

After consideration of all relevant material presented, including the Committee's recommendation, and other information, it is hereby found that this interim final rule, as hereinafter set forth, will tend to effectuate the declared policy of the Act.

Pursuant to 5 U.S.C. 553, it is also found and determined upon good cause that it is impracticable, unnecessary, and contrary to the public interest to give preliminary notice prior to putting this rule into effect and that good cause exists for not postponing the effective date of this rule until 30 days after publication in the **Federal Register** because: (1) Handlers are already shipping potatoes from the 2006–2007 crop and want to take advantage of the grade relaxation aspect of this rule as soon as possible; (2) handlers are aware of this rule, which was recommended at a public meeting; and (3) this rule provides a 60-day comment period and any additional comments received will be considered prior to finalization of this rule.

List of Subjects in 7 CFR Part 948

Marketing agreements, Potatoes, Reporting and recordkeeping requirements.

■ For the reasons set forth in the preamble, 7 CFR part 948 is amended as follows:

PART 948—IRISH POTATOES GROWN IN COLORADO

■ 1. The authority citation for 7 CFR part 948 continues to read as follows:

Authority: 7 U.S.C. 601–674.

■ 2. In § 948.386, paragraphs (a)(3) and (a)(4), and the introductory text of paragraph (b) are revised to read as follows:

§ 948.386 Handling regulation.

* * * * *

(a) * * *

(3) *1 1/2-inch minimum to 2 1/4-inch maximum diameter (Size B).* U.S. Commercial grade or better: *Provided*, That round, red-skinned varieties shall grade U.S. No. 1 or better.

(4) *1-inch minimum diameter to 1 3/4-inch maximum diameter.* U.S. Commercial grade or better: *Provided*, That round, red-skinned varieties shall grade U.S. No. 1 or better.

* * * * *

(b) *Maturity (skinning) requirements.* From August 1 through October 31 shall be:

* * * * *

Dated: December 21, 2006.
Lloyd C. Day,
Administrator, Agricultural Marketing Service.
 [FR Doc. 06–9897 Filed 12–21–06; 4:59 pm]
BILLING CODE 3410–02–P

NUCLEAR REGULATORY COMMISSION

**10 CFR Part 72
 RIN 3150–AH98**

List of Approved Spent Fuel Storage Casks: HI–STORM 100 Revision 3; Withdrawal of Direct Final Rule

AGENCY: U.S. Nuclear Regulatory Commission.

ACTION: Direct final rule; withdrawal.

SUMMARY: The U.S. Nuclear Regulatory Commission (NRC) is withdrawing a direct final rule that would have revised the Holtec International HI–STORM 100 cask system listing within the “List of Approved Spent Fuel Storage Casks” to include Amendment No. 3 to the Certificate of Compliance. The NRC is taking this action because it has received significant adverse comments in response to the direct final rule. These significant adverse comments shall be considered as comments to the companion proposed rule that was published concurrently with the direct final rule.

FOR FURTHER INFORMATION CONTACT: Jayne M. McCausland, Office of Federal and State Materials and Environmental Management Programs, U.S. Nuclear Regulatory Commission, Washington, DC 20555, telephone (301) 415–6219 (e-mail: *jmm2@nrc.gov*).

SUPPLEMENTARY INFORMATION: On October 16, 2006 (71 FR 60659), the NRC published in the **Federal Register** a direct final rule amending its regulations in 10 CFR 72.214 to revise the Holtec International HI–STORM 100 cask system listing within the “List of Approved Spent Fuel Storage Casks” to include Amendment No. 3 to the Certificate of Compliance Number 1014 (CoC No. 1014). Amendment No. 3 modifies the present cask system design by revising: Technical Specification (TS) 3.1.3 to eliminate cooling of the Multi-Purpose Canister (MPC) cavity prior to reflood with water, as part of cask unloading operations; TS 3.3.1 to allow linear interpolation between minimal soluble boron concentrations, for certain fuel enrichments in the MPC–32/32F; Appendix B, Section 1, to make modifications to the definitions of fuel debris, damaged fuel assembly, and

non-fuel hardware; and Appendix B, Section 2, to permit the storage of pressurized water reactor fuel assemblies with annular fuel pellets in the top and bottom 12 inches of the active fuel length. Amendment No. 3 also revises CoC No. 1014 to incorporate minor editorial corrections. The direct final rule was to become effective on January 2, 2007. The NRC also concurrently published a companion proposed rule on October 16, 2006 (71 FR 60672).

In the direct final rule, NRC stated that if any significant adverse comments were received, a notice of timely withdrawal of the direct final rule would be published in the **Federal Register** and the direct final rule would not take effect.

The NRC received significant adverse comments on the direct final rule; therefore, the NRC is withdrawing the direct final rule. These significant adverse comments shall be considered as comments to the companion proposed rule that was published concurrently with the direct final rule. The NRC will not initiate a second comment period on the companion proposed rule.

Dated at Rockville, Maryland, this 14th day of December, 2006.

For the Nuclear Regulatory Commission.

Luis A. Reyes,
Executive Director for Operations.
 [FR Doc. E6–22109 Filed 12–26–06; 8:45 am]
BILLING CODE 7590–01–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2006–26675; Directorate Identifier 2006–NM–203–AD; Amendment 39–14864; AD 2006–26–06]

RIN 2120–AA64

Airworthiness Directives; Boeing Model 777–200 and –300 Series Airplanes Equipped with Rolls-Royce Engines

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule; request for comments.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Boeing Model 777–200 and –300 series airplanes equipped with Rolls-Royce engines. This AD requires repetitive inspections to detect cracks of the outer V-blades of the thrust reverser, and

corrective action if necessary. This AD also provides for optional terminating action for the repetitive inspections. This AD results from reports of cracked outer V-blades in the thrust reversers. We are issuing this AD to prevent separation of a thrust reverser from the airplane during normal reverse thrust or during a refused takeoff, which could result in impact damage to other airplane areas. If a thrust reverser separates from the airplane during a refused takeoff, the engine could produce forward thrust, resulting in unexpected thrust asymmetry and a possible runway excursion.

DATES: This AD becomes effective January 11, 2007.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of January 11, 2007.

We must receive comments on this AD by February 26, 2007.

ADDRESSES: Use one of the following addresses to submit comments on this AD.

- *DOT Docket Web site:* Go to <http://dms.dot.gov> and follow the instructions for sending your comments electronically.

- *Government-wide rulemaking Web site:* Go to <http://www.regulations.gov> and follow the instructions for sending your comments electronically.

- *Mail:* Docket Management Facility; U.S. Department of Transportation, 400 Seventh Street SW., Nassif Building, Room PL-401, Washington, DC 20590.

- *Fax:* (202) 493-2251.

- *Hand Delivery:* Room PL-401 on the plaza level of the Nassif Building, 400 Seventh Street, SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207, for service information identified in this AD.

FOR FURTHER INFORMATION CONTACT: Gary Oltman, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 917-6443; fax (425) 917-6590.

SUPPLEMENTARY INFORMATION:

Discussion

We have received reports of cracked outer V-blades in the Rolls-Royce engine thrust reversers on Boeing Model 777-200 and -300 series airplanes. The cracks were found in the top and bottom ends of the V-blade. The outer V-blade engages the aft end of the engine fan case to transmit fore and aft loads from the thrust reversers. Initial analysis of

the V-blade did not include bending loads, and consequently the fatigue margins were not sufficient. A thrust reverser that separates from the airplane during normal reverse thrust or refused takeoff can damage other airplane areas. If a thrust reverser separates from the airplane during a refused takeoff, the engine could produce forward thrust, resulting in unexpected thrust asymmetry and a possible runway excursion.

Relevant Service Information

We have reviewed Boeing Special Attention Service Bulletin 777-78-0064, Revision 1, dated November 30, 2006. The service bulletin describes procedures for doing repetitive detailed inspections to detect cracks in the outer V-blade of the thrust reverser, replacing cracked V-blades with serviceable parts, and sending a report of the inspection results to Boeing. The compliance time for the initial inspection ranges from 250 to 6,000 flight cycles after the effective date of the AD, depending on the number of flight cycles on the V-blade, with repetitive intervals not to exceed 2,000 flight cycles from the last detailed inspection.

We have also reviewed Boeing Special Attention Service Bulletin 777-78-0061, dated July 6, 2006, which describes procedures for doing a special detailed inspection to detect cracks in the outer V-blade of the thrust reverser in addition to a special detailed (eddy current or fluorescent penetrant) inspection to detect cracks in the fay surface area of the lower chord of the torque box where the outer V-blade attaches. If a crack is found in the outer V-blade, a new configuration V-blade is installed. If no crack is found, the V-blade is changed and installed with new support brackets at the top and bottom ends. The service bulletin also specifies contacting Boeing for repair instructions for cracks found in the torque box lower chord.

Accomplishment of the actions specified in Special Attention Service Bulletin 777-78-0061 eliminates the need for the repetitive inspections of Special Attention Service Bulletin 777-78-0064.

FAA's Determination and Requirements of This AD

The unsafe condition described previously is likely to exist or develop on other airplanes of the same type design. For this reason, we are issuing this AD to prevent separation of a thrust reverser from the airplane during normal reverse thrust or during a refused takeoff, which could result in impact damage to other airplane areas.

If a thrust reverser separates from the airplane during a refused takeoff, the engine could produce forward thrust, resulting in unexpected thrust asymmetry and a possible runway excursion. This AD requires accomplishing the actions specified in Boeing Special Attention Service Bulletin 777-78-0064 described previously. This AD also provides for an optional terminating action for the repetitive inspections.

Difference Between the AD and Service Information

Boeing Special Attention Service Bulletin 777-78-0061 specifies to contact the manufacturer for instructions on how to repair certain conditions, but this AD requires repairing those conditions, if accomplished, in one of the following ways:

- Using a method that we approve; or
- Using data that meet the

certification basis of the airplane, and that have been approved by an Authorized Representative for the Boeing Commercial Airplanes Delegation Option Authorization Organization whom we have authorized to approve repair methods.

Interim Action

We consider this AD interim action. We are considering mandating the optional terminating action specified in Boeing Special Attention Service Bulletin 777-78-0061, which terminates the repetitive inspections required by this AD. However, the planned compliance time for this terminating action would allow enough time to provide notice and opportunity for prior public comment on the merits of the actions.

FAA's Determination of the Effective Date

Since an unsafe condition exists that requires the immediate adoption of this AD, we have found that notice and opportunity for public comment before issuing this AD are impracticable, and that good cause exists to make this AD effective in less than 30 days.

Comments Invited

This AD is a final rule that involves requirements that affect flight safety and was not preceded by notice and an opportunity for public comment; however, we invite you to submit any relevant written data, views, or arguments regarding this AD. Send your comments to an address listed in the **ADDRESSES** section. Include "Docket No. FAA-2006-26675; Directorate Identifier 2006-NM-203-AD" at the beginning of

your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the AD that might suggest a need to modify it.

We will post all comments we receive, without change, to <http://dms.dot.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this AD. Using the search function of that Web site, anyone can find and read the comments in any of our dockets, including the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You may review the DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR 19477-78), or you may visit <http://dms.dot.gov>.

Examining the Docket

You may examine the AD docket on the Internet at <http://dms.dot.gov>, or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the DOT street address stated in the **ADDRESSES** section. Comments will be available in the AD docket shortly after the Docket Management System receives them.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in subtitle VII, part A, subpart III, section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on

the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

2006-26-06 Boeing: Amendment 39-14864. Docket No. FAA-2006-26675; Directorate Identifier 2006-NM-203-AD.

Effective Date

- (a) This AD becomes effective January 11, 2007.

Affected ADs

- (b) None.

Applicability

(c) This AD applies to Boeing Model 777-200 and -300 series airplanes, certificated in any category, equipped with Rolls-Royce engines; as identified in Boeing Special Attention Service Bulletin 777-78-0064, Revision 1, dated November 30, 2006.

Unsafe Condition

(d) This AD results from reports of cracked outer V-blades in the thrust reversers. We are issuing this AD to prevent separation of a thrust reverser from the airplane during normal reverse thrust or during a refused

takeoff, which could result in impact damage to other airplane areas. If a thrust reverser separates from the airplane during a refused takeoff, the engine could produce forward thrust, resulting in unexpected thrust asymmetry and a possible runway excursion.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

Repetitive Inspections

(f) Do the detailed inspections to detect cracks in the outer V-blade of the thrust reversers. Do the inspections in accordance with the Accomplishment Instructions of Boeing Special Attention Service Bulletin 777-78-0064, Revision 1, dated November 30, 2006. Do the inspections at the applicable times specified in paragraph 1.E. of the service bulletin; except, where the service bulletin specifies an initial compliance time after the date on the service bulletin, this AD requires compliance within the specified time after the effective date of this AD. Do applicable corrective actions before further flight in accordance with the service bulletin or paragraph (h) of this AD.

(g) Actions done before the effective date of this AD in accordance with Boeing Special Attention Service Bulletin 777-78-0064, dated August 7, 2006, are acceptable for compliance with the requirements of paragraph (f) of this AD.

Report

(h) At the applicable time specified in paragraph (h)(1) or (h)(2) of this AD, send a report of the findings (both positive and negative) of each inspection required by paragraph (f) of this AD to the Manager, Seattle Aircraft Certification Office (ACO), FAA. The report must include the information specified in Appendix A of Boeing Special Attention Service Bulletin 777-78-0064, Revision 1, dated November 30, 2006. Under the provisions of the Paperwork Reduction Act (44 U.S.C. 3501 *et seq.*), the Office of Management and Budget (OMB) has approved the information collection requirements contained in this AD and has assigned OMB Control Number 2120-0056.

(1) For all inspections done after the effective date of this AD: Send the report within 10 days after the inspection.

(2) For any inspection done before the effective date of this AD: Send the report within 10 days after the effective date of this AD.

Optional Terminating Action

(i) Accomplishment of the applicable inspections and related investigative/corrective actions, in accordance with the Accomplishment Instructions of Boeing Special Attention Service Bulletin 777-78-0061, dated July 6, 2006, terminates the requirements of this AD; except, where the service bulletin specifies to contact the manufacturer for appropriate action, repair before further flight using a method approved in accordance with the procedures specified in paragraph (j) of this AD.

Alternative Methods of Compliance (AMOCs)

(j)(1) The Manager, Seattle ACO, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) Before using any AMOC approved in accordance with § 39.19 on any airplane to which the AMOC applies, notify the appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

(3) An AMOC that provides an acceptable level of safety may be used for any repair required by this AD, if it is approved by an Authorized Representative for the Boeing Commercial Airplanes Delegation Option Authorization Organization who has been authorized by the Manager, Seattle ACO, to make those findings. For a repair method to be approved, the repair must meet the certification basis of the airplane.

Material Incorporated by Reference

(k) You must use Boeing Special Attention Service Bulletin 777-78-0064, Revision 1, dated November 30, 2006, to perform the actions that are required by this AD, unless the AD specifies otherwise. If the optional terminating action is accomplished, you must use Boeing Special Attention Service Bulletin 777-78-0061, dated July 6, 2006, to perform the optional terminating actions specified in this AD, unless the AD specifies otherwise. The Director of the Federal Register approved the incorporation by reference of these documents in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207, for a copy of this service information. You may review copies at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., Room PL-401, Nassif Building, Washington, DC; on the Internet at <http://dms.dot.gov>; or at the National Archives and Records Administration (NARA). For information on the availability of this material at the NARA, call (202) 741-6030, or go to http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Renton, Washington, on December 14, 2006.

Stephen P. Boyd,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E6-22040 Filed 12-26-06; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2006-23659; Directorate Identifier 2005-NM-236-AD; Amendment 39-14863; AD 2006-26-05]

RIN 2120-AA64

Airworthiness Directives; Fokker Model F27 Mark 100, 200, 300, 400, 500, 600, and 700 Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for all Fokker Model F27 Mark 100, 200, 300, 400, 500, 600, and 700 airplanes. This AD requires revising the Limitations section of the airplane flight manual regarding the use of continuous ignition, fuel filter heating, and resetting circuit breakers during flight in certain conditions such as icing. This AD results from reports of power loss on one or both engines in icing conditions. We are issuing this AD to advise the flightcrew that continuous ignition will not reduce the probability of power loss, and what action they must take to avoid this hazard. Loss of power in one or more engines during flight, if not prevented, could result in loss of control of the airplane.

DATES: This AD becomes effective January 31, 2007.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of January 31, 2007.

ADDRESSES: You may examine the AD docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC.

Contact Fokker Services B.V., P.O. Box 231, 2150 AE Nieuw-Vennep, the Netherlands, for service information identified in this AD.

FOR FURTHER INFORMATION CONTACT: Tom Rodriguez, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1137; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:**Examining the Docket**

You may examine the airworthiness directive (AD) docket on the Internet at

<http://dms.dot.gov> or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the street address stated in the **ADDRESSES** section.

Discussion

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to all Fokker Model F27 Mark 100, 200, 300, 400, 500, 600, and 700 airplanes. That NPRM was published in the **Federal Register** on January 24, 2006 (71 FR 3792). That NPRM proposed to require revising the Limitations section of the airplane flight manual regarding the use of continuous ignition, fuel filter heating, and resetting circuit breakers during flight in certain conditions such as icing.

Comments

We provided the public the opportunity to participate in the development of this AD. We received no comments on the NPRM or on the determination of the cost to the public.

Clarification of Note 1

Note 1 of the NPRM stated that the revision to the Limitations section of the Fokker F27 Airplane Flight Manual (AFM) could be done by inserting a copy of Manual Change Notification—Operational Documentation (MCNO) MCNO-F27-020, dated June 1, 2004, into the Normal Procedures, Abnormal Procedures, and Emergency Procedures sections of the Fokker F27 AFM. We have clarified Note 1 of this AD to state that the revision can be done by inserting a copy of that MCNO into the Limitations section, as specified in paragraph (f) of this AD.

Conclusion

We have carefully reviewed the available data, and determined that air safety and the public interest require adopting the AD with the change described previously. We have determined that this change will neither increase the economic burden on any operator nor increase the scope of the AD.

Costs of Compliance

This AD affects about 27 airplanes of U.S. registry. The revision takes about 1 work hour per airplane, at an average labor rate of \$65 per work hour. Based on these figures, the estimated cost of the AD for U.S. operators is \$1,755, or \$65 per airplane.