 5 61 93 36 96; fax: +33 5 61 93 44 51; email: account.airworth-eas@airbus.com; Internet <http://www.airbus.com>.

For B/E Aerospace service information identified in this proposed AD, contact B/E Aerospace Inc., 10800 Pflumm Road, Lenexa, KS 66215; telephone: 913-338-9800; fax: 913-469-8419; Internet <http://beaerospace.com/home/globalsupport>.

You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for

and locating Docket No. FAA-2015-3989; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone: 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Sanjay Ralhan, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone: 425-227-1405; fax: 425-227-1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA-2015-3989; Directorate Identifier 2014-NM-250-AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2015-0117, dated June 24, 2015; corrected August 7, 2015 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for all Airbus Model A318; A319; A320; and A321 series airplanes. The MCAI states:

Reports have been received indicating premature ageing of certain chemical oxygen generators, Part Number (P/N) 117042-XX (XX representing any numerical value), manufactured by B/E Aerospace. Some operators reported that when they tried to activate generators, some older units failed to activate. Given the number of failed units reported, all generators manufactured in 1999, 2000 and 2001 were considered unreliable.

This condition, if not corrected, could lead to failure of the generator to activate and

consequently not deliver oxygen during an emergency, possibly resulting in injury to aeroplane occupants.

To address this potential unsafe condition, Airbus issued Alert Operators Transmission (AOT) A35N006-14, making reference to B/E Aerospace Service Information Letter (SIL) D1019-01 (currently at Revision 1) and B/E Aerospace Service Bulletin (SB) 117042-35-001.

Consequently, EASA issued AD * * * (later revised) to require identification and replacement of the affected oxygen generators.

Since EASA AD 2014-0275R1 [<http://ad.easa.europa.eu/ad/2014-0275R1>] was issued, and following new investigation results, EASA have decided to introduce a life limitation concerning all P/N 117042-XX chemical oxygen generators, manufactured by B/E Aerospace.

For the reason described above, this [EASA] AD retains the requirements of the EASA AD 2014-0275R1, which is superseded, expands the scope of the [EASA] AD to include chemical oxygen generators manufactured after 2001, and requires their removal from service before exceeding 10 years since date of manufacture.

You may examine the MCAI in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-3989.

Related Service Information Under 1 CFR Part 51

Airbus has issued Alert AOT A35N006-14, dated December 10, 2014, including Appendix 01.

B/E Aerospace Inc. has issued Service Bulletin 117042-35-001, dated December 10, 2014.

This service information describes procedures to replace certain passenger chemical oxygen generators. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section of this NPRM.

FAA’s Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of these same type designs.

Costs of Compliance

We estimate that this proposed AD affects 953 airplanes of U.S. registry.

We also estimate that it would take about 2 work-hours per product to comply with the basic requirements of this proposed AD. The average labor rate is \$85 per work-hour. Required parts would cost about \$390 per product. Based on these figures, we estimate the cost of this proposed AD on U.S. operators to be \$533,680, or \$560 per product.

Paperwork Reduction Act

A federal agency may not conduct or sponsor, and a person is not required to respond to, nor shall a person be subject to penalty for failure to comply with, a collection of information subject to the requirements of the Paperwork Reduction Act unless that collection of information displays a current valid OMB control number. The control number for the collection of information required by this proposed AD is 2120-0056. The paperwork cost associated with this proposed AD has been detailed in the Costs of Compliance section of this document and includes time for reviewing instructions, as well as completing and reviewing the collection of information. Therefore, all reporting associated with this proposed AD is mandatory. Comments concerning the accuracy of this burden and suggestions for reducing the burden should be directed to the FAA at 800 Independence Ave. SW., Washington, DC 20591, ATTN: Information Collection Clearance Officer, AES-200.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations

for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

Airbus: Docket No. FAA-2015-3989; Directorate Identifier 2014-NM-250-AD.

(a) Comments Due Date

We must receive comments by December 3, 2015.

(b) Affected ADs

None.

(c) Applicability

This AD applies to the airplanes, certificated in any category, identified in paragraphs (c)(1), (c)(2), (c)(3), and (c)(4) of this AD; all manufacturer serial numbers, except those that have embodied Airbus modification 33125 (gaseous system for all oxygen containers) in production.

(1) Airbus Model A318-111, -112, -121, and -122 airplanes.

(2) Airbus Model A319-111, -112, -113, -114, -115, -131, -132, and -133 airplanes.

(3) Airbus Model A320-211, -212, -214, -231, -232, and -233 airplanes.

(4) Airbus Model A321-111, -112, -131, -211, -212, -213, -231, and -232 airplanes.

(d) Subject

Air Transport Association (ATA) of America Code 35, Oxygen.

(e) Reason

This AD was prompted by reports of premature aging of certain passenger chemical oxygen generators that resulted in the generators failing to activate. We are issuing this AD to prevent failure of the passenger chemical oxygen generator to activate and consequently not deliver oxygen during an emergency, possibly resulting in injury to the airplane occupants.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Part Number Inspection

Within 30 days after the effective date of this AD, do a one-time inspection of passenger chemical oxygen generators, part numbers (P/N) 117042-02 (15 minutes (min)-2 masks), 117042-03 (15 min-3 masks), 117042-04 (15 min-4 masks), 117042-22 (22 min-2 masks), 117042-23 (22 min-3 masks), and 117042-24 (22 min-4 masks) to determine the date of manufacture as specified in Airbus Alert Operators Transmission (AOT) A35N006-14, dated December 10, 2014, including Appendix 01. Refer to figure 1 and figure 2 to paragraph (g) of this AD for the location of the date. A review of airplane maintenance records is acceptable for the inspection required by paragraph (g) of this AD, provided the date of manufacture can be conclusively determined by that review.

Figure 1 to paragraph (g) of this AD - location of date (MM-YY)

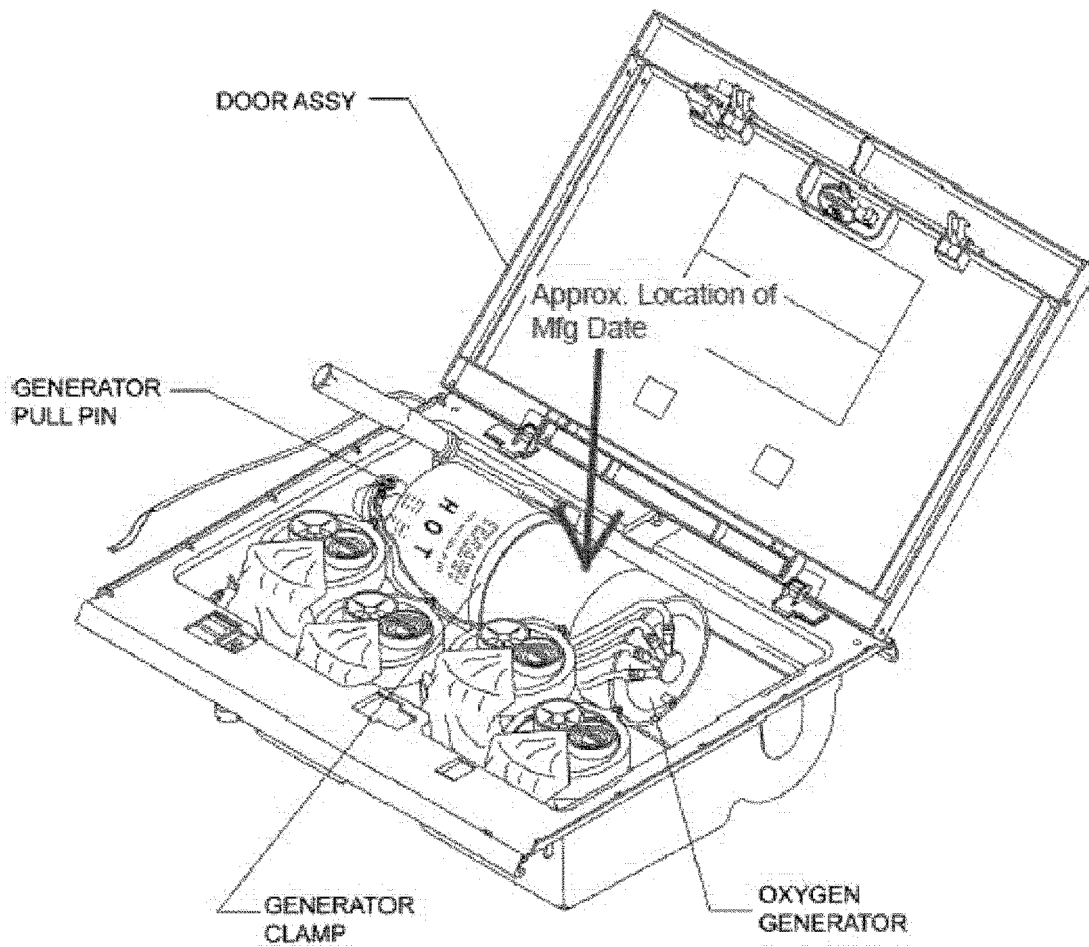
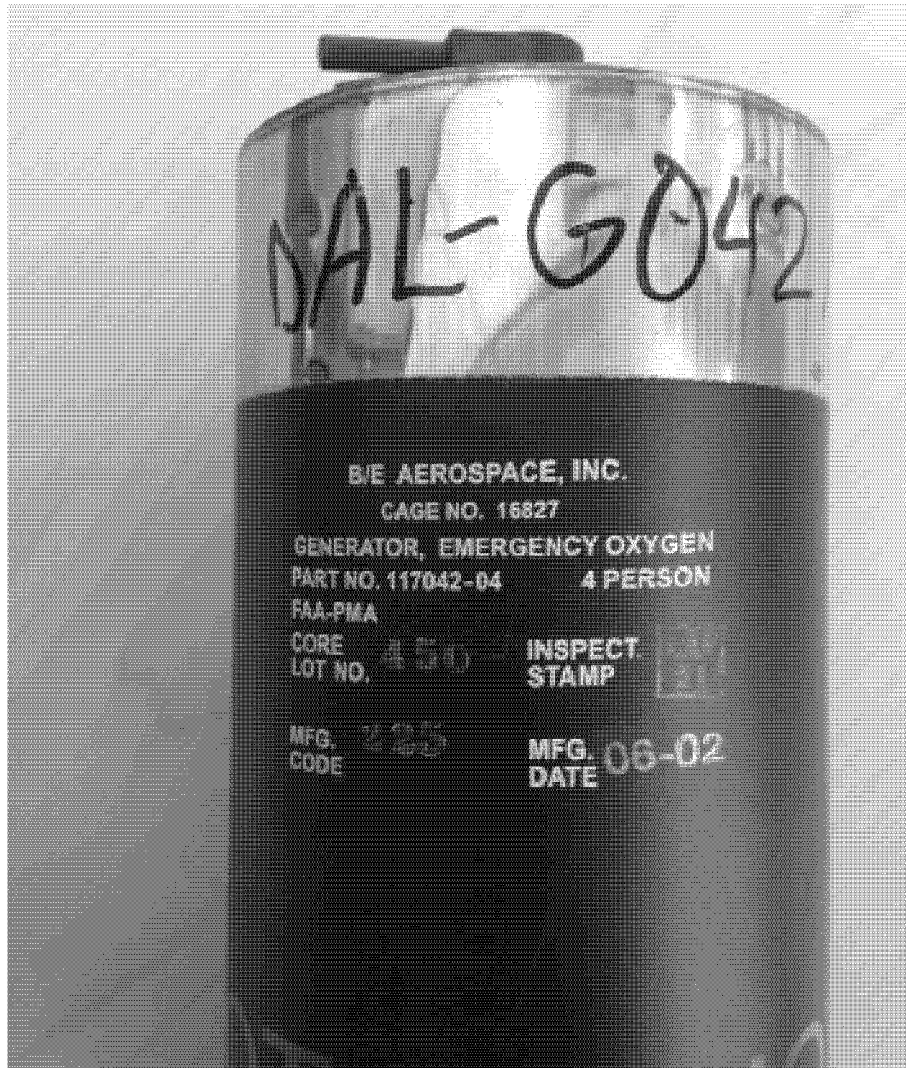


Figure 2 to paragraph (g) of this AD - MFG.DATE (06-02 = June 2002) example



(h) Replacement of Passenger Chemical Oxygen Generators Manufactured in 1999, 2000, or 2001

If, during any inspection required by paragraph (g) of this AD, any passenger chemical oxygen generator having a date of manufacture in 1999, 2000, or 2001 is found: At the applicable time specified in paragraph (h)(1), (h)(2), or (h)(3) of this AD, remove and replace the affected passenger chemical oxygen generator with a serviceable unit, in accordance with the Accomplishment Instructions of B/E Aerospace Service Bulletin 117042-35-001, dated December 10, 2014 (for 15 minute passenger chemical oxygen generators); and Airbus AOT A35N006-14, dated December 10, 2014, including Appendix 01 (for 22 minute passenger chemical oxygen generators).

(1) For passenger chemical oxygen generators that have a date of manufacture in 1999: Within 30 days after the effective date of this AD.

(2) For passenger chemical oxygen generators that have a date of manufacture in 2000: Within 6 months after the effective date of this AD.

(3) For passenger chemical oxygen generators that have a date of manufacture in 2001: Within 12 months after the effective date of this AD.

(i) Replacement of Passenger Chemical Oxygen Generators Manufactured in 2002 and Later

If, during any inspection required by paragraph (g) of this AD, any passenger chemical oxygen generator having a date specified in table 1 to paragraph (i) of this AD is found: At the applicable time specified in table 1 to paragraph (i) of this AD, remove and replace the affected passenger chemical oxygen generator with a serviceable unit, in accordance with the Accomplishment Instructions of B/E Aerospace Service Bulletin 117042-35-001, dated December 10, 2014 (for 15 minute passenger chemical oxygen generators) and Airbus AOT A35N006-14, dated December 10, 2014, including Appendix 01 (for 22 minute passenger chemical oxygen generators).

TABLE 1 TO PARAGRAPH (i) OF THIS AD—REPLACEMENT COMPLIANCE TIMES

Year of manufacture	Compliance time
2002	Within 12 months after the effective date of this AD.
2003	Within 16 months after the effective date of this AD.
2004	Within 20 months after the effective date of this AD.
2005	Within 24 months after the effective date of this AD.
2006	Within 28 months after the effective date of this AD.
2007	Within 32 months after the effective date of this AD.
2008	Within 36 months after the effective date of this AD.

TABLE 1 TO PARAGRAPH (i) OF THIS AD—REPLACEMENT COMPLIANCE TIMES—Continued

Year of manufacture	Compliance time
2009	Before exceeding 10 years since date of manufacture of the passenger chemical oxygen generator.

(j) Definition of Serviceable

For the purpose of this AD, a serviceable unit is a passenger chemical oxygen generator having P/N 117042-XX with a manufacturing date not older than 10 years, or any other approved part number, provided that the generator has not exceeded the life limit established for that generator by the manufacturer.

(k) Reporting

At the applicable time specified in paragraph (k)(1) or (k)(2) of this AD, submit a report of the findings (both positive and negative) of the inspection required by paragraph (g) of this AD, in accordance with paragraph 7., "Reporting," of Airbus AOT A35N006-14, dated December 10, 2014, including Appendix 01. The report must include the information specified in Appendix 1 of Airbus AOT A35N006-14, dated December 10, 2014.

(1) If the inspection was done on or after the effective date of this AD: Submit the report within 30 days after the inspection.

(2) If the inspection was done before the effective date of this AD: Submit the report within 30 days after the effective date of this AD.

(l) Parts Installation Limitation

As of the effective date of this AD, no person may install a passenger chemical oxygen generator, unless it is determined, prior to installation, that the oxygen generator is a serviceable unit as specified in paragraph (j) of this AD.

(m) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Sanjay Ralhan, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone: 425-227-1405; fax: 425-227-1149. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of

the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Contacting the Manufacturer*: For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA; or the European Aviation Safety Agency (EASA); or Airbus's EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(3) *Reporting Requirements*: A federal agency may not conduct or sponsor, and a person is not required to respond to, nor shall a person be subject to a penalty for failure to comply with, a collection of information subject to the requirements of the Paperwork Reduction Act unless that collection of information displays a current valid OMB Control Number. The OMB Control Number for this information collection is 2120-0056. Public reporting for this collection of information is estimated to be approximately 5 minutes per response, including the time for reviewing instructions, completing and reviewing the collection of information. All responses to this collection of information are mandatory. Comments concerning the accuracy of this burden and suggestions for reducing the burden should be directed to the FAA at: 800 Independence Ave. SW., Washington, DC 20591, Attn: Information Collection Clearance Officer, AES-200.

(n) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA Airworthiness Directive EASA AD 2015-0117, dated June 24, 2015; corrected August 7, 2015, for related information. This MCAI may be found in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-3989.

(2) For Airbus service information identified in this proposed AD, contact Airbus, Airworthiness Office—EIAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email account.airworth-eas@airbus.com; Internet <http://www.airbus.com>. For BE service identified in this proposed AD, contact B/E Aerospace Inc., 10800 Pflumm Road, Lenexa, KS 66215; telephone: 913-338-9800; fax: 913-469-8419; Internet <http://beaerospace.com/home/globalsupport>. You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Issued in Renton, Washington, on October 6, 2015.

Jeffrey E. Duven,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2015-26220 Filed 10-16-15; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2015-3988; Directorate Identifier 2015-NM-005-AD]

RIN 2120-AA64

Airworthiness Directives; Bombardier, Inc. Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to supersede Airworthiness Directive (AD) 2014-17-51, for certain Bombardier, Inc. Model CL-600-2B16 airplanes. AD 2014-17-51 currently requires inspecting the inboard flap fasteners of the hinge-box forward fitting at Wing Station (WS) 76.50 and WS 127.25 to determine the orientation and condition of the fasteners, as applicable, and replacement or repetitive inspections of the fasteners if necessary. AD 2014-17-51 also provides for optional terminating action for the requirements of that AD. Since we issued AD 2014-17-51, we have determined that additional action is necessary. This proposed AD would also require accomplishment of the previously optional terminating action. We are proposing this AD to detect and correct incorrectly oriented or fractured fasteners, which could result in premature failure of the fasteners attaching the inboard flap hinge-box forward fitting; failure of the fasteners could lead to the detachment of the flap hinge box and the flap surface, and consequent loss of control of the airplane.

DATES: We must receive comments on this proposed AD by December 3, 2015.

ADDRESSES: You may send comments by any of the following methods:

- *Federal eRulemaking Portal*: Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax*: 202-493-2251.

- *Mail*: U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

- *Hand Delivery*: U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.