

(ii) The specimen set must have a minimum of ten (10) flaws which provide an acoustic response similar to PWSCC indications. All flaws must be greater than 10 percent of the nominal pipe wall thickness. A minimum of 20 percent of the total flaws must initiate from the inside surface and 20 percent from the outside surface. At least 20 percent of the flaws must be in the depth ranges of 10–30 percent through wall thickness and at least 20 percent within a depth range of 31–50 percent through wall thickness. At least 20 percent and no more than 60 percent of the flaws must be oriented axially.

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Dated at Rockville, Maryland, this 24th day of July 2009.

For the Nuclear Regulatory Commission.

Bruce S. Mallett,

Acting Executive Director for Operations.

[FR Doc. E9–18546 Filed 8–4–09; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2009–0509 Directorate Identifier 2009–CE–029–AD; Amendment 39–15985; AD 2009–16–02]

RIN 2120–AA64

Airworthiness Directives; Pilatus Aircraft Limited Model PC–7 Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) issued by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

This Airworthiness Directive (AD) is prompted due to reported corrosion on the bolts and in the bores of the attachment fittings for the engine mounting frame. The corrosion is caused by damaged cadmium plating of the bolts or damaged surface finish of the attachment fitting.

Such a condition, if left uncorrected, could lead to crack initiation at the bolt and the fitting bore and subsequently to the failure of the engine attachment fitting.

We are issuing this AD to require actions to correct the unsafe condition on these products.

DATES: This AD becomes effective September 9, 2009.

On September 9, 2009, the Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD.

ADDRESSES: You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at Document Management Facility, U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Doug Rudolph, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4059; fax: (816) 329–4090.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on June 5, 2009 (74 FR 26994). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states:

This Airworthiness Directive (AD) is prompted due to reported corrosion on the bolts and in the bores of the attachment fittings for the engine mounting frame. The corrosion is caused by damaged cadmium plating of the bolts or damaged surface finish of the attachment fitting.

Such a condition, if left uncorrected, could lead to crack initiation at the bolt and the fitting bore and subsequently to the failure of the engine attachment fitting.

In order to correct and control the situation, this AD requires a visual inspection of the relevant bolts and fittings. Additionally, the replacement of the bolts is required.

Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM or on the determination of the cost to the public.

Conclusion

We reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed.

Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use

different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow FAA policies. Any such differences are highlighted in a NOTE within the AD.

Costs of Compliance

We estimate that this AD will affect 10 products of U.S. registry. We also estimate that it will take about 4.5 work-hours per product to comply with the basic requirements of this AD. The average labor rate is \$80 per work-hour. Required parts will cost about \$300 per product.

Based on these figures, we estimate the cost of this AD to the U.S. operators to be \$6,600 or \$660 per product.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

- (1) Is not a “significant regulatory action” under Executive Order 12866;
- (2) Is not a “significant rule” under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and

(3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD Docket.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new AD:

2009-16-02 Pilatus Aircraft Limited:

Amendment 39-15985; Docket No. FAA-2009-0509; Directorate Identifier 2009-CE-029-AD.

Effective Date

(a) This airworthiness directive (AD) becomes effective September 9, 2009.

Affected ADs

(b) None.

Applicability

(c) This AD applies to Model PC-7 airplanes, all manufacturer serial numbers, certificated in any category.

Subject

(d) Air Transport Association of America (ATA) Code 53: Fuselage.

Reason

(e) The mandatory continuing airworthiness information (MCAI) states:

This Airworthiness Directive (AD) is prompted due to reported corrosion on the

bolts and in the bores of the attachment fittings for the engine mounting frame. The corrosion is caused by damaged cadmium plating of the bolts or damaged surface finish of the attachment fitting.

Such a condition, if left uncorrected, could lead to crack initiation at the bolt and the fitting bore and subsequently to the failure of the engine attachment fitting.

In order to correct and control the situation, this AD requires a visual inspection of the relevant bolts and fittings. Additionally, the replacement of the bolts is required.

Actions and Compliance

(f) Unless already done, do the following actions:

(1) Visually inspect the bolts and the bores (with boroscope) of the attachment fittings for the engine mounting frame following paragraph 3.A of PILATUS Aircraft Ltd. Pilatus PC-7 Service Bulletin No. 53-006, dated November 17, 2008, at whichever of the following occurs later:

(i) Upon accumulating 5,000 hours total time-in-service (TIS) or 5 years from the date of manufacture, whichever occurs first; or

(ii) Within the next 6 months after September 9, 2009 (the effective date of this AD).

(2) If no sign of corrosion is found during the inspection required in paragraph (f)(1) of this AD, before further flight, replace the bolts. Repetitively inspect thereafter at intervals not to exceed every 5 years following PILATUS Aircraft Ltd. Pilatus PC-7 Maintenance Manual Chapter 05-10-20, page 4, dated November 30, 2008.

(3) If any sign of corrosion is found during any of the inspections required in paragraphs (f)(1) and (f)(2) of this AD, before further flight, do the corrective actions following paragraph 3.A. of PILATUS Aircraft Ltd. Pilatus PC-7 Service Bulletin No. 53-006, dated November 17, 2008. Repetitively inspect thereafter at intervals not to exceed every 5 years following PILATUS Aircraft Ltd. Pilatus PC-7 Maintenance Manual Chapter 05-10-20, page 4, dated November 30, 2008.

FAA AD Differences

Note: This AD differs from the MCAI and/or service information as follows: No differences.

Other FAA AD Provisions

(g) The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Doug Rudolph, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4059; fax: (816) 329-4090. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) Airworthy Product: For any requirement in this AD to obtain corrective

actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) Reporting Requirements: For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act (44 U.S.C. 3501 et seq.), the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

Related Information

(h) Refer to MCAI FOCA AD HB-2009-004, dated May 12, 2009; PILATUS Aircraft Ltd. Pilatus PC-7 Service Bulletin No. 53-006, dated November 17, 2008; and Pilatus PC-7 Maintenance Manual Chapter 05-10-20, page 4, dated November 30, 2008, for related information.

Material Incorporated by Reference

(i) You must use PILATUS Aircraft Ltd. Pilatus PC-7 Service Bulletin No. 53-006, dated November 17, 2008; and Pilatus PC-7 Maintenance Manual Chapter 05-10-20, page 4, dated November 30, 2008, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact PILATUS AIRCRAFT LTD., Customer Service Manager, CH-6371 STANS, Switzerland; telephone: +41 (0)41 619 62 08; fax: +41 (0)41 619 73 11; Internet: <http://www.pilatus-aircraft.com/>, or e-mail: snolan@pilatus-aircraft.com.

(3) You may review copies of the service information incorporated by reference for this AD at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the Central Region, call (816) 329-3768.

(4) You may also review copies of the service information incorporated by reference for this AD at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Kansas City, Missouri, on July 23, 2009.

Kim Smith,

Manager, Small Airplane Directorate, Aircraft Certification Service.

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