

**Related Information**

(h) Refer to MCAI European Aviation Safety Agency Airworthiness Directive 2006-0115, dated May 10, 2006; and the Dassault Service Bulletins listed in Table 1 of this AD, for related information.

Issued in Renton, Washington, on August 31, 2007.

**Stephen P. Boyd,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. E7-18045 Filed 9-12-07; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2007-29174; Directorate Identifier 2007-NM-125-AD]

RIN 2120-AA64

**Airworthiness Directives; Boeing Model 737-100, -200, -200C, -300, -400, and -500 Series Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** The FAA proposes to adopt a new airworthiness directive (AD) for certain Boeing Model 737-100, -200, -200C, -300, -400, and -500 series airplanes. This proposed AD would require repetitive inspections to detect cracking of the body station 303.9 frame, and corrective action if necessary. This proposed AD also provides for optional terminating action for the repetitive inspections. This proposed AD results from reports of cracks found at the cutout in the web of body station frame 303.9 inboard of stringer 16L. We are proposing this AD to detect and correct such cracking, which could prevent the left forward entry door from sealing correctly, and could cause in-flight decompression of the airplane.

**DATES:** We must receive comments on this proposed AD by October 29, 2007.

**ADDRESSES:** Use one of the following addresses to submit comments on this proposed AD.

- *DOT Docket Web site:* Go to <http://dms.dot.gov> and follow the instructions for sending your comments electronically.

- *Government-wide rulemaking Web site:* Go to <http://www.regulations.gov> and follow the instructions for sending your comments electronically.

- *Mail:* U.S. Department of Transportation, Docket Operations, M-

30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

- *Fax:* (202) 493-2251.
- *Hand Delivery:* Room W12-140 on the ground floor of the West Building, 1200 New Jersey Avenue, SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207, for the service information identified in this proposed AD.

**FOR FURTHER INFORMATION CONTACT:**

Howard Hall, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 917-6430; fax (425) 917-6590.

**SUPPLEMENTARY INFORMATION:****Comments Invited**

We invite you to submit any relevant written data, views, or arguments regarding this proposed AD. Send your comments to an address listed in the **ADDRESSES** section. Include the docket number "FAA-2007-29174; Directorate Identifier 2007-NM-125-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the proposed AD. We will consider all comments received by the closing date and may amend the proposed AD in light of those comments.

We will post all comments we receive, without change, to <http://dms.dot.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed AD. Using the search function of that web site, anyone can find and read the comments in any of our dockets, including the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You may review DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR 19477-78), or you may visit <http://dms.dot.gov>.

**Examining the Docket**

You may examine the AD docket on the Internet at <http://dms.dot.gov>, or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Operations office (telephone (800) 647-5527) is located on the

ground floor of the West Building at the DOT street address stated in the **ADDRESSES** section. Comments will be available in the AD docket shortly after the Docket Management System receives them.

**Discussion**

We have received reports of cracks found at the cutout in the web of the body station 303.9 frame inboard of stringer 16L on seven Boeing Model 737 "classic" airplanes. The cracks were found on airplanes that had accumulated between 37,562 and 64,000 total flight cycles. Such cracking, if not corrected, could prevent the left forward entry door from sealing correctly, and could cause in-flight decompression of the airplane.

**Relevant Service Information**

We have reviewed two service bulletins related to this action. The service bulletins are similar but affect different groups of airplanes.

Boeing Alert Service Bulletin 737-53A1188, Revision 2, dated May 9, 2007, for certain Model 737-300, -400, and -500 series airplanes, describes the following actions:

- Repetitive high-frequency eddy current (HFEC) and detailed inspections to detect cracking in the station 303.9 web and doubler around the cutouts for door stop straps at stringers 15L and 16L.

- A repair/preventive change, which includes installing a new web, doubler, and stop fitting assemblies; changing the shape of the web cutout; and doing an eddy current inspection.

Service Bulletin 737-53A1188 specifies a threshold for the initial inspection of 10,000 total flight cycles and a grace period of 2,250 flight cycles.

Boeing Alert Service Bulletin 737-53A1197, dated August 25, 2006, for certain Model 737-100, -200, -200C, -300, -400, and -500 series airplanes, describes the following actions:

- Repetitive ultrasound inspections of the slot-shaped cutout in the web for the door stop strap at stringer 16L.

- Repetitive HFEC inspections of the web along the upper edge and lower edge of the doubler around the doorstop strap at stringer 16L.

- Repetitive detailed inspections of the web around the doubler for the cutout at stringer 16.

- A repair/preventive change, which involves installing a new web and doubler.

Service Bulletin 737-53A1197 specifies a threshold for the initial inspection of 30,000 total flight cycles and a grace period of 2,250 flight cycles.

For both service bulletins, a repair/preventive change eliminates the need

for the repetitive inspections. For airplanes on which the repair/preventive change was previously done according to the original version or Revision 1 of Alert Service Bulletin 737-53A1188, replacing the existing kit with a new kit (in accordance with Revision 2) is necessary to eliminate the need for the repetitive inspections.

Accomplishing the actions specified in the service bulletins is intended to adequately address the unsafe condition.

**FAA’s Determination and Requirements of the Proposed AD**

We have evaluated all pertinent information and identified an unsafe

condition that is likely to exist or develop on other airplanes of this same type design. For this reason, we are proposing this AD, which would require accomplishing the actions specified in the service information described previously, except as discussed below.

**Difference Between Proposed AD and Service Information**

The service bulletins specify to contact the manufacturer for instructions on how to repair certain conditions, but this proposed AD would require repairing those conditions in one of the following ways:

- Using a method that we approve; or

- Using data that meet the certification basis of the airplane, and that have been approved by an Authorized Representative for the Boeing Commercial Airplanes Delegation Option Authorization Organization whom we have authorized to make those findings.

**Costs of Compliance**

There are about 2,765 airplanes of the affected design in the worldwide fleet. The following table provides the estimated costs, depending on airplane configuration, for U.S. operators to comply with this proposed AD.

ESTIMATED COSTS

Action	Work hours	Average labor rate per hour	Parts	Cost per airplane	Number of U.S.-registered airplanes	Fleet cost
Inspection .....	1 to 4 .....	\$80	None .....	\$80 to \$320, per inspection cycle.	1,154 .....	\$92,320 to \$369,280, per inspection cycle.
Repair/preventive change, if done.	12 to 30 .....	80	\$564 to \$2,236	\$1,524 to \$4,636 .....	Up to 1,154 .....	Up to \$5,349,944.

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, “General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

We have determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not

have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed regulation:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket. See the ADDRESSES section for a location to examine the regulatory evaluation.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Safety.

**The Proposed Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

**Boeing:** Docket No. FAA-2007-29174; Directorate Identifier 2007-NM-125-AD.

**Comments Due Date**

- (a) The FAA must receive comments on this AD action by October 29, 2007.

**Affected ADs**

- (b) None.

**Applicability**

- (c) This AD applies to the airplanes, certificated in any category, identified in Table 1 of this AD.

TABLE 1.—APPLICABILITY

Boeing model—	As identified in Boeing Alert Service Bulletin—
737-100, -200, and -200C series airplanes .....	737-53A1197, dated August 25, 2006.
737-300, -400, and -500 series airplanes .....	737-53A1188, Revision 2, dated May 9, 2007, or 737-53A1197, dated August 25, 2006.

**Unsafe Condition**

(d) This AD results from reports of cracks found at the cutout in the web of body station frame 303.9 inboard of stringer 16L. We are issuing this AD to detect and correct such cracking, which could prevent the left forward entry door from sealing correctly, and could cause in-flight decompression of the airplane.

**Compliance**

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

**Repetitive Inspections: Service Bulletin 737-53A1188**

(f) For airplanes identified in Boeing Alert Service Bulletin 737-53A1188, Revision 2, dated May 9, 2007, including airplanes modified by the repair/preventive change specified in the original version, dated April 9, 1998, or Revision 1, dated March 18, 1999, of the service bulletin: Do detailed and high frequency eddy current (HFEC) inspections in the web and doubler around the slotted holes in the frame web at stringers 15L and 16L, in accordance with the Accomplishment Instructions of the service bulletin. Do the inspections at the applicable time specified in paragraph 1.E. of the service bulletin, except as provided by paragraph (h) of this AD. Do all applicable corrective actions before further flight in accordance with the service bulletin, except as provided by paragraph (i) of this AD. Repeat the inspections at intervals not to exceed 4,500 flight cycles until accomplishment of the repair/preventive change in accordance with the service bulletin, which terminates the repetitive inspection requirements. A repair/preventive change done in accordance with the original version or Revision 1 of the service bulletin does not terminate the repetitive inspections, but the repetitive inspections may be terminated after the existing kit is replaced with a new kit in accordance with Revision 2 of the service bulletin, paragraph 3.B., Part II, step 3, or Part III, step 3.

**Repetitive Inspections: Service Bulletin 737-53A1197**

(g) For airplanes identified in Boeing Alert Service Bulletin 737-53A1197, dated August 25, 2006: Do an ultrasound inspection of the slot-shaped cutout in the web for the door stop strap at stringer 16L, an HFEC inspection of the web along the upper and lower edges of the doubler around the doorstop strap at stringer 16L, and a detailed inspection of the web around the doubler for the cutout at stringer 16L, in accordance with the Accomplishment Instructions of the service bulletin. Do the inspections at the applicable time specified in paragraph 1.E. of the service bulletin, except as provided by paragraph (h) of this AD. Do all applicable corrective actions before further flight in accordance with the service bulletin, except as provided by paragraph (i) of this AD. Repeat the inspections at intervals not to exceed 4,500 flight cycles, until accomplishment of the repair/preventive change in accordance with the service

bulletin, which terminates the repetitive inspections.

**Exceptions to Service Bulletin Specifications**

(h) Where Boeing Alert Service Bulletin 737-53A1188, Revision 2, dated May 9, 2007; and Boeing Alert Service Bulletin 737-53A1197, dated August 25, 2006, specify a compliance time after release of the service bulletin, this AD requires compliance within the specified time after the effective date of this AD.

(i) Where Boeing Alert Service Bulletin 737-53A1188, Revision 2, dated May 9, 2007; and Boeing Alert Service Bulletin 737-53A1197, dated August 25, 2006, specify to contact Boeing for appropriate action, including repair of damage outside the scope of the service bulletin, repair using a method approved in accordance with the procedures specified in paragraph (j) of this AD.

**Alternative Methods of Compliance (AMOCs)**

(j)(1) The Manager, Seattle Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(3) An AMOC that provides an acceptable level of safety may be used for any repair required by this AD, if it is approved by an Authorized Representative for the Boeing Commercial Airplanes Delegation Option Authorization Organization who has been authorized by the Manager, Seattle ACO, to make those findings. For a repair method to be approved, the repair must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

Issued in Renton, Washington, on August 31, 2007.

**Stephen P. Boyd,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. E7-18049 Filed 9-12-07; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF COMMERCE****Bureau of Economic Analysis****15 CFR Part 806**

[Docket No. 07 0301041-7043-02]

RIN 0691-AA63

**Direct Investment Surveys: BE-11, Annual Survey of U.S. Direct Investment Abroad**

**AGENCY:** Bureau of Economic Analysis, Commerce.

**ACTION:** Notice of proposed rulemaking.

**SUMMARY:** This proposed rule would amend regulations concerning the reporting requirements for the BE-11, Annual Survey of U.S. Direct Investment Abroad. The BE-11 survey is conducted annually and is a sample survey that obtains financial and operating data covering the overall operations of U.S. parent companies and their foreign affiliates. Currently, banks are excluded from coverage. BEA proposes to expand the reporting requirements on the BE-11 annual survey so that U.S. parent companies that are banks, foreign affiliates of bank parents, and bank foreign affiliates of nonbank parents will be reportable. A few minor changes will be required to the instructions on Form BE-11A, Report for U.S. Reporter, so it can be used to collect bank as well as nonbank data. BEA is now implementing a new, specialized Form BE-11B for foreign affiliates of bank parents and bank foreign affiliates of nonbank parents.

**DATES:** Comments on this proposed rule will receive consideration if submitted in writing on or before 5 p.m. November 13, 2007.

**ADDRESSES:** You may submit comments, identified by RIN 0691-AA63, and referencing the agency name (Bureau of Economic Analysis), by any of the following methods:

- *Federal eRulemaking Portal:* <http://www.regulations.gov>. Follow the instructions for submitting comments. For agency, select "Commerce Department—all."
- *E-mail:* [David.Galler@bea.gov](mailto:David.Galler@bea.gov).
- *Fax:* Office of the Chief, Direct Investment Division, (202) 606-5318.
- *Mail:* Office of the Chief, Direct Investment Division, U.S. Department of Commerce, Bureau of Economic Analysis, BE-50, Washington, DC 20230.
- *Hand Delivery/Courier:* Office of the Chief, Direct Investment Division, U.S. Department of Commerce, Bureau of Economic Analysis, BE-50, Shipping and Receiving, Section M100, 1441 L Street, NW., Washington, DC 20005.

Public Inspection: Comments may be inspected at BEA's offices, 1441 L Street, NW., Room 7005, between 8:30 a.m. and 5 p.m., Eastern Time Monday through Friday.

**FOR FURTHER INFORMATION CONTACT:** David H. Galler, Chief, Direct Investment Division (BE-50), Bureau of Economic Analysis, U.S. Department of Commerce, Washington, DC 20230; phone (202) 606-9835.

**SUPPLEMENTARY INFORMATION:** This proposed rule would amend 15 CFR Part 806.14 to set forth the reporting requirements for the BE-11, Annual