

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Safety.

**The Proposed Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

**Gulfstream Aerospace Corporation:** Docket No. FAA–2006–24951; Directorate Identifier 2005–NM–184–AD.

**Comments Due Date**

(a) The FAA must receive comments on this AD action by July 21, 2006.

**Affected ADs**

(b) None.

**Applicability**

(c) This AD applies to the following Gulfstream Aerospace Corporation airplanes, certificated in any category:

TABLE 1.—APPLICABILITY

| Model                   | Serial Nos.                  |
|-------------------------|------------------------------|
| GV series airplanes ..  | 674 through 693 inclusive.   |
| GV–SP series airplanes. | 5001 through 5072 inclusive. |

**Unsafe Condition**

(d) This AD results from a report indicating that the wiring harness outer shield and insulation on the primary conductors may have been inadvertently cut due to an

improper method used to remove the wiring outer jacket. We are issuing this AD to prevent the loss of the hardover prevention system (HOPS) in the roll axis due to a short circuit in the wiring harness of the aileron force link assembly, which could result in reduced controllability of the airplane.

**Compliance**

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

**Repair**

(f) Within 12 months after the effective date of this AD, repair the force link assembly wire harness by doing all actions specified in the Accomplishment Instructions of the applicable service information identified in Table 2 of this AD, except as required by paragraph (g) of this AD.

TABLE 2.—SERVICE INFORMATION

| For model—                   | Use—                                                       |
|------------------------------|------------------------------------------------------------|
| GV–SP series airplanes ..... | Gulfstream G500 Customer Bulletin 14, dated June 23, 2005. |
| GV–SP series airplanes ..... | Gulfstream G550 Customer Bulletin 14, dated June 23, 2005. |
| GV series airplanes .....    | Gulfstream GV Customer Bulletin 135, dated June 23, 2005.  |

**Note 1:** The Gulfstream customer bulletins identified in Table 2 of this AD include Vought Service Bulletin SB-VAIGV/GVSP–27–PG0098, dated May 9, 2005, as an additional source of service information for the repair.

**Exception to Service Bulletin Specifications**

(g) During the inspection of the environmental seal around the installed wires required by paragraph (f) of this AD: If any nick or other damage is found, repair before further flight using a method approved by the Manager, Atlanta Aircraft Certification Office (ACO), FAA. For a repair method to be approved by the Manager, Atlanta ACO, as required by this paragraph, the Manager's approval letter must specifically refer to this AD.

**Alternative Methods of Compliance (AMOCs)**

(h)(1) The Manager, Atlanta ACO, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) Before using any AMOC approved in accordance with § 39.19 on any airplane to which the AMOC applies, notify the appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

Issued in Renton, Washington, on May 26, 2006.

**Jeffrey E. Duven,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. E6–8711 Filed 6–5–06; 8:45 am]

**BILLING CODE 4910–13–P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

**[Docket No. FAA–2006–24952; Directorate Identifier 2006–NM–107–AD]**

**RIN 2120–AA64**

**Airworthiness Directives; Boeing Model 767 Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** The FAA proposes to adopt a new airworthiness directive (AD) for all Boeing Model 767 airplanes. This proposed AD would require repetitive detailed inspections of the wire bundles, power display unit (PDU) wiring, and wire attaching hardware,

supports, and sleeving located in the forward and aft lower lobe cargo compartments and corrective actions as necessary. This proposed AD results from a fire in the forward lower lobe cargo compartment found shortly after airplane arrival. We are proposing this AD to detect and correct damage to wires in the forward and aft lower lobe cargo compartments, which could result in a potential short circuit and consequent fire in the forward and aft lower lobe cargo compartments.

**DATES:** We must receive comments on this proposed AD by July 21, 2006.

**ADDRESSES:** Use one of the following addresses to submit comments on this proposed AD.

- DOT Docket Web site: Go to <http://dms.dot.gov> and follow the instructions for sending your comments electronically.

- Government-wide rulemaking Web site: Go to <http://www.regulations.gov> and follow the instructions for sending your comments electronically.

- Mail: Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL–401, Washington, DC 20590.

- Fax: (202) 493–2251.

- Hand Delivery: Room PL–401 on the plaza level of the Nassif Building, 400 Seventh Street, SW., Washington,

DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207, for the service information identified in this proposed AD.

**FOR FURTHER INFORMATION CONTACT:**

Elias Natsiopoulos, Aerospace Engineer, Systems and Equipment Branch, ANM-130S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 917-6478; fax (425) 917-6590.

**SUPPLEMENTARY INFORMATION:**

**Comments Invited**

We invite you to submit any relevant written data, views, or arguments regarding this proposed AD. Send your comments to an address listed in the **ADDRESSES** section. Include the docket number "FAA-2006-24952; Directorate Identifier 2006-NM-107-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the proposed AD. We will consider all comments received by the closing date and may amend the proposed AD in light of those comments.

We will post all comments we receive, without change, to <http://dms.dot.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed AD. Using the search function of that web site, anyone can find and read the comments in any of our dockets, including the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You may review DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR 19477-78), or you may visit <http://dms.dot.gov>.

**Examining the Docket**

You may examine the AD docket on the Internet at <http://dms.dot.gov>, or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the DOT street address stated in the **ADDRESSES** section. Comments will be available in the AD docket shortly after the Docket Management System receives them.

**Discussion**

We have received a report of a fire in the forward lower lobe cargo compartment of a Boeing Model 767-300 series airplane, found 20 minutes after arrival. The flightcrew and passengers had deplaned before the start of the fire, and the cleaning crew and all other personnel were evacuated without incident. Investigation revealed that flammable debris had accumulated below the 13L and 14L power drive units (PDUs), but the source of ignition was not positively identified. During subsequent inspections of the operator's fleet of Model 767 airplanes, crushed and chafed PDU power supply cables were found. This condition, if not corrected, could result in a potential short circuit and consequent fire in the forward and aft lower lobe cargo compartments.

**Relevant Service Information**

We have reviewed Boeing Service Bulletin 767-25-0376 (for Model 767-200, -300, and -300F series airplanes) and Boeing Service Bulletin 767-25-0377 (for Model 767-400ER series airplanes), both dated November 17, 2005. The service bulletins describe procedures for doing repetitive detailed inspections for damage to the wire bundles, PDU wiring, and wire attaching hardware, supports, and sleeving located in the forward and aft lower lobe cargo compartments; and doing corrective actions as necessary. The corrective actions include repairing any damage to the wire bundles, PDU wiring, and wire attaching hardware, supports, and sleeving. Accomplishing the actions specified in the service information is intended to adequately address the unsafe condition.

**FAA's Determination and Requirements of the Proposed AD**

We have evaluated all pertinent information and identified an unsafe condition that is likely to exist or develop on other airplanes of this same type design. For this reason, we are proposing this AD, which would require accomplishing the actions specified in the service information described previously.

**Clarification of Inspection Terminology**

The "detailed visual inspection" specified in the service bulletins is referred to as a "detailed inspection" in this proposed AD. Boeing has included the definition for a detailed inspection in Note 4 of the service bulletins.

**Costs of Compliance**

There are about 857 airplanes of the affected design in the worldwide fleet.

This proposed AD would affect about 374 airplanes of U.S. registry. The proposed inspections would take about 6 work hours per airplane, at an average labor rate of \$80 per work hour. Based on these figures, the estimated cost of the proposed AD for U.S. operators is \$179,520, or \$480 per airplane, per inspection cycle.

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

We have determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Safety.

## The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

**Boeing:** Docket No. FAA–2006–24952; Directorate Identifier 2006–NM–107–AD.

#### Comments Due Date

(a) The FAA must receive comments on this AD action by July 21, 2006.

#### Affected ADs

(b) None.

#### Applicability

(c) This AD applies to all Model 767–200, –300, –300F, and –400ER series airplanes, certificated in any category.

#### Unsafe Condition

(d) This AD results from a fire in the forward lower lobe cargo compartment found shortly after airplane arrival. We are issuing this AD to detect and correct damage to wires in the forward and aft lower lobe cargo compartments, which could result in a potential short circuit and consequent fire in the forward and aft lower lobe cargo compartments.

#### Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

#### Repetitive Inspections and Corrective Actions if Applicable

(f) Within 36 months after the effective date of this AD, do detailed inspections for damage to the wire bundles, power drive unit wiring, and wire attaching hardware, supports, and sleeving located in the forward and aft lower lobe cargo compartments; and do all applicable corrective actions before further flight after the inspections; by accomplishing all of the actions specified in the Accomplishment Instructions of Boeing Service Bulletin 767–25–0376 (for Model 767–200, –300, and –300F series airplanes) or Boeing Service Bulletin 767–25–0377 (for Model 767–400ER series airplanes), both dated November 17, 2005, as applicable. Repeat the inspections thereafter at intervals not to exceed 24,000 flight hours or 72 months, whichever occurs first.

#### Alternative Methods of Compliance (AMOCs)

(g)(1) The Manager, Seattle Aircraft Certification Office (ACO), FAA, has the

authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) Before using any AMOC approved in accordance with § 39.19 on any airplane to which the AMOC applies, notify the appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

Issued in Renton, Washington, on May 26, 2006.

**Jeffrey E. Duven,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*  
[FR Doc. E6–8708 Filed 6–5–06; 8:45 am]

**BILLING CODE 4910–13–P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. 2001–NM–110–AD]

**RIN 2120–AA64**

#### Airworthiness Directives; Saab Model SAAB SF340A and SAAB 340B Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Proposed rule; withdrawal.

**SUMMARY:** This action withdraws a notice of proposed rulemaking (NPRM) that proposed a new airworthiness directive (AD), applicable to all Saab Model SAAB SF340A and SAAB 340B series airplanes. That action would have required modifying or replacing flight data recorders (FDR) of a certain model. Since the issuance of the NPRM, the Federal Aviation Administration (FAA) has received new data that the identified unsafe condition has been corrected on all airplanes that would have been subject to the NPRM. Accordingly, the proposed rule is withdrawn.

**FOR FURTHER INFORMATION CONTACT:** Dan Rodina, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98055–4056; telephone (425) 227–2125; fax (425) 227–1149.

**SUPPLEMENTARY INFORMATION:** A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to add a new airworthiness directive (AD), applicable to all Saab Model SAAB SF340A and SAAB 340B series airplanes, was published in the **Federal Register** as a Notice of Proposed Rulemaking (NPRM) on August 17, 2001 (66 FR 43128). The proposed rule would have required modifying or replacing

flight data recorders (FDR) of a certain model. That action resulted from reports of a number of incidents in which flight data have been lost from the FDR.

The proposed actions were intended to prevent loss of flight data from the FDR, which could hamper discovery of the cause of an accident, preventing the Federal Aviation Administration from developing and mandating actions to prevent additional accidents caused by the same unsafe condition.

#### Actions That Occurred Since the NPRM Was Issued

Since the issuance of that NPRM, the FDR manufacturer, L–3 Communications Corporation, notified the FAA that the only affected operator flying in the U.S. has accomplished the action proposed in the NPRM, and no unsafe condition exists or is likely to exist. In the event that a Saab Model SAAB–Fairchild SF340A (SAAB/SF340A), or SAAB 340B airplane is imported into the U.S., the FAA issued a Special Airworthiness Information Bulletin (NM–06–40, April 14, 2006) to alert U.S. operators of the need to install the replacement FDR.

#### Explanation of Change to Model Designation

We have revised the model reference in the above paragraph to identify model designations as published in the most recent type certificate data sheet for the affected models.

#### FAA's Conclusions

Upon further consideration, the FAA has determined that the actions that would have been required by the NPRM have already been done on all affected airplanes, and the identified unsafe condition has been corrected. Accordingly, the proposed rule is hereby withdrawn.

Withdrawal of this NPRM constitutes only such action, and does not preclude the agency from issuing another action in the future, nor does it commit the agency to any course of action in the future.

#### Regulatory Impact

Since this action only withdraws a notice of proposed rulemaking, it is neither a proposed nor a final rule and therefore is not covered under Executive Order 12866, the Regulatory Flexibility Act, or DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979).

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.