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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2007-0157; Directorate Identifier 2001-NE-23-AD; Amendment 39-15469; AD 2008-08-16]

RIN 2120-AA64

Airworthiness Directives; Turbomeca Makila 1A and 1A1 Turboshaft Engines

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: The FAA is superseding an existing airworthiness directive (AD) for Turbomeca Makila 1A, 1A1, and 1A2 turboshaft engines. That AD currently requires replacing certain digital electronic control units (DECUs) and electronic control units (ECUs) with modified DECUs and ECUs. This AD applies only to Makila 1A and 1A1 turboshaft engines, and requires replacing the selector-comparator board in the ECU with a board incorporating Turbomeca modification TU 250. This AD results from recent unexplained reversions of the ECU to the 65% N1 back-up mode. We are issuing this AD to prevent dual-engine continued operation at 65% N1 after reversion of the ECU to the 65% N1 back-up mode due to temporary loss of N2 speed signal, which could lead to inability to continue safe flight, emergency autorotation landing, or an accident.

DATES: This AD becomes effective May 21, 2008.

ADDRESSES: The Docket Operations office is located at Docket Management Facility, U.S. Department of Transportation, 1200 New Jersey Avenue, SE., West Building Ground Floor, Room W12-140, Washington, DC 20590-0001.

You can get the service information identified in this AD from Turbomeca, 40220 Tarnos, France; telephone (33) 05 59 74 40 00; fax (33) 05 59 74 45 15.

FOR FURTHER INFORMATION CONTACT:

James Lawrence, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; e-mail: james.lawrence@faa.gov; telephone (781) 238-7176; fax (781) 238-7199.

SUPPLEMENTARY INFORMATION: The FAA proposed to amend 14 CFR part 39 by superseding AD 2002-15-05, Amendment 39-12833 (67 FR 49859, August 1, 2002), with a proposed AD. The proposed AD applies to Turbomeca Makila 1A and 1A1 turboshaft engines. We published the proposed AD in the **Federal Register** on November 15, 2007 (72 FR 64172). That action proposed to require replacing the selector-comparator board in the ECU with a board incorporating Turbomeca modification TU 250.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647-5527) is provided in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

Comments

We provided the public the opportunity to participate in the development of this AD. We received no comments on the proposal or on the determination of the cost to the public. We also found we needed to clarify the unsafe condition statement from "We are issuing this AD to prevent dual-engine reversion of the ECU to the 65% N1 back-up mode, which could lead to inability to continue safe flight, emergency autorotation landing, or an accident" to "We are issuing this AD to prevent dual-engine continued operation at 65% N1 after reversion of the ECU to the 65% N1 back-up mode due to temporary loss of N2 speed signal, which could lead to inability to

continue safe flight, emergency autorotation landing, or an accident".

Conclusion

We have carefully reviewed the available data and determined that air safety and the public interest require adopting the AD with the change described previously.

Makila 1A2 Turboshaft Engines Excluded From This AD

Although Makila 1A2 turboshaft engines, which were also listed in the previous AD, might be affected by this unsafe condition, EASA is reviewing the need to mandate a corrective action. Depending on the review outcome, we might address those engines in another AD action.

Costs of Compliance

We estimate that this AD will affect 10 Makila 1A and 1A1 turboshaft engines installed on helicopters of U.S. registry. We also estimate that it will take about 1 work-hour per engine to perform the actions, and that the average labor rate is \$80 per work-hour. Required parts will cost about \$3,500 per engine. Based on these figures, we estimate the total cost of the AD to U.S. operators to be \$35,800.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will

not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866;
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a summary of the costs to comply with this AD and placed it in the AD Docket. You may get a copy of this summary at the address listed under **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by removing Amendment 39-12833 (67 FR 49859, August 1, 2002), and by adding a new airworthiness directive, Amendment 39-15469, to read as follows:

2008-08-16 Turbomeca: Amendment 39-15469. Docket No. FAA-2007-0157; Directorate Identifier 2001-NE-23-AD.

Effective Date

- (a) This airworthiness directive (AD) becomes effective May 21, 2008.

Affected ADs

- (b) This AD supersedes AD 2002-15-05, Amendment 39-12833.

Applicability

- (c) This AD applies to Turbomeca Makila 1A and 1A1 turboshaft engines. These engines are installed on, but not limited to, Eurocopter France model AS 332C, AS 332L, and AS 332L1 helicopters.

Unsafe Condition

- (d) This AD results from recent unexplained reversion of the electronic control unit (ECU) to the 65% N1 back-up

mode. The actions specified in this AD are intended to prevent dual-engine continued operation at 65% N1 after reversion of the ECU to the 65% N1 back-up mode due to temporary loss of N2 speed signal, which could lead to inability to continue safe flight, emergency autorotation landing, or an accident.

Compliance

- (e) You are responsible for having the actions required by this AD performed before June 30, 2008, unless the actions have already been done.

(f) Replace the selector-comparator board in the ECU with a board incorporating Turbomeca Modification TU 250. Information on Modification TU 250 can be found in Turbomeca Mandatory Service Bulletin No. 298 73 0250, dated March 23, 2007.

Alternative Methods of Compliance

- (g) The Manager, Engine Certification Office, FAA, has the authority to approve alternative methods of compliance for this AD if requested using the procedures found in 14 CFR 39.19.

Related Information

(h) European Aviation Safety Agency AD 2007-0144, dated May 18, 2007, also addresses the subject of this AD.

(i) Contact James Lawrence, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; e-mail: james.lawrence@faa.gov; telephone (781) 238-7176; fax (781) 238-7199, for more information about this AD.

Issued in Burlington, Massachusetts, on April 8, 2008.

Peter A. White,

Assistant Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. E8-8083 Filed 4-15-08; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71

[Docket No. FAA-2008-0003; Airspace Docket No. 08-ASW-1]

Amendment of Class E Airspace; Lexington, OK

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Direct final rule; confirmation of effective date, correction.

SUMMARY: This action confirms the effective date and makes a correction to the direct final rule that establishes Class E airspace at Muldrow Army Heliport, Lexington, OK, published in the **Federal Register** February 15, 2008 (73 FR 8795) Docket No. FAA-2008-0003. In the airspace description of the

rule, the geographic coordinates were incorrect, and reference to Notice to Airmen and Airport/Facility Directory should be removed. This action corrects those errors.

DATES: *Effective Dates:* 0901 UTC April 10, 2008. The Director of the Federal Register approves this incorporation by reference action under Title 1, Code of Federal Regulations, part 51, subject to the annual revision of FAA Order 7400.9 and publication of conforming amendments.

FOR FURTHER INFORMATION CONTACT: Gary Mallett, Central Service Center, System Support Group, Federal Aviation Administration, Southwest Region, 2601 Meacham Blvd., Fort Worth, TX 76193-0530; telephone (817) 222-4949.

SUPPLEMENTARY INFORMATION:

History

The FAA published a direct final rule with request for comments in the **Federal Register** February 15, 2008, (73 FR 8795), Docket No. FAA-2008-0003. Subsequent to publication, the FAA found that the geographic coordinates for the Heliport were incorrect, and the sentence referencing the Notice to Airmen and Airport/Facility Directory should not have been included in the airspace description of this action.

The FAA uses the direct final rule procedure for non-controversial rules where the FAA believes that there will be no adverse public comment. This direct final rule advised the public that no adverse comments were anticipated, and that unless a written adverse comment, or a written notice of intent to submit an adverse comment, was received within the comment period, the regulation would become effective on April 10, 2008. No adverse comments were received; thus, this notice confirms that the direct final rule will become effective on this date.

Correction

■ In the **Federal Register** dated February 15, 2008, in **Federal Register** Docket No. FAA-2008-0003, on page 8796, column 2, line 31, correct to read: (Lat. 35°01'00" N., long. 97°14'01" W.

■ On page 8796, column 2, line 39, remove the following:

“This Class E5 airspace is effective during specific dates and times established in advance by Notice to Airmen. The effective date and time will thereafter be continuously published in the Airport/Facility Directory.”

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