

# Rules and Regulations

Federal Register

Vol. 73, No. 146

Tuesday, July 29, 2008

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## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2007-29240; Directorate Identifier 2007-CE-076-AD; Amendment 39-15618; AD 2008-15-06]

RIN 2120-AA64

#### Airworthiness Directives; Cessna Aircraft Company Models 175 and 175A Airplanes

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for certain Cessna Aircraft Company (Cessna) Models 175 and 175A airplanes. This AD requires you to check the airplane logbook to determine if the original engine mounting brackets have been replaced. If the original engine mounting brackets are still installed, this AD requires you to repetitively inspect those brackets for cracks and replace any cracked engine mounting bracket. After replacing all four original engine mounting brackets, no further

action will be required by this AD. This AD was prompted by a report that the engine became detached from the firewall during landing on one of the affected airplanes. We are issuing this AD to detect and correct cracks in the engine mounting brackets, which could result in failure of the engine mounting bracket. This failure could lead to the engine detaching from the firewall.

**DATES:** This AD becomes effective on September 2, 2008.

On September 2, 2008, the Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD.

**ADDRESSES:** For service information identified in this AD, contact Cessna Aircraft Company, Product Support, P.O. Box 7706, Wichita, Kansas 67277; telephone: (316) 517-5800; fax: (316) 942-9006.

To view the AD docket, go to U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590, or on the Internet at <http://www.regulations.gov>. The docket number is FAA-2007-29240; Directorate Identifier 2007-CE-076-AD.

**FOR FURTHER INFORMATION CONTACT:** Gary Park, Aerospace Engineer, 1801 Airport Road, Room 100, Wichita, Kansas 67209; telephone: (316) 946-4123; fax: (316) 946-4107.

#### SUPPLEMENTARY INFORMATION:

##### Discussion

On March 31, 2008, we issued a proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to certain Cessna Models 175 and 175A

airplanes. This proposal was published in the **Federal Register** as a notice of proposed rulemaking (NPRM) on April 8, 2008 (73 FR 19017). The NPRM proposed to require you to check the airplane logbook to determine if the original engine mounting brackets have been replaced. If the original engine mounting brackets are still installed, the NPRM proposed to require you to repetitively inspect those brackets for cracks and replace any cracked engine mounting bracket. After replacing all four original engine mounting brackets, no further action would be required by the NPRM.

#### Comments

We provided the public the opportunity to participate in developing this AD. We received no comments on the proposal or on the determination of the cost to the public.

#### Conclusion

We have carefully reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed except for minor editorial corrections. We have determined that these minor corrections:

- Are consistent with the intent that was proposed in the NPRM for correcting the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM.

#### Costs of Compliance

We estimate that this AD affects 1,218 airplanes in the U.S. registry.

We estimate the following costs to do the inspection:

| Labor cost                                   | Parts cost           | Total cost per airplane | Total cost on U.S. operators |
|--|----------------------|-------------------------|------------------------------|
| 7.5 work-hours × \$80 per hour = \$600 ..... | Not applicable ..... | \$600                   | \$730,800                    |

We estimate the following costs to do any necessary replacements:

| Labor cost   | Parts cost   | Total cost per airplane                               |
|--|--|---|
| 3 work-hours per bracket × \$80 per hour = \$240 per bracket. 4 brackets per airplane × \$240 per bracket = \$960. | \$200 per bracket. 4 × \$200 = \$800 for all 4 brackets. | \$440 per bracket. \$1,760 to replace all 4 brackets. |

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106 describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this AD.

**Regulatory Findings**

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and

responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a summary of the costs to comply with this AD (and other information as included in the Regulatory Evaluation) and placed it in the AD Docket. You may get a copy of this summary by sending a request to us at the address listed under **ADDRESSES**. Include "Docket No. FAA-2007-29240; Directorate Identifier 2007-CE-076-AD" in your request.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**Adoption of the Amendment**

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

2. FAA amends § 39.13 by adding the following new AD:

**2008-15-06 Cessna Aircraft Company:**  
Amendment 39-15618; Docket No. FAA-2007-29240; Directorate Identifier 2007-CE-076-AD.

**Effective Date**

(a) This AD becomes effective on September 2, 2008.

**Affected ADs**

(b) None.

**Applicability**

(c) This AD applies to the following airplane models and serial numbers that are certificated in any category:

| Model          | Serial Nos.                        | Year manufactured |
|----------------|------------------------------------|-------------------|
| (1) 175 .....  | 55001 through 55703 .....          | 1958.             |
| (2) 175 .....  | 55704 through 56238 .....          | 1959.             |
| (3) 175 .....  | 28700A, 626, and 640 .....         | 1958 and 1959.    |
| (4) 175A ..... | 691, and 56239 through 56777 ..... | 1960.             |

**Unsafe Condition**

(d) A report that the engine became unattached from the firewall during landing on one of the affected airplanes prompts this AD. We are issuing this AD to detect and

correct cracks in the engine mounting brackets, which could result in failure of the engine mounting bracket. This failure could lead to the engine detaching from the firewall.

**Compliance**

(e) To address this problem, you must do the following, unless already done:

| Actions  | Compliance  | Procedures   |
|--|---|--|
| (1) Check the airplane logbook to determine if all four of the original engine mounting brackets have been replaced.   | Within the next 30 days after September 2, 2008 (the effective date of this AD).  | The owner/operator holding at least a private pilot certificate as authorized by section 43.7 may do this action.  |
| (2) If you can positively determine that all four of the original engine mounting brackets have been replaced, no further action is required.  | Not applicable .....  | Make an entry into the aircraft logbook showing compliance with this portion of the AD in accordance with 14 CFR 43.9. The owner/operator holding at least a private pilot certificate as authorized by section 43.7 may do this action. |
| (3) If you cannot positively determine that all four of the original engine mounting brackets have been replaced, inspect each of the upper and lower engine mounting brackets on both the left and right sides for cracks as specified in the service bulletin. | Initially inspect within the next 12 months after September 2, 2008 (the effective date of this AD). If no cracks are found, repetitively inspect thereafter at intervals not to exceed 500 hours time-in-service (TIS) until all four of the original engine mounting brackets are replaced. | Follow Cessna Single Engine Service Bulletin SEB07-2, Revision 2, dated June 18, 2007.   |
| (4) If cracks are found in any of the engine mounting brackets during any inspection required in paragraph (e)(3) of this AD, replace the cracked engine mounting bracket(s).  | Before further flight after the inspection in which cracks are found. Replacing the cracked engine mounting bracket terminates the repetitive inspection required in paragraph (e)(3) of this AD only for the replaced engine mounting bracket.   | Follow Cessna Single Engine Service Bulletin SEB07-2, Revision 2, dated June 18, 2007.   |

| Actions  | Compliance  | Procedures   |
|--|---|--|
| (5) To terminate the repetitive inspections required in paragraph (e)(3) of this AD, you may replace all four original engine mounting brackets.   | At any time before or after the initial inspection required in paragraph (e)(3) of this AD. | Follow Cessna Single Engine Service Bulletin SEB07-2, Revision 2, dated June 18, 2007. |
| (6) Dispose of every replaced bracket following 14 CFR 43.10, paragraph (c)(6), which states the following: "Mutilation. The part may be mutilated to deter its installation in a type certificated product. The mutilation must render the part beyond repair and incapable of being reworked to appear to be airworthy". | Before further flight after the engine mounting bracket is removed for replacement.         | Not applicable.  |

#### Alternative Methods of Compliance (AMOCs)

(f) The Manager, Wichita Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Gary Park, Aerospace Engineer, FAA, Wichita ACO, 1801 Airport Road, Room 100, Wichita, Kansas 67209; telephone: 316-946-4123; fax: 316-946-4107; e-mail address: [gary.park@faa.gov](mailto:gary.park@faa.gov). Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

#### Material Incorporated by Reference

(g) You must use Cessna Single Engine Service Bulletin SEB07-2, Revision 2, dated June 18, 2007, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Cessna Aircraft Company, Product Support, P.O. Box 7706, Wichita, Kansas 67277.

(3) You may review copies at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Kansas City, Missouri 64106; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: [http://www.archives.gov/federal\\_register/code\\_of\\_federal\\_regulations/ibr\\_locations.html](http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html).

Issued in Kansas City, Missouri, on July 15, 2008.

**Kim Smith,**

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. E8-16583 Filed 7-28-08; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 71

[Docket No. FAA-2008-0460; Airspace Docket No. 08-AAL-18]

#### Establishment of Class E Airspace; Venetie, AK

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** This action establishes Class E airspace at Venetie, AK to provide adequate controlled airspace to contain aircraft executing Standard Instrument Approach Procedures (SIAPs) and Obstacle Departure Procedures (ODPs). Two SIAPs and an ODP are being developed for the Venetie Airport at Venetie Alaska. This action establishes Class E airspace upward from 700 feet (ft.) and 1,200 ft. above the surface at the Venetie Airport, Venetie, AK.

**DATES:** *Effective Date:* 0901 UTC, September 25, 2008. The Director of the Federal Register approves this incorporation by reference action under title 1, Code of Federal Regulations, part 51, subject to the annual revision of FAA Order 7400.9 and publication of conforming amendments.

**FOR FURTHER INFORMATION CONTACT:** Gary Rolf, AAL-538G, Federal Aviation Administration, 222 West 7th Avenue, Box 14, Anchorage, AK 99513-7587; telephone number (907) 271-5898; fax: (907) 271-2850; e-mail: [gary.ctr.rolf@faa.gov](mailto:gary.ctr.rolf@faa.gov). Internet address: <http://www.alaska.faa.gov/at>.

#### SUPPLEMENTARY INFORMATION:

#### History

On Thursday, May 29, 2008, the FAA proposed to amend part 71 of the Federal Aviation Regulations (14 CFR part 71) to establish Class E airspace upward from 700 ft. and 1,200 ft. above the surface at Venetie, AK (73 FR 30820). The action was proposed in order to create Class E airspace

sufficient in size to contain aircraft while executing instrument procedures for the Venetie Airport. Class E controlled airspace extending upward from 700 ft. and 1,200 ft. above the surface in the Venetie Airport area is established by this action.

Interested parties were invited to participate in this rulemaking proceeding by submitting written comments on the proposal to the FAA. No comments were received. The rule is adopted as proposed.

The area will be depicted on aeronautical charts for pilot reference. The coordinates for this airspace docket are based on North American Datum 83. The Class E airspace areas designated as 700/1,200 ft. transition areas are published in paragraph 6005 of FAA Order 7400.9R, *Airspace Designations and Reporting Points*, signed August 15, 2007, and effective September 15, 2007, which is incorporated by reference in 14 CFR 71.1. The Class E airspace designations listed in this document will be published subsequently in the Order.

#### The Rule

This amendment to 14 CFR part 71 establishes Class E airspace at the Venetie Airport, Alaska. This Class E airspace is established to accommodate aircraft executing instrument procedures, and will be depicted on aeronautical charts for pilot reference. The intended effect of this rule is to provide adequate controlled airspace for Instrument Flight Rules (IFR) operations at the Venetie Airport, Venetie, Alaska.

The FAA has determined that this regulation only involves an established body of technical regulations for which frequent and routine amendments are necessary to keep them operationally current. It, therefore—(1) Is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034; February 26, 1979); and (3) does not warrant preparation of a regulatory evaluation as the anticipated