

circumstances that would warrant the use of different or additional scenarios or a planning horizon of more than nine quarters. Thus, under the Stress Test Rule the OCC reserves the authority to require a covered institution to use different or additional scenarios and/or planning horizons the agency may deem appropriate. For example, a covered institution may conduct business activities or have risk exposures that would encounter stress under conditions that differ materially from those that would generate stress for other institutions. The OCC expects such situations to be rare and anticipates making every effort to distribute the same scenarios to all covered institutions.

In addition to the minimum three scenarios, the OCC may require a covered institution with significant trading activities to include factors related to trading and counterparty risk in its stress test. Typically, these factors might include additional shocks to specific market prices, interest rates, rate spreads, or other key market variables consistent with historical or hypothetical adverse market events.

#### IV. Development and Distribution

As one part of the process of developing scenarios, the OCC will gather information from outside entities and develop themes for the stress test scenarios, including the identification of potentially material vulnerabilities or salient risks to the financial system, and consider potential paths for individual variables. The outside entities may include academic experts, staffs of international organizations, foreign supervisors, financial institutions that regularly provide forecasts, and other private sector risk analysts that regularly conduct stress tests based on U.S. and global economic and financial scenarios. The OCC will use the information gathered in this manner to inform its consideration of potential risks and scenarios.

The OCC, the Board of Governors of the Federal Reserve System (Board), and the Federal Deposit Insurance Corporation (FDIC) (Agencies) expect to consult closely to develop scenarios for stress testing. Absent specific supervisory concerns, the OCC anticipates that the annual stress test scenarios distributed by the OCC will be the same as or nearly identical to the scenarios developed by the Board for the supervisory stress tests conducted by the Board under Section 165(i)(1). This would mean the same economic and financial variables following the same paths as used in the scenarios for the Board's supervisory stress tests.

Although the Agencies generally expect to consult closely on scenario development, they may have different views of risks that should be reflected in the stress test scenarios used by covered institutions for the annual stress test. The OCC may distribute scenarios to covered institutions that differ in certain respects from those distributed by the FDIC and the Board if necessary to better reflect specific OCC concerns. The OCC expects such situations to be extremely rare, however, and anticipates making every effort to avoid differences in the scenarios required by each agency.

The OCC anticipates that the stress test scenarios will be revised annually as appropriate to ensure that each scenario remains relevant under prevailing economic and industry conditions. These yearly revisions will enable the scenarios to capture evolving risks and vulnerabilities. The need to ensure that scenarios do not become outdated because of economic and financial developments makes a lengthy process of review and comment concerning scenarios prior to distribution each year impractical. However, the process of consultation with the Board and the FDIC, as well as the ongoing interaction of OCC staff with public and private sector experts to obtain views on salient risks and to obtain suggestions for the behavior of key economic variables, should ensure that the stress conditions reflected in the scenarios are well suited to their purpose.

The scenario development process culminates with the distribution of the scenarios to all covered institutions no later than November 15 of each year. The scenario descriptions provided to covered institutions will include values for economic and financial variables depicting the paths those variables follow under the scenarios. The OCC believes that distribution of the scenarios by November 15 aligns with similar processes at the FDIC and the Board.

Dated: October 21, 2013.

**Thomas J. Curry,**

*Comptroller of the Currency.*

[FR Doc. 2013-25421 Filed 10-25-13; 8:45 am]

**BILLING CODE 4810-33-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2013-0832; Directorate Identifier 2012-NM-047-AD; Amendment 39-17612; AD 2013-20-06]

RIN 2120-AA64

#### Airworthiness Directives; Airbus Airplanes

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule; request for comments.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for all Airbus Model A340-211 -212, -213, -311, -312, -313, -541, and -642 airplanes. This AD requires revising the maintenance program to incorporate certain maintenance requirements and airworthiness limitations. This AD was prompted by a determination that existing maintenance requirements are not adequate to address the unsafe condition. We are issuing this AD to address the aging effects of aircraft systems. Such aging effects could change the characteristics of systems life-limited components leading to an increased potential for failure, which, in isolation or in combination with one or more other specific failures or events, could result in failure of certain life limited parts, which could reduce the structural integrity or the controllability of the airplane.

**DATES:** This AD becomes effective November 12, 2013.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of November 12, 2013.

We must receive comments on this AD by December 12, 2013.

**ADDRESSES:** You may send comments by any of the following methods:

- Federal eRulemaking Portal: Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- Fax: (202) 493-2251.

- Mail: U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

- Hand Delivery: U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this AD, contact Airbus SAS, Airworthiness Office—EAL, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 45 80; email [airworthiness.A330-A340@airbus.com](mailto:airworthiness.A330-A340@airbus.com); Internet <http://www.airbus.com>. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the MCAI, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

### FOR FURTHER INFORMATION CONTACT:

Vladimir Ulyanov, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone (425) 227-1138; fax (425) 227-1149.

### SUPPLEMENTARY INFORMATION:

#### Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, has issued EASA Airworthiness Directive 2012-0021, dated January 30, 2012 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for the specified products. The MCAI states:

The mandatory instructions and airworthiness limitations applicable to the Ageing Systems Maintenance (ASM) are specified in Airbus A340 [Airworthiness Limitations Section] ALS Part 4, which is approved by the European Aviation Safety Agency (EASA).

The revision 02 of Airbus A340 ALS Part 4 introduces more restrictive maintenance requirements and/or airworthiness limitations. Failure to comply with the instructions of ALS Part 4 could result in an unsafe condition.

This [EASA] AD requires the implementation of the maintenance requirements and/or airworthiness limitations as specified in Airbus A340 ALS

Part 4 revision 02, approved on 12 October 2011. \* \* \*

The unsafe condition is the aging effects of aircraft systems. Such aging effects could change the characteristics of systems life-limited components leading to an increased potential for failure, which, in isolation or in combination with one or more other specific failures or events, could result in failure of certain life limited parts, which could reduce the structural integrity or the controllability of the airplane. You may obtain further information by examining the MCAI in the AD docket.

### Relevant Service Information

Airbus has issued A340 Airworthiness Limitations Section (ALS), Part 4, Aging Systems Maintenance, Revision 02, dated October 12, 2011. The actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

### Related Rulemaking

Certain maintenance requirements specified in A340 ALS, Part 4—Aging Systems Maintenance, Revision 02, dated October 12, 2011, are already required by other ADs. Therefore, accomplishing the actions required by this AD will terminate the requirements of the following ADs for Model A340 airplanes.

- AD 2003-14-11, Amendment 39-13230 (68 FR 41521, July 14, 2003).
- AD 2004-11-08, Amendment 39-13654 (69 FR 31874, June 8, 2004).
- AD 2004-13-25, Amendment 39-13707 (69 FR 41394, July 9, 2004).
- AD 2004-18-14, Amendment 39-13793 (69 FR 55326, September 14, 2004).
- AD 2007-05-12, Amendment 39-14973 (72 FR 10057, March 7, 2007).
- AD 2008-06-07, Amendment 39-15419 (73 FR 13103, March 12, 2008; corrected April 15, 2008 (73 FR 20367)).
- AD 2012-04-07, Amendment 39-16963 (77 FR 12989, March 5, 2012).

### FAA’s Determination and Requirements of This AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are issuing this AD because we evaluated all pertinent information and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

This AD requires revisions to certain operator maintenance documents to include new inspections. Compliance with these inspections is required by 14 CFR 91.403(c). For airplanes that have been previously modified, altered, or repaired in the areas addressed by these inspections, the operator may not be able to accomplish the inspections described in the revisions. In this situation, to comply with 14 CFR 91.403(c), the operator must request approval for an alternative method of compliance according to paragraph (j)(1) of this AD. The request should include a description of changes to the required inspections that will ensure the continued operational safety of the airplane.

There are no products of this type currently registered in the United States. However, this rule is necessary to ensure that the described unsafe condition is addressed if any of these products are placed on the U.S. Register in the future.

### FAA’s Determination of the Effective Date

Since there are currently no domestic operators of this product, notice and opportunity for public comment before issuing this AD are unnecessary.

### Comments Invited

This AD is a final rule that involves requirements affecting flight safety, and we did not precede it by notice and opportunity for public comment. We invite you to send any written relevant data, views, or arguments about this AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA-2013-0832; Directorate Identifier 2012-NM-047-AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this AD. We will consider all comments received by the closing date and may amend this AD because of those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this AD.

### Costs of Compliance

We estimate that this AD affects 0 airplanes of U.S. registry.

We estimate the following costs to comply with this AD:

ESTIMATED COSTS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Maintenance program revision.	2 work-hours × \$85 per hour = \$170 per revision ....	\$0	\$170 per revision .....	\$0

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**Adoption of the Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

- 2. The FAA amends § 39.13 by adding the following new AD:

**2013–20–06 Airbus:** Amendment 39–17612. Docket No. FAA–2013–0832; Directorate Identifier 2012–NM–047–AD.

**(a) Effective Date**

This airworthiness directive (AD) becomes effective November 12, 2013.

**(b) Affected ADs**

This AD affects the ADs specified in paragraphs (b)(1) through (b)(7) of this AD:  
 (1) AD 2003–14–11, Amendment 39–13230 (68 FR 41521, July 14, 2003);  
 (2) AD 2004–11–08, Amendment 39–13654 (69 FR 31874, June 8, 2004);  
 (3) AD 2004–13–25, Amendment 39–13707 (69 FR 41394, July 9, 2004);  
 (4) AD 2004–18–14, Amendment 39–13793 (69 FR 55326, September 14, 2004);  
 (5) AD 2007–05–12, Amendment 39–14973 (72 FR 10057, March 7, 2007);  
 (6) AD 2008–06–07, Amendment 39–15419 (73 FR 13103, March 12, 2008; corrected April 15, 2008 (73 FR 20367)); and  
 (7) AD 2012–04–07, Amendment 39–16963 (77 FR 12989, March 5, 2012).

**(c) Applicability**

This AD applies to Airbus Model A340–211, –212, –213, –311, –312, –313, –541, and –642 airplanes; certificated in any category; all manufacturer serial numbers.

**(d) Subject**

Air Transport Association (ATA) of America Code 05, Time Limits/Maintenance Checks.

**(e) Reason**

This AD was prompted by a determination that existing maintenance requirements are not adequate to address the unsafe condition. We are issuing this AD to address the aging effects of aircraft systems. Such aging effects could change the characteristics of systems life-limited components leading to an increased potential for failure, which, in isolation or in combination with one or more other specific failures or events, could result

in failure of certain life limited parts, which could reduce the structural integrity or the controllability of the airplane.

**(f) Compliance**

You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

**(g) Maintenance Program Revision**

Within 6 months after the effective date of this AD, revise the maintenance program by incorporating A340 Airworthiness Limitations Section (ALS), Part 4—Aging Systems Maintenance, Revision 02, dated October 12, 2011. Comply with all applicable instructions and airworthiness limitations included in A340 ALS, Part 4—Aging Systems Maintenance, Revision 02, dated October 12, 2011. The initial compliance times for the actions are within the applicable compliance times specified in the Record of Revisions pages of A340 ALS, Part 4—Aging Systems Maintenance, Revision 02, dated October 12, 2011, or within 6 months after the effective date of this AD, whichever is later.

**(h) Alternative Actions or Intervals**

After accomplishing the revision required by paragraph (g) of this AD, no alternative actions (e.g., inspections) or intervals may be used unless the actions or intervals are approved as an alternative method of compliance (AMOC) in accordance with the procedures specified in paragraph (j)(1) of this AD.

**(i) Terminating Action for Other ADs**

Accomplishing the revision of the maintenance program and complying with all applicable instructions and airworthiness limitations required by paragraph (g) of this AD terminates the requirements of the ADs specified in paragraphs (i)(1) through (i)(7) of this AD for Model A340 airplanes only.

- (1) AD 2003–14–11, Amendment 39–13230 (68 FR 41521, July 14, 2003).
- (2) AD 2004–11–08, Amendment 39–13654 (69 FR 31874, June 8, 2004).
- (3) AD 2004–13–25, Amendment 39–13707 (69 FR 41394, July 9, 2004).
- (4) AD 2004–18–14, Amendment 39–13793 (69 FR 55326, September 14, 2004).
- (5) AD 2007–05–12, Amendment 39–14973 (72 FR 10057, March 7, 2007).
- (6) AD 2008–06–07, Amendment 39–15419 (73 FR 13103, March 12, 2008; corrected April 15, 2008 (73 FR 20367)).
- (7) AD 2012–04–07, Amendment 39–16963 (77 FR 12989, March 5, 2012).

**(j) Other FAA AD Provisions**

The following provisions also apply to this AD:

- (1) *Alternative Methods of Compliance (AMOCs):* The Manager, International

Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Vladimir Ulyanov, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone (425) 227-1138; fax (425) 227-1149. Information may be emailed to: [9-ANM-116-AMOC-REQUESTS@faa.gov](mailto:9-ANM-116-AMOC-REQUESTS@faa.gov). Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) **Airworthy Product:** For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

#### (k) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) European Aviation Safety Agency (EASA) Airworthiness Directive 2012-0021, dated January 30, 2012, for related information. You may examine the MCAI on the Internet at <http://www.regulations.gov> by searching for and locating it in Docket No. FAA-2013-0832.

(2) Service information identified in this AD that is not incorporated by reference may be obtained at the addresses specified in paragraphs (l)(3) and (l)(4) of this AD.

#### (l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the following service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use the following service information to do the actions required by this AD, unless the AD specifies otherwise.

(i) A340 Airworthiness Limitations Section (ALS), Part 4—Aging Systems Maintenance, Revision 02, dated October 12, 2011. The revision date is not identified on the title page of this document.

(ii) Reserved.

(3) For service information identified in this AD, contact Airbus SAS, Airworthiness Office—EAL, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 45 80; email [airworthiness.A330-A340@airbus.com](mailto:airworthiness.A330-A340@airbus.com); Internet <http://www.airbus.com>.

(4) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.

(5) You may also review copies of the service information that is incorporated by

reference at the National Archives and Records Administration (NARA). For information on the availability of this material at an NARA facility, call 202-741-6030, or go to [http://www.archives.gov/federal\\_register/code\\_of\\_federal\\_regulations/ibr\\_locations.html](http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html).

Issued in Renton, Washington, on September 17, 2013.

**Ross Landes,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2013-23899 Filed 10-25-13; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

**[Docket No. FAA-2013-0539; Directorate Identifier 2012-NM-145-AD; Amendment 39-17616; AD 2013-20-10]**

**RIN 2120-AA64**

#### Airworthiness Directives; Airbus Airplanes

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** We are superseding airworthiness directive (AD) 2000-12-11, for certain Model A300 B4-600 and Model A300 B4-600R series airplanes. AD 2000-12-11 required repetitive inspections to detect cracks in the bolt holes inboard and outboard of rib 9 on the bottom booms of the front and rear wing spars, and repair if necessary. This new AD reduces the initial inspection compliance time and repetitive inspection interval. This AD was prompted by a fleet survey and an updated fatigue and damage tolerance analysis indicating a high risk for fatigue cracking on the front and rear spar bottom booms. We are issuing this AD to detect and correct fatigue cracks in the bolt holes of the wing spars, which could result in reduced structural integrity of a wing spar.

**DATES:** This AD becomes effective December 2, 2013.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of December 2, 2013.

**ADDRESSES:** You may examine the AD docket on the Internet at <http://www.regulations.gov/#/docketDetail;D=FAA-2013-0539>; or in person at the U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room

W12-140, 1200 New Jersey Avenue SE., Washington, DC.

For service information identified in this AD, contact Airbus SAS—EAW (Airworthiness Office), 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email [account.airworth-eas@airbus.com](mailto:account.airworth-eas@airbus.com); Internet <http://www.airbus.com>. You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

**FOR FURTHER INFORMATION CONTACT:** Dan Rodina, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; phone: (425) 227-2125; fax: (425) 227-1149.

#### SUPPLEMENTARY INFORMATION:

##### Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. The NPRM published in the **Federal Register** on July 3, 2013 (78 FR 40069), and proposed to supersede AD 2000-12-11, Amendment 39-11789 (65 FR 37853, June 19, 2000). The NPRM proposed to correct an unsafe condition for the specified products.

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, has issued EASA Airworthiness Directive 2012-0138, dated July 26, 2012 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for the specified products. The MCAI states:

Full fatigue tests carried out by the manufacturer revealed crack initiation from the bolts holes at inboard and outboard of rib 9, on the front and rear spar bottom booms. Similar cracks at the same area were reported by A300-600 aeroplane operators.

This condition, if not detected and corrected, could affect the structural integrity of the aeroplane.

To address this unsafe condition, [Direction Générale de l'Aviation Civile] DGAC France issued AD 94-208-169(B)R2 ([http://ad.easa.europa.eu/blob/19942082tb-superseded.pdf/AD\\_F-1994-208-169R2\\_2](http://ad.easa.europa.eu/blob/19942082tb-superseded.pdf/AD_F-1994-208-169R2_2)) [which corresponds to FAA AD 95-07-05, Amendment 39-9187 (60 FR 17990, April 10, 1995)] to require an ultrasonic inspection of holes inboard and outboard of rib 9 on the front and rear spar bottom booms on Left Hand and Right Hand wings.

Since that [DGAC] AD was issued, a fleet survey and updated Fatigue and Damage