

**FAA AD Differences**

**Note:** This AD differs from the MCAI and/or service information as follows: No differences.

**Other FAA AD Provisions**

(g) The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, New York Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Rocco Viselli, Aerospace Engineer, Airframe and Propulsion Branch, ANE-171, FAA, New York ACO, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone (516) 228-7331; fax (516) 794-5531. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) Reporting Requirements: For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

**Related Information**

(h) Refer to MCAI Canadian Airworthiness Directive CF-2008-25, dated July 3, 2008; and Bombardier Service Bulletin 670BA-24-011, Revision C, dated November 28, 2005; for related information.

**Material Incorporated by Reference**

(i) You must use Bombardier Service Bulletin 670BA-24-011, Revision C, dated November 28, 2005, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Bombardier, Inc., 400 Côte-Vertu Road West, Dorval, Québec H4S 1Y9, Canada; telephone 514-855-5000; fax 514-855-7401; e-mail [thd.crj@aero.bombardier.com](mailto:thd.crj@aero.bombardier.com); Internet <http://www.bombardier.com>.

(3) You may review copies of the service information that is incorporated by reference at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221 or 425-227-1152.

(4) You may also review copies of the service information at the National Archives and Records Administration (NARA). For information on the availability of this

material at NARA, call 202-741-6030, or go to: [http://www.archives.gov/federal\\_register/code\\_of\\_federal\\_regulations/ibr\\_locations.html](http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html).

Issued in Renton, Washington, on November 26, 2008.

**Ali Bahrami,**

*Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. E8-29077 Filed 12-10-08; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

**[Docket No. FAA-2007-0175; Directorate Identifier 2007-NM-184-AD; Amendment 39-15766; AD 2008-25-08]**

**RIN 2120-AA64**

**Airworthiness Directives; Boeing Model 757 Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for certain Boeing Model 757 airplanes. This AD requires changing the wiring of the fuel boost pump and doing other specified actions. This AD results from reports of short circuits in an electrical connector at the wing-to-body electrical disconnect panel. We are issuing this AD to prevent a short circuit of the electrical connector for the fuel boost pump, which could cause the instruments for fuel, flap, slat, and aileron systems to malfunction and create a potential ignition source inside the fuel tanks. A potential ignition source inside the fuel tank in combination with flammable fuel vapors could result in a fuel tank explosion and consequent loss of the airplane.

**DATES:** This AD is effective January 15, 2009.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of January 15, 2009.

**ADDRESSES:** For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H-65, Seattle, Washington 98124-2207; telephone 206-544-5000, extension 1, fax 206-766-5680; e-mail [me.boecom@boeing.com](mailto:me.boecom@boeing.com); Internet <https://www.myboeingfleet.com>.

**Examining the AD Docket**

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800-647-5527) is the Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

**FOR FURTHER INFORMATION CONTACT:** Philip Sheridan, Aerospace Engineer, Systems and Equipment Branch, ANM-130S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 917-6441; fax (425) 917-6590.

**FOR FURTHER INFORMATION CONTACT:**

Philip Sheridan, Aerospace Engineer, Systems and Equipment Branch, ANM-130S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 917-6441; fax (425) 917-6590.

**SUPPLEMENTARY INFORMATION:****Discussion**

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an airworthiness directive (AD) that would apply to certain Boeing Model 757 airplanes. That NPRM was published in the **Federal Register** on November 9, 2007 (72 FR 63512). That NPRM proposed to require changing the wiring of the fuel boost pump and doing other specified actions.

**Comments**

We gave the public the opportunity to participate in developing this AD. We considered the comments received.

**Request To Incorporate Revised Service Information**

Boeing asks that we refer to Boeing Special Attention Service Bulletins 757-28-0095 and 757-28-0096, both Revision 1, both dated June 4, 2008, in the final rule. Boeing Special Attention Service Bulletins 757-28-0095 and 757-28-0096, both dated June 18, 2007, were referred to in the NPRM as the appropriate sources of service information for accomplishing the actions specified.

We have reviewed Revision 1 of these referenced service bulletins and we agree with the commenter since no additional work is necessary on airplanes changed in accordance with the original issue of the referenced service information. Revision 1 of these service bulletins clarifies certain procedures and certain routing and splice locations. We have added Revision 1 of these service bulletins to the applicability specified in paragraphs (c)(1) and (c)(2) of this AD, and to paragraph (f) of this AD, as the appropriate sources of service

information for accomplishing the actions specified. In addition, we have added credit to paragraph (f) for previously accomplishing the actions using the original issue of the service bulletins.

#### Request To Change Airplane Manufacturer's Name

Boeing asks that the airplane manufacturer's name specified in the product identification section of the regulatory text of the NPRM be changed from "Airbus" to "Boeing."

We agree that the airplane manufacturer's name should be changed, as this was an inadvertent error in the NPRM; we have changed the name in that paragraph of the AD accordingly.

#### Request To Allow Alternate Routing of Wiring

Continental Airlines (CAL) refers to Figure 2 of Boeing Special Attention Service Bulletin 757-28-0095, dated June 18, 2007, and states that it would be easier and more appropriate when doing the wire modification to utilize one of the open holes in the panel instead of splicing the wires for this location. CAL adds that the splice locations for the left-hand aft and right-hand aft boost pumps, as shown in Boeing Special Attention Service Bulletins 757-28-0095 and 757-28-0096, both dated June 18, 2007, are incorrect and the splices cannot be accomplished in those areas. CAL recommends that these service bulletins be revised with the proper alternative rework instructions for the subject discrepancies.

We have reviewed the referenced service information and we do not agree that the splices cannot be accomplished in the locations referred to in the comment. In addition, we have determined that clarification of certain routing and splice locations is helpful, and that clarification is provided in Revision 1 of the referenced service bulletins, as noted previously. We infer that CAL is also asking us to allow alternative routing of the wiring to that specified in the referenced service bulletins. Under the provisions of paragraph (g) of this AD, we will consider requests for approval of an AMOC if sufficient data are submitted to substantiate that the alternative routing of the wiring would provide an acceptable level of safety. We have made no change to the AD in this regard.

#### Conclusion

We reviewed the relevant data, considered the comments received, and

determined that air safety and the public interest require adopting the AD with the changes described previously. We also determined that these changes will not increase the economic burden on any operator or increase the scope of the AD.

#### Costs of Compliance

There are 1,697 airplanes of the affected design in the worldwide fleet. This AD affects about 673 airplanes of U.S. registry. The required actions take up to 12 work hours per airplane, at an average labor rate of \$80 per work hour. Based on these figures, the estimated cost of the AD for U.S. operators is up to \$646,080, or up to \$960 per airplane.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General Requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979), and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

You can find our regulatory evaluation and the estimated costs of compliance in the AD Docket.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new AD:

**2008-25-08 Boeing:** Amendment 39-15766. Docket No. FAA-2007-0175; Directorate Identifier 2007-NM-184-AD.

#### Effective Date

(a) This airworthiness directive (AD) is effective January 15, 2009.

#### Affected ADs

(b) None.

#### Applicability

(c) This AD applies to the airplanes identified in paragraphs (c)(1) and (c)(2) of this AD, certificated in any category.

(1) Boeing Model 757-200, -200PF, and -200CB series airplanes, as identified in Boeing Special Attention Service Bulletin 757-28-0095, Revision 1, dated June 4, 2008.

(2) Boeing Model 757-300 series airplanes, as identified in Boeing Special Attention Service Bulletin 757-28-0096, Revision 1, dated June 4, 2008.

#### Unsafe Condition

(d) This AD results from reports of short circuits in an electrical connector at the wing-to-body electrical disconnect panel. We are issuing this AD to prevent a short circuit of the electrical connector for the fuel boost pump, which could cause the instruments for the fuel, flap, slat, and aileron systems to malfunction and create a potential ignition source inside the fuel tank. A potential ignition source inside the fuel tank in combination with flammable fuel vapors could result in a fuel tank explosion and consequent loss of the airplane.

#### Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

#### Fuel Boost Pump Wiring Change

(f) Within 60 months after the effective date of this AD, change the wiring of the fuel boost pump and do all other specified actions as applicable, by accomplishing all of the applicable actions specified in the Accomplishment Instructions of Boeing Special Attention Service Bulletin 757-28-

0095 (for Model 757–200, –200PF, and –200CB series airplanes) or 757–28–0096 (for Model 757–300 series airplanes), both Revision 1, both dated June 4, 2008; as applicable. The other specified actions must be done before further flight after changing the fuel boost pump wiring. Actions accomplished before the effective date of this AD in accordance with Boeing Special Attention Service Bulletin 757–28–0095 or 757–28–0096, both dated June 18, 2007, are considered acceptable for compliance with the corresponding actions in this paragraph.

#### Alternative Methods of Compliance (AMOCs)

(g)(1) The Manager, Seattle Aircraft Certification Office (ACO), FAA, ATTN: Philip Sheridan, Aerospace Engineer, Systems and Equipment Branch, ANM–130S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057–3356; telephone (425) 917–6441; fax (425) 917–6590; has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19.

(2) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

#### Material Incorporated by Reference

(h) You must use Boeing Special Attention Service Bulletin 757–28–0095, Revision 1, dated June 4, 2008; or Boeing Special Attention Service Bulletin 757–28–0096, Revision 1, dated June 4, 2008; as applicable; to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H–65, Seattle, Washington 98124–2207; telephone 206–544–5000, extension 1, fax 206–766–5680; e-mail [me.boecom@boeing.com](mailto:me.boecom@boeing.com); Internet <https://www.myboeingfleet.com>.

(3) You may review copies of the service information that is incorporated by reference at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425–227–1221 or 425–227–1152.

(4) You may also review copies of the service information at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202–741–6030, or go to: [http://www.archives.gov/federal\\_register/code\\_of\\_federal\\_regulations/ibr\\_locations.html](http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html).

Issued in Renton, Washington, on November 28, 2008.

**Ali Bahrami,**

*Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. E8–29079 Filed 12–10–08; 8:45 am]

**BILLING CODE 4910–13–P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

**[Docket No. FAA–2008–1274; Directorate Identifier 2008–NM–197–AD; Amendment 39–15764; AD 2008–25–06]**

**RIN 2120–AA64**

#### **Airworthiness Directives; Airbus Model A319, A320, and A321 Series Airplanes Equipped With International Aero Engines (IAE) Model V2500–A1 Engines or Model V25xx–A5 Series Engines**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule; request for comments.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

\* \* \* Airbus has advised that an incorrect part number has been introduced in the IPC (illustrated parts catalog) \* \* \* for the rear engine mount barrel nut. This problem affects Airbus A319, A320 and A321 models with IAE (International Aero Engine) V2500–A5 engines.

The part number introduced in error is not certificated for the IAE V2500–A5 engine installation and, if installed, may fail in service.

\* \* \* \* \*

Failure of the rear engine mount barrel nut could result in reduced structural integrity of the rear engine mount and possible separation of the engine from the airplane, and a consequent hazard to the airplane and persons and property on the ground. This AD requires actions that are intended to address the unsafe condition described in the MCAI.

**DATES:** This AD becomes effective December 26, 2008.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of December 26, 2008.

We must receive comments on this AD by January 12, 2009.

**ADDRESSES:** You may send comments by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* (202) 493–2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–40, 1200 New Jersey Avenue, SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647–5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

**FOR FURTHER INFORMATION CONTACT:** Tim Dulin, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057–3356; telephone (425) 227–2141; fax (425) 227–1149.

#### SUPPLEMENTARY INFORMATION:

##### Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, has issued EASA Emergency Airworthiness Directive 2008–0191–E, dated October 20, 2008 (referred to after this as “the MCAI”), to correct an unsafe condition for the specified products. The MCAI states:

DGAC–France AD 90–079–009(B) R1 [and corresponding FAA AD 90–22–08, amendment 39–6781 (corrected on November 8, 1990, 55 FR 47028)] introduced an inspection and daily rectification action on Airbus A320–231 aircraft in order to prevent failure of one or more of the engine rear mount barrel nuts.

Subsequent to the above problem, Airbus has advised that an incorrect part number has been introduced in the IPC (illustrated parts catalog) (reference 71–22–11 Figure 80B item 180) for the rear engine mount barrel nut. This problem affects Airbus A319, A320 and A321 models with IAE (International Aero Engine) V2500–A5 engines.