

28–2148, Revision 03, dated June 2, 2009, have been accomplished; and have modification 07633 done in production; or on which the actions specified in Airbus Service Bulletin A310–36–2015 have been done; no further action is required by this AD.

(m) Credit for Previous Actions

This paragraph provides credit for modifications required by paragraphs (g), (i), (j), and (k) of this AD, if the modifications were performed before the effective date of this AD using Airbus Mandatory Service Bulletin A310–28–2148, Revision 04, dated April 14, 2010.

(n) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Dan Rodina, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, Washington 98057–3356; telephone (425) 227–1225; fax (425) 227–1149. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD. AMOCs approved previously in accordance with AD 2008–01–05, Amendment 39–15330 (73 FR 2795, January 16, 2008), are approved as AMOCs for the corresponding provisions of this AD.

(2) *Airworthy Product*: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(o) Related Information

Refer to MCAI European Aviation Safety Agency (EASA) Airworthiness Directive 2011–0005, dated January 17, 2011, and the following service information for related information.

(1) Airbus Service Bulletin A310–28–2148, Revision 01, dated October 29, 2002.

(2) Airbus Service Bulletin A310–28–2148, Revision 02, dated March 9, 2007.

(3) Airbus Mandatory Service Bulletin A310–28–2148, Revision 05, dated August 3, 2010.

(4) Airbus Mandatory Service Bulletin A310–28–2148, Revision 06, dated August 31, 2011.

(p) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the following service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use the following service information to do the actions required by this AD, unless the AD specifies otherwise.

(3) The following service information was approved for IBR on May 15, 2012.

(i) Airbus Mandatory Service Bulletin A310–28–2148, Revision 05, dated August 3, 2010.

(ii) Airbus Mandatory Service Bulletin A310–28–2148, Revision 06, dated August 31, 2011.

(4) The following service information was approved for IBR on February 20, 2008 (73 FR 2795, January 16, 2008).

(i) Airbus Service Bulletin A310–28–2148, Revision 02, dated March 9, 2007.

(5) The following service information was approved for IBR on September 3, 2004 (69 FR 45578, July 30, 2004).

(i) Airbus Service Bulletin A310–28–2148, Revision 01, dated October 29, 2002.

(6) For service information identified in this AD, contact Airbus SAS—EAW (Airworthiness Office), 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email account.airworth-eas@airbus.com; Internet <http://www.airbus.com>.

(7) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington. For information on the availability of this material at the FAA, call 425–227–1221.

(8) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at an NARA facility, call 202–741–6030, or go to http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Renton, Washington, on January 26, 2012.

Kalene C. Yanamura,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2012–8220 Filed 4–9–12; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2011–1342; Directorate Identifier 2011–CE–038–AD; Amendment 39–16996; AD 2012–06–15]

RIN 2120–AA64

Airworthiness Directives; DG Flugzeugbau GmbH Sailplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for DG Flugzeugbau GmbH Models DG–500 Elan Orion, DG–500 Elan Trainer, DG–500/20 Elan, and DG–500/22 Elan sailplanes and Models DG–500M and DG–500MB powered sailplanes. This AD results from mandatory continuing airworthiness information (MCAI) issued by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as incorrect re-installation of the rear cockpit securing rope for the headrest of the rear seat during maintenance, which could cause the rear seat to interfere with the control stick of the sailplane. We are issuing this AD to require actions to address the unsafe condition on these products.

DATES: This AD is effective May 15, 2012.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of May 15, 2012.

ADDRESSES: You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at Document Management Facility, U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590.

For service information identified in this AD, contact DG Flugzeugbau GmbH, Otto-Lilienthal-Weg 2, 76646 Bruchsal, Federal Republic of Germany; telephone: +49 (0) 7251 3020140; fax: +49 (0) 7251 3020149; Internet: <http://www.dg-flugzeugbau.de/tech-mitteilungen-e.html>; email: dirks@dg-flugzeugbau.de. You may review copies of the referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329–4148.

FOR FURTHER INFORMATION CONTACT: Jim Rutherford, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4165; fax: (816) 329-4090; email: jim.rutherford@faa.gov.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a supplemental notice of proposed rulemaking (SNPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That SNPRM was published in the **Federal Register** on January 17, 2012 (77 FR 2236). That SNPRM proposed to correct an unsafe condition for the specified products. The MCAI states:

Several occurrences have been reported of incorrect re-installation of rear cockpit securing rope for the headrest of the rear seat during maintenance. In one of these occurrences, the aeroplane suffered an accident. The technical investigations following this accident have revealed that the rear cockpit headrest securing rope was too long, which caused the rear seat to interfere with the control stick of the aeroplane.

This condition if not detected and corrected, could lead to loss of control of the aeroplane.

To address this unsafe condition, DG Flugzeugbau have developed a modification to be accomplished in accordance with the Working Instruction No. 1 for Technical Note (TN) 348/20 in issue 3, dated 13 September 2011, for the English language version and in issue 2, dated 22 October 2008, for the German language version (English version revised at issue 3 to correct a translation discrepancy), which aims to prevent wrong re-installation of the headrest securing rope. TN 500/05 embodies this Working Instruction.

For the reasons described above, this AD requires a one-time inspection of the length of the rear cockpit headrest securing rope and, in case of discrepancy, readjustment of the length. In addition, this AD requires the installation of a modified headrest securing rope with snap hook.

You may obtain further information by examining the MCAI in the AD docket.

Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the SNPRM (77 FR 2236, January 17, 2012) or on the determination of the cost to the public.

Conclusion

We reviewed the relevant data and determined that air safety and the public interest require adopting the AD as proposed except for minor editorial changes. We have determined that these minor changes:

- Are consistent with the intent that was proposed in the SNPRM (77 FR

2236, January 17, 2012) for correcting the unsafe condition; and

- Do not add any additional burden upon the public than was already proposed in the SNPRM (77 FR 2236, January 17, 2012).

Costs of Compliance

We estimate that this AD will affect 16 products of U.S. registry. We also estimate that it will take about 2.5 work-hours per product to comply with the basic requirements of this AD. The average labor rate is \$85 per work-hour. Required parts will cost about \$1,088 per product.

Based on these figures, we estimate the cost of this AD on U.S. operators to be \$20,808, or \$1,300.50 per product.

In addition, we estimate that any necessary follow-on actions will take about 0.5 work-hour, for a cost of \$42.50 per product. We have no way of determining the number of products that may need these actions.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),

(3) Will not affect intrastate aviation in Alaska, and

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the SNPRM (77 FR 2236, January 17, 2012), the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new AD:

2012-06-15 DG Flugzeugbau GmbH:

Amendment 39-16996; Docket No. FAA-2011-1342; Directorate Identifier 2011-CE-038-AD.

(a) Effective Date

This airworthiness directive (AD) becomes effective May 15, 2012.

(b) Affected ADs

None.

(c) Applicability

This AD applies to DG Flugzeugbau GmbH Models DG-500 Elan Orion, DG-500 Elan Trainer, DG-500/20 Elan, and DG-500/22 Elan sailplanes and Models DG-500M and DG-500MB powered sailplanes, all serial numbers, that are:

- (i) Equipped with a headrest on the rear seat; and
- (ii) Certificated in any category.

(d) Subject

Air Transport Association of America (ATA) Code 25: Equipment/Furnishing

(e) Reason

This AD was prompted by mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as incorrect re-installation of the rear cockpit securing rope for the headrest of the rear seat during maintenance. We are issuing this AD to correct the length of the rear cockpit headrest securing rope, which if too long, could cause the rear seat to interfere with the control stick of the sailplane and could result in loss of control of the sailplane.

(f) Actions and Compliance

Unless already done, do the following actions:

(1) Within the next 30 days after May 15, 2012 (the effective date of this AD), inspect the rear cockpit headrest securing rope to determine the length. Do the inspection as specified in Instruction No. 2 of DG Flugzeugbau GmbH Technical Note No. 500/05, dated September 19, 2011.

(i) If the length of the rear cockpit headrest securing rope is more than 450 millimeters (mm) or less than 400 mm, before further flight after the inspection required in paragraph (f)(1) of this AD, adjust the length of the rear cockpit headrest securing rope to a length between 400 mm and 450 mm as shown in Sketch 2 of DG Flugzeugbau GmbH Working Instruction No. 1 for TN348/20, Issue 3, dated September 13, 2011. After doing the adjustment, do the action required in paragraph (f)(2) of this AD.

(ii) If the length of the rear cockpit headrest securing rope is between 400 mm and 450 mm, do the action required in paragraph (f)(2) of this AD.

(2) Within 3 months after May 15, 2012 (the effective date of this AD), replace the rear cockpit headrest securing rope with a rear cockpit headrest securing rope with a snap hook. Do the replacement following DG Flugzeugbau GmbH Working Instruction No. 1 for TN348/20, Issue 3, dated September 13, 2011, as specified in Instruction No. 3 of DG Flugzeugbau GmbH Technical Note No. 500/05, dated September 19, 2011.

(3) Replacement of the rear cockpit headrest securing rope with a rear cockpit headrest securing rope with a snap hook done before May 15, 2012 (the effective date of this AD) following DG Flugzeugbau GmbH Working Instruction No. 1 for TN348/20, Issue 2, is considered acceptable for compliance with paragraph (f)(2) of this AD.

(4) Although the European Aviation Safety Agency (EASA) MCAI and DG Flugzeugbau GmbH Technical Note No. 500/05, dated September 19, 2011, allows the inspection required in paragraph (f)(1) of this AD to be done by a pilot-owner, the U.S. regulatory system requires all actions required by this AD be done by a certified mechanic.

(g) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures

found in 14 CFR 39.19. Send information to ATTN: Jim Rutherford, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4165; fax: (816) 329-4090; email: jim.rutherford@faa.gov. Before using any approved AMOC on any sailplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product*: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements*: For any reporting requirement in this AD, a federal agency may not conduct or sponsor, and a person is not required to respond to, nor shall a person be subject to a penalty for failure to comply with a collection of information subject to the requirements of the Paperwork Reduction Act unless that collection of information displays a current valid OMB Control Number. The OMB Control Number for this information collection is 2120-0056. Public reporting for this collection of information is estimated to be approximately 5 minutes per response, including the time for reviewing instructions, completing and reviewing the collection of information. All responses to this collection of information are mandatory. Comments concerning the accuracy of this burden and suggestions for reducing the burden should be directed to the FAA at: 800 Independence Ave. SW., Washington, DC 20591, Attn: Information Collection Clearance Officer, AES-200.

(h) Related Information

Refer to MCAI EASA AD No.: 2011-0191, dated September 30, 2011; DG Flugzeugbau GmbH Technical Note No. 500/05, dated September 19, 2011; and DG Flugzeugbau GmbH Working Instruction No. 1 for TN348/20, Issue 3, dated September 13, 2011, for related information.

(i) Material Incorporated by Reference

(1) You must use the following service information to do the actions required by this AD, unless the AD specifies otherwise. The Director of the Federal Register approved the incorporation by reference (IBR) under 5 U.S.C. 552(a) and 1 CFR part 51.

(i) DG Flugzeugbau GmbH Technical Note No. 500/05, dated September 19, 2011, and

(ii) DG Flugzeugbau GmbH Working Instruction No. 1 for TN348/20, Issue 3, dated September 13, 2011.

(2) For service information identified in this AD, contact DG Flugzeugbau GmbH, Otto-Lilienthal-Weg 2, 76646 Bruchsal, Federal Republic of Germany; telephone: +49 (0) 7251 3020140; fax: +49 (0) 7251 3020149; Internet: <http://www.dg-flugzeugbau.de/tech-mitteilungen-e.html>; email: dirks@dg-flugzeugbau.de.

(3) You may review copies of the service information at the FAA, Small Airplane

Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

(4) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at an NARA facility, call 202-741-6030, or go to http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Kansas City, Missouri, on March 19, 2012.

Earl Lawrence,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2012-7003 Filed 4-9-12; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2011-1113; Directorate Identifier 2009-SW-53-AD; Amendment 39-17005; AD 2012-06-24]

RIN 2120-AA64

Airworthiness Directives; Sikorsky Aircraft Corporation Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for Sikorsky Aircraft Corporation (Sikorsky) Model S-92A helicopters. This AD was prompted by the discovery of tail rotor blade assemblies (blades) manufactured with mislocated aluminum wire mesh, leaving portions of the graphite torque tube (spar) region unprotected from a lightning strike. The actions are intended to detect mislocated blade wire mesh and to prevent spar delamination, loss of the blade tip cap during a lightning strike, blade imbalance, loss of a blade, and subsequent loss of control of the helicopter.

DATES: This AD is effective May 15, 2012.

The Director of the Federal Register approved the incorporation by reference of certain documents listed in this AD as of May 15, 2012.

ADDRESSES: For service information identified in this AD, contact Sikorsky Aircraft Corporation, Attn: Manager, Commercial Technical Support, mailstop s581a, 6900 Main Street, Stratford, CT 06614; telephone (800) 562-4409; email