

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by removing Airworthiness Directive (AD) 2007–16–08, Amendment 39–15147 (72 FR 44728, August 9, 2007), and adding the following new AD:

The Boeing Company: Docket No. FAA–2015–0498; Directorate Identifier 2014–NM–152–AD.

(a) Comments Due Date

The FAA must receive comments on this AD action by May 14, 2015.

(b) Affected ADs

This AD replaces AD 2007–16–08, Amendment 39–15147 (72 FR 44728, August 9, 2007).

(c) Applicability

This AD applies to all Boeing Model 747–100, 747–100B, 747–100B SUD, 747–200B, 747–200C, 747–300, 747–400, 747–400D, and 747SR series airplanes, certificated in any category.

(d) Subject

Air Transport Association (ATA) of America Code 53, Fuselage.

(e) Unsafe Condition

This AD was prompted by reports of cracks found on the station 800 frame on the left-side and right-side main entry doors (MED), at the forward and aft inner chord strap and angles, which are outside the inspection area of AD 2007–16–08, Amendment 39–15147 (72 FR 44728, August 9, 2007). We are issuing this AD to detect and correct fatigue cracks that could extend and fully sever the frame, which could result in development of skin cracks that could lead to rapid depressurization of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Inspections of Station 800 Frame Assembly Between Stringer 14 and Stringer 30

Except as required by paragraph (i) of this AD, at the applicable time specified in paragraph 1.E., “Compliance,” of Boeing Alert Service Bulletin 747–53A2451, Revision 2, dated June 13, 2014: Do a detailed inspection for cracking in the inner chord strap, angles, and exposed web adjacent to the inner chords, and do surface and open hole high-frequency eddy current (HFEC) inspections for cracking in the inner chord strap and angles of the station 800 frame assembly between stringer 14 and stringer 30, in accordance with the Accomplishment Instructions of Boeing Alert Service Bulletin 747–53A2451, Revision 2, dated June 13, 2014. Repeat the inspections

at the applicable times specified in paragraph 1.E., “Compliance,” of Boeing Alert Service Bulletin 747–53A2451, Revision 2, dated June 13, 2014.

(h) Repair of Cracking

If any cracking is found during any inspection required by paragraph (g) of this AD, before further flight, repair the cracking using a method approved in accordance with the procedures specified in paragraph (k) of this AD.

(i) Exception to the Service Information

(1) Where Boeing Alert Service Bulletin 747–53A2451, Revision 2, dated June 13, 2014, specifies a compliance time “after the Revision 2 date of this service bulletin,” this AD requires compliance within the specified time after the effective date of this AD.

(2) The Condition column of paragraph 1.E., “Compliance,” of the Boeing Alert Service Bulletin 747–53A2451, Revision 2, dated June 13, 2014, refers to total flight cycles “as of the Revision 2 date of this service bulletin.” This AD, however applies to airplanes with the specified total flight cycles or total flight hours as of the effective date of this AD.

(j) Credit for Previous Actions

This paragraph provides credit for the inspections and repairs of the inner chord strap and angles of the station 800 frame assembly between stringer 14 and stringer 18 required by paragraphs (g) and (h) of this AD, if those actions were performed before the effective date of this AD using Boeing Alert Service Bulletin 747–53A2451, Revision 1, dated November 10, 2005.

(k) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Seattle Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the ACO, send it to the attention of the person identified in paragraph (k)(1) of this AD. Information may be emailed to: 9-ANM-Seattle-ACO-AMOC-Requests@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(3) An AMOC that provides an acceptable level of safety may be used for any repair required by this AD if it is approved by the Boeing Commercial Airplanes Organization Designation Authorization (ODA) that has been authorized by the Manager, Seattle ACO, to make those findings. For a repair method to be approved, the repair must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

(4) AMOCs approved for AD 2007–16–08, Amendment 39–15147 (72 FR 44728, August 9, 2007), are approved as AMOCs for the corresponding provisions of this AD.

(l) Related Information

(1) For more information about this AD, contact Bill Ashforth, Aerospace Engineer, Airframe Branch, ANM–120S, FAA, Seattle Aircraft Certification Office (ACO), 1601 Lind Avenue SW., Renton, WA 98057–3356; phone: 425–917–6432; fax: 425–917–6590; email: bill.ashforth@faa.gov.

(2) For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H–65, Seattle, WA 98124–2207; telephone 206–544–5000, extension 1; fax 206–766–5680; Internet <https://www.myboeingfleet.com>. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221.

Issued in Renton, Washington, on March 19, 2015.

Michael Kaszycki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2015–07081 Filed 3–27–15; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2015–0676; Directorate Identifier 2014–NM–164–AD]

RIN 2120–AA64

Airworthiness Directives; Bombardier, Inc. Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for certain Bombardier, Inc. Model BD–700–1A10 and BD–700–1A11 airplanes. This proposed AD was prompted by a report of several events where pilots experienced difficulty in lateral control of the airplane after doing a climb through heavy rain conditions and a determination that the cause was water ingress in the aileron control pulley assembly. This proposed AD would require, for certain airplanes, inspecting for correct clearance and rework if necessary, and, for certain other airplanes, installing a cover for the aileron pulley assembly. We are proposing this AD to prevent water ingress in the aileron control pulley assembly, which could freeze in cold conditions and result in reduced control of the airplane.

DATES: We must receive comments on this proposed AD by May 14, 2015.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.
- *Fax:* 202-493-2251.
- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.
- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Bombardier, Inc., 400 Côte-Vertu Road West, Dorval, Québec H4S 1Y9, Canada; telephone 514-855-5000; fax 514-855-7401; email thd.crj@aero.bombardier.com; Internet <http://www.bombardier.com>. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-0676; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Fabio Buttitta, Aerospace Engineer, Airframe and Mechanical Systems Branch, ANE-171, FAA, New York Aircraft Certification Office (ACO), 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516-228-7303; fax 516-794-5531.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA-2015-0676; Directorate Identifier

2014-NM-164-AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

Transport Canada Civil Aviation (TCCA), which is the aviation authority for Canada, has issued Canadian Airworthiness Directive CF-2014-23, dated July 18, 2014 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for certain Bombardier, Inc. Model BD-700-1A10 and BD-700-1A11 airplanes. The MCAI states:

There have been several reports whereby pilots have experienced difficulty in lateral control following climb through heavy rain conditions. In each event, the pilots were able to overcome this difficulty without disconnecting the aileron control. An investigation has determined that the root cause of the restricted movement of the aileron was due to water ingress into the wing root aileron control pulley assembly through a gap on the wing-to-fuselage fairing resulting in freezing of the aileron control system.

If not corrected, this condition could result in reduced lateral control of the aeroplane.

This [Canadian] AD mandates [for certain airplanes] the incorporation of a cover for the aileron pulley assembly [and inspection and rework if necessary] to prevent water ingress in the aileron control pulley assembly [and for certain other airplanes, mandates an inspection and rework if necessary].

The inspection involves doing a general visual inspection for correct clearance. You may examine the MCAI in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-0676.

Related Service Information Under 1 CFR Part 51

Bombardier has issued the following service information:

- Service Bulletin 700-1A11-27-034, Revision 04, dated September 4, 2014;
- Service Bulletin 700-27-076, Revision 04, dated September 4, 2014;
- Service Bulletin 700-27-5004, Revision 04, dated September 4, 2014; and
- Service Bulletin 700-27-6004, Revision 04, dated September 4, 2014.

This service information describes procedures, for certain airplanes, for installing a cover for the No. 1 aileron pulley, including an inspection for correct clearance and rework, and for certain other airplanes, for an inspection for correct clearance and rework. The actions described in this service information are intended to correct the unsafe condition identified in the MCAI. This service information is reasonably available; see **ADDRESSES** for ways to access this service information.

FAA’s Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of the same type design.

Costs of Compliance

We estimate that this proposed AD affects 60 airplanes of U.S. registry.

We also estimate that it would take about 9 work-hours per product to comply with the basic requirements of this proposed AD. The average labor rate is \$85 per work-hour. Based on these figures, we estimate the cost of this proposed AD on U.S. operators to be \$45,900, or \$765 per product.

According to the manufacturer, some of the costs of this proposed AD may be covered under warranty, thereby reducing the cost impact on affected individuals. We do not control warranty coverage for affected individuals. As a result, we have included all costs in our cost estimate.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for

safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

Bombardier, Inc.: Docket No. FAA–2015–0676; Directorate Identifier 2014–NM–164–AD.

(a) Comments Due Date

We must receive comments by May 14, 2015.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Bombardier, Inc. Model BD–700–1A10 and BD–700–1A11 airplanes, certificated in any category, having serial

numbers 9002 through 9520 inclusive and 9998.

(d) Subject

Air Transport Association (ATA) of America Code 27, Flight Controls.

(e) Reason

This AD was prompted by a report of several events where pilots experienced difficulty in lateral control of the airplane after doing a climb through heavy rain conditions and a determination that the cause was water ingress in the aileron control pulley assembly. We are issuing this AD to prevent water ingress in the aileron control pulley assembly, which could freeze in cold conditions and result in reduced control of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Installation of Cover for the Aileron Pulley Assembly

Except as provided by paragraph (j) of this AD, for airplanes on which a cover for the No. 1 aileron pulley was not installed as of the effective date of this AD: Within 150 flight cycles after the effective date of this AD, install a cover for the No. 1 aileron pulley, including doing a general visual inspection for correct clearance and rework as applicable, in accordance with paragraph C., “PART B—Modification,” of the Accomplishment Instructions of the applicable service bulletins identified in paragraphs (g)(1) and (g)(2) for this AD.

(1) For Model BD–700–1A10 airplanes: Bombardier Service Bulletin 700–27–076, Revision 04, dated September 4, 2014; or 700–27–6004, Revision 04, dated September 4, 2014.

(2) For Model BD–700–1A11 airplanes: Bombardier Service Bulletin 700–1A11–27–034, Revision 04, dated September 4, 2014; or 700–27–5004, Revision 04, dated September 4, 2014.

(h) Inspection and Rework

Except as provided by paragraph (j) of this AD, for airplanes that have incorporated a cover for the No. 1 aileron pulley using the applicable service information identified in paragraphs (h)(1) and (h)(2) of this AD as of the effective date of this AD: Within 150 flight cycles after the effective date of this AD, do a general visual inspection for correct clearance and, before further flight, rework, as applicable, in accordance with paragraph B., “PART A—Inspection and Rework,” of the Accomplishment Instructions of the applicable service information identified in paragraphs (g)(1) and (g)(2) of this AD.

(1) For Model BD–700–1A10 airplanes: Bombardier Service Bulletin 700–27–076, dated March 5, 2012; or 700–27–6004, dated March 5, 2012.

(2) For Model BD–700–1A11 airplanes: Bombardier Service Bulletin 700–1A11–27–034, dated March 5, 2012; or 700–27–5004, dated March 5, 2012.

(i) Re-Identification of Overwing Panels

Except as provided by paragraph (j) of this AD, for airplanes on which the Service Non-

Incorporated Engineering Orders (SNIEO) or Service Requests for Product Support Action (SRPSA) that are listed in table 2 of paragraph 1.A., “Effectivity,” in the service information identified in paragraphs (i)(1), (i)(2), or (i)(3) of this AD have been incorporated: Within 150 flight cycles from the effective date of this AD, do the re-identification of the overwing panels, in accordance with paragraph 2.B(2)(g) of the Accomplishment Instructions of the applicable service information identified in paragraphs (g)(1) and (g)(2) of this AD.

(1) Bombardier Service Bulletin 700–27–076, Revision 04, dated September 4, 2014.

(2) Bombardier Service Bulletin 700–27–6004, Revision 04, dated September 4, 2014.

(3) Bombardier Service Bulletin 700–1A11–27–034, Revision 04, dated September 4, 2014.

(j) Exception to the Requirements of Paragraphs (g), (h), and (i) of this AD

Airplanes on which the SRPSA, as listed in table 1 of paragraph 1.A., “Effectivity,” in the service information identified in paragraph (j)(1), (j)(2), or (j)(3) of this AD has been accomplished as of the effective date of this AD, meet the intent of paragraphs (g), (h), and (i) of this AD and no further action is required.

(1) Bombardier Service Bulletin 700–27–076, Revision 04, dated September 4, 2014.

(2) Bombardier Service Bulletin 700–27–6004, Revision 04, dated September 4, 2014.

(3) Bombardier Service Bulletin 700–1A11–27–034, Revision 04, dated September 4, 2014.

(k) Credit for Previous Actions

This paragraph provides credit for actions required by paragraphs (g), (h), and (i) of this AD, if those actions were performed before the effective date of this AD using the applicable service information identified in paragraphs (k)(1) through (k)(8) of this AD, which are not incorporated by reference in this AD.

(1) Bombardier Service Bulletin 700–1A11–27–034, Revision 01, dated July 16, 2012.

(2) Bombardier Service Bulletin 700–1A11–27–034, Revision 02, dated June 17, 2014.

(3) Bombardier Service Bulletin 700–27–076, Revision 01, dated July 16, 2012.

(4) Bombardier Service Bulletin 700–27–076, Revision 02, dated June 17, 2014.

(5) Bombardier Service Bulletin 700–27–5004, Revision 01, dated July 16, 2012.

(6) Bombardier Service Bulletin 700–27–5004, Revision 02, dated June 17, 2014.

(7) Bombardier Service Bulletin 700–27–6004, Revision 01, dated July 16, 2012.

(8) Bombardier Service Bulletin 700–27–6004, Revision 02, dated June 17, 2014.

(l) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, New York Aircraft Certification Office (ACO), ANE–170, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal

inspector or local Flight Standards District Office, as appropriate. If sending information directly to the ACO, send it to ATTN: Program Manager, Continuing Operational Safety, FAA, New York ACO, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516-228-7300; fax 516-794-5531. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Contacting the Manufacturer:* For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, New York ACO, ANE-170, FAA; or Transport Canada Civil Aviation (TCCA); or Bombardier, Inc.'s TCCA Design Approval Organization (DAO). If approved by the DAO, the approval must include the DAO-authorized signature.

(m) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) Canadian Airworthiness Directive CF-2014-23, dated July 18, 2014, for related information. This MCAI may be found in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-0676.

(2) For service information identified in this AD, contact Bombardier, Inc., 400 Côte Vertu Road West, Dorval, Québec H4S 1Y9, Canada; telephone 514-855-5000; fax 514-855-7401; email thd.crj@aero.bombardier.com; Internet <http://www.bombardier.com>. You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Issued in Renton, Washington, on March 19, 2015.

Michael Kaszyscki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2015-07072 Filed 3-27-15; 8:45 am]

BILLING CODE 4910-13-P

ENVIRONMENTAL PROTECTION AGENCY

40 CFR Part 52

[EPA-R07-OAR-2015-0170; FRL-9925-23-Region 7]

Approval and Promulgation of Implementation Plans; State of Missouri, Control of Sulfur Emissions From Stationary Boilers

AGENCY: Environmental Protection Agency (EPA).

ACTION: Proposed rule.

SUMMARY: The Environmental Protection Agency (EPA) is proposing to approve

an amendment to the State Implementation Plan (SIP) submitted by the State of Missouri on October 17, 2013, related to the Missouri rule "Control of Sulfur Emissions from Stationary Boilers." The SIP revision is administrative and provides clarity on the applicability of emission limits and removes definitions originally included in this rule which have been moved to the "Definitions and Common Reference Tables" rule.

DATES: Comments on this proposed action must be received in writing by April 29, 2015.

ADDRESSES: Submit your comments, identified by Docket ID No. EPA-R07-OAR-2015-0170, by mail to Larry Gonzalez, Environmental Protection Agency, Air Planning and Development Branch, 11201 Renner Boulevard, Lenexa, Kansas 66219. Comments may also be submitted electronically or through hand delivery/courier by following the detailed instructions in the **ADDRESSES** section of the direct final rule located in the rules section of this **Federal Register**.

FOR FURTHER INFORMATION CONTACT:

Larry Gonzalez, Environmental Protection Agency, Air Planning and Development Branch, 11201 Renner Boulevard, Lenexa, Kansas 66219 at 913-551-7041, or by email at gonzalez.larry@epa.gov.

SUPPLEMENTARY INFORMATION: EPA is proposing to approve an amendment to the SIP submitted by the State of Missouri on October 17, 2013, related to Missouri rule 10 CSR 10-5.570 "Control of Sulfur Emissions from Stationary Boilers." The SIP revision is administrative and provides clarity on the applicability of emission limits specified at 10 CSR 10-5.570(3)(A)2. Additionally, the amendment removes definitions originally included in 10 CSR 10-5.570 which have been moved to 10 CSR 10.6.020 "Definitions and Common Reference Tables".

In the final rules section of the **Federal Register**, EPA is approving the state's SIP revision as a direct final rule without prior proposal because the Agency views this as a noncontroversial revision amendment and anticipates no relevant adverse comments to this action. A detailed rationale for the approval is set forth in the direct final rule. If no relevant adverse comments are received in response to this action, no further activity is contemplated in relation to this action. If EPA receives relevant adverse comments, the direct final rule will be withdrawn and all public comments received will be addressed in a subsequent final rule based on this proposed action. EPA will

not institute a second comment period on this action. Any parties interested in commenting on this action should do so at this time. Please note that if EPA receives adverse comment on part of this rule and if that part can be severed from the remainder of the rule, EPA may adopt as final those parts of the rule that are not the subject of an adverse comment. For additional information, see the direct final rule which is located in the rules section of this **Federal Register**.

List of Subjects in 40 CFR Part 52

Environmental protection, Air pollution control, Carbon monoxide, Incorporation by reference, Intergovernmental relations, Lead, Nitrogen dioxide, Ozone, Particulate matter, Reporting and recordkeeping requirements, Sulfur oxides, Volatile organic compounds.

Dated: March 17, 2015.

Mark Hague,

Acting Regional Administrator, Region 7.

[FR Doc. 2015-07125 Filed 3-27-15; 8:45 am]

BILLING CODE 6560-50-P

ENVIRONMENTAL PROTECTION AGENCY

40 CFR Part 52

[EPA-R06-OAR-2015-0033; FRL-9925-20-Region 6]

Approval and Promulgation of Implementation Plans; Texas; Public Participation for Air Quality Permit Applications

AGENCY: Environmental Protection Agency (EPA).

ACTION: Proposed rule.

SUMMARY: The Environmental Protection Agency (EPA) is proposing to approve two provisions submitted by the State of Texas as revisions to the Texas State Implementation Plan (SIP) on July 2, 2010, specific to the applicability of the public notice requirements to applications for Plant-Wide Applicability (PAL) permits and standard permits for concrete batch plants without enhanced controls. Today's proposal and the accompanying direct final action will complete the rulemaking process started in our December 13, 2012, proposal and approve the public notice provisions into the Texas SIP. The EPA is proposing to convert the public notice applicability provisions for Texas Flexible Permits from a final conditional approval to a full approval. The EPA is proposing approval of these revisions