

electromechanical gust lock system to become unscrewed. We are proposing this AD to prevent unscrewing of the spring cartridge clevis from jamming the elevator, which could lead to reduced controllability of the airplane.

Compliance: (e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

Restatement of Requirements of AD 2003–09–03

Inspection

(f) For Model EMB–135BJ airplanes: Within 30 days after May 14, 2003 (the effective date of AD 2003–09–03), perform a general visual inspection of each spring cartridge of the elevator gust lock system to determine if the lock washer projection correctly fits the slots in the cartridge flange, in accordance with EMBRAER Service Bulletin 145LEG–27–0006, dated December 9, 2002. Before further flight, replace any discrepant spring cartridge with a new part having the same part number, in accordance with the service bulletin; or replace the spring cartridge, part number (P/N) KDP2611, with a new, improved spring cartridge, P/N KDP4235, as specified in paragraph (h) of this AD. After the effective date of this AD, only the replacement specified in paragraph (h) may be accomplished. Repeat the inspection at intervals not to exceed 800 flight hours until the replacement of the spring cartridge is accomplished as required by paragraph (h). Although the service bulletin recommends that operators report inspection results to EMBRAER, this AD does not require such a report.

Note 1: For the purposes of this AD, a general visual inspection is: “A visual examination of an interior or exterior area, installation, or assembly to detect obvious damage, failure, or irregularity. This level of inspection is made from within touching distance unless otherwise specified. A mirror may be necessary to ensure visual access to all surfaces in the inspection area. This level of inspection is made under normally available lighting conditions such as daylight, hangar lighting, flashlight, or droplight and may require removal or opening of access panels or doors. Stands, ladders, or platforms may be required to gain proximity to the area being checked.”

(g) For airplanes not identified in paragraph (f) of this AD: At the applicable time specified in paragraph (g)(1) or (g)(2) of this AD, perform a general visual inspection of each spring cartridge of the elevator gust lock system to determine if the lock washer projection correctly fits the slots in the cartridge flange, in accordance with EMBRAER Service Bulletin 145–27–0098, dated December 9, 2002. Repeat the inspection at intervals not to exceed 800 flight hours after the initial inspection until the replacement of the spring cartridge, P/N KDP2611, with a new, improved spring cartridge, P/N KDP4235, is done as specified in paragraph (h) of this AD. Although the service bulletin recommends that operators report inspection results to EMBRAER, this AD does not require such a report.

(1) For airplanes equipped with an operational electromechanical gust lock system on the elevator: Inspect within 30 days after May 14, 2003, in accordance with PART I of the service bulletin. Before further flight, replace any discrepant spring cartridge with a new part having the same part number, in accordance with PART I of the service bulletin; or do the replacement specified in paragraph (h) of this AD. After the effective date of this AD, only the replacement specified in paragraph (h) may be accomplished.

(2) For airplanes that are not equipped with an operational electromechanical gust lock system on the elevator, but that are equipped with provisions for the system: Inspect within 60 days after May 14, 2003, in accordance with PART II of the service bulletin. Before further flight, replace any discrepant spring cartridge with a new part having the same part number, in accordance with PART II of the service bulletin; or do the replacement specified in paragraph (h) of this AD. After the effective date of this AD, only the replacement specified in paragraph (h) may be accomplished. Alternatively, removal of the spring cartridges terminates the repetitive inspection requirement of this AD during the time the cartridges are removed.

New Requirements of This AD

Replacement of Spring Cartridge

(h) Within 5,500 flight hours or 36 months after the effective date of this AD, whichever comes first, replace the spring cartridge, P/N KPD2611, with a new, improved spring cartridge, P/N KDP4235, in accordance with the Accomplishment Instructions of EMBRAER Service Bulletin 145LEG–27–0012, Revision 01, dated April 12, 2004 (for Model EMB–135BJ airplanes); or Service Bulletin 145–27–0102, Revision 02, dated January 20, 2005 (for Model EMB–135ER, –135KE, –135KL, –135LR, –145, –145ER, –145MR, –145LR, –145XR, –145MP, and –145EP airplanes); as applicable. Accomplishing this replacement terminates the repetitive inspections required by paragraphs (f) and (g) of this AD.

Parts Installation

(i) As of the effective date of this AD, no person may install a spring cartridge, P/N KPD2611, on any airplane.

Cartridge Replacement According to Previous Issue of Service Bulletin

(j) Spring cartridge replacements accomplished before the effective date of this AD in accordance with EMBRAER Service Bulletin 145LEG–27–0012, dated March 2, 2004; or Service Bulletin 145–27–0102, dated December 23, 2003, or Revision 01, dated April 12, 2004; are considered acceptable for compliance with the corresponding action required by this AD.

Alternative Methods of Compliance (AMOCs)

(k)(1) The Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) AMOCs approved previously according to AD 2003–09–03, amendment 39–13132, are approved as AMOCs for the corresponding provisions of this AD.

Related Information

(l) Brazilian airworthiness directive 2003–01–03R1, dated July 26, 2004, also addresses the subject of this AD.

Issued in Renton, Washington, on August 2, 2005.

Kevin Mullin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.
[FR Doc. 05–15592 Filed 8–5–05; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2005–22035; Directorate Identifier 2005–NM–016–AD]

RIN 2120–AA64

Airworthiness Directives; Airbus Model A300 B2 and B4 Series Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to adopt a new airworthiness directive (AD) for all Airbus Model A300 B2 and B4 series airplanes. This proposed AD would require an inspection to determine the part number of all angle of attack (AOA) sensors, and repetitive replacement of the AOA sensors with new or overhauled AOA sensors if necessary. This proposed AD would also provide an optional terminating action for the repetitive replacements. This proposed AD is prompted by reports of several false stall warnings associated with stick-shaker activation, occurring during take-off. We are proposing this AD to prevent false stall warnings associated with stick-shaker activation, which could result in increased pilot workload as the pilot tries to determine the cause of the stall warning and possible reduction in the pilot's ability to control the airplane.

DATES: We must receive comments on this proposed AD by September 7, 2005.

ADDRESSES: Use one of the following addresses to submit comments on this proposed AD.

- **DOT Docket Web site:** Go to <http://dms.dot.gov> and follow the instructions for sending your comments electronically.

• *Government-wide Rulemaking Web site:* Go to <http://www.regulations.gov> and follow the instructions for sending your comments electronically.

• *Mail:* Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., Nassif Building, Room PL-401, Washington, DC 20590.

• *Fax:* (202) 493-2251.

• *Hand Delivery:* Room PL-401 on the plaza level of the Nassif Building, 400 Seventh Street SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France, for service information identified in this proposed AD.

FOR FURTHER INFORMATION CONTACT: Tim Backman, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2797; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to submit any relevant written data, views, or arguments regarding this proposed AD. Include the docket number "FAA-2005-22035; Directorate Identifier 2005-NM-016-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the proposed AD. We will consider all comments received by the closing date and may amend the proposed AD in light of those comments.

We will post all comments we receive, without change, to <http://dms.dot.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed AD. Using the search function of that Web site, anyone can find and read the comments in any of our dockets, including the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You may review the DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR

19477-78), or you may visit <http://dms.dot.gov>.

Examining the Docket

You may examine the AD docket on the Internet at <http://dms.dot.gov>, or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the DOT street address stated in the **ADDRESSES** section. Comments will be available in the AD docket shortly after the Docket Management System receives them.

Discussion

The Direction Générale de l'Aviation Civile (DGAC), which is the airworthiness authority for France, notified us that an unsafe condition may exist on certain Airbus Model A300 B2 and B4 series airplanes. The DGAC advises that it has received reports of several false stall warnings associated with stick-shaker activation, occurring during take-off. Investigation revealed that defective angle of attack (AOA) sensors caused the false stall warnings. This condition, if not corrected, could result in increased pilot workload as the pilot tries to determine the cause of the stall warning and possible reduction in the pilot's ability to control the airplane.

Relevant Service Information

Airbus has issued Service Bulletin A300-34-0176, Revision 01, dated February 3, 2004, which describes the following procedures:

- Inspecting zone 120 to determine the part number (P/N) of all three AOA sensors.

- Repetitively replacing any Honeywell AOA sensor having P/N 965-4020-007 with a new or overhauled AOA sensor.

Accomplishing the actions specified in the service information is intended to adequately address the unsafe condition. The DGAC mandated the service information and issued French airworthiness directive F-2003-457 R1, dated December 22, 2004, to ensure the continued airworthiness of these airplanes in France.

Airbus has also issued Service Bulletin A300-34-0092, Revision 03,

dated November 2, 2004. Service Bulletin A300-34-0092 describes the following procedures:

- Replacing Honeywell "pencil" AOA sensors having P/N 965-4020-007 with "vane" AOA sensors between frame (FR)18 and FR19.

- Replacing the current detectors in relay boxes 252VU and 107VU with new current detectors.

Airbus Service Bulletin A300-34-0092 also specifies that accomplishing the modification in that service bulletin cancels the actions specified in Airbus Service Bulletin A300-34-0176.

FAA's Determination and Requirements of the Proposed AD

These airplane models are manufactured in France and are type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the DGAC has kept the FAA informed of the situation described above. We have examined the DGAC's findings, evaluated all pertinent information, and determined that we need to issue an AD for products of this type design that are certificated for operation in the United States.

Therefore, we are proposing this AD, which would require accomplishing the actions specified in the service information described previously, except as discussed under "Difference Between the Proposed AD and Service Bulletin."

Difference Between the Proposed AD and Service Bulletin

Operators should note that, although the Accomplishment Instructions of the Airbus Service Bulletin A300-34-0176, Revision 01, dated February 3, 2004, describe procedures for reporting inspection findings, this proposed AD would not require that action. We do not need this information from operators.

Costs of Compliance

The following table provides the estimated costs for U.S. operators to comply with this proposed AD.

ESTIMATED COSTS

Action	Work hours	Average labor rate per hour	Parts	Cost per airplane	Number of U.S.-registered airplanes	Fleet cost
Inspection	1	\$65	None	\$65	20	\$1,300.

ESTIMATED COSTS—Continued

Action	Work hours	Average labor rate per hour	Parts	Cost per airplane	Number of U.S.-registered airplanes	Fleet cost
Replacement if necessary, per replacement cycle.	2	65	\$3,300 (\$1,100 per sensor)	3,430	20	\$68,600 per replacement cycle.
Optional terminating action	7	65	\$8,780	9,235	20	\$184,700.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in subtitle VII, part A, subpart III, section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We have determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

Airbus: Docket No. FAA-2005-22035; Directorate Identifier 2005-NM-016-AD.

Comments Due Date

(a) The Federal Aviation Administration must receive comments on this AD action by September 7, 2005.

Affected ADs

(b) None.

Applicability: (c) This AD applies to all Airbus Model A300 B2-1A, B2-1C, B2K-3C, and B2-203 airplanes; and Model A300 B4-2C, B4-103, and B4-203 airplanes; certificated in any category.

Unsafe Condition

(d) This AD was prompted by reports of several false stall warnings associated with stick-shaker activation, occurring during take-off. We are issuing this AD to prevent false stall warnings associated with stick-shaker activation, which could result in increased pilot workload as the pilot tries to determine the cause of the stall warning and possible reduction in the pilot's ability to control the airplane.

Compliance: (e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

Inspection and Repetitive Replacement, if Necessary

(f) Within 4,500 flight hours or 36 months after the effective date of this AD, whichever is first: Inspect zone 120 to determine the part number of all three angle of attack (AOA) sensors, in accordance with the Accomplishment Instructions of Airbus Service Bulletin A300-34-0176, Revision 01, dated February 3, 2004. If no Honeywell AOA sensor having part number (P/N) 965-4020-007 is found, then no further action is

required by this paragraph. If any Honeywell AOA sensor having P/N 965-4020-007 is found, before further flight, replace the AOA sensor with a new or overhauled AOA sensor having P/N 965-4020-007, in accordance with the service bulletin. Repeat the replacement thereafter at intervals not to exceed 8,000 flight hours or 96 months, whichever is first. Accomplishing the actions specified in paragraph (g) of this AD terminates the repetitive replacements.

Optional Terminating Action

(g) Replacement of all Honeywell AOA sensors having P/N 965-4020-007 between frame (FR)18 and FR19 with "vane type" AOA sensors; and replacement of the current detectors in relay boxes 252VU and 107VU with new current detectors; in accordance with the Accomplishment Instructions of Airbus Service Bulletin A300-34-0092, Revision 03, dated November 2, 2004; terminate the repetitive replacements required by paragraph (f) of this AD.

No Reporting Requirement

(h) Although Airbus Service Bulletin A300-34-0176, Revision 01, dated February 3, 2004, specifies to submit certain information to the manufacturer, this AD does not include that requirement.

Parts Installation

(i) As of the effective date of this AD, no person may install an AOA sensor having P/N 965-4020-007 on any airplane, unless it is new or overhauled and is repetitively inspected as required by paragraph (f) of this AD.

Credit for Previously Accomplished Actions

(j) Actions done before the effective date of this AD in accordance with Airbus Service Bulletin A300-34-0176, dated July 9, 2003, are acceptable for compliance with the corresponding requirements of paragraph (f) of this AD.

Credit for Optional Terminating Action

(k) Actions done before the effective date of this AD in accordance with Airbus Service Bulletin A300-34-092, Revision 2, dated July 18, 1985, are acceptable for compliance with the requirements of paragraph (g) of this AD.

Alternative Methods of Compliance (AMOCs)

(l) The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

Related Information

(m) French airworthiness directive F-2003-457 R1, dated December 22, 2004, also addresses the subject of this AD.

Issued in Renton, Washington, on August 2, 2005.

Kevin Mullin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.
[FR Doc. 05-15593 Filed 8-5-05; 8:45 am]

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DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2005-22036; Directorate Identifier 2005-NM-009-AD]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A300 B2 and B4 Series Airplanes; Model A300 B4-600, B4-600R, and F4-600R Series Airplanes, and Model C4-605R Variant F Airplanes (Collectively Called A300-600 Series Airplanes); and Model A310 Series Airplanes; Equipped With General Electric CF6-80A3 or CF6-80C2 Engines

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to adopt a new airworthiness directive (AD) for certain Airbus airplane models, as specified above. This proposed AD would require installing electro-pneumatic locking bar devices (TRAS lock systems) in the engine nacelles, installing a dedicated and shielded electrical circuit that is segregated from the existing thrust reverser control system, and performing related investigative/corrective actions if necessary. This proposed AD is prompted by the manufacturer's reassessment of the thrust reverser systems in the Airbus airplane models specified above, which showed that the thrust reverser could inadvertently deploy in flight under certain conditions. We are proposing this AD to prevent inadvertent deployment of thrust reversers in flight, which could result in reduced controllability of the airplane.

DATES: We must receive comments on this proposed AD by September 7, 2005.

ADDRESSES: Use one of the following addresses to submit comments on this proposed AD.

- *DOT Docket Web Site:* Go to <http://dms.dot.gov> and follow the instructions for sending your comments electronically.

- *Government-Wide Rulemaking Web Site:* Go to <http://www.regulations.gov> and follow the instructions for sending your comments electronically.

- *Mail:* Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., Nassif Building, room PL-401, Washington, DC 20590.

- *By Fax:* (202) 493-2251.

- *Hand Delivery:* Room PL-401 on the plaza level of the Nassif Building, 400 Seventh Street SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France.

You can examine the contents of this AD docket on the Internet at <http://dms.dot.gov>, or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., room PL-401, on the plaza level of the Nassif Building, Washington, DC. This docket number is FAA-2005-22036; the directorate identifier for this docket is 2005-NM-009-AD.

FOR FURTHER INFORMATION CONTACT: Tim Backman, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2797; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:**Comments Invited**

We invite you to submit any relevant written data, views, or arguments regarding this proposed AD. Send your comments to an address listed under **ADDRESSES**. Include "Docket No. FAA-2005-22036; Directorate Identifier 2005-NM-009-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the proposed AD. We will consider all comments submitted by the closing date and may amend the proposed AD in light of those comments.

We will post all comments we receive, without change, to <http://dms.dot.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed AD. Using the search function of our docket Web site, anyone can find and read the comments in any of our dockets, including the name of the individual

who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You can review the DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR 19477-78), or you can visit <http://dms.dot.gov>.

Examining the Docket

You can examine the AD docket on the Internet at <http://dms.dot.gov>, or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the DOT street address stated in the **ADDRESSES** section. Comments will be available in the AD docket shortly after the DMS receives them.

Discussion

The Direction Générale de l'Aviation Civile (DGAC), which is the airworthiness authority for France, notified us that an unsafe condition may exist on certain Airbus Model A300 B2 and B4 series airplanes; Model A300 B4-600, B4-600R, and F4-600R series airplanes, and Model C4-605R Variant F airplanes (collectively called A300-600 series airplanes); and Model A310 series airplanes; equipped with General Electric CF6-80A3 or CF6-80C2 engines. The DGAC advises that the manufacturer has reassessed the thrust reverser systems of those airplanes and has determined that the thrust reverser could inadvertently deploy in flight. The manufacturer has developed a new, improved design of the thrust reversers, which provides an independent system to prevent deployment of the thrust reversers in flight. Inadvertent deployment of the thrust reversers in flight could result in reduced controllability of the airplane.

Relevant Service Information

Airbus has issued the following service bulletins (SBs), which describe procedures for installing electro-pneumatic locking bar devices (TRAS lock system) and a dedicated and shielded electrical circuit that is segregated from the existing thrust reverser control system. The new electrical circuit commands the locking bar devices (TRAS locks), which would be installed in the engine nacelles. Implementation of the following SBs is intended to provide an independent system to prevent inadvertent deployment of thrust reversers during flight.