

(2) If any repair identified in the "Condition" column of table 8 of paragraph 1.E., "Compliance," of Boeing Alert Service Bulletin 737-53A1214, Revision 4, dated December 16, 2011, is found and the "Reference" column refers to Appendix E of that service bulletin: At the applicable times specified in table 8 of paragraph 1.E., "Compliance," of Boeing Alert Service Bulletin 737-53A1214, Revision 4, dated December 16, 2011, except as required by paragraph (r)(2) of this AD, remove the repair and replace with a new repair, in accordance with Appendix E of Boeing Alert Service Bulletin 737-53A1214, Revision 4, dated December 16, 2011.

(3) If any non-SRM (structural repair manual) repair is found and the repair does not have FAA-approved damage tolerance inspections, except as required by paragraph (r)(2) of this AD, at the applicable time specified in table 7 of Paragraph 1.E., "Compliance," of Boeing Alert Service Bulletin 737-53A1214, Revision 4, dated December 16, 2011: Contact the Boeing Commercial Airplanes Organization Designation Authorization (ODA) that has been authorized by the Manager, Seattle Aircraft Certification Office, for damage tolerance inspections. Do those damage tolerance inspections at the times given using a method approved in accordance with the procedures specified in paragraph (u) of this AD.

(r) Retained Exceptions to the Service Information

This paragraph restates the exceptions to the service information required by paragraph (r) of AD 2012-18-13, Amendment 39-17190 (77 FR 57990, September 19, 2012).

(1) Where Boeing Alert Service Bulletin 737-53A1214, Revision 4, dated December 16, 2011, specifies to contact Boeing for repair instructions: Before further flight, repair using a method approved in accordance with the procedures specified in paragraph (u) of this AD.

(2) Where Boeing Alert Service Bulletin 737-53A1214, Revision 4, dated December 16, 2011, specifies a compliance time "after the date of Revision 1 to this service bulletin," "from the date of Revision 3 of this service bulletin," "after the date of Revision 3 to this service bulletin," or "of the effective date of AD 99-08-23," this AD requires compliance within the specified compliance time after October 24, 2012 (the effective date of AD 2012-18-13, Amendment 39-17190 (77 FR 57990, September 19, 2012)).

(3) Access and restoration procedures specified in the Accomplishment Instructions of Boeing Alert Service Bulletin 737-53A1214, Revision 4, dated December 16, 2011, are not required by this AD. Operators may do those procedures following their maintenance practices.

(4) Where table 1 of paragraph 1.E., "Compliance" of Boeing Alert Service Bulletin 737-53A1214, Revision 4, dated December 16, 2011, specifies a compliance time relative to actions done "in accordance with paragraph (a)(2) of AD 99-08-23," this AD requires compliance within the specified compliance time relative to actions specified in paragraph (g)(2) of this AD.

(5) Where the Condition columns in tables 2, 3, 5, and 6 of paragraph 1.E., "Compliance," of Boeing Alert Service Bulletin 737-53A1214, Revision 4, dated December 16, 2011, refer to total flight cycles, this AD applies to the airplanes with the specified total flight cycles as of October 24, 2012 (the effective date of AD 2012-18-13, Amendment 39-17190 (77 FR 57990, September 19, 2012)).

(s) Retained Terminating Action With Revised Paragraph Reference

This paragraph restates the terminating action specified in paragraph (s) of AD 2012-18-13, Amendment 39-17190 (77 FR 57990, September 19, 2012), with a revised paragraph reference. Accomplishment of the requirements in paragraph (k) of this AD terminates the requirements of paragraphs (g) through (j) of this AD.

(t) Credit for Previous Actions

This paragraph restates the credit for previous actions specified by paragraph (t) of AD 2012-18-13, Amendment 39-17190 (77 FR 57990, September 19, 2012). This paragraph provides credit for the actions required by paragraphs (k) through (s) of this AD, if the actions were performed before the effective date of this AD using the service bulletins specified in paragraphs (t)(1) through (t)(4) of this AD.

(1) Boeing Alert Service Bulletin 737-53A1214, dated June 17, 1999.

(2) Boeing Alert Service Bulletin 737-53A1214, Revision 1, dated June 22, 2000.

(3) Boeing Alert Service Bulletin 737-53A1214, Revision 2, dated May 24, 2001.

(4) Boeing Alert Service Bulletin 737-53A1214, Revision 3, dated January 19, 2011.

(u) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Seattle Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the ACO, send it to the attention of the person identified in the Related Information section of this AD. Information may be emailed to: 9-ANM-Seattle-ACO-AMOC-Requests@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(3) An AMOC that provides an acceptable level of safety may be used for any repair required by this AD if it is approved by the Boeing Commercial Airplanes ODA that has been authorized by the Manager, Seattle ACO, to make those findings. For a repair method to be approved, the repair must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

(4) AMOCs approved previously in accordance with AD 99-08-23, Amendment 39-11132 (64 FR 19879, April 23, 1999), are approved as AMOCs for the corresponding provisions of this AD.

(5) AMOCs approved previously in accordance with AD 2012-18-13, Amendment 39-17190 (77 FR 57990, September 19, 2012), are approved as AMOCs for the corresponding provisions of this AD.

(v) Related Information

(1) For more information about this AD, contact Alan Pohl, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue SW., Renton, Washington 98057-3356; phone: (425) 917-6440; fax: (425) 917-6590; email: alan.pohl@faa.gov.

(2) For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H-65, Seattle, Washington 98124-2207; telephone 206-544-5000, extension 1; fax 206-766-5680; Internet <https://www.myboeingfleet.com>. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.

Issued in Renton, Washington, on January 2, 2013.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2013-00186 Filed 1-8-13; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2012-1217; Directorate Identifier 2012-NE-39-AD]

RIN 2120-AA64

Airworthiness Directives; International Aero Engines AG Turbofan Engines

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for certain International Aero Engines AG (IAE), V2525-D5 and V2528-D5 turbofan engines, with a certain number (No.) 4 bearing internal scavenge tube and a certain No. 4 bearing external scavenge tube installed. This proposed AD was prompted by a report of an engine under-cowl fire and commanded in-flight shutdown. This proposed AD would require replacement of certain part number (P/N) No. 4 bearing internal scavenge tubes, and alignment checks of certain P/N No. 4 bearing external scavenge tubes. We are proposing this AD to prevent engine fire and damage to the airplane.

DATES: We must receive comments on this proposed AD by March 11, 2013.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.
- *Fax:* 202-493-2251.
- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact International Aero Engines, 628 Hebron Avenue, Suite 400, Glastonbury, CT 06033; phone: 860-368-3823; fax: 860-755-6876. You may view the referenced service information at the FAA, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803. For information on the availability of this material at the FAA, call 781-238-7125.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received and other information. The street address for the Docket Office (phone: 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Martin Adler, Aerospace Engineer, Engine & Propeller Directorate, FAA, 12 New England Executive Park, Burlington, MA 01803; phone: 781-238-7157; fax: 781-238-7199; email: martin.adler@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposal. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2012-1217; Directorate Identifier 2012-NE-39-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will

consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

We received a report of a fire warning on an IAE V2525 turbofan engine shortly after takeoff. The engine experienced an under-cowl fire and a commanded in flight shutdown. Investigation revealed that this event was caused by failure of the No. 4 bearing internal scavenge tube due to high stress. A misalignment of the No. 4 bearing external scavenge tube was noted to be a contributing factor. This proposed AD would direct the replacement of all No. 4 bearing internal scavenge tubes, P/N 2A2074-01. This proposed AD would also require checking the alignment of the No. 4 bearing external scavenge tube, P/N 6A5254, and if it fails the check, replacement of the external scavenge tube. These conditions, if not corrected, could result in engine fire and damage to the airplane.

Relevant Service Information

We reviewed IAE Service Bulletin (SB) No. V2500-ENG-72-0630, Revision 1, dated September 20, 2012. The SB describes procedures for replacement of the No. 4 bearing internal scavenge tube and for verification of proper alignment of the No. 4 bearing external scavenge tube.

FAA's Determination

We are proposing this AD because we evaluated all the relevant information and determined the unsafe condition described previously is likely to exist or develop in other products of the same type design.

Proposed AD Requirements

This proposed AD would require the replacement of the No. 4 bearing internal scavenge tube, P/N 2A2074-01, at the next combustor module-level exposure. This AD would also require verification of the alignment and installation of the No. 4 bearing external scavenge tube, P/N 6A5254, relative to the tube-to-boss elbow, P/N 2A2514 or P/N 2A3951-01, on the No. 4 bearing internal scavenge tube, P/N 2A2074-01.

Differences Between the Proposed AD and the Service Information

The SB requires replacement of the No. 4 bearing internal scavenge tube, P/N 2A2074-01, at each combustor module-level exposure. This AD would require replacement at each combustor module-level exposure after 10,000 cycles.

Interim Action

We consider this proposed AD interim action. The design approval holder is currently developing a modification that will address the unsafe condition identified in this AD. Once this modification is developed, approved, and available, we might consider additional rulemaking.

Costs of Compliance

We estimate that this proposed AD would affect 123 engines installed on airplanes of U.S. registry. We estimate that it would take 1.5 hours per engine to replace the No. 4 bearing internal scavenge tube, and 3 hours per engine to replace the No. 4 bearing external scavenge tube. Required parts would cost \$25,251 per engine. The average labor rate is \$85 per hour. Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be \$3,152,921.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and

responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

(1) Is not a “significant regulatory action” under Executive Order 12866,

(2) Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),

(3) Will not affect intrastate aviation in Alaska to the extent that it justifies making a regulatory distinction, and

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

International Aero Engines AG: Docket No. FAA–2012–1217; Directorate Identifier 2012–NE–39–AD.

(a) Comments Due Date

We must receive comments by March 11, 2013.

(b) Affected ADs

None.

(c) Applicability

This AD applies to International Aero Engines AG (IAE), V2525–D5 and V2528–D5 turbofan engines, serial numbers V20001 through V20285, with number (No.) 4 bearing internal scavenge tube, part number (P/N) 2A2074–01 and No. 4 bearing external scavenge tube, P/N 6A5254 installed.

(d) Unsafe Condition

This AD was prompted by a report of an engine under-cowl fire, commanded in-flight shutdown, and damage to the airplane. We are issuing this AD to prevent engine fire and damage to the airplane.

(e) Compliance

Comply with this AD within the compliance times specified, unless already done.

(f) No. 4 Bearing Internal Scavenge Tube, P/N 2A2074–01, Replacement

Replace the No. 4 bearing internal scavenge tube, P/N 2A2074–01, at each combustor module-level exposure after the No. 4 bearing internal scavenge tube has accumulated 10,000 flight cycles (FCs) since new. If the FCs on the tube cannot be confirmed, replace the tube at each combustor module-level exposure.

(g) No. 4 Bearing External Scavenge Tube, P/N 6A5254, Installation

At each installation, check the alignment of the No. 4 bearing external scavenge tube, P/N 6A5254, in accordance with paragraphs 3.A. PART 2, of IAE NMSB No. V2500–ENG–72–0630, Revision 1, dated September 20, 2012. If the tube is misaligned, replace with a new tube.

(h) Definitions

Combustor module level exposure is defined as separation of the combustor case and the compressor case flanges.

(i) Alternative Methods of Compliance (AMOCs)

The Manager, Engine Certification Office, FAA, may approve AMOCs to this AD. Use the procedures found in 14 CFR 39.19 to make your request.

(j) Related Information

(1) For more information about this AD, contact Martin Adler, Aerospace Engineer, Engine Certification Office, FAA, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; email: martin.adler@faa.gov, phone: 781–238–7779; fax: 781–238–7199.

(2) For service information identified in this AD, contact International Aero Engines AG, 628 Hebron Avenue, Suite 400, Glastonbury, CT 06033; phone: 860–368–3823; fax: 860–755–6876. You may view this service information at the FAA, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803. For information on the availability of this material at the FAA, call 781–238–7125.

Issued in Burlington, Massachusetts, on December 28, 2012.

Colleen M. D’Alessandro,

Assistant Manager, Engine & Propeller Directorate, Aircraft Certification Service.

[FR Doc. 2013–00212 Filed 1–8–13; 8:45 am]

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DEPARTMENT OF COMMERCE

National Oceanic and Atmospheric Administration

15 CFR Part 922

Boundary Expansion of Cordell Bank and Gulf of the Farallones National Marine Sanctuaries; Intent To Prepare Draft Environmental Impact Statement; Scoping Meetings

AGENCY: Office of National Marine Sanctuaries (ONMS), National Ocean Service (NOS), National Oceanic and Atmospheric Administration (NOAA), Department of Commerce (DOC).

ACTION: Correction.

SUMMARY: On December 21, 2012, NOAA published a notice of intent in the **Federal Register** to revise the boundaries of Cordell Bank and Gulf of the Farallones national marine sanctuaries. This document makes a correction to the dates of the scoping meetings. The end of the scoping period remains March 1, 2013.

DATES: NOAA will accept public comments on the notice of intent published at 77 FR 75601 (December 21, 2012) through March 1, 2013.

Dates for scoping meetings are:

(1) January 24, 2013 at the Bodega Bay Grange Hall.

(2) February 12, 2013 at the Point Arena High School.

(3) February 13, 2013 at the Gualala Community Center.

ADDRESSES: You may submit comments on this document, identified by NOAA–NOS–2012–0228, by any of the following methods:

- **Electronic Submission:** Submit all electronic public comments via the Federal e-Rulemaking Portal. Go to www.regulations.gov `#!/docketDetail;D=NOAA-NOS-2012-0228`, click the “Comment Now!” icon, complete the required fields, and enter or attach your comments.

- **Mail:** Maria Brown, Sanctuary Superintendent, Gulf of the Farallones National Marine Sanctuary, 991 Marine Drive, The Presidio, San Francisco, CA 94129.

Instructions: Comments sent by any other method, to any other address or individual, or received after the end of the comment period, may not be considered by NOAA. All comments received are a part of the public record and will generally be posted for public viewing on www.regulations.gov without change. All personal identifying information (e.g., name, address, etc.), confidential business information, or otherwise sensitive information