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## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2004-19534; Directorate Identifier 2004-NM-99-AD; Amendment 39-14198; AD 2005-15-09]

RIN 2120-AA64

#### Airworthiness Directives; Airbus Model A300 B2 and B4 Series Airplanes; Model A300 B4-600, B4-600R, and F4-600R Series Airplanes, and Model A300 C4-605R Variant F Airplanes (Collectively Called A300-600 Series Airplanes); and Model A310-200 and -300 Series Airplanes

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain Airbus airplane models, as specified above. This AD requires modifying the thermal insulation system of certain fuselage frames, and modifying the fuselage drainage system. This AD also requires revising the FAA-approved maintenance inspection program to include inspections for corrosion or cracking in the subject areas. This AD is prompted by reports of corrosion in the lower part of the pressure bulkhead at certain fuselage frames. We are issuing this AD to prevent accumulation of

condensation in the insulation blankets of certain fuselage frames, which could cause corrosion that could result in reduced structural integrity of the fuselage and consequent rapid decompression of the airplane.

**DATES:** This AD becomes effective August 30, 2005.

The incorporation by reference of certain publications listed in the AD is approved by the Director of the Federal Register as of August 30, 2005.

**ADDRESSES:** For service information identified in this AD, contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France.

**Docket:** The AD docket contains the proposed AD, comments, and any final disposition. You can examine the AD docket on the Internet at <http://dms.dot.gov>, or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the U.S. Department of Transportation, 400 Seventh Street SW., room PL-401, Washington, DC. This docket number is FAA-2004-19534; the directorate identifier for this docket is 2004-NM-99-AD.

**FOR FURTHER INFORMATION CONTACT:** Dan Rodina, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2125; fax (425) 227-1149.

**SUPPLEMENTARY INFORMATION:** The FAA proposed to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) with an AD for certain Airbus Model A300 B2 and B4 series airplanes; Model A300 B4-600, B4-600R, and F4-600R series airplanes, and Model A300 C4-605R Variant F airplanes (collectively called A300-600 series airplanes); and Model A310-200 and

-300 series airplanes. That proposed AD was published as a supplemental notice of proposed rulemaking (NPRM) in the **Federal Register** on May 18, 2005 (70 FR 28491). The supplemental NPRM proposed to require modifying the thermal insulation system of certain fuselage frames, and modifying the fuselage drainage system. The supplemental NPRM also proposed to require revising the FAA-approved maintenance inspection program to include inspections for corrosion or cracking in the subject areas.

#### Comments

We provided the public the opportunity to participate in the development of this AD. No comments have been submitted on the supplemental NPRM or on the determination of the cost to the public. Comments submitted on the original NPRM were addressed in the supplemental NPRM.

#### Explanation of Change to Applicability

We have revised the applicability of this AD to identify model designations as published in the most recent type certificate data sheet for the affected models.

#### Conclusion

We have carefully reviewed the available data, and determined that air safety and the public interest require adopting the AD with the change described previously. We have determined that this change will neither increase the economic burden on any operator nor increase the scope of the AD.

#### Costs of Compliance

The following table provides the estimated costs for U.S. operators to comply with this AD, at an average labor rate of \$65 per work hour.

Action	Models	Work hours	Parts	Cost per airplane	Number of U.S.-registered airplanes	Fleet cost
Modifying the Thermal Insulation System .....	A300 B2/B4 series .....	5	\$567	\$892	23	\$20,516
Modifying the Thermal Insulation System .....	A300-600 series .....	4	567	827	116	95,932
Modifying the Thermal Insulation System .....	A310-200 and -300 series .....	4	567	827	47	38,869
Modifying the Fuselage Drainage System .....	A300 B2/B4 series .....	38	1,857	4,327	23	99,521
Modifying the Fuselage Drainage System .....	A300-600 series .....	36	1,378	3,718	116	431,288
Modifying the Fuselage Drainage System .....	A310-200 and -300 series .....	27	1,451	3,206	47	150,682

Action	Models	Work hours	Parts	Cost per airplane	Number of U.S.-registered airplanes	Fleet cost
Maintenance Program Revision .....	All .....	.....	(1)	65	186	12,090

<sup>1</sup> None.

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, “General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

(1) Is not a “significant regulatory action” under Executive Order 12866;

(2) Is not a “significant rule” under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and

(3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**Adoption of the Amendment**

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

**2005-15-09 Airbus:** Amendment 39-14198. Docket No. FAA-2004-19534; Directorate Identifier 2004-NM-99-AD.

**Effective Date**

(a) This airworthiness directive (AD) becomes effective August 30, 2005.

**Affected ADs**

(b) None.

**Applicability**

(c) This AD applies to Airbus Model A300 B2-1A, B2-1C, B2K-3C, B4-2C, B4-103, and B4-203 airplanes; A300 B4-601, B4-603, B4-620, B4-622, B4-605R, B4-622R, F4-605R, F4-622R, and C4-605R Variant F airplanes; and A310-203, -204, -221, -222, -304, -322, -324, and -325 airplanes; certificated in any category; except those on which both Airbus Modifications 5946 and 8057 were done during production.

**Unsafe Condition**

(d) This AD was prompted by reports of corrosion in the lower part of the pressure bulkhead at fuselage frames (FR) 39 and 54. We are issuing this AD to prevent accumulation of condensation in the insulation blankets of certain fuselage FRs, which could cause corrosion that could result in reduced structural integrity of the fuselage and consequent rapid decompression of the airplane.

**Compliance**

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

**Modification of Thermal Insulation and Fuselage Drainage Systems**

(f) Within 22 months after the effective date of this AD, modify the thermal insulation system of applicable fuselage frames and modify the fuselage drainage system, by doing all actions in the Accomplishment Instructions of the applicable service bulletins specified in Table 1 of this AD.

TABLE 1.—RELEVANT SERVICE BULLETINS

For Airbus models—	Modify the thermal insulation system according to Airbus Service Bulletin—	And modify the fuselage drainage system according to Airbus Service Bulletin—
A300 B2-1A, B2-1C, B2K-3C, B4-2C, B4-103, and B4-203.	A300-21-0116, Revision 03, dated January 29, 2004.	A300-53-0201, Revision 05, dated July 15, 2004.
A300 B4-601, B4-603, B4-620, B4-622, B4-605R, B4-622R, F4-605R, F4-622R, and C4-605R Variant F.	A300-21-6025, Revision 02, dated January 29, 2004.	A300-53-6008, Revision 05, dated July 15, 2004.
A310-203, -204, -221, -222, -304, -322, -324, and -325.	A310-21-2041, Revision 03, dated January 29, 2004.	A300-53-2027, Revision 04, dated July 15, 2004.

**Modifications Accomplished According to Previous Issues of Service Bulletins**

(g) Modifications accomplished before the effective date of this AD according to the

service bulletins listed in Table 2 are considered acceptable for compliance with the corresponding action specified in paragraph (f) of this AD.

TABLE 2.—ADDITIONAL SERVICE BULLETINS

Models	Airbus Service Bulletin	Revision level	Date
A300 B2-1A, B2-1C, B2K-3C, B4-2C, B4-103, and B4-203 .....	A300-21-116 .....	1	March 24, 1992.
A300 B2-1A, B2-1C, B2K-3C, B4-2C, B4-103, and B4-203 .....	A300-21-0116 .....	02	June 13, 2003.
A300 B2-1A, B2-1C, B2K-3C, B4-2C, B4-103, and B4-203 .....	A300-53-0201 .....	04	May 2, 2003.
A300 B4-601, B4-603, B4-620, B4-622, B4-605R, F4-605R, F4-622R, B4-622R, and C4-605R Variant F.	A300-21-6025 .....	01	June 13, 2003.
A300 B4-601, B4-603, B4-620, B4-622, B4-605R, F4-605R, F4-622R, B4-622R, and C4-605R Variant F.	A300-53-6008 .....	02	August 10, 1989.
A300 B4-601, B4-603, B4-620, B4-622, B4-605R, F4-605R, F4-622R, B4-622R, and C4-605R Variant F.	A300-53-6008 .....	03	November 6, 1990.
A300 B4-601, B4-603, B4-620, B4-622, B4-605R, F4-605R, F4-622R, B4-622R, and C4-605R Variant F.	A300-53-6008 .....	04	April 28, 2003.
A310-203, -204, -221, -222, -304, -322, -324, and -325 .....	A310-21-2041 .....	1	December 10, 1990.
A310-203, -204, -221, -222, -304, -322, -324, and -325 .....	A310-21-2041 .....	02	June 13, 2003.
A310-203, -204, -221, -222, -304, -322, -324, and -325 .....	A310-53-2027 .....	2	November 6, 1990.
A310-203, -204, -221, -222, -304, -322, -324, and -325 .....	A310-53-2027 .....	03	May 2, 2003.

**Maintenance Program Revision**

(h) Within 90 days after doing the actions required by paragraph (f) of this AD, or within 90 days after the effective date of this AD, whichever is later: Incorporate into the FAA-approved maintenance inspection program repetitive detailed inspections for corrosion or cracking of fuselage structure from FR 38.2 to 39, and at FR 54, as applicable, as described in Airbus Maintenance Planning Document Task Numbers 538295-0603-1 (for Airbus Model A300 B2-1A, B2-1C, B2K-3C, B4-2C, B4-103, and B4-203 airplanes), and 531531-01-1 and 531533-01-1 (for Airbus Model A300 B4-601, B4-603, B4-620, B4-622, B4-605R, F4-605R, F4-622R, B4-622R, and C4-605R Variant F airplanes; and A310-203, -204, -221, -222, -304, -322, -324, and -325 airplanes). Then, thereafter, comply with the applicable requirements.

**Note 1:** For the purposes of this AD, a detailed inspection is: "An intensive

examination of a specific item, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at an intensity deemed appropriate. Inspection aids such as mirror, magnifying lenses, etc., may be necessary. Surface cleaning and elaborate procedures may be required."

**Alternative Methods of Compliance (AMOCs)**

(i) The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

**Related Information**

(j) French airworthiness directive 2003-317(B), dated August 20, 2003, also addresses the subject of this AD.

**Material Incorporated by Reference**

(k) You must use the service information that is specified in Table 3 of this AD to perform the actions that are required by this AD, unless the AD specifies otherwise. The Director of the Federal Register approves the incorporation by reference of these documents in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. To get copies of the service information, contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. To view the AD docket, go to the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW, room PL-401, Nassif Building, Washington, DC. To review copies of the service information, go to the National Archives and Records Administration (NARA). For information on the availability of this material at the NARA, call (202) 741-6030, or go to [http://www.archives.gov/federal\\_register/code\\_of\\_federal\\_regulations/ibr\\_locations.html](http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html).

TABLE 3.—MATERIAL INCORPORATED BY REFERENCE

Airbus Service Bulletin	Revision level	Date
A300-21-0116 .....	Revision 03 .....	January 29, 2004.
A300-21-6025 .....	Revision 02 .....	January 29, 2004.
A300-53-0201 .....	Revision 05 .....	July 15, 2004.
A300-53-6008 .....	Revision 05 .....	July 15, 2004.
A310-21-2041 .....	Revision 03 .....	January 29, 2004.
A310-53-2027 .....	Revision 04 .....	July 15, 2004.

Issued in Renton, Washington, on July 13, 2005.

Ali Bahrami,

Manager, Transport Airplane Directorate,  
Aircraft Certification Service.

[FR Doc. 05-14397 Filed 7-25-05; 8:45 am]

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## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2004-19175; Directorate Identifier 2003-NM-246-AD; Amendment 39-14197; AD 2005-15-08]

RIN 2120-AA64

#### Airworthiness Directives; Boeing Model 747-100B SUD, -200B, -300, -400, and -400D Series Airplanes

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain Boeing Model 747-100B SUD, -200B, -300, -400, and -400D series airplanes. This AD requires repetitive inspections for cracking in fuselage stringers 8L, 8R, 10L, and 10R at body stations 460, 480, and 500 frame locations; and repair if necessary. This AD is prompted by findings of cracking in fuselage stringers 8L, 8R, 10L, and 10R at body stations 460, 480, and 500 frame locations. We are issuing this AD to detect and correct fatigue cracking in certain fuselage stringers, which, if left undetected, could result in fuselage skin cracking that reduces the structural integrity of the skin panel, and consequent rapid depressurization of the airplane.

**DATES:** This AD becomes effective August 30, 2005.

The incorporation by reference of a certain publication listed in the AD is approved by the Director of the **Federal Register** as of August 30, 2005.

**ADDRESSES:** For service information identified in this AD, contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207.

**Docket:** The AD docket contains the proposed AD, comments, and any final disposition. You can examine the AD docket on the Internet at <http://dms.dot.gov>, or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at

the U.S. Department of Transportation, 400 Seventh Street SW., room PL-401, Washington, DC. This docket number is FAA-2004-19175; the directorate identifier for this docket is 2003-NM-246-AD.

**FOR FURTHER INFORMATION CONTACT:** Nick Kusz, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 917-6432; fax (425) 917-6590.

**SUPPLEMENTARY INFORMATION:** The FAA proposed to amend 14 CFR part 39 with an AD for certain Boeing Model 747-100B SUD, -200B, -300, -400, and -400D series airplanes. That action, published in the **Federal Register** on September 28, 2004 (69 FR 57884), proposed to require repetitive inspections for cracking in fuselage stringers 8L, 8R, 10L, and 10R at body stations 460, 480, and 500 frame locations; and repair if necessary.

#### Comments

We provided the public the opportunity to participate in the development of this AD. We have considered the comments that have been submitted on the proposed AD.

#### Support for the Proposed AD

One commenter concurs with the FAA's compliance recommendations specified in the proposed AD. A second commenter, the manufacturer, requested that the compliance time be changed to match the referenced service bulletin; the commenter later submitted a comment stating that it reanalyzed the data and now concurs with the compliance time specified in the proposed AD.

#### Request for Clarification of the Compliance Time

One commenter states that paragraph (f) of the proposed AD specifies repeating the inspection at intervals not to exceed 3,000 flight cycles until the optional terminating action is accomplished. The commenter adds that the referenced service bulletin recommends inspections at specific thresholds that equate to a 3,000-flight-cycle interval, until the airplane accumulates 25,000 flight cycles. The commenter also notes that the referenced service bulletin recommends that airplanes having more than 25,000 total flight cycles be inspected at intervals not to exceed 2,000 flight cycles, and adds that the proposed AD does not seem to address this situation. The commenter asks that the preamble in the proposed AD clearly specify that the 3,000-flight-cycle interval cited in

paragraph (f) replaces the threshold values in the referenced service bulletin.

Although we acknowledge the commenter's concern, the difference in compliance times was explained in the proposed AD. In the section titled "Differences Between the Proposed AD and Service Information" of the preamble of the proposed AD, we define the difference in compliance times, as follows: "The manufacturer reanalyzed the service problem and has advised the FAA that the reanalysis has resulted in threshold and repetitive inspection intervals different from the service bulletin. This resulted in simplified initial thresholds and an increased number of flight cycles between repetitive inspections." That section of the preamble of the proposed AD is not restated in the final rule; therefore, we made no change to the final rule in this regard.

#### Request for Optional Open-Hole and Surface High Frequency Eddy Current (HFEC) Inspections To Extend Repetitive Inspection Intervals

One commenter states that, subsequent to the release of the referenced service bulletin, Boeing advised the commenter of optional open-hole and surface HFEC inspections that could be performed in addition to the specified detailed inspections. The commenter adds that these optional inspections would allow extending the repetitive inspection interval to 4,000 flight cycles, until the accumulation of 25,000 total flight cycles on the airplane. The commenter asks that the FAA consult with Boeing about this alternative inspection process and, if appropriate, include that option in the final rule.

Although we acknowledge that the optional inspections may be a viable alternative to the detailed inspections, we have confirmed with the manufacturer that while an open-hole and surface HFEC inspection may be accomplished, there are no existing procedures available. Therefore, we do not agree to add the optional inspections and extend the repetitive inspection interval in this final rule. Paragraph (i) of this AD provides affected operators the opportunity to apply for an alternative method of compliance (AMOC) and to present data to justify adding the optional inspections and extending the repetitive inspection interval. In addition, if the referenced service information is revised to add the optional inspections, we may approve it as an AMOC to the final rule, if appropriate. We have made no change to the final rule in this regard.