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#### Related Information

(h) Refer to the following documents for related information:

(1) MCAI EASA AD No.: 2010-0107, dated June 11, 2010;

(2) Grob Aircraft Service Bulletin MSB 869-24/1, dated July 20, 2009;

(3) Grob Aircraft Service Letter SL-869-01, dated June 9, 2009;

(4) Grob Aircraft AG G 103 C Twin III SL Pilot's Operating Handbook (POH) (dated December 1991), pages 0.2A, 0.3, 0.4, and 4.9, Revision 6, dated July 20, 2009; and

(5) Grob Aircraft AG G 103 C Twin III SL Maintenance Manual (dated December 1991), page 6.12, Revision 9, dated May 24, 2002; and pages 0.1A, 0.2, 0.3, 4.2, and 6.6, Revision 10, dated December 15, 2006.

(i) For service information related to this AD, contact GROB Aircraft AG, Lettenbachstrasse 9, 86874 Tussenhausen-Mattsies, Head of Customer Service and Support, Germany; telephone: +49 (0) 8268-998-139; fax: +49 (0) 8268-998-200; e-mail [productsupport@grob-aircraft.com](mailto:productsupport@grob-aircraft.com); Internet: <http://www.grob-aircraft.eu>. You may review copies of the referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call 816-329-4148.

#### Material Incorporated by Reference

(i) You must use Grob Aircraft Service Bulletin No. MSB-869-24/1, dated July 20, 2009; Grob Aircraft Service Letter SL-869-01, dated June 9, 2009; Grob Aircraft AG G 103 C Twin III SL Pilot's Operating Handbook (POH) (dated December 1991), pages 0.2A, 0.3, 0.4, and 4.9, Revision 6, dated July 20, 2009; and Grob Aircraft AG G 103 C Twin III SL Maintenance Manual (dated December 1991), page 6.12, Revision 9, dated May 24, 2002; and pages 0.1A, 0.2, 0.3, 4.2, and 6.6, Revision 10, dated December 15, 2006; to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact GROB Aircraft AG, Lettenbachstrasse 9, 86874 Tussenhausen-Mattsies, Head of Customer Service and Support, Germany; telephone: +49 (0) 8268-

998-139; fax: +49 (0) 8268-998-200; e-mail [productsupport@grob-aircraft.com](mailto:productsupport@grob-aircraft.com); Internet: <http://www.grob-aircraft.eu>.

(3) You may review copies of the referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

(4) You may also review copies of the service information incorporated by reference for this AD at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: [http://www.archives.gov/federal\\_register/code\\_of\\_federal\\_regulations/ibr\\_locations.html](http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html).

Issued in Kansas City, Missouri, on April 22, 2011.

**John Colomy,**

*Acting Manager, Small Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2011-10388 Filed 5-10-11; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

**[Docket No. FAA-2011-0037; Directorate Identifier 2010-NM-273-AD; Amendment 39-16691; AD 2011-10-10]**

**RIN 2120-AA64**

#### **Airworthiness Directives; Airbus Model A300 B4-600, B4-600R, and F4-600R Series Airplanes, and Model C4-605R Variant F Airplanes (Collectively Called A300-600 Series Airplanes)**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

[T]he FAA has published SFAR 88 (Special Federal Aviation Regulation 88).

In their letters referenced 04/00/02/07/01-L296, dated March 4th, 2002, and 04/00/02/07/03-L024, dated February 3rd, 2003, the JAA [Joint Aviation Authorities] recommended the application of a similar regulation to the National Aviation Authorities (NAA).

Under this regulation, all holders of type certificates for passenger transport aircraft \* \* \* are required to conduct a design review against explosion risks.

During improvement of the protection of fuel pump wiring against short-circuit by accomplishment of Airbus Service Bulletin (SB) A300-24-6094, a study led by the manufacturer concluded that the harness, installed through the wing panel needed to be protected to prevent possible damage in case of chafing which could potentially lead to short-circuit [and intermittent function or loss of the inner tank fuel pump. Loss of both inner tank fuel pumps could result in inability to use the remaining fuel supply in the inner tank. A short-circuit could also result in an ignition source in a flammable leakage zone].

\* \* \* \* \*

We are issuing this AD to require actions to correct the unsafe condition on these products.

**DATES:** This AD becomes effective June 15, 2011.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of June 15, 2011.

**ADDRESSES:** You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Dan Rodina, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-2125; fax (425) 227-1149.

#### **SUPPLEMENTARY INFORMATION:**

##### **Discussion**

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on February 7, 2011 (76 FR 6581). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states:

[T]he FAA has published SFAR 88 (Special Federal Aviation Regulation 88).

In their letters referenced 04/00/02/07/01-L296, dated March 4th, 2002, and 04/00/02/07/03-L024, dated February 3rd, 2003, the JAA [Joint Aviation Authorities] recommended the application of a similar regulation to the National Aviation Authorities (NAA).

Under this regulation, all holders of type certificates for passenger transport aircraft with either a passenger capacity of 30 or more, or a payload capacity of 3,402 kg (7,500 lb) or more, which have received their certification since January 1st, 1958, are required to conduct a design review against explosion risks.

During improvement of the protection of fuel pump wiring against short-circuit by

accomplishment of Airbus Service Bulletin (SB) A300–24–6094, a study led by the manufacturer concluded that the harness, installed through the wing panel needed to be protected to prevent possible damage in case of chafing which could potentially lead to short-circuit [and intermittent function or loss of the inner tank fuel pump. Loss of both inner tank fuel pumps could result in inability to use the remaining fuel supply in the inner tank. A short-circuit could also result in an ignition source in a flammable leakage zone].

For the reasons stated above, this [EASA] AD requires the replacement of bushes in the hydraulic reservoir panel.

You may obtain further information by examining the MCAI in the AD docket.

#### Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM or on the determination of the cost to the public.

#### Conclusion

We reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed.

#### Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow our FAA policies. Any such differences are highlighted in a NOTE within the AD.

#### Costs of Compliance

We estimate that this AD will affect 120 products of U.S. registry. We also estimate that it will take about 13 work-hours per product to comply with the basic requirements of this AD. The average labor rate is \$85 per work-hour. Required parts will cost about \$266 per product. Where the service information lists required parts costs that are covered under warranty, we have assumed that there will be no charge for these parts. As we do not control warranty coverage for affected parties, some parties may incur costs higher than estimated here. Based on these figures, we estimate the cost of this AD to the U.S. operators to be \$164,520, or \$1,371 per product.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

*For the reasons discussed above, I certify this AD:*

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647–5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new AD:

**2011–10–10 Airbus:** Amendment 39–16691. Docket No. FAA–2011–0037; Directorate Identifier 2010–NM–273–AD.

#### Effective Date

- (a) This airworthiness directive (AD) becomes effective June 15, 2011.

#### Affected ADs

- (b) None.

#### Applicability

(c) This AD applies to all Airbus Model A300 B4–601, B4–603, B4–620, B4–622, B4–605R, B4–622R, F4–605R, F4–622R, and C4–605R Variant F airplanes, certificated in any category, all certified models, all serial numbers, except airplanes on which Airbus Mandatory Service Bulletin A300–24–6102 (Airbus Modification 13381) has been embodied.

#### Subject

- (d) Air Transport Association (ATA) of America Code 24: Electrical Power.

#### Reason

(e) The mandatory continuing airworthiness information (MCAI) states: [T]he FAA has published SFAR 88 (Special Federal Aviation Regulation 88). In their letters referenced 04/00/02/07/01–L296, dated March 4th, 2002, and 04/00/02/07/03–L024, dated February 3rd, 2003, the JAA [Joint Aviation Authorities] recommended the application of a similar regulation to the National Aviation Authorities (NAA).

Under this regulation, all holders of type certificates for passenger transport aircraft \* \* \* are required to conduct a design review against explosion risks.

During improvement of the protection of fuel pump wiring against short-circuit by accomplishment of Airbus Service Bulletin (SB) A300–24–6094, a study led by the manufacturer concluded that the harness, installed through the wing panel needed to be protected to prevent possible damage in case of chafing which could potentially lead to short-circuit [and intermittent function or loss of the inner tank fuel pump. Loss of both inner tank fuel pumps could result in

inability to use the remaining fuel supply in the inner tank. A short-circuit could also result in an ignition source in a flammable leakage zone].

\* \* \* \* \*

#### Compliance

(f) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

#### Actions

(g) Within 30 months after the effective date of this AD, install Teflon bushes in the hydraulic reservoir panel at the lower left-hand side in accordance with the Accomplishment Instructions of Airbus Mandatory Service Bulletin A300-24-6102, Revision 01, dated September 24, 2010.

#### Credit for Actions Accomplished in Accordance With Previous Service Information

(h) Actions done before the effective date of this AD in accordance with Airbus Mandatory Service Bulletin A300-24-6102, dated August 13, 2009, are acceptable for compliance with the corresponding requirements of this AD.

#### FAA AD Differences

**Note 1:** This AD differs from the MCAI and/or service information as follows: No differences.

#### Other FAA AD Provisions

(i) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to *Attn:* Dan Rodina, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-2125; fax (425) 227-1149. Information may be e-mailed to: *9-ANM-116-AMOC-REQUESTS@faa.gov*. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

#### Related Information

(j) Refer to MCAI European Aviation Safety Agency Airworthiness Directive 2010-0225,

dated November 5, 2010; and Airbus Mandatory Service Bulletin A300-24-6102, Revision 01, dated September 24, 2010; for related information.

#### Material Incorporated by Reference

(k) You must use Airbus Mandatory Service Bulletin A300-24-6102, Revision 01, dated September 24, 2010, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Airbus SAS-EAW (Airworthiness Office), 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; e-mail *account.airworth-eas@airbus.com*; Internet *http://www.airbus.com*.

(3) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.

(4) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: *http://www.archives.gov/federal\_register/code\_of\_federal\_regulations/ibr\_locations.html*.

Issued in Renton, Washington, on April 26, 2011.

**Kalene C. Yanamura,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2011-10817 Filed 5-10-11; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2010-0706; Directorate Identifier 2010-NM-064-AD; Amendment 39-16683; AD 2011-10-02]

**RIN 2120-AA64**

#### **Airworthiness Directives; The Boeing Company Model 747-400, 747-400D, and 747-400F Series Airplanes Equipped With General Electric CF6-80C2 or Pratt & Whitney PW4000 Series Engines**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for the products listed above. This AD requires modifying certain thrust reverser control

system wiring to the flap control unit (FCU). This AD was prompted by a report of automatic retraction of the leading edge flaps due to indications transmitted to the FCU from the thrust reverser control system during takeoff. We are issuing this AD to prevent automatic retraction of the leading edge flaps during takeoff, which could result in reduced climb performance and consequent collision with terrain and obstacles or forced landing of the airplane.

**DATES:** This AD is effective June 15, 2011.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of June 15, 2011.

**ADDRESSES:** For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H-65, Seattle, Washington 98124-2207; telephone 206-544-5000, extension 1; fax 206-766-5680; e-mail *me.boecom@boeing.com*; Internet *https://www.myboeingfleet.com*. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.

#### Examining the AD Docket

You may examine the AD docket on the Internet at *http://www.regulations.gov*; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (*phone:* 800-647-5527) is Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

#### FOR FURTHER INFORMATION CONTACT:

Tung Tran, Aerospace Engineer, Propulsion Branch, ANM-140S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; *phone:* 425-917-6505; fax 425-917-6590; *e-mail:* *tung.tran@faa.gov*.

#### SUPPLEMENTARY INFORMATION:

##### Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an airworthiness directive (AD) that would apply to the