which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAAapproved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) Reporting Requirements: For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120–0056.

Related Information

(h) Refer to MCAI Brazilian Airworthiness Directives 2006–11–02R3 and 2006–11–03R3, both effective June 13, 2008; EMBRAER Service Bulletins 170–22–0003 and 190–22– 0002, both Revision 01, both dated November 5, 2007; EMBRAER Service Bulletin 170–31– 0019, Revision 01, dated June 25, 2007; and EMBRAER Service Bulletin 190–31–0009, Revision 02, dated June 29, 2007; for related information.

Material Incorporated by Reference

(i) You must use the applicable service information specified in Table 1 of this AD to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Empresa Brasileira de Aeronautica S.A. (EMBRAER), P.O. Box 343—CEP 12.225, Sao Jose dos Campos—SP, Brazil.

(3) You may review copies at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741–6030, or go to: http:// www.archives.gov/federal-register/cfr/ibrlocations.html.

TABLE 1—MATERIAL INCORPORATED BY REFERENCE

EMBRAER Service Bulletin—	Revision—	Dated—	
170–22– 0003.	01	November 5, 2007.	
170–31– 0019.	01	June 25, 2007.	
190–22– 0002.	01	November 5, 2007.	
190–31– 0009.	02	June 29, 2007.	

Issued in Renton, Washington, on August 6, 2008.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. E8–19143 Filed 8–27–08; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2008-0621; Directorate Identifier 2008-NM-015-AD; Amendment 39-15653; AD 2008-17-15]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 737–600, –700, –800, and –900 Series Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT. **ACTION:** Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for certain Boeing Model 737-600, -700, -800, and -900 series airplanes. This AD requires installing hot short protector (HSP) support brackets and equipment for the fuel quantity indicating system (FQIS) fuel densitometer and other specified actions as applicable. This AD also requires revising the Airworthiness Limitations (AWLs) section of the Instructions for Continued Airworthiness to incorporate AWL No. 28-AWL-07. This AD results from fuel system reviews conducted by the manufacturer. We are issuing this AD to prevent the center tank fuel densitometer from overheating and becoming a potential ignition source inside the center fuel tank, which, in combination with flammable fuel vapors, could result in a center fuel tank explosion and consequent loss of the airplane.

DATES: This AD is effective October 2, 2008.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of October 2, 2008.

ADDRESSES: For service information identified in this AD, contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124–2207.

Examining the AD Docket

You may examine the AD docket on the Internet at *http:// www.regulations.gov*; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800–647–5527) is the Document Management Facility, U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT:

Georgios Roussos, Aerospace Engineer, Systems and Equipment Branch, ANM– 130S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057–3356; telephone (425) 917–6482; fax (425) 917–6590.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an airworthiness directive (AD) that would apply to certain Boeing Model 737-600, -700, -800, and -900 series airplanes. That NPRM was published in the Federal Register on June 9, 2008 (73 FR 32491). That NPRM proposed to require installing hot short protector (HSP) support brackets and equipment for the fuel quantity indicating system (FQIS) fuel densitometer and other specified actions as applicable. That NPRM proposed to also require revising the Airworthiness Limitations (AWLs) section of the Instructions for Continued Airworthiness to incorporate AWL No. 28-AWL-07.

Comments

We gave the public the opportunity to participate in developing this AD. We considered the single comment received. Boeing concurs with the NPRM.

Change to Final Rule Regarding Later Revisions of Service Information

We removed all references to the use of "later revisions" of the applicable service information from this AD to be consistent with FAA and Office of the **Federal Register** policies. We may consider approving the use of later revisions of the service information as an alternative method of compliance with this AD, as provided by paragraph (k) of this AD.

Conclusion

We reviewed the relevant data, considered the comment received, and determined that air safety and the public interest require adopting the AD with the change described previously. We also determined that this change will not increase the economic burden on any operator or increase the scope of the AD.

Costs of Compliance

We estimate that this AD affects 13 airplanes of U.S. registry. The following

ESTIMATED COSTS

table provides the estimated costs, at an average labor rate of \$80 per work hour, for U.S. operators to comply with this AD.

Action	Work hours	Parts	Cost per airplane	Number of U.Sreg- istered air- planes	Fleet cost
Installation of HSP support brackets and equipment. AWLs revision	Up to 16	Up to \$14,698	Up to \$15,978	13	Up to \$207,714
	1	None	\$80	13	\$1,040

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

(1) Is not a "significant regulatory action" under Executive Order 12866,

(2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979), and

(3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

You can find our regulatory evaluation and the estimated costs of compliance in the AD Docket.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

2008–17–15 Boeing: Amendment 39–15653. Docket No. FAA–2008–0621; Directorate Identifier 2008–NM–015–AD.

Effective Date

(a) This airworthiness directive (AD) is effective October 2, 2008.

Affected ADs

(b) None.

Applicability

(c) This AD applies to Boeing Model 737– 600, -700, -800, and -900 series airplanes, certificated in any category; as identified in Boeing Alert Service Bulletin 737–28A1221, Revision 1, dated November 9, 2007.

Note 1: This AD requires revisions to certain operator maintenance documents to include new inspections. Compliance with these inspections is required by 14 CFR 91.403(c). For airplanes that have been previously modified, altered, or repaired in the areas addressed by these inspections, the operator may not be able to accomplish the inspections described in the revisions. In this situation, to comply with 14 CFR 91.403(c), the operator must request approval for an alternative method of compliance (AMOC) according to paragraph (k) of this AD. The request should include a description of changes to the required inspections that will ensure the continued operational safety of the airplane.

Unsafe Condition

(d) This AD results from fuel system reviews conducted by the manufacturer. We are issuing this AD to prevent the center tank fuel densitometer from overheating and becoming a potential ignition source inside the center fuel tank, which, in combination with flammable fuel vapors, could result in a center fuel tank explosion and consequent loss of the airplane.

Compliance

(e) Comply with this AD within the compliance times specified, unless already done.

Installation of the Hot Short Protector (HSP)

(f) Within 60 months after the effective date of this AD, install the HSP support brackets and equipment for the fuel quantity indicating system (FQIS) fuel densitometer and do all the other specified actions as applicable, by accomplishing all of the applicable actions specified in the Accomplishment Instructions of Boeing Alert Service Bulletin 737–28A1221, Revision 1, dated November 9, 2007.

Airworthiness Limitations (AWLs) Revision for AWL No. 28–AWL–07

(g) Concurrently with accomplishing the actions required by paragraph (f) of this AD, revise the AWLs section of the Instructions for Continued Airworthiness (ICA) by incorporating AWL No. 28–AWL–07 of Subsection F, "AIRWORTHINESS LIMITATIONS—FUEL SYSTEM AWLS," of Section 9 of the Boeing 737–600/700/800/900 Maintenance Planning Data (MPD) Document, D626A001–CMR, Revision March 2007 R2 (hereafter referred to as "the MPD").

No Alternative Critical Design Configuration Control Limitations (CDCCLs)

(h) After accomplishing the action specified in paragraph (g) of this AD, no alternative CDCCLs may be used unless the CDCCLs are approved as an alternative methods of compliance (AMOC) in accordance with the procedures specified in paragraph (k) of this AD.

Credit for Actions Done According to Previous Issue of Service Bulletin

(i) Actions done before the effective date of this AD in accordance with Boeing Alert Service Bulletin 737–28A1221, dated January 14, 2007, are acceptable for compliance with the requirements of paragraph (f) of this AD.

Terminating Action for AWLs Revision

(j) Incorporating AWL No. 28–AWL–07 into the AWLs section of the ICA in accordance with paragraph (g)(3) of AD 2008–10–10, amendment 39–15516, terminates the action required by paragraph (g) of this AD.

AMOCs

(k)(1) The Manager, Seattle Aircraft Certification Office, FAA, ATTN: Georgios Roussos, Aerospace Engineer, Systems and Equipment Branch, ANM–130S, 1601 Lind Avenue SW., Renton, Washington 98057– 3356; telephone (425) 917–6482; fax (425) 917–6590; has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19.

(2) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

Material Incorporated by Reference

(l) You must use Boeing Alert Service Bulletin 737–28A1221, Revision 1, dated November 9, 2007; and Airworthiness Limitation 28–AWL–07 of Section 9 of the Boeing 737–600/700/800/900 Maintenance Planning Data (MPD) Document, D626A001– CMR, Revision March 2007 R2; to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124–2207.

(3) You may review copies of the service information incorporated by reference at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202–741–6030, or go to: http:// www.archives.gov/federal_register/ code_of_federal_regulations/ ibr_locations.html.

Issued in Renton, Washington, on August 12, 2008.

Michael J Kaszycki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. E8–19367 Filed 8–27–08; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2008-0148; Directorate Identifier 2007-NM-299-AD; Amendment 39-15655; AD 2008-17-17]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 747 Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT. **ACTION:** Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for all Boeing Model 747 airplanes. This AD requires a one-time inspection of certain fuselage skins at section 41 to find any external doublers that cover the inspection areas and to identify the external doublers that end on a stringer and those that do not, and related investigative and corrective actions if necessary. This AD results from reports of cracks found at fastener locations in the fuselage skins at section 41. We are issuing this AD to detect and correct fuselage skin cracks at fastener locations along the skin-to-stringer attachments, which could join together and become large and consequently result in rapid decompression of the cabin.

DATES: This AD is effective October 2, 2008.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of October 2, 2008.

ADDRESSES: For service information identified in this AD, contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124–2207.

Examining the AD Docket

You may examine the AD docket on the Internet at http:// www.regulations.gov; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800–647–5527) is the Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Ivan Li, Aerospace Engineer, Airframe Branch, ANM–120S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057–3356; telephone (425) 917–6437; fax (425) 917–6590.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an airworthiness directive (AD) that would apply to all Boeing Model 747 airplanes. That NPRM was published in the **Federal Register** on February 8, 2008 (73 FR 7486). That NPRM proposed to require a one-time inspection of certain fuselage skins at section 41 to find any external doublers that cover the inspection areas and to identify the external doublers that end on a stringer and those that do not, and related investigative and corrective actions if necessary.

Comments

We gave the public the opportunity to participate in developing this AD. We considered the comments received from the two commenters.

Support for the NPRM

Boeing concurs with the contents of the NPRM.

Request to Either Delay Issuance of the AD or Extend the Compliance Time

Japan Airlines (JAL) requests that we delay issuance of the AD until Boeing has published a revision to Boeing Alert Service Bulletin 747-53A2704, dated October 4, 2007, to extend the compliance time from 16,000 total flight cycles to 25,000 total flight cycles for accomplishing the general visual inspection for external doublers in Areas 2 and 3. If we cannot delay issuance of the AD, then JAL requests that we revise the compliance time accordingly in this AD. JAL states that Boeing has advised that only the inspection of Area 1 must be done before 16,000 total flight cycles, and that Areas 2 and 3 should be inspected together with the high frequency eddy current inspection of the skin-to-stringer attachments before 25,000 total flight cycles. JAL asserts that Boeing will revise the compliance time in the next revision to the service bulletin.

We do not agree to delay issuance of this AD, or to revise the compliance time for inspecting for external doublers in Areas 2 and 3. We have coordinated with Boeing, and Boeing has no plans, at this time, to revise the service bulletin. Boeing also has confirmed that the inspection of Areas 1, 2, and 3 was left at 16,000 total flight cycles for simplicity of the compliance table in the service bulletin. Therefore, we agree that, for Groups 1 through 5 airplanes,