

Proposed Rules

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This section of the FEDERAL REGISTER contains notices to the public of the proposed issuance of rules and regulations. The purpose of these notices is to give interested persons an opportunity to participate in the rule making prior to the adoption of the final rules.

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2015-0909; Directorate Identifier 96-ANE-24-AD]

RIN 2120-AA64

Airworthiness Directives; AlliedSignal Inc. and Rajay Inc. Oil Scavenge Pumps

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Proposed rule; withdrawal.

SUMMARY: The FAA is withdrawing a notice of proposed rulemaking (NPRM). The NPRM proposed a new airworthiness directive (AD) that had applied to AlliedSignal oil scavenge pumps, part numbers (P/Ns) 101633-01 and -02 and Rajay Inc. oil scavenge pumps, P/Ns 1025-1 and -2 installed on Continental Motors, Inc. (CMI) IO-470 and TSIO-520 reciprocating engines and on Lycoming Engines, Inc., (Lycoming) IO-360, IO-540, and O-360 reciprocating engines. We have found no service difficulties with these model oil scavenge pumps when installed on the affected engines. Accordingly, we withdraw the proposed rule.

DATES: As of June 19, 2015, the proposed rule published February 20, 1997 at 62 FR 7730 is withdrawn.

FOR FURTHER INFORMATION CONTACT: Richard McCauley, Aerospace Engineer, Seattle Aircraft Certification Office, FAA, Transport Directorate, 1601 Lind Avenue SW., Renton, WA 98055-4056; phone: 425-917-6502; fax: 425-917-6590; email: richard.mccauley@faa.gov.

SUPPLEMENTARY INFORMATION: The FAA proposed to amend 14 CFR part 39 with a proposed AD (62 FR 7730, February 20, 1997). The proposed AD had applied to AlliedSignal oil scavenge pumps, P/Ns 101633-01 and -02 and Rajay Inc. oil scavenge pumps, P/Ns 1025-1 and -2 installed on CMI IO-470 and TSIO-520 reciprocating engines and on

Lycoming IO-360, IO-540, and O-360 reciprocating engines. The NPRM proposed to require initial and repetitive inspections of the oil scavenge pump for the security of the snap ring installation, snap ring and washer wear, and shaft groove wear, and their replacement, if necessary, with serviceable parts. The proposed action was prompted by reports of severe wear on the end plate of the oil scavenge pump. The proposed actions were intended to prevent oil scavenge pump snap ring failure causing severe wear on the pump end plate, which could result in loss of engine oil and subsequent engine shutdown.

Since we issued the NPRM (62 FR 7730, February 20, 1997), additional information became available after the public comment period closed on April 21, 1997.

Upon further consideration, we hereby withdraw the proposed rule for the following reason:

- We reviewed service difficulty reports, AD databases, and airplane manufacturer's data and found no unsafe condition in the last 19 years of service history associated with these oil scavenge pumps installed on the affected engines.

Withdrawal of the NPRM (62 FR 7730, February 20, 1997) constitutes only such action, and does not preclude the agency from issuing another notice in the future, nor does it commit the agency to any course of action in the future.

Since this action only withdraws a notice of proposed rulemaking, it is neither a proposed nor a final rule. Therefore, Executive Order 12866, the Regulatory Flexibility Act, or DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979) do not cover this withdrawal.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Withdrawal

Accordingly, the notice of proposed rulemaking, Docket No. FAA-2015-0909; Directorate Identifier 96-ANE-24-AD, published in the **Federal Register** on February 20, 1997 (62 FR 7730), is withdrawn.

Issued in Burlington, Massachusetts, on June 10, 2015.

Ann C. Mollica,

Acting Directorate Manager, Engine & Propeller Directorate, Aircraft Certification Service.

[FR Doc. 2015-14993 Filed 6-18-15; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2015-0486; Directorate Identifier 2015-NE-07-AD]

RIN 2120-AA64

Airworthiness Directives; Pratt & Whitney Canada Corp. Turboshift Engines

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for certain Pratt & Whitney Canada Corp. (P&WC) PT6B-37A turboshaft engines. This proposed AD was prompted by reports of incorrect engine torque for PT6B-37A engines. This proposed AD would require initial and repetitive inspections until replacement of the No. 10 bearing, and eventual replacement of the No. 9 bearing, both located in the engine reduction gearbox (RGB) assembly. We are proposing this AD to prevent axial migration of the No. 10 bearing in the engine RGB assembly, which could lead to engine overtorque, failure of the engine, in-flight shutdown, and loss of the rotorcraft.

DATES: We must receive comments on this proposed AD by August 18, 2015.

ADDRESSES: You may send comments by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for *submitting comments*.
- *Mail:* Docket Management Facility, U.S. Department of Transportation, 1200 New Jersey Avenue SE., West Building Ground Floor, Room W12-140, Washington, DC 20590-0001.

- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

- Fax: 202-493-2251.

For service information identified in this proposed AD, contact Pratt & Whitney Canada Corp., 1000 Marie-Victorin, Longueuil, Quebec, Canada, J4G 1A1; phone: 800-268-8000; fax: 450-647-2888; Web site: www.pwc.ca. You may view this service information at the FAA, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, MA. For information on the availability of this material at the FAA, call 781-238-7125.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-0486; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the mandatory continuing airworthiness information (MCAI), the regulatory evaluation, any comments received, and other information. The address for the Docket Office (phone: 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:

Barbara Caufield, Aerospace Engineer, Engine Certification Office, FAA, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; phone: 781-238-7146; fax: 781-238-7199; email: barbara.caufield@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2015-0486; Directorate Identifier 2015-NE-07-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed AD.

Discussion

The Transport Canada Civil Aviation, which is the aviation authority for Canada, has issued Canada AD CF-2015-01, dated January 20, 2015

(referred to hereinafter as "the MCAI"), to correct an unsafe condition for the specified products. The MCAI states:

Five incidences of incorrect engine torque indication have been reported for PT6B-37A engine installations on AW119MKII helicopters. A lower than actual engine torque indication due to a faulty indication system, particularly on a helicopter being operated at max allowable torque (90 to 110%) range, may result in undetected over-torque condition.

Repeated over-torque conditions that are undetected and consequently are not corrected in accordance with conditional inspection requirements of original equipment manufacturer (OEM) Instructions for Continued Airworthiness (ICAs), may have a negative impact on the operational safety of the aircraft. Investigation by P&WC has determined the root cause of the subject torque indication anomaly to be the axial migration of part number (P/N) 3310433-03 bearings at the engine torque sensing gear location.

The axial migration of the No. 10 bearing is caused by non-optimal bearing internal clearance. This migration may cause an erroneous torque reading, possibly leading to engine overtorque and engine failure. We are also requiring replacement of the No. 9 bearing since it may also migrate, has the same part number as a No. 10 bearing, and could be installed in the same location as a No. 10 bearing.

You may obtain further information by examining the MCAI in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-0486.

Related Service Information Under 1 CFR Part 51

P&WC has issued Service Bulletin (SB) No. PT6B-72-39095, Revision No. 3, dated December 29, 2014. The service information describes procedures for inspecting affected bearings. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section of this NPRM.

Other Related Service Information

P&WC has also issued SB No. PT6B-72-39092, Revision No. 4, dated December 29, 2014. The service information describes procedures for removing affected bearings.

FAA's Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of Canada, and is approved for operation in the United States. Pursuant to our bilateral

agreement with Canada, they have notified us of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all information provided by Transport Canada Civil Aviation and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design. This proposed AD would require initial and repetitive inspections until replacement of the No. 10 bearing, as well as eventual replacement of the No. 9 bearing, in the engine RGB assembly.

Costs of Compliance

We estimate that this proposed AD affects 83 engines installed on rotorcraft of U.S. registry. We estimate that it would take about 3 hours per engine to perform the initial and repetitive inspections to comply with this proposed AD. We also estimate that it would take about 1 hour per engine to replace the affected bearings. The average labor rate is \$85 per hour. Required parts cost about \$49,800 per engine. Based on these figures, we estimate the cost of this proposed AD on U.S. operators to be \$4,161,620.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

(1) Is not a “significant regulatory action” under Executive Order 12866,

(2) Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),

(3) Will not affect intrastate aviation in Alaska to the extent that it justifies making a regulatory distinction, and

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

Pratt & Whitney Canada Corp.: Docket No. FAA-2015-0486; Directorate Identifier 2015-NE-07-AD.

(a) Comments Due Date

We must receive comments by August 18, 2015.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Pratt & Whitney Canada Corp. (P&WC) PT6B-37A turboshaft engines with engine serial numbers identified in Table 1 of paragraph 4, Appendix, in P&WC Service Bulletin (SB) No. PT6B-72-39095, Revision No. 3, dated December 29, 2014.

(d) Reason

This AD was prompted by reports of incorrect engine torque for PT6B-37A turboshaft engines. We are issuing this AD to prevent axial migration of the No. 10 bearing in the engine reduction gearbox (RGB) assembly, which could lead to engine overtorque, failure of the engine, in-flight shutdown, and loss of the rotorcraft.

(e) Actions and Compliance

Comply with this AD within the compliance times specified, unless already done.

(1) Initial Inspection

(i) Within 50 flight hours (FHs) time in service after the effective date of this AD, inspect the No. 10 bearing, part number (P/N) 3310433-03, in the RGB assembly for axial movement. Use paragraphs 3.A. to 3.C. in the Accomplishment Instructions in P&WC SB No. PT6B-72-39095, Revision No. 3, dated December 29, 2014, to do the inspection. If the bearing fails the inspection, replace the No. 9 and No. 10 bearings before further flight.

(2) Repetitive Inspection

(i) For engines with 500 FHs or less total time since new (TSN), repeat the inspection required by paragraph (e)(1) of this AD every 100 FHs time since last inspection (TSLI) until 500 hours total TSN, and, thereafter, every 200 FHs TSLI until removal.

(ii) For engines with more than 500 FHs total TSN perform the inspection required by paragraph (e)(1) to this AD within 200 FHs TSLI, and, thereafter, every 200 FHs TSLI until removal.

(3) Removal and Replacement of Affected Bearings

(i) For engine serial numbers (S/Ns) PCE-PU0192, PU0193, PU0201, PU0208, PU0209, PU0212, PU0213, PU0214, PU0216, PU0219, and PU0220, remove the No. 9 and No. 10 bearings, P/N 3310433-03, within 450 FHs or 42 months after the effective date of this AD, whichever occurs first, and replace with parts eligible for installation.

(ii) For all engine S/Ns identified in Applicability paragraph (c) of this AD, other than those listed in paragraph (e)(3)(i) of this AD, remove the No. 9 and No. 10 bearings, P/N 3310433-03, and replace with parts eligible for installation within 42 months after the effective date of this AD.

(iii) Replacement of the No. 9 and No. 10 bearing, P/N 3310433-03, with the No. 9 and No. 10 bearing, P/N 3310233-03 or P/N 3310533-03, is terminating action for this AD.

(f) Reporting Requirements

You do not have to contact your Local Field Service Representative as discussed in paragraph 3.C.(3) of P&WC SB No. PT6B-72-39095, Revision No. 3, dated December 29, 2014.

(g) Credit for Previous Action

If you previously replaced the No. 9 and No. 10 bearings in accordance with the instructions contained in P&WC SB No. PT6B-72-39092, Revision No. 2, dated August 8, 2014, or earlier revisions, then you have complied with this AD.

(h) Alternative Methods of Compliance (AMOCs)

The Manager, Engine Certification Office, FAA, may approve AMOCs for this AD. Use the procedures found in 14 CFR 39.19 to make your request. You may email your request to: ANE-AD-AMOC@faa.gov.

(i) Related Information

(1) For more information about this AD, contact Barbara Caufield, Aerospace Engineer, Engine Certification Office, FAA, Engine & Propeller Directorate, 12 New

England Executive Park, Burlington, MA 01803; phone: 781-238-7146; fax: 781-238-7199; email: barbara.caufield@faa.gov.

(2) Refer to MCAI Transport Canada AD CF-2015-01, dated January 20, 2015, for more information. You may examine the MCAI in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating it in Docket No. FAA-2015-0486.

(3) P&WC SB No. PT6B-72-39092, Revision No. 4, dated December 29, 2014, and SB No. PT6B-72-39095, Revision No. 3, dated December 29, 2014, can be obtained from P&WC using the contact information in paragraph (i)(4) of this proposed AD.

(4) For service information identified in this proposed AD, contact Pratt & Whitney Canada Corp., 1000 Marie-Victorin, Longueuil, Quebec, Canada, J4G 1A1; phone: 800-268-8000; fax: 450-647-2888; Internet: www.pwc.ca.

(5) You may view this service information at the FAA, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, MA. For information on the availability of this material at the FAA, call 781-238-7125.

Issued in Burlington, Massachusetts, on June 10, 2015.

Ann C. Mollica,

Acting Directorate Manager, Engine & Propeller Directorate, Aircraft Certification Service.

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DEPARTMENT OF THE TREASURY

Internal Revenue Service

26 CFR Part 1

[REG-102648-15]

RIN 1545-BM66

Suspension of Benefits Under the Multiemployer Pension Reform Act of 2014

AGENCY: Internal Revenue Service (IRS), Treasury.

ACTION: Notice of proposed rulemaking, notice of proposed rulemaking by cross-reference to temporary regulations, and notice of public hearing.

SUMMARY: This document contains proposed regulations relating to multiemployer pension plans that are projected to have insufficient funds, at some point in the future, to pay the full benefits to which individuals will be entitled under the plans (referred to as plans in “critical and declining status”). The Multiemployer Pension Reform Act of 2014 (“MPRA”) amended the Internal Revenue Code to incorporate suspension of benefits provisions that permit these multiemployer plans to reduce pension benefits payable to participants and beneficiaries if certain